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REPORT**



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NASA CR-2631

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**NASA/MSFC MULTILAYER DIFFUSION MODELS
AND COMPUTER PROGRAMS - VERSION 5**

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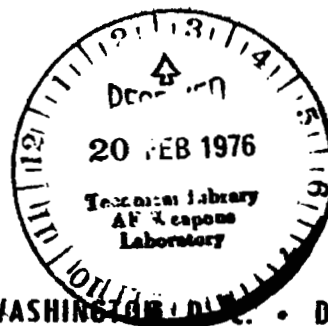
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16. ABSTRACT <p>The purpose of this report is to document the latest developments in the transport and diffusion models and algorithms developed for use by NASA in predicting concentrations and dosages downwind from normal and abnormal launches of rocket vehicles and the associated computer programs for use in performing the calculations. This report consists of:</p> <ul style="list-style-type: none"> • A description of the mathematical specifications and procedures used in the Preprocessor Program to calculate rocket exhaust cloud rise, cloud dimensions and other input parameters to the transport and diffusion models • A description of the revised mathematical specifications for the NASA/MSFC Multilayer Diffusion Models • Descriptions of the Preprocessor and Version 5 of the NASA/MSFC Multilayer Diffusion Models Programs • Users' instructions for implementing the Preprocessor and Multilayer Diffusion Models Programs • Worked example problems illustrating the use of the models and computer programs 					
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FOREWORD

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SECTION 1

INTRODUCTION

1.1 BACKGROUND

Work under two concurrent NASA contracts (NAS8-21453 and NAS8-30503) resulted in publication of the NASA Handbook for Estimating Toxic Fuel Hazards (Dumbauld, Bjorklund, Cramer and Record, 1970). The handbook was accompanied by a computer program (NASA/MSFC Multilayer Diffusion Models Program) specifically designed for research-planning estimates of downwind dispersions from normal and abnormal launches of rocket vehicles as well as from accidental cold spills and leaks of toxic fuels. The transport and diffusion models that formed the basis of the program were extensions of the generalized concentration, dosage and deposition model concept originally developed for the U. S. Army (Cramer, et al., 1964; 1967) to include a multilayer tropospheric construct with provision for changes in atmospheric structure along the cloud trajectory downwind from the launch site. As experience was gained in the application of the NASA/MSFC Multilayer Diffusion Models Program, revisions were made to simplify its use and to incorporate the latest available advances in diffusion technology. Version 2 of the Program, designed for use in operational support of vehicle launches, was completed in 1973. A technical report by Dumbauld, Bjorklund and Bowers (1973) describes the updated transport, diffusion and cloud rise models and the operation of the computer program.

Since 1973, work has continued on the development and refinement of the models and the computer program.

1.2 PURPOSE

The purpose of this report is to document:

- The latest refinements and developments in the transport and diffusion models

- Version 5 of the NASA/MSFC Multilayer Diffusion Models Program
- The Preprocessor Program used in conjunction with Version 5 of the Multilayer Program

The majority of the changes in the NASA/MSFC Multilayer Diffusion Models Program have been made to further simplify the use of the models and computer program so that users can apply the techniques for hazard estimation with only a minimum knowledge of the technical aspects of diffusion meteorology and associated disciplines. The simplification of program use was accomplished in two ways. First, a Preprocessor Program was designed to automate the calculation of the:

- Effective height of the stabilized exhaust cloud
- Distribution of exhaust products in the meteorological layers containing the cloud
- Meteorological inputs to the transport and diffusion models

Use of the Preprocessor Program essentially requires only surface meteorological and radiosonde data be available for input. Second, the NASA/MSFC Multilayer Diffusion Models Program was redesigned for more efficient operation and an extensive capability for automated plotting of isopleths and centerline profiles of concentration, dosage and deposition due to gravitational settling and precipitation scavenging is now provided to assist users in interpretation of computational results.

In addition to simplifications in the use of computer programs, technical features of the transport and diffusion models have been altered to more accurately reflect observations of cloud rise during actual launches of solid-fueled vehicles and the physical aspects of the transport and diffusion process. For example, alterations have been made in assumptions regarding the physical shape of the ground cloud at time of cloud stabilization and in the diffusion models to account for partial reflection (or partial absorption) of material at air-surface interfaces.

1.3 ORGANIZATION OF THE REPORT

The main body of the report contains five sections. Section 2 contains a description of the models and algorithms contained in the Preprocessor Program. The generalized transport and diffusion models for calculating concentration, dosage and deposition contained in the NASA/MSFC Multilayer Diffusion Models Program are described in Section 3. A brief description of the Preprocessor and Diffusion Models Programs is given in Section 4 and Section 5 contains some sample problems and their solutions obtained using the Programs.

There are five appendices to the report. Appendix A contains user instructions for implementing the Preprocessor Program and Appendix B contains user instructions for implementing the NASA/MSFC Multilayer Diffusion Models Program. Computer program listings for both the programs are given in Appendix C. Appendix D contains example computer output listings for the sample problems described in Section 5 of the main text. Finally, a derivation of the vertical term for Model 4 of the NASA/MSFC Multilayer Diffusion Models Program is presented in Appendix E.

SECTION 2

PREPROCESSOR MODELS AND ALGORITHMS

A Preprocessor Program has been developed to simplify the calculation of source and meteorological inputs required by the NASA/MSFC Multilayer Diffusion Models Program. The models and algorithms incorporated in the Preprocessor Program for calculating the requisite model inputs are described in this section of the report. In its present form, the Preprocessor Program has the capability of processing data for the Space Shuttle, Titan III, Delta-Thor and Minuteman II vehicles.

2.1 FUEL PROPERTIES AND VEHICLE RISE DATA

Properties of the vehicle fuel and rise data are required to calculate cloud rise and the source strength distribution of atmospheric pollutants in the troposphere resulting from normal and abnormal launches. Two types of abnormal launches have been hypothesized for the Space Shuttle and Titan III vehicles. In one situation, it is assumed that a single solid engine of the Titan or Space Shuttle zero stage ignites and burns over the normal engine firing period while the vehicle remains in a hold-down configuration on the pad (single-engine burn). In the second pad-abort situation, it is assumed that an on-pad explosion ruptures the casings of the two solid engines and that all the solid propellant then falls to the ground in the vicinity of the pad and burns at a constant rate over a 5-minute period (slow burn). In both pad-abort situations, it is further assumed that the other vehicle stages are unaffected by the burning of the zero-stage solid propellant and do not therefore contribute in any way to the combustion products or heat released during the on-pad aborts. The net affects of this latter assumption is to minimize the cloud rise and maximize the concentration of the pollutants produced by the burning of the solid propellant. Slow-burn incidents in which fuel burns over a 5-minute period are also hypothesized for the Delta-Thor and Minuteman II vehicles. In the case of the Delta-Thor, both the

solid propellant from the six castor engines and liquid fuel from the zero-stage engine are assumed to contribute to the heat available for cloud rise and to pollutant concentrations. The solid propellant from all three stages of the Minuteman II is assumed to fall to the ground near the launch site as the vehicle is destroyed just after clearing the launch silo.

Fuel expenditure and heat content data incorporated in the Preprocessor Program for normal and abnormal launches of the four types of vehicles are given in Table 2-1. Fuel expenditure rates for normal launches were obtained by averaging fuel consumption over the approximate period from lift-off until the vehicle is about 3 kilometers above the surface. For single-engine burns, the fuel expenditure rate was calculated by dividing the total propellant weight in a single engine by the normal firing period shown in the table. Similarly for slow burns in the vicinity of the launch area, the fuel expenditure rate was obtained by dividing the total weight of all solid engines, and the zero stage liquid engine in case of the Delta-Thor, by the total assumed burn time of 300 seconds.

The fuel heat contents used in calculating cloud rise are estimates based on the best available information and are subject to change as additional information becomes available. In case of the Tital III vehicle, the fuel heat content is based on estimates from two-phase (gas and solid) flow accounting approximately for heat gains from afterburning and heat losses due to radiation, and on the experience to date in predicting cloud rise for a number of actual launches using the instantaneous cloud rise model described in Section 2.2 below. Lacking better information, the same heat content has also been used for the solid engines of the Space Shuttle vehicle. A heat content value of 500 calories per gram was used for the liquid fuel in the zero stage engines of the Space Shuttle and Delta-Thor vehicles because previous experience in predicting cloud rise from Saturn launches indicates this value may be appropriate. The heat contents of the solid fuel for the Delta-Thor and Minuteman II are based on early estimates of the heat content for solid fueled engines and are probably low. However, experience in predicting cloud rise from a limited number of Delta-Thor

TABLE 2-1

FUEL EXPENDITURE AND HEAT CONTENT DATA

Property	Vehicle Type			
	Space Shuttle	Titan III	Delta-Thor	Minuteman II

(a) Normal Launch

Fuel Expenditure Rate (g sec ⁻¹)				
Solid Engine	9.385 x 10 ⁶	4.174 x 10 ⁶	6.05 x 10 ⁵	3.771 x 10 ⁵
Liquid Engine	1.531 x 10 ⁶		3.13 x 10 ⁵	
Effective Fuel Heat Content (cal g ⁻¹)				
Solid Fuel	2500	2500	691	691
Liquid Fuel	500		500	

(b) Single Engine Burn

Fuel Expenditure Rate (g sec ⁻¹)	3.753 x 10 ⁶	1.742 x 10 ⁶		
Normal Firing Period (sec)	122	122		
Effective Fuel Heat Content (cal g ⁻¹)	1274	1036		

TABLE 2-1

FUEL EXPENDITURE AND HEAT CONTENT DATA
(CONTINUED)

Property	Vehicle Type			
	Space Shuttle	Titan III	Delta-Thor	Minuteman II

(c) Slow Burn

Fuel Expenditure Rate (g sec ⁻¹)				
Solid Engine	3.052 x 10 ⁶	1.301 x 10 ⁶	7.462 x 10 ⁴	2.87 x 10 ⁷
Liquid Engine			2.212 x 10 ⁵	
Total Burn Time (sec)	300	300	300	300
Effective Fuel Heat Content (cal g ⁻¹)	1000	1000	1000	1000

launches using the heat contents in Table 2-1 and a combination of the instantaneous and continuous cloud rise models indicate serious errors are not being made in the rise prediction. In any case, the conservative low estimates of heat available for cloud rise for the Delta-Thor and Minuteman II vehicles have the effect of minimizing cloud rise and maximizing ground-level concentrations. There is no experience in predicting cloud rise from launch aborts of any of the four vehicle types. The heat contents shown in Table 2-1 for single-engine burns of the Space Shuttle and Titan vehicles are based on the estimate that about 1500 calories per gram are available from the fuel burn from a single engine, but that some of the available heat is dissipated in heating and vaporizing 1.26×10^3 kilograms per second of water used to spray the launch pad during the burn. The heat content of 1000 calories per gram hypothesized for slow burns of fuel from the zero-stages of the vehicle is thought to be a realistic estimate of the heat available for cloud rise from the burning of unconfined fuel.

The fraction by weight of pollutants comprising the rocket exhaust products of the four types of vehicles used in the calculation of the vertical distribution of pollutants in the lower troposphere are shown in Table 2-2. Two sets of fractions are given for the Minuteman II because in an abnormal launch all three stages are hypothesized as being destroyed and in normal launches only the first stage contributes to the pollutant distribution in the lower troposphere. Recent indications (Cicerone, Stedman and Stolarski, 1973) are that CO in the exhaust plume may quickly oxidize to CO₂, in which case no toxic CO problem exists. However, because the oxidation of CO to CO₂ has not been verified by direct measurements, the distribution of CO in the lower troposphere is calculated in the Preprocessor Program.

It should be noted that the values of heat content and fuel expenditure rates in Table 2-1 and the fraction of pollutants by weight in Table 2-2 may differ from the values assigned as constants in the Preprocessor Program described in Appendix A. This occurs because, in the Program, the constants are manipulated in different ways to calculate cloud rise and the vertical distribution of pollutants. Thus, after the

TABLE 2-2

POLLUTANT COMPOSITION OF FUEL (Fraction by Weight)

Pollutant	Vehicle Type				
	Space Shuttle	Titan III	Delta-Thor	Minuteman II	
				Normal	Abnormal
HCl	0.207	0.210	0.208	0.197	0.204
CO*					
Solid Engine	0.280	0.279	0.223	0.220	0.219
Liquid Engine			0.473		
CO ₂		0.029			
Al ₂ O ₃	0.304	0.304	0.378	0.277	0.280

* May be oxidized to CO₂

TABLE 2-3

VEHICLE RISE DATA

Power-Law Coefficients	Vehicle Type			
	Space Shuttle	Titan III	Delta-Thor	Minuteman II
a	0.664	0.635	1.321	0.440
b	0.485	0.484	0.395	0.479

calculations are completed by the Preprocessor Program, the values shown in Tables 2-1 and 2-2 have been, in effect, used in the calculations.

The altitude-time curves of the various vehicles are also required to calculate the buoyant rise of the ground cloud of exhaust products. A simple power-law relationship is sufficiently accurate to describe the altitude-time curve in the first several thousand meters near the surface. A logarithmic least-squares regression curve of the form

$$\text{Time} = a (\text{Altitude})^b \quad (2-1)$$

where time is in seconds and altitude in meters was fitted to the altitude-time information for all vehicles. The resulting values of the coefficients a and b obtained from the fitting procedure are given in Table 2-3.

2.2 CLOUD RISE FORMULAS

The burning of rocket engines during launches and on-pad aborts results in the formation of a cloud of hot exhaust products which subsequently rises and entrains ambient air until an equilibrium with ambient conditions is reached. For normal launches, the cloud is formed principally by the forced ascent of hot, turbulent exhaust products that have been deflected laterally and vertically by the launch pad hardware and the ground surface. In the case of normal launches of solid-fueled vehicles or vehicles with a number of solid boosters, vehicle hold-down times are minimal and vehicle residence times in the lowest kilometer of the atmosphere are relatively short. The exhaust products contained in the stabilized ground cloud are therefore emitted over a time period of from about 10 to 30 seconds. Experience to date shows that the buoyant rise from vehicles with solid-fueled first stages is best predicted by using a cloud-rise model for instantaneous sources and rise from vehicles having liquid-fueled first stages is best predicted using a cloud-rise model for continuous sources. Thus, an instantaneous source cloud-rise model is used in the Preprocessor Program to predict buoyant rise of the ground cloud for normal

launches of the Space Shuttle, Titan III and Minuteman II vehicles. Limited experience in predicting the buoyant rise of the ground cloud generated by normal launches of the Delta-Thor vehicle with its large liquid-fueled first stage and six solid-fueled boosters indicates that an average of the rise predicted using the instantaneous and continuous source cloud rise models is appropriate. While no cloud rise data are available for on-pad aborts of any of the four vehicle types, cloud rise data from static-tests of liquid fueled rocket engines indicate that the use of a cloud rise model for continuous sources is appropriate in these cases.

The instantaneous and continuous cloud-rise models used in the Preprocessor Program are based on work by Briggs (1969, 1970). Derivations of these models also appear in the report by Dumbauld, et al. (1973). Only the simplified forms of the models used in the Preprocessor Program are given below.

2.2.1 Instantaneous Source Cloud-Rise Model

The maximum cloud rise z_{mI} from an instantaneous source in a thermally stable atmosphere is given by the expression

$$z_{mI} = \left[\frac{8 F_I}{\gamma_I^3 s} \right]^{1/4} \quad (2-2)$$

where

F_I = initial buoyancy term

$$= \frac{3 g Q_I}{4 c_p T \pi \rho}$$

g = gravitational acceleration (9.8 m sec^{-2})

Q_I = effective heat released (cal)

c_p = specific heat of air at constant pressure
($\text{cal g}^{-1} \text{ } ^\circ\text{K}$)

T = ambient air temperature ($^\circ\text{K}$) at the surface
reference height z_R

ρ = density of ambient air (g m^{-3}) at z_R

$$s = \frac{g}{T} \frac{\partial \Phi}{\partial z} \approx \frac{g}{T} \frac{\Delta \Phi}{\Delta z}$$

$\frac{\Delta \Phi}{\Delta z}$ = vertical gradient of ambient virtual potential temperature

γ_I = entrainment constant = 0.64

The time t_{lk} required for the center of the ground cloud to reach an altitude z_k is given by the expression

$$t_{lk} = s^{-1/2} \arccos \left(1 - \frac{(z_k)^4}{K} \right); \frac{(z_k)^4}{K} > 2 \quad (2-3)$$

where

$$K = \frac{3 Q_I}{\rho c_p \pi \gamma_I^3 \frac{\Delta \Phi}{\Delta z}}$$

The time t^* required for the cloud to reach the stabilization height z_{mI} is thus given by the expression

$$t^* = \pi s^{-1/2} \quad (2-4)$$

In calculating z_{mI} from Equation (2-2) the effective heat released for cloud rise Q_I is calculated from the relationship

$$Q_I = Q_F t_R \{z_{mI}\} \quad (2-5)$$

where

$$\begin{aligned} Q_F &= \text{rate of heat released by the burning fuel} \\ &= H \cdot W \\ H &= \text{fuel heat content} \end{aligned}$$

W = fuel expenditure rate

$$\begin{aligned} t_R \{z_{mI}\} &= \text{time in seconds for the vehicle to} \\ &\quad \text{reach the height of the centroid of the} \\ &\quad \text{stabilized ground cloud } z_{mI} \\ &= a (z_{mI})^b \end{aligned} \quad (2-6)$$

where a and b are the power-law coefficients from Table 2-3.

According to the above formulas, the following quantities are interrelated: the calculated maximum cloud rise z_{mI} , the height over which the virtual potential temperature gradient $\Delta\Phi / \Delta z$ is measured, and the value of $t_R \{z_{mI}\}$ used in obtaining Q_I . The final value of maximum cloud rise must therefore be found through iteration of Equation (2-1). The Preprocessor Program accepts radiosonde data which provides the temperature, pressure and relative humidity profiles as functions of height in the troposphere and the surface density ρ for use in the cloud rise calculation. The virtual potential temperature Φ_k at each radiosonde measurement height z_k is obtained from the expression (Tabata, 1973)

$$\Phi_k = T_k \left[\frac{1 + 1.61 w_k}{1 + w_k} \right] \left[\frac{1000}{P_k} \right]^{0.288}$$

where

T_k = Air temperature at z_k in degrees absolute

P_k = barometric pressure at z_k (mb)

w_k = mixing ratio at z_k

$$= \frac{0.622 (RH)_k e_{s;k}}{P_k - (RH)_k e_{s;k}}$$

$(RH)_k$ = relative humidity at z_k expressed as a fraction

$$\begin{aligned}
e_{s; k} &= \text{saturation vapor pressure at } z_k \\
&= 10^{(c-dx-ex^2)} \\
x &= 1000/T_k
\end{aligned}$$

and c, d and e are constants given by

$$\begin{aligned}
c &= 8.42926604 \\
d &= 1.82717843 \\
e &= 0.071208271
\end{aligned}$$

The iteration process then begins by assuming that the first level above the surface reference height ($z_k \{k=1\} = z_R$) at which a radiosonde measurement is available ($z_k \{k=2\}$) is equal to z_{mI} and solving Equation (2-6) for $t_R \{z_{mI} = z_2\}$ and Equation (2-5) for Q_I . The lapse rate of virtual potential temperature between the surface reference height and the first height above the surface is then obtained from the expression

$$\frac{\Delta \Phi}{\Delta z} = \frac{\Phi_k \{k=2\} - \Phi_k \{k=1\}}{z_k \{k=2\} - z_R}$$

and Equation (2-2) solved for z_{mI} . If the value of z_{mI} calculated from Equation (2-6) is less than the value $z_k \{k=2\}$, then 10 meters is subtracted from $z_k \{k=2\}$ and the process repeated using the same value of $\Delta\Phi/\Delta z$ until the estimated value of z_{mI} is within ± 10 meters of the value of z_{mI} calculated from Equation (2-2). If on the first iteration step the value of z_{mI} calculated from Equation (2-6) exceeds the value $z_k \{k=2\}$, then the value of $z_k \{k=3\}$ is used as an estimate of z_{mI} and the iteration process is continued by increasing k until the value of z_{mI} from Equation (2-6) exceeds the estimated value of z_{mI} . However, for each iteration where $k > 2$, a least-square regression fit of the form

$$\frac{\Delta \psi}{\Delta z} = \frac{\sum_{k=1}^K \left\{ \left[z_k - \left(\sum_{k=1}^K z_k / K \right) \right] \left[\phi_k - \left(\sum_{k=1}^K \phi_k / K \right) \right] \right\}}{\sum_{k=1}^K \left[z_k - \left(\sum_{k=1}^K z_k / K \right) \right]^2} \quad (2-7)$$

is used, at the suggestion of Dr. Briscoe Stephens of NASA/MSFC, to obtain an estimate of the vertical potential temperature gradient. As a result of this iteration procedure, the Preprocessor Program obtains an estimated final height of the cloud centroid within ± 10 meters of the exact value that should be calculated.

It should be noted that Equation (2-2) is for use when the atmosphere is thermally stable. For this reason, if the virtual potential temperature gradient is ever less than 3.322×10^{-4} degrees per meter, the gradient is set equal to 3.322×10^{-4} degrees per meter. Experience has shown that use of this device yields nearly the same final cloud rise as the cloud rise that would have been obtained had a formula for an adiabatic atmosphere been used in the calculation.

2.2.2 Continuous Source Cloud Rise Model

The maximum cloud rise z_{mc} from a continuous source is

$$z_{mc} = \left(\frac{6 F_c}{\bar{u}_c \gamma_c^2 s} \right)^{1/3} \quad (2-8)$$

where

$$\begin{aligned} F_c &= \text{buoyancy flux} \\ &= \frac{g Q_c}{\pi \rho c_p T} \\ Q_c &= \text{effective heat rate (cal sec}^{-1}\text{)} \end{aligned}$$

$$\begin{aligned}
&= H \bullet W \\
\gamma_c &= \text{entrainment constant} = 0.5 \\
\bar{u}_c &= \text{height-weighted mean wind speed} \\
&\quad \text{between the surface and } z_{mc} \\
&= \frac{\sum_{k=1}^K (\bar{u}_k) + \bar{u}_j}{z_{mc} - z_1} \\
\bar{u}_k &= \left(z_{k+1} - z_k \right) \left(\frac{u_{k+1} + u_k}{2} \right) \\
\bar{u}_j &= \left\{ \frac{\frac{(u_{K+1} - u_K)(z_{mc} - z_K)}{(z_{K+1} - z_K)} + 2 u_K}{2} \right\} (z_{mc} - z_K) \\
K &= \text{index of the last radiosonde} \\
&\quad \text{measurement point below the} \\
&\quad \text{height of the cloud centroid at} \\
&\quad \text{stabilization } z_{mc}
\end{aligned}$$

and u_k are wind speeds at the radiosonde observation heights z_k . The time t_{ck} required for the centroid of the ground cloud to reach an altitude z_k for continuous sources is given by the expression

$$t_{ck} = s^{-1/2} \arccos \left(1 - \frac{(z_k)^3}{J} \right); \left((z_k)^3 / J \right) > 2 \quad (2-9)$$

where

$$J = \frac{3 Q_c z_k}{\rho_c \pi \gamma_c^2 \frac{\Delta \Phi}{\Delta z} \sum_{i=1}^k (\bar{u}_i)}$$

The time t^* required for the cloud to reach the stabilization height z_{mc} is identical to the time for instantaneous sources given by Equation (2-4).

As in the case of instantaneous sources, iteration is required to solve for the height z_{mc} because it is interrelated with the height over which $\Delta\Phi/\Delta z$ is measured and with the wind speed \bar{u}_c . The same iteration procedure described in Section 2.2.1 above for instantaneous sources is used in the Preprocessor routine for calculating z_{mc} .

2.3 CALCULATION OF SOURCE MODEL INPUT PARAMETERS

The Preprocessor Program is used to calculate the dimensions and spatial position of the stabilized ground cloud and the distribution of exhaust products within the cloud. In its present form, the Preprocessor Program can be used only with Models 3 and 4 (see Section 3) to calculate the source model input parameters for those portions of the stabilized ground cloud that are contained within the surface mixing layer. The source geometry identified with Models 3 and 4 is described in Section 5. In brief, Model 3 assumes that the portion of the stabilized ground cloud in the surface mixing layer is all contained within a spherical volume source. In Model 4, the stabilized ground cloud in the surface mixing layer is assumed to be contained in a number of cylindrical volume sources extending from the ground surface to the top of the mixing layer. Extension of the Preprocessor Program to include calculations of source inputs for all the models in the NASA/MSFC Multi-layer Diffusion Models Program is currently in process with emphasis on implementation for the REEDA system of NASA/MSFC.

2.3.1 Calculation of the Dimensions and Spatial Position of the Stabilized Ground Cloud

In the Preprocessor Program, the cloud radius at any height z during cloud rise is calculated by the expression

$$r\{z\} = \left\{ \begin{array}{ll} \gamma z & ; \quad z \leq z_m \\ \gamma(2z_m - z) & ; \quad z_m < z \leq 2z_m \\ 0 & ; \quad z > 2z_m \end{array} \right\} \quad (2-10)$$

where γ is equal γ_I or γ_C and z_m is equal to z_{mI} or z_{mC} depending on whether the source is instantaneous (I) or continuous (C).

The alongwind, crosswind and vertical source dimensions in the surface mixing layer for Model 3 are calculated under the following assumptions:

- The alongwind, crosswind and vertical distributions of exhaust products within the stabilized cloud are Gaussian
- The concentration of exhaust products at a distance of one radius from the cloud centroid is 10 percent of the concentration at the centroid

Thus the standard deviations of the alongwind ($\sigma_{x0}\{K=1\}$), crosswind ($\sigma_{y0}\{K=1\}$) and vertical ($\sigma_{z0}\{K=1\}$) distribution which define the source dimensions at the point of cloud stabilization are calculated from the relationships

$$\sigma_{x0}\{K=1\} = \sigma_{y0}\{K=1\} = r\{z_m\} / 2.15 = \gamma z_m / 2.15 \quad (2-11)$$

$$\sigma_{z0}\{K=1\} = \left\{ \begin{array}{ll} r\{z_m\} / 2.15 & ; z_{TK}\{K=1\} > z_m + r\{z_m\} \\ \frac{z_{TK}\{K=1\} - z_m + r\{z_m\}}{4.3} & ; z_m - r\{z_m\} < z_{TK}\{K=1\} \leq z_m + r\{z_m\} \end{array} \right\} \quad (2-12)$$

where

$$z_{TK} \{K=1\} = \text{depth of the surface mixing layer for Model 3}$$

In the special case where the bulk of the cloud is above the surface mixing layer ($z_{tK} \{K=1\} \leq z_m - r \{z_m\}$), the Preprocessor Program prints a message indicating that calculations are terminated for Model 3 since its use is invalid in the surface mixing layer for this case. The effective height $H_K \{K=1\}$ of the stabilized cloud centroid in the surface mixing layer for Model 3 is given by

$$H_K \{K=1\} = \left\{ \begin{array}{l} z_m \quad ; z_{TK} \{K=1\} \geq z_m + r \{z_m\} \\ \frac{z_{TK} \{K=1\} + z_m - r \{z_m\}}{2} ; z_m - r \{z_m\} < z_{TK} \{K=1\} < z_m + r \{z_m\} \end{array} \right\} \quad (2-13)$$

In the case of Model 4, the Preprocessor Program assumes that the surface mixing layer, for which the depth is defined as $z_{TL} \{L=1\}$, is comprised of one or more sublayers K with boundaries at heights in the surface mixing layer specified in the radiosonde message. Alongwind and crosswind source dimensions are defined in each K sublayer for normal launches by

$$\sigma_{xo} \{K\} = \sigma_{yo} \{K\} = \left\{ \begin{array}{l} \frac{\gamma z_m}{2.15} \quad ; z' \leq z_m \\ \frac{\gamma (2z_m - z')}{2.15} \quad ; z' > z_m, \gamma z' > 200 \\ \frac{200}{2.15} = 93 \quad ; z' > z_m, \gamma z' \leq 200 \end{array} \right\} \quad (2-14)$$

and for abnormal launches by

$$\sigma_{xo}\{K\} = \sigma_{yo}\{K\} = \left\{ \begin{array}{ll} \frac{\gamma z_m}{2.15} & ; z' \leq z_m \\ \frac{\gamma (2z_m - z')}{2.15} & ; z_m < z' \leq 2z_m \\ 0 & ; z' > 2z_m \end{array} \right\}$$

$$z' = \text{midpoint of the } K^{\text{th}} \text{ sublayer}$$

$$= (z_{TK} + z_{BK}) / 2$$

$$z_{TK} = \text{height of the top of the } K^{\text{th}} \text{ sublayer}$$

$$z_{BK} = \text{height of the base of the } K^{\text{th}} \text{ sublayer}$$

The corresponding vertical source dimension for Model 4 is zero since the model assumes a rectangular source distribution in the sublayers and uses error functions to simulate a vertical line source between sublayer boundaries which is comprised of an infinite number of (vertical) point sources (see Section 3.4 below).

For Model 4, the spatial position in the horizontal plane of the cloud in any sublayer K at t^* , the time of cloud stabilization, with respect to the origin at the launch pad is assumed to be given by

$$A_K = 90 - \tan^{-1} (y_k / x_k) \quad (2-15)$$

$$R_K = (x_k^2 + y_k^2)^{1/2} \quad (2-16)$$

where

$$y_k = y_{k-1} - R_k \cos(\bar{\theta}) \quad (2-17)$$

$$x_k = x_{k-1} - R_k \sin(\bar{\theta}) \quad (2-18)$$

$$\text{For } t_{I, k+1} \text{ or } t_{c, k+1} < t^* : \quad (2-19)$$

$$\bar{\theta} = (\theta_{k+1} + \theta_k) / 2$$

$$R_k = \left\{ \begin{array}{l} (t_{I, k+1} - t_{I, k}) (u_{k+1} + u_k) / 2 ; \text{ instantaneous source} \\ \frac{(t_{c, k+1} - t_{c, k}) \sum_{i=1}^k (z_{i+1} - z_i) (u_{i+1} + u_i) / 2}{\sum_{i=1}^k (z_{i+1} - z_i)} ; \text{ continuous source} \end{array} \right\} \quad (2-20)$$

where θ_k and u_k are respectively the wind direction and wind speed from radiosonde measurements at the k^{th} height and $t_{I, k}$ and $t_{c, k}$ are the times respectively from Equations (2-3) and (2-9) for the cloud to pass through the k^{th} height. When

$$t_{I, k+1} \text{ or } t_{c, k+1} > t^* :$$

$$\bar{\theta} = (\theta_m + \theta_k) / 2 \quad (2-21)$$

where

$$\theta_m = \left[\theta' / (z_{k+1} - z_k) \right] [z_m - z_k] + \theta_k \quad (2-22)$$

$$\theta' = \left\{ \begin{array}{l} (\theta_{k+1} - \theta_k) ; \quad |\theta_{k+1} - \theta_k| < 180 \\ (\theta_{k+1} - \theta_k - 360); \quad \theta_{k+1} - \theta_k > 180 \\ (\theta_{k+1} - \theta_k + 360); \quad \theta_{k+1} - \theta_k < -180 \end{array} \right\} \quad (2-23)$$

and

$$R_k = \begin{cases} R_I' (t^* - t_{L,k+1}) ; \text{ instantaneous source} \\ R_c' (t^* - t_{c,k+1}) ; \text{ continuous source} \end{cases} \quad (2-24)$$

where

$$R_I' = \left[(z_m - z_k) / (z_{k+1} - z_k) \right] \left[(u_{k+1} - u_k) / 2 \right] + u_k \quad (2-25)$$

$$R_c' = \left\{ \left[(z_m - z_k) / (z_{k+1} - z_k) \right] \left[(u_{k+1} - u_k) / 2 \right] + u_k \right\} \{ z_m - z_k \} \\ + \sum_{i=1}^k \left[(z_{i+1} - z_i) (u_{i+1} + u_i) / 2 \right] \{ z_m \} \quad (2-26)$$

For Model 3, only the last position of the stabilized cloud, obtained by stepping through Equations (2-15) to (2-26), is used to identify the position of the cloud at stabilization.

2.3.2 Calculation of the Distribution of Exhaust Products within the Stabilized Cloud

The fraction by weight of pollutant material in the stabilized ground cloud in each of the K sublayers within the surface mixing layer for Model 4 is given by the expression

$$F\{K\} = \begin{cases} Q P \{z_{TK}\} & ; K = 1 \\ Q (P \{z_{TK}\} - P \{z_{BK}\}) ; 1 < K < z_{TL} \{L = 1\} \end{cases} \quad (2-27)$$

where

$$Q = \text{total weight of exhaust products in the stabilized ground cloud}$$

$$= W \cdot t_R \{z_m\} \cdot FM \quad (2-28)$$

FM = fraction by weight of pollutant material in the fuel
from Table 2-2

$$P \{z_{TK}\} = \frac{1}{\sqrt{2\pi} \sigma} \int_{-\infty}^{z_{TK}} \exp \left[-1/2 \left(\frac{z - z_m}{\sigma} \right)^2 \right] dz \quad (2-29)$$

$$= \Phi \left\{ \frac{z_{TK} - z_m}{\sigma} \right\}$$

$$\sigma = r \{z = z_m\} / 2.15$$

$$P \{z_{BK}\} = \frac{1}{\sqrt{2\pi} \sigma} \int_{-\infty}^{z_{BK}} \exp \left[-1/2 \left(\frac{z - z_m}{\sigma} \right)^2 \right] dz \quad (2-30)$$

$$= \Phi \left\{ \frac{z_{BK} - z_m}{\sigma} \right\}$$

$$\Phi(t) = \frac{1}{\sqrt{2\pi}} \int_{-\infty}^t \exp \left(\frac{-\xi^2}{2} \right) d\xi$$

Inspection of Equations (2-29) and (2-30) shows that a Gaussian vertical distribution of material is assumed about the height of the stabilized ground cloud z_m . If the height z_m is less than the depth of the surface mixing layer z_{TL} ($L = 1$) and the launch is normal, the vehicle exhaust trail will insert more material into the surface mixing

layer than given by Equation (2-27). In this case, the total fraction of material in the K^{th} sublayer is given by

$$F_T \{K\} = \left\{ \begin{array}{l} F\{K\} + W \cdot FM (t_R \{z_{TK}\} - t_R \{z_m\}) ; t_R \{z_{BK}\} < t_R \{z_m\}, z_m < z_n \{L=1\} \\ F\{K\} + W \cdot FM (t_R \{z_{TK}\} - t_R \{z_{BK}\}) ; t_R \{z_{BK}\} > t_R \{z_m\}, z_m < z_{TL} \{L=1\} \end{array} \right\} \quad (2-31)$$

In case of Model 3, only the total source strength of the pollutant in the surface mixing layer $F_T \{K = 1\}$ is required. Thus,

$$F_T \{K = 1\} = Q P \{z_{TK} \{K = 1\}\} \quad (2-32)$$

where $z_{TK} \{K=1\}$ is the top of the surface mixing layer, Q is defined by Equation (2-28) above and $P \{z_{TK} \{K=1\}\}$ is defined by Equation (2-29).

Since the desired concentration units for HCl, CO and CO_2 are parts per million, the complete expression for Q_K is

$$Q_K = F_T \{K\} \left(\frac{10^3 \text{ mg}}{\text{g}} \right) \left(\frac{22.4}{M} \right) \left(\frac{T}{273.16} \right) \left(\frac{1013.2}{P} \right) \quad (2-33)$$

where

M = molecular weight of HCl, CO, or CO_2
 T = ambient air temperature ($^{\circ}\text{K}$)
 P = ambient pressure (mb)

For Al_2O_3 , the desired concentration units are milligrams per cubic meter, so that

$$Q_K = F_T \{K\} (10^3 \text{ mg/g}) \quad (2-34)$$

2.4 CALCULATION OF METEOROLOGICAL MODEL INPUT PARAMETERS

The majority of meteorological model input parameters required by the Preprocessor Program, and by Models 3 and 4 described in Section 3 below are obtained from radiosonde observations. These include:

$$u_k = \text{wind speed (m sec}^{-1}\text{)}$$

$$\theta_k = \text{wind direction (deg)}$$

$$T_k = \text{ambient air temperature (}^{\circ}\text{K)}$$

$$(\text{RH})_k = \text{ambient relative humidity expressed as a percentage}$$

$$P_k = \text{atmospheric pressure (mb)}$$

$$\rho = \text{surface air density (gm}^{-3}\text{)}$$

In addition, the use of Models 3 and 4 to predict concentrations and dosages in the surface mixing layer requires the following meteorological parameters:

$\sigma'_{AL} \{L=1\}$ or $\sigma'_{AK} \{K=1\}$ = mean layer standard deviation of the wind azimuth angle in radians for the surface mixing layer for Model 4 (L notation) and Model 3 (K notation).

$\sigma'_{EL} \{L=1\}$ or $\sigma'_{EK} \{K=1\}$ = mean layer standard deviation of the wind elevation angle in radians for the surface mixing layer for Model 4 and Model 3.

α_L or α_K = lateral diffusion coefficient in the surface mixing layer for Model 4 and Model 3.

β_L or β_K = vertical diffusion coefficient in the surface mixing layer for Model 4 and Model 3.

$z_{TL} \{L=1\}$ or $z_{TK} \{K=1\}$ = depth of the surface mixing layer for Model 4 and Model 3.

The standard deviation of the wind azimuth angle in the surface mixing layer, in the absence of turbulence induced in the layer by the rocket vehicle itself, is proportional to the source function time τ and the wind speed profile in the mixing layer. Dumbauld, et al. (1970) give the following relationships:

$$\sigma'_A \{\tau\} \sim \sigma'_{AR} \left(\frac{\tau}{\tau_0} \right)^{1/5} ; \tau < \tau_0 < 600 \text{ seconds} \quad (2-35)$$

and

$$\sigma'_A \{z\} \sim \sigma'_{AR} \left(\frac{z}{z_R} \right)^{-p} \quad (2-36)$$

where

σ'_{AR} = standard deviation of the wind azimuth angle for the reference time τ_o measured at a reference height z_R

p = power-law coefficient of the wind speed profile in the surface mixing layer

Assuming that the source function time τ is equivalent to the time t^* required for stabilization of the ground cloud, the mean standard deviation of the wind azimuth angle in the surface mixing layer is

$$\sigma'_{AL}\{L=1\} = \sigma'_{AR} \left(\frac{t^*}{\tau_o} \right)^{1/5} \frac{\left(z_{TL}\{L=1\}^{1-p} - z_R^{1-p} \right)}{\left(z_{TL}\{L=1\} - z_R \right)^{(1-p)} (z_R)^{-p}} \quad (2-37)$$

For physically reasonable combinations of z_{TL} , p and t^* , which are interrelated because of their joint dependence on atmospheric stability in the surface mixing layer, the value of $\sigma'_{AL}\{L=1\}$ falls within the range

$$\frac{\sigma'_{AR}\{\tau_o = 600\}}{4} < \sigma'_{AL}\{L=1\} < \frac{\sigma'_{AR}\{\tau_o = 600\}}{1.5} \quad (2-38)$$

when the reference standard deviation σ'_{AR} is measured near the surface ($2 \text{ m} \leq z_R \leq 20 \text{ m}$) over a reference time of 10 minutes. The passage of the rocket vehicle through the surface layer introduces turbulence that may persist for 30 minutes or longer, depending on the stability. This vehicle-generated turbulence enhances ambient turbulence levels and thus acts to increase the value of $\sigma'_{AL}\{L=1\}$. For these reasons, the simple expression

$$\sigma'_{AL} \{L = 1\} = \frac{\sigma'_{AR} \{\tau_o = 600\}}{2} \quad (2-39)$$

is used in the Preprocessor Program to define the mean layer standard deviation of the wind azimuth angle.

The standard deviation of the wind elevation angle is not related to the source function time, but is approximately related to the value of $\sigma'_A \{\tau_o\}$ by the expression (Cramer, et al., 1964)

$$\sigma'_{ER} \approx \frac{\sigma'_{AR} \{\tau_o = 600 \text{ seconds}\}}{3} \quad (2-40)$$

and to the wind profile in the surface mixing layer by

$$\sigma'_E \{z\} \approx \left\{ \begin{array}{ll} \sigma'_{ER} \left(\frac{z}{z_R} \right)^{0.3 - p} & ; \text{unstable conditions} \\ \sigma'_{ER} \left(\frac{z}{z_R} \right)^{-p} & ; \text{neutral and stable conditions} \end{array} \right\} \quad (2-41)$$

Therefore, under unstable atmospheric conditions, σ'_E increases with height. Also, while σ'_E decreases with height under neutral and stable conditions, vehicle-induced turbulence will persist longer under these conditions and will tend to increase ambient turbulent levels. For these reasons, the Preprocessor Program assumes turbulence is isentropic in the surface mixing layer such that

$$\sigma'_{EL} \{L=1\} = \sigma'_{AL} \{L=1\} = \frac{\sigma'_{AR} \{\tau_o = 600\}}{2} \quad (2-42)$$

Thus , the only turbulence input required for program operation is $\sigma'_{AR} \{ \tau_o = 600 \text{ seconds} \}$, the reference standard deviation of the wind azimuth angle measured over a 10-minute period at a height near the surface. At Kennedy Space Center, these measurements are available from bi-directional vanes mounted on towers throughout the launch complex areas.

In the NASA/MSFC Multilayer Diffusion Models Program, values must be assigned to the lateral α and vertical β power-law diffusion coefficients. Experience in application of the models to predict dispersion from nearly-instantaneous sources has shown that α and β normally vary insignificantly from unity. The diffusion experiments discussed by Pasquill (1974) indicate that α and β for an instantaneous source are essentially independent of atmospheric stability. For short travel distances, the experimental evidence suggests that α is greater than or equal to unity. Even for travel distances as long as 500 kilometers, a value of unity (Pasquill 1974, p. 224) appears to provide a reasonably good approximation to the data. Before vertical cloud expansion is restricted at the top of the surface mixing layer, the diffusion experiments analyzed by Pasquill show that β is also equal to unity. At larger downwind distances, the experiments indicate a less rapid increase of cloud vertical expansion, which probably reflects the restriction on vertical growth at the top of the surface mixing layer. Since the NASA/MSFC Multilayer Diffusion Models provide for restricting vertical growth at the top of the surface mixing layer, a value for β of unity would provide a good estimate of vertical expansion rates. For these reasons, the Preprocessor Program sets all power-law diffusion coefficients in the surface mixing layer to unity.

The depth of the surface mixing layer, $z_{TL} \{L = 1\}$ for Model 4 and $z_{TK} \{K = 1\}$ for Model 3, is not automatically derived from meteorological data available to the Preprocessor Program at the present time, but must be specified as an input to the program. In the input list to the Preprocessor Program, the depth of the surface mixing layer is denoted by H_m . An experienced meteorologist can estimate the

depth of the surface mixing layer through inspection of the radiosonde data and a general knowledge of the mesoscale and synoptic scale weather patterns at a particular site.

SECTION 3

THE NASA/MSFC MULTILAYER DIFFUSION MODEL

The meteorological structure in the lower troposphere is usually comprised of several layers with distinctive wind, temperature and humidity fields. Horizontal spatial variations in wind regimes can also occur in the surface layer, usually as a consequence of changes in terrain or at land-water interfaces. The models described below have been designed to accommodate to these variations of meteorological structure in the lower troposphere. The vertical stratification problem in the troposphere is handled by applying the models to individual layers in which the meteorological structure is reasonably homogeneous. Layer boundaries are placed at the points of major discontinuities in the vertical profiles of wind, temperature, and humidity. For simplicity, it is assumed that, in general, there is no flux of material across layer boundaries due to turbulent mixing. Provision is made, however, for the flux of material across layer boundaries as a result of gravitational settling or precipitation scavenging and, in Model 4, as a result of breakdown in meteorological layer structure. Changes in the meteorological structure of layers, at some arbitrary time or downwind distance from the point of release, are accommodated by stopping the transport and diffusion processes in the layers affected by the change in structure, calculating new sets of initial source and meteorological input parameters and re-starting the transport and diffusion process with the new inputs. The provisions in the use of Model 4 for permitting changes in initial conditions are also ideally suited for calculating concentration and dosage fields in the surface mixing layer when portions of the cloud of exhaust products are located at varying spatial positions at cloud stabilization. The use of this feature of Model 4 is further explained in Section 3.4 below.

The major changes in the model construct from those appearing in the report by Dumbauld, Bjorklund and Bowers (1973) are that provision has been made in the Vertical Terms of Models 3, 4 and 6 for partial reflection of material at the ground or water surface. For convenience, the mathematical specifications for all the various layer models contained in the NASA/MSFC Multilayer Diffusion Models Program are given below.

3.1 MODEL 1

In this layer model, the source extends vertically through the entire layer and turbulent mixing is occurring. It is assumed that the vertical distribution of toxic material is uniform with height and that the distributions of toxic material along the x- and y-layer coordinates are Gaussian.

3.1.1 Dosage Equation for Model 1

The dosage equation for Model 1 in the K^{th} layer is

$$D_K \{x_K, y_K, z_K\} = \frac{Q_K}{\sqrt{2\pi} \bar{u}_K \sigma_{yK} (z_{TK} - z_{BK})} \left\{ \exp \left(\frac{-y_K^2}{2\sigma_{yK}^2} \right) \right\} \quad (3-1)$$

In the above expression

Q_K = the source strength in units of mass

z_{TK} = height of the top of the K^{th} layer

z_{BK} = height of the base of the K^{th} layer

The quantity \bar{u}_K in Equation (3-1) is the mean cloud transport speed in meters per second in the K^{th} layer. In the surface layer ($K = 1$), the wind speed-height profile is defined according to the power-law expression

$$\bar{u} \{z_K, K = 1\} = \bar{u}_R \left(\frac{z_K \{K = 1\}}{z_R} \right)^p \quad (3-2)$$

where

$$\begin{aligned} \bar{u}_R &= \text{mean wind speed measured at the reference height } z_R \\ p &= \text{power-law exponent for the wind speed profile in the surface layer} \\ &= \log \left(\frac{\bar{u}_{TK} \{K=1\}}{\bar{u}_R} \right) / \log \left(\frac{z_{TK} \{K=1\}}{z_R} \right) \\ \bar{u}_{TK} \{K=1\} &= \text{mean wind speed at the top of the surface layer } z_{TK} \{K=1\} \\ z_K \{K=1\} &= \text{height in the surface layer} \end{aligned}$$

Thus, in the surface layer, the mean cloud transport speed is defined by the expression

$$\begin{aligned} \bar{u}_K \{K=1\} &= \frac{\bar{u}_R}{(z_{TK} \{K=1\} - z_R) z_R^p} \int_{z_R}^{z_{TK}} (z_K \{K=1\})^p dz \\ &= \frac{\bar{u}_R \left[(z_{TK} \{K=1\})^{1+p} - (z_R)^{1+p} \right]}{(z_{TK} \{K=1\} - z_R) (z_R)^p (1+p)} \end{aligned}$$

In layers above the surface layer ($K > 1$), the wind speed-height profile is assumed linear and defined by the expression

$$\bar{u} \{z_K, K > 1\} = \bar{u}_{BK} + \left(\frac{\bar{u}_{TK} - \bar{u}_{BK}}{z_{TK} - z_{BK}} \right) (z_K - z_{BK}) \quad (3-3)$$

where

\bar{u}_{TK} = mean wind speed at the top of the layer z_{TK}

\bar{u}_{BK} = mean wind speed at the base of the layer z_{BK}

In the K^{th} layer ($K > 1$), the mean cloud transport speed is given by the expression

$$\bar{u}_K \{K > 1\} = (\bar{u}_{TK} + \bar{u}_{BK}) / 2$$

The standard deviation of the crosswind dosage distribution is defined by the expression

$$\sigma_{yK} = \left\{ \left[\sigma'_{AK} \{ \tau \} x_{ry} \left(\frac{x_K + x_{yK} - x_{ry} (1 - \alpha_K)}{\alpha_K x_{ry}} \right)^{\alpha_K} \right]^2 + \left[\frac{\Delta \theta'_K x_K}{4.3} \right]^2 \right\}^{1/2} \quad (3-4)$$

where

$\sigma'_{AK} \{ \tau \}$ = mean layer standard deviation of the wind azimuth angle in radians for the cloud stabilization time τ

In the surface layer ($K = 1$),

$$\sigma'_{AK} \{ \tau, K=1 \} = \frac{\sigma'_{AR} \{ \tau \} \left[(z_{TK} \{K=1\})^{m+1} - (z_R)^{m+1} \right]}{(m+1)(z_{TK} \{K=1\} - z_R)(z_R)^m} \quad (3-5)$$

where

$\sigma'_{AR} \{ \tau \}$ = standard deviation of the wind azimuth angle in radians
at height z_R and for the cloud stabilization time τ

$$= \sigma_{AR} \{ \tau_o \} \left(\frac{\tau}{\tau_o} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$\sigma_{AR} \{ \tau_o \}$ = standard deviation of the wind azimuth angle in degrees
at height z_R and for the reference time period τ_o

m = power-law exponent for the vertical profile of the
standard deviation of the wind azimuth angle in the
surface layer

$$= \log \left(\frac{\sigma'_{ATK} \{ \tau, K=1 \}}{\sigma'_{AR} \{ \tau \}} \right) / \log \left(\frac{z_{TK} \{ K=1 \}}{z_R} \right)$$

$$\sigma'_{ATK} \{ \tau, K=1 \} = \sigma_{ATK} \{ \tau_o, K=1 \} \left(\frac{\tau}{\tau_o} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$\sigma_{ATK} \{ \tau_o, K=1 \}$ = standard deviation of the wind azimuth angle in degrees
at the top of the surface layer z_{TK} for the reference
time period τ_o

For layers above the surface ($K > 1$),

$$\sigma'_{ATK} \{ \tau, K > 1 \} = \left(\sigma'_{ABK} \{ \tau \} + \sigma'_{ABK} \{ \tau \} \right) / 2 \quad (3-6)$$

where

$$\sigma'_{ATK} \{ \tau \} = \sigma_{ATK} \{ \tau_o \} \left(\frac{\tau}{\tau_o} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$$\sigma_{ATK} \{ \tau_o \} = \text{standard deviation of the wind azimuth angle in degrees at the top of the layer for the reference time period } \tau_o$$

$$\sigma'_{ABK} \{ \tau \} = \sigma_{ABK} \{ \tau_o \} \left(\frac{\tau}{\tau_o} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$$\sigma_{ABK} \{ \tau_o \} = \text{standard deviation of the wind azimuth angle in degrees at the base of the layer for the reference time period } \tau_o$$

$$x_K = \text{downwind distance from the source}$$

$$y_K = \text{crosswind distance from the axis of the cloud}$$

$$x_{yK} = \text{crosswind virtual distance}$$

$$= \frac{\sigma_{yo} \{ K \}}{\sigma_{AK} \{ \tau \}} - x_{Ry}$$

$$\text{when } \sigma_{yo} \{ K \} \leq \sigma'_{AK} \{ \tau \} x_{ry}$$

$$= \alpha_K x_{ry} \left(\frac{\sigma_{yo} \{ K \}}{\sigma'_{AK} \{ \tau \} x_{ry}} \right)^{1/\alpha_K} - x_{Ry} + x_{ry} (1 - \alpha_K)$$

when $\sigma_{y0} \{K\} > \sigma'_{AK} \{ \tau \} x_{ry}$

$\sigma_{y0} \{K\}$ = standard deviation of the lateral source dimension
in the layer at downwind distance x_{Ry}

x_{Ry} = distance from the source at which $\sigma_{y0} \{K\}$ is measured

x_{ry} = distance over which rectilinear crosswind expansion
occurs downwind from an ideal point source

α_K = lateral diffusion coefficient in the layer

$\Delta\theta'_K$ = vertical wind direction shear in the layer

$$= \left(\theta_{TK} - \theta_{BK} \right) \left(\frac{\pi}{180} \right)$$

θ_{TK} = mean wind direction in degrees at the top of the layer

θ_{BK} = mean wind direction in degrees at the base of the layer

3.1.2 Concentration Equation for Model 1

The maximum concentration for Model 1 in the K^{th} layer is given by the expression

$$x_K \{x_K, y_K, z_K\} = \frac{D_K \bar{u}_K}{\sqrt{2\pi} \sigma_{xK}} \quad (3-7)$$

where

σ_{xK} = standard deviation of the alongwind concentration
distribution in the layer

$$= \left[\left(\frac{L\{x_K\}}{4.3} \right)^2 + \sigma_{xo}^2 \{K\} \right]^{1/2} \quad (3-8)$$

$L\{x_K\}$ = alongwind cloud length for a point source in the layer at the distance x_K from the source

$$= \left\{ \begin{array}{ll} \frac{0.28 (\Delta \bar{u}_K) (x_K)}{\bar{u}_K} ; \Delta \bar{u}_K \geq 0 \\ \frac{0.28 (|\Delta \bar{u}_K|) (x_K)}{\bar{u}_K} ; \Delta \bar{u}_K < 0, \frac{\Delta \Phi}{\Delta z} \{K\} < 0 \\ 0 ; \Delta \bar{u}_K < 0, \frac{\Delta \Phi}{\Delta z} \{K\} \geq 0 \end{array} \right\} \quad (3-9)$$

$\Delta \bar{u}_K$ = vertical wind-speed shear in the layer

$\Delta \bar{u}_K \{K=1\}$ = $\bar{u}_{TK} \{K=1\} - \bar{u}_R$

$\Delta \bar{u}_K \{K>1\}$ = $\bar{u}_{TK} - \bar{u}_{BK}$

$\frac{\Delta \Phi}{\Delta z} \{K\}$ = lapse rate of virtual potential temperature in the layer

$\sigma_{xo} \{K\}$ = standard deviation of the alongwind source dimension in the layer at the point of cloud stabilization

The above equation for $L \{x_K\}$ is based on the theoretical and empirical results reported by Tyldesley and Wallington (1965) who analyzed ground-level concentration measurements made at distances of 5 to 120 kilometers downwind from instantaneous line-source releases.

The maximum centerline concentration for Model 1 in the K^{th} layer is given by the expression

$$x_{CK} \{x_K, y_K = 0, z_K\} = x_K \left\{ \exp \left(\frac{-y_K^2}{2 \sigma_{yK}^2} \right) \right\}^{-1} \quad (3-10)$$

The average alongwind concentration is defined as

$$\bar{x}_K = D_K / t_{pK} \quad (3-11)$$

where

$$\begin{aligned} t_{pK} &= \text{cloud passage time in seconds in the } K^{th} \text{ layer} \\ &\cong 4.3 \sigma_{xK} / \bar{u}_K \end{aligned}$$

The time mean alongwind concentration in the K^{th} layer is defined by the expression

$$x_K \{x_K, y_K, z_K; T_A\} = \frac{D_K}{T_A} \left\{ \operatorname{erf} \left(\frac{\bar{u}_K T_A}{2 \sqrt{2} \sigma_{xK}} \right) \right\} \quad (3-12)$$

where

$$T_A = \text{time in seconds over which concentration is to be averaged}$$

The time mean alongwind concentration is equivalent to the average alongwind concentration when t_{pK} equals T_A .

3.2 MODEL 2

Layer Model 2 refers to the same source configuration as Model 1 in which the source extends vertically through the entire depth of the layer and the distribution of toxic material is uniform with height. In Model 2, however, it is assumed that no turbulent mixing is occurring. Consequently, there is no dilution of the cloud due to turbulent expansion. The dosage and concentration equations for Model 2 are given by Equations (3-1) and (3-7), respectively, with the following substitutions:

$$\sigma_{yK} = \sigma_{y0} \{K\} \quad (3-13)$$

$$\sigma_{xK} = \sigma_{x0} \{K\} \quad (3-14)$$

3.3 MODEL 3

This layer model differs from Models 1 and 2 in that the vertical extent of the source is less than the depth of the layer. The model equation thus contains vertical expansion terms.

3.3.1 Dosage Equation for Model 3

The dosage equation for Model 3 in the K^{th} layer is given by the expression

$$D_K \{x_K, y_K, z_{BK} < z_K < z_{TK}\} = \frac{Q_K}{2\pi \sigma_{yK} \sigma_{zK} \bar{u}_K} \left\{ \exp \left[-\frac{1}{2} \left(\frac{y_K}{\sigma_{yK}} \right)^2 \right] \right\} \quad (3-15)$$

$$\left\{ \sum_{i=0}^{\infty} \left[\gamma_r^i \left[\exp \left(-\frac{1}{2} \left(\frac{2i(z_{TK} - z_{BK}) + (H_K - z_K)}{\sigma_{zK}} \right)^2 \right) \right] \right] \right\}$$

$$\begin{aligned}
& + \gamma_r^{i+1} \left[\exp \left(- \frac{1}{2} \left(\frac{2i(z_{TK} - z_{BK}) + (H_K - 2z_{BK} + z_K)}{\sigma_{zK}} \right)^2 \right) \right] \\
& + \sum_{i=1}^{\infty} \left[\gamma_r^i \left[\exp \left(- \frac{1}{2} \left(\frac{2i(z_{TK} - z_{BK}) - (H_K - z_K)}{\sigma_{zK}} \right)^2 \right) \right] \right. \\
& \left. + \gamma_r^{i-1} \left[\exp \left(- \frac{1}{2} \left(\frac{2i(z_{TK} - z_{BK}) - (H_K - 2z_{BK} + z_K)}{\sigma_{zK}} \right)^2 \right) \right] \right] \Bigg\} \quad (3-15)
\end{aligned}$$

where

- Q_K = source strength or total mass of material in the layer
 H_K = effective source height or height of the centroid of the stabilized cloud
 σ_{zK} = standard deviation of the vertical dosage distribution in the layer
 γ_r = fraction of material reflected at the surface z_{BK} (1 for complete reflection and 0 for no reflection) and 0^0 is defined to be equal to unity for convenience in writing Equation (3-15)

The remaining terms are the same as those in Equation (3-1) for Model 1.

The standard deviation of the vertical dosage distribution is defined by the expression

$$\sigma_{zK} = \sigma'_{EK} x_{rz} \left(\frac{x_K + x_{zK} - x_{rz} (1 - \beta_K)}{\beta_K x_{rz}} \right)^{\beta_K} \quad (3-16)$$

where

σ'_{EK} = mean standard deviation of the wind elevation angle
in radians for the layer

x_{zK} = vertical virtual distance in the layer

β_K = vertical diffusion coefficient in the layer

x_{rz} = distance over which rectilinear vertical expansion
occurs downwind from an ideal point source

In the surface layer ($K = 1$),

$$\sigma'_{EK} \{K=1\} = \frac{\sigma_{ER} \left[(z_{TK} \{K=1\})^{q+1} - (z_R)^{q+1} \right]}{(q+1) (z_{TK} \{K=1\} - z_R) (z_R)^q} \left(\frac{\pi}{180} \right) \quad (3-17)$$

where

σ_{ER} = standard deviation of the wind elevation angle in degrees
at the height z_R

q = power-law exponent for the vertical profile of the
standard deviation of the wind elevation angle in the
surface layer

$$= \log \left(\frac{\sigma_{ETK} \{K=1\}}{\sigma_{ER}} \right) / \log \left(\frac{z_{TK} \{K=1\}}{z_R} \right)$$

$\sigma_{ETK} \{K=1\}$ = standard deviation of the wind elevation angle in degrees at the top of the surface layer

Above the surface layer ($K > 1$),

$$\sigma'_{EK} \{K > 1\} = (\sigma_{ETK} + \sigma_{EBK}) \left(\frac{\pi}{360} \right)$$

where

σ_{ETK} = standard deviation of the wind elevation angle in degrees at the top of the layer

σ_{EBK} = standard deviation of the wind elevation angle in degrees at the base of the layer

The vertical virtual distance x_{zK} is given by the expression

$$\left\{ \begin{array}{l} \frac{\sigma_{zo} \{K\}}{\sigma'_{EK}} - x_{Rz} \quad ; \quad \sigma_{zo} \{K\} \leq \sigma'_{EK} x_{rz} \\ \beta_K x_{rz} \left(\frac{\sigma_{zo} \{K\}}{\sigma'_{EK} x_{rz}} \right)^{1/\beta_K} - x_{Rz} + x_{rz} (1 - \beta_K) ; \quad \sigma_{zo} \{K\} > \sigma'_{EK} x_{rz} \end{array} \right\}$$

where

$\sigma_{zo} \{K\}$ = standard deviation of the vertical dosage distribution at x_{Rz}

x_{Rz} = distance from the source at which $\sigma_{zo} \{K\}$ is measured in the K^{th} layer

3.3.2

Concentration Equation for Model 3

The concentration equation for Model 3 is the same as that for Model 1 which is given by Equation (3-7) in Section 3.1.2 with D_K from Equation (3-15). Equation (3-10) in Section 3.1.2 also gives the maximum centerline concentration for Model 3. Similarly, average and time mean alongwind concentrations for Model 3 are given by Equations (3-11) and (3-12) with D_K from Equation (3-15).

3.4 MODEL 4

Model 4, the layer-breakdown model, may be used to calculate concentration and dosage fields resulting from changes in the meteorological layer structure. Model 4 may also be used to determine concentration and dosage fields in the surface mixing layer downwind from a source in which the source strength varies with height in the layer and/or where there are spatial differences in source locations within a layer. The application of Model 4 requires the following assumption:

- The boundary between adjacent layers or sublayers is eliminated and the layers are replaced by a single layer L
- Turbulent mixing is occurring in layer L
- The material in each of the layers or sublayers is initially uniformly distributed in the vertical
- Reflection occurs at the upper and lower boundaries of layer L

The selection of Model 4 for layer breakdown calculations or to accommodate vertical source strength variations and/or spatial differences in source locations in the surface mixing layer is controlled in the computer program by selection of certain options (see Appendix B) available in the input configuration. If no special provision is made and Model 4 is specified for use, the program assumes that the function of the model is to accommodate to vertical source strength variations or

spatial differences in source locations. For example, the surface mixing layer can be divided into several sublayers where the source strength, although assumed to be vertically uniform in the K^{th} sublayer, increases with height in subsequent layers. Also, because of wind speed and direction shear in the sublayers, portions of the exhaust cloud may be located at different positions in the horizontal plane at the time of cloud stabilization. In this case, Model 4 calculates the contribution from each sublayer to the composite concentration and dosage fields in the surface mixing layer by permitting turbulent mixing across the initial sublayer boundaries.

If Model 4 is to be used to predict the concentration and dosage fields downwind from a change in meteorological structure, the program option ISKIP(7) must be properly set, the input parameter NBK must be initialized, and the meteorological parameters for the new L^{th} layer and the time t^* at which layer breakdown occurs must be specified (see Appendix B).

3.4.1 Dosage Equation for Model 4

The dosage equation for Model 4 for the contribution from the portion of the cloud in the K^{th} layer to the receptor position in the layer L is given by the expression

$$D_{LK} = \frac{Q_K}{2\sqrt{2\pi} \bar{u}_L \sigma_{yLK} (z_{TK} - z_{BK})} \left\{ \exp \left[-\frac{1}{2} \left(\frac{y_L}{\sigma_{yLK}} \right)^2 \right] \right\} \\ \left(\sum_{i=0}^{\infty} \left[\gamma_r^i \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) + z_{TK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) - z_{BL} + z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \right. \right. \\ \left. \left. + \gamma_r^{i+1} \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) - 2z_{BL} + z_{TK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) + 2z_{BL} - z_{BK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \right] \right) \quad (3-18)$$

$$\begin{aligned}
& + \sum_{i=1}^{\infty} \left[\gamma_r^i \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) - z_{BK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) + z_{TK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \right. \\
& \left. + \gamma_r^{i-1} \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) + 2z_{BL} - z_{BK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) - 2z_{BL} + z_{TK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \right] \quad (3-18)
\end{aligned}$$

where, again for convenience, 0^0 is defined equal to unity.

The total contribution to a receptor position in layer L is calculated by summing the contributions from all K layers. A derivation of the vertical term (terms following the summation signs) in Equation (3-18) is presented in Appendix E. In the above expression

$$\begin{aligned}
Q_K &= \text{source strength in units of mass for the source in the layer K} \\
\bar{u}_L &= \text{mean cloud transport speed in the } L^{\text{th}} \text{ layer} \\
&= \frac{\sum_{K=1}^n \left\{ (z_{TK} - z_{BK}) (\bar{u}_{TK} + \bar{u}_{BK}) / 2 \right\}}{\sum_{K=1}^n (z_{TK} - z_{BK})} \quad (3-19) \\
\bar{u}_{TK} &= \text{wind speed at the top of the } K^{\text{th}} \text{ layer} \\
\bar{u}_{BK} &= \text{wind speed at the base of the } K^{\text{th}} \text{ layer}
\end{aligned}$$

and n is the number of sublayers in layer L.

The crosswind distance from the axis of the cloud to a receptor y_L (defined positive to the right looking downwind) is given by the expression

$$y_L = (y_j - y_{SK}) \sin \theta'_L - (x_j - x_{SK}) \cos \theta'_L \quad (3-20)$$

where

x_j, y_j = position of the receptor with respect to the origin of the reference coordinate system with the y axis positive northward and the x axis positive eastward

x_{SK}, y_{SK} = coordinates of the cloud centroid in the K^{th} layer at time t^* with respect to the origin of the reference coordinate system

$$x_{SK} = x_i - \bar{u}_K t^* \sin \theta'_K$$

$$y_{SK} = y_i - \bar{u}_K t^* \cos \theta'_K$$

x_i, y_i = coordinates of the source in the K^{th} layer with respect to the origin of the reference coordinate system

$$\theta'_L = \frac{\sum_{K=1}^n \{ (z_{TK} - z_{BK}) (\theta'_K) \}}{\sum_{K=1}^n (z_{TK} - z_{BK})} \quad (3-21)$$

$$\theta'_K = (\theta_{TK} + \theta_{BK}) \left(\frac{\pi}{360} \right)$$

θ_{TK} = wind direction at the top of the K^{th} layer in degrees

θ_{BK} = wind direction at the base of the K^{th} layer in degrees

The standard deviation of the crosswind dosage distribution σ_{yLK} in the L^{th} layer is defined by the expression

$$\sigma_{yLK} = \left\{ \left[\sigma'_{AL} \{ \tau \} x_{ry} \left(\frac{x_L + x_{yKL}^* - x_{ry} (1 - \alpha_L)}{\alpha_L x_{ry}} \right)^{\alpha_L} \right]^2 + \left[\frac{\Delta \theta'_L x_L}{4.3} \right]^2 \right\}^{1/2} \quad (3-22)$$

where

$\sigma'_{AL} \{ \tau \}$ = mean layer standard deviation of the wind azimuth angle in radians for the effective cloud stabilization time τ

In the surface layer ($L = 1$),

$$\sigma'_{AL} \{ \tau, L=1 \} = \frac{\sigma'_{ARL} \{ \tau \} \left[(z_{TL} \{ L=1 \})^{m_L + 1} - (z_R)^{m_L + 1} \right]}{(m_L + 1) (z_{TL} \{ L=1 \} - z_R) (z_R)^{m_L}} \quad (3-23)$$

where

$\sigma'_{ARL} \{ \tau \}$ = standard deviation of the wind azimuth angle in radians at height z_R and for time τ

$$= \sigma_{\text{ARL}} \{ \tau_0 \} \left(\frac{\tau}{\tau_0} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$$\sigma_{\text{ARL}} \{ \tau_0 \} = \text{standard deviation of the wind azimuth angle in degrees at height } z_R \text{ and for the reference time period } \tau_0$$

$$m_L = \text{power-law exponent for the vertical profile of the standard deviation of the wind azimuth angle in the surface layer } L = 1$$

$$m_L = \frac{\log \left(\frac{\sigma_{\text{ATL}} \{ \tau, L=1 \}}{\sigma_{\text{ARL}} \{ \tau \}} \right)}{\log \left(\frac{z_{\text{TL}} \{ L=1 \}}{z_R} \right)}$$

$$\sigma'_{\text{ATL}} \{ \tau, L=1 \} = \sigma_{\text{ATL}} \{ \tau_0, L=1 \} \left(\frac{\tau}{\tau_0} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$$\sigma_{\text{ATL}} \{ \tau_0, L=1 \} = \text{standard deviation of the wind azimuth angle in degrees at the top of the surface layer } z_{\text{TL}} \text{ for the reference time period } \tau_0$$

For layers above the surface layer ($L > 1$),

$$\sigma'_{\text{AL}} \{ \tau, L > 1 \} = \left(\sigma'_{\text{ATL}} \{ \tau \} + \sigma'_{\text{ABL}} \{ \tau \} \right) / 2 \quad (3-24)$$

where

$$\sigma'_{\text{ATL}} \{ \tau \} = \sigma_{\text{ATL}} \{ \tau_0 \} \left(\frac{\tau}{\tau_0} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$\sigma_{ATL} \{ \tau_o \}$ = standard deviation of the wind azimuth angle in degrees
at the top of the layer for the reference time period
 τ_o

$$\sigma'_{ABL} \{ \tau \} = \sigma_{ABL} \{ \tau_o \} \left(\frac{\tau}{\tau_o} \right)^{1/5} \left(\frac{\pi}{180} \right)$$

$\sigma_{ABL} \{ \tau_o \}$ = standard deviation of the wind azimuth angle in degrees
at the base of the layer for the reference time τ_o

The wind-direction shear in radians in the layer is
given by the expression

$$\Delta \theta'_L = \left(\theta_{TL} - \theta_{BL} \right) \left(\frac{\pi}{180} \right)$$

where

θ_{TL} = mean wind direction in degrees at the top of the
layer z_{TL}

θ_{BL} = mean wind direction in degrees at the base of the layer
 z_{BL}

The crosswind virtual distance in the L^{th} layer due to source
(cloud) originating in the K^{th} layer is given by the expression

$$x_{yKL}^* = x_{ry} \left(\frac{\sigma_{yKL}^*}{\sigma_{AL} \{r\} x_{ry}} \right)^{1/\alpha_L} + x_{ry} (1 - \alpha_L)$$

where

- σ_{yKL}^* = crosswind source dimension in Layer L due to source (cloud) originating in the K^{th} layer
- = $\left\{ \left[(\sigma_{xK}^*)^2 \sin^2 (\theta'_K - \theta'_L) \right] + \left[(\sigma_{yK}^*)^2 \cos^2 (\theta'_K - \theta'_L) \right] \right\}^{1/2}$
- σ_{xK}^* = alongwind standard deviation of the dosage distribution in the K^{th} layer at time t^*
- σ_{yK}^* = crosswind standard deviation of the dosage distribution in the K^{th} layer at time t^*
- α_L = lateral diffusion coefficient in the layer
- x_{ry} = distance over which rectilinear crosswind expansion occurs downwind from an ideal point source

The downwind distance from the point where the change in layer structure occurs for the source (cloud) in the K^{th} layer to the point where the dosage is to be calculated x_L is given by the expression

$$x_L = - (x_j - x_{SK}) \sin \theta'_L - (y_j - y_{SK}) \cos \theta'_L \quad (3-25)$$

The standard deviation of the vertical dosage distribution σ_{zLK} in the L^{th} layer is defined by the expression

$$\sigma_{zLK} = \sigma_{EL}^j x_{rz} \left(\frac{x_L}{x_{rz}} \right)^{\beta_L} \quad (3-26)$$

where

$$\begin{aligned} \sigma_{EL}^j &= \text{mean standard deviation of the wind elevation angle} \\ &\quad \text{in radians for the layer} \\ \beta_L &= \text{vertical diffusion coefficient in the layer} \\ x_{rz} &= \text{distance over which rectilinear vertical expansion} \\ &\quad \text{occurs downwind of an ideal point source} \end{aligned}$$

In the surface layer ($L = 1$),

$$\sigma_{EL}^j \{L=1\} = \frac{\sigma_{ERL} \left[(z_{TL} \{L=1\})^{q_L+1} - (z_R)^{q_L+1} \right]}{(q_L+1) (z_{TL} \{L=1\} - z_R) (z_R)^{q_L}} \left(\frac{\pi}{180} \right) \quad (3-27)$$

where

$$\begin{aligned} \sigma_{ERL} &= \text{standard deviation of the wind elevation angle in degrees} \\ &\quad \text{at the reference height } z_R \\ q_L &= \text{power-law exponent for the vertical profile of the} \\ &\quad \text{standard deviation of the wind elevation angle in the} \\ &\quad \text{surface layer} \\ &= \log \left(\frac{\sigma_{ETL} \{L=1\}}{\sigma_{ERL}} \right) / \log \left(\frac{z_{TL} \{L=1\}}{z_R} \right) \\ \sigma_{ETL} \{L=1\} &= \text{standard deviation of the wind elevation angle in} \\ &\quad \text{degrees at the top of the layer } z_{TL} \end{aligned}$$

Above the surface layer ($L > 1$),

$$\sigma'_{EL} \{L > 1\} = \left(\sigma_{ETL} + \sigma_{EBL} \right) \left(\frac{\pi}{360} \right) \quad (3-28)$$

where

σ_{ETL} = standard deviation of the wind elevation angle in degrees at the top of the layer z_{TL}

σ_{EBL} = standard deviation of the wind elevation angle in degrees at the base of the layer z_{BL}

3.4.2 Concentration Equation for Model 4

The maximum concentration equation for Model 4 is given by the expression

$$x_{LK} \{x_L, y_L, z_L\} = \frac{D_{LK} \bar{u}_L}{\sqrt{2\pi} \sigma_{xLK}} \quad (3-29)$$

where

σ_{xLK} = standard deviation of the cloud alongwind concentration distribution in the layer

$$= \left\{ \left(\frac{L \{x_{LK}\}}{4.3} \right)^2 + \left(\sigma_{xKL}^* \right)^2 \right\}^{1/2}$$

$L \{x_{LK}\}$ = alongwind cloud length of a point source at distance x_L

$$= \left\{ \begin{array}{ll} \frac{0.28 (\Delta \bar{u}_L) x_L}{\bar{u}_L} & ; \Delta \bar{u}_L \geq 0 \\ \frac{0.28 (|\Delta \bar{u}_L|) (x_L)}{\bar{u}_L} & ; \Delta \bar{u}_L < 0 ; \frac{\Delta \Phi}{\Delta z} \{L\} < 0 \\ 0 & ; \Delta \bar{u}_L < 0 ; \frac{\Delta \Phi}{\Delta z} \{L\} \geq 0 \end{array} \right\} \quad (3-30)$$

$\Delta \bar{u}_L$ = vertical wind speed shear in the layer

$$= \bar{u}_{TL} - \bar{u}_{BL}$$

$\frac{\Delta \Phi}{\Delta z} \{L\}$ = lapse rate of potential temperature in the layer

σ_{xKL}^* = alongwind source dimension in layer L due to source (cloud) originating in the Kth layer

$$= \left\{ \left[(\sigma_{xK}^*)^2 \cos^2 (\theta_K' - \theta_L') \right] + \left[(\sigma_{yK}^*)^2 \sin^2 (\theta_K' - \theta_L') \right] \right\}^{1/2}$$

The maximum centerline concentration for Model 4 in the Lth layer is given by the expression

$$\begin{aligned} x_{CLK} \{x_{LK}, y_{LK}=0, z_{LK}\} &= x_{LK} / \langle \text{LATERAL TERM} \rangle \\ &= x_{LK} \left\{ \exp \left[- \left(\frac{y_L^2}{2\sigma_{yLK}^2} \right) \right] \right\}^{-1} \end{aligned} \quad (3-31)$$

The average alongwind concentration at the cloud centerline is defined as

$$\bar{x}_{LK} = D_{LK} / t_{pL} \quad (3-32)$$

where

$$\begin{aligned} t_{pL} &= \text{cloud passage time in seconds in the } L^{\text{th}} \text{ layer} \\ &= 4.3 \sigma_{xLK} / \bar{u}_L \end{aligned}$$

The time mean alongwind concentration in the L^{th} layer for averaging time T_A is defined by the expression

$$\bar{\chi}_K \{x_{LK}, y_{LK}, z_{LK}; T_A\} = \frac{D_{LK}}{T_A} \left\{ \text{erf} \left(\frac{\bar{u}_L T_A}{2 \sqrt{2} \sigma_{xLK}} \right) \right\} \quad (3-33)$$

3.5 MODEL 5

This model is used to calculate the amount of material deposited on the surface by precipitation scavenging in the K^{th} layer. The ground-level deposition WD_K due to precipitation scavenging, for the case in which the vertical distribution of toxic material in the layer is uniform with height, is given by the expression

$$\begin{aligned} WD_K \{x_K, y_K, z=0\} &= \frac{\Lambda Q_K}{\sqrt{2\pi} \sigma_{yK} \bar{u}_K} \left\{ \exp \left[-\frac{1}{2} \left(\frac{y_K}{\sigma_{yK}} \right)^2 \right] \right\} \\ &\quad \left\{ \exp \left[-\Lambda \left(\frac{x_K}{\bar{u}_K} - t_1 \right) \right] \right\} \end{aligned} \quad (3-34)$$

where

$$\begin{aligned} Q_K &= \text{source strength in units of mass for the source in layer } K \\ t_1 &= \text{time precipitation begins} \\ \Lambda &= \text{percent of material removed per unit time} \end{aligned}$$

The principal assumptions made in deriving the above expression are:

- The rate of precipitation is steady over an area that is large compared to the horizontal dimension of the cloud of toxic material
- The precipitation originates at a level above the top of the toxic cloud so that hydrometeors pass vertically through the entire cloud
- The time duration of the precipitation is sufficiently long so that the entire alongwind length of the toxic cloud passes over the point x

Engelmann (see Slade, 1968, pp. 208-221) discusses the general problems of calculating the amount of material removed by precipitation scavenging and recommends values of the coefficient Λ that may be combined with precipitation rates to obtain estimates of total surface deposition. Other useful information may be obtained from the proceedings of the 1970 Symposium on Precipitation Scavenging (Engelmann and Slinn, 1970), from Pellett (1974) and from Knutson, *et al.*, (1974).

When changes in layer structure occur at time t^* , the contribution to ground deposition WD_{LK} due to precipitation scavenging in the K^{th} layer is given by the expression

$$WD_{LK} \{x_L, y_L, z=0\} = \frac{\Lambda Q_K}{\sqrt{2\pi} \sigma_{yLK} \bar{u}_L} \left\{ \exp \left[-\frac{1}{2} \left(\frac{y_L}{\sigma_{yLK}} \right)^2 \right] \right\} \quad (3-35)$$

$$\left\{ \exp \left[-\Lambda \left(\frac{x_L}{\bar{u}_L} + t^* - t_1 \right) \right] \right\}$$

Maximum ground-level deposition at a point $(x_L, y_L, z=0)$ assuming no previous cloud depletion due to scavenging, can be obtained by setting the second exponential term in Equation (3-35) to unity. Total ground deposition is obtained by summing the contributions from all layers through which precipitation is falling at points on the reference grid coordinate system. The height of the top of the uppermost layer through which precipitation is falling z_{lim} must be supplied as an input to the computer program.

The dosage or concentration at a point in space, assuming precipitation scavenging occurs, is obtained by multiplying the appropriate dosage or concentration equation by the exponential term in Equation (3-34) or (3-35) containing the coefficient Λ .

3.6 MODEL 6

This model is used to calculate the ground deposition due to gravitational settling. The basic source configuration is an area source of finite lateral extent and unit vertical extent. Other source configurations are treated by summing the deposition at the ground resulting from a number of basic sources arranged to simulate the desired configuration. The model is essentially a tilted plume model in which the effects of wind shear are taken into account. The axis of a particle or droplet cloud of a given settling velocity intersects the ground plane at a distance from the source and at an angle from the mean surface wind direction that are proportional to the total angular wind shear and the residence time of the settling material in the layers between the source and the ground surface. In any layer, the inclination of the cloud axis from the horizontal is given by $\tan^{-1} V_s / \bar{u}$, where V_s is the particle or droplet settling velocity and \bar{u} is the mean transport wind speed in the layer. In all cases, material released in the K^{th} layer and dispersed upwards by turbulence is assumed to be reflected downwards at the interface of the K^{th} and $(K + 1)^{th}$ layers. The basic model is used to calculate the ground-level deposition pattern for a single value of the settling velocity. The total deposition

pattern is obtained by summing the results for all settling velocities representative of the particle or droplet-size distribution of the released material on a reference coordinate grid system.

In the computer program, provision is made for calculating deposition from a source which fills the layer in the vertical and for a source in which the vertical extent is less than the depth of the layer. These models are described below.

I

3.6.1 Gravitational Deposition Model for a Source that Extends Vertically Through the Entire Layer

Ground-level deposition by gravitational settling for a source that extends vertically through the entire layer and in which the material is uniformly distributed in the vertical is calculated by summing contributions from a number of elementary sources in the K^{th} layer. Deposition at the surface for a single elementary source at height H_{nK} in the layer is given by the expression

$$DEP_{nK} = \frac{f_i Q_K T_K}{2\pi \sigma_{ynK} \zeta_K} \left\{ M_{nK} + N_{nK} \right\} \left\{ \exp \left[- \frac{1}{2} \left(\frac{y_s}{\sigma_{ynK}} \right)^2 \right] \right\} \quad (3-36)$$

where

- f_i = fraction of particles or droplets with settling velocity V_s
- Q_K = source emission rate in layer K (g sec^{-1})
- T_K = source emission time in layer K
- ζ_K = number of elementary sources in layer K for simulating a uniform vertical distribution

- y_s = lateral distance from the deposition axis of particles or droplets with settling velocity V_s
- = $R_s \sin \phi_s$
- R_s = radial distance in the horizontal plane from the source to a receptor
- ϕ_s = angle between the axis of the ground-level deposition pattern and the radial connecting source and receptor for settling velocity V_s

The terms M_{nK} and N_{nK} are vertical terms that include provision for reflection from the boundary between the K^{th} and $(K+1)^{th}$ layers. These terms are defined by the expressions

$$\begin{aligned}
 M_{nK} + N_{nK} = & (1-\gamma_r) \left\{ \left[\frac{\bar{\beta}_K H_{nK} + (1-\bar{\beta}_K) V_s x_s / \bar{u}_{nK}}{\sigma'_{EnK}(x_s)^{1+\bar{\beta}_K}} \right] \left[\exp \left(-\frac{1}{2} \left(\frac{H_{nK} - (V_s x_s / \bar{u}_{nK})}{\sigma'_{EnK}(x_s)^{1+\bar{\beta}_K}} \right)^2 \right) \right] \right. \\
 & + \sum_{a=1}^{\infty} \gamma_r^{a-1} \left\{ \left[\frac{\bar{\beta}_K (2a z_{TK} - H_{nK}) - (1-\bar{\beta}_K) V_s x_s / \bar{u}_{nK}}{\sigma'_{EnK}(x_s)^{1+\bar{\beta}_K}} \right] \left[\exp \left(-\frac{1}{2} \left(\frac{2a z_{TK} - H_{nK} + (V_s x_s / \bar{u}_{nK})}{\sigma'_{EnK}(x_s)^{1+\bar{\beta}_K}} \right)^2 \right) \right] \right. \\
 & \left. \left. + \gamma_r \left[\frac{\bar{\beta}_K (2a z_{TK} + H_{nK}) + (1-\bar{\beta}_K) V_s x_s / \bar{u}_{nK}}{\sigma'_{EnK}(x_s)^{1+\bar{\beta}_K}} \right] \left[\exp \left(-\frac{1}{2} \left(\frac{2a z_{TK} + H_{nK} - (V_s x_s / \bar{u}_{nK})}{\sigma'_{EnK}(x_s)^{1+\bar{\beta}_K}} \right)^2 \right) \right] \right\} \right\} \quad (3-37)
 \end{aligned}$$

where

$$x_s = R_s \cos \phi_s$$

$$\bar{u}_{nK} = \text{mean wind transport speed in the layer between } H_{nK} \text{ and the ground}$$

$$= \frac{(X_{nK}^2 + Y_{nK}^2)^{1/2}}{H_{nK}} V_s$$

$$X_{nK} = \frac{\bar{u}_{HK}}{V_s b_K} \left\{ \sin \left[b_K (H_{nK} - z_{BK}) + S\theta'_{K-1} \right] - \sin(S\theta'_{K-1}) \right\} \\ + \sum_{i=1}^{K-1} \left\{ \frac{\bar{u}_i}{V_s b_i} \left[\sin(S\theta'_i) - \sin(S\theta'_{i-1}) \right] \right\}$$

$$Y_{nK} = \frac{\bar{u}_{HK}}{V_s b_K} \left\{ \cos \left[b_K (H_{nK} - z_{BL}) + S\theta'_{K-1} \right] - \cos(S\theta'_{K-1}) \right\} \\ + \sum_{i=1}^{K-1} \left\{ \frac{-\bar{u}_i}{V_s b_i} \left[\cos(S\theta'_i) - \cos(S\theta'_{i-1}) \right] \right\}$$

$$S\theta'_{K-1} = \sum_{i=1}^{K-1} \Delta\theta'_i$$

$$S\theta'_K = \sum_{i=1}^K \Delta\theta'_i$$

$$b_K = \frac{S\theta'_K - S\theta'_{K-1}}{z_{TK} - z_{BK}}$$

The quantity \bar{u}_{HK} is the mean layer wind speed between the height H_{nK} and the base of the K^{th} layer. The following expressions define the mean layer wind speeds in the surface layer ($K = 1$) and the layers above the surface layer ($K > 1$):

$$\bar{u}_{HK}\{K=1\} = \frac{\bar{u}_R \left[(H_{nK}\{K=1\})^{1+p} - (z_R)^{1+p} \right]}{(1+p) (H_{nK}\{K=1\} - z_R) (z_R)^p} \quad (3-38)$$

$$\bar{u}_{HK}\{K>1\} = \left[\left(\frac{\bar{u}_{TK} - \bar{u}_{BK}}{z_{TK} - z_{BK}} \right) \left(\frac{H_{nK} - z_{BK}}{2} \right) \right] + \left[\frac{\bar{u}_{BK}}{2} \right] \quad (3-39)$$

The mean standard deviation of the wind elevation angle in radians in the layer between H_{nK} and the base of the K^{th} layer is given by the expressions

$$\sigma'_{EnK}\{K=1\} = \frac{\sigma_{ER} \left[(H_{nK}\{K=1\})^{1+q} - (z_R)^{1+q} \right]}{(1+q) (H_{nK}\{K=1\} - z_R) (z_R)^q} \left(\frac{\pi}{180} \right) \quad (3-40)$$

$$\begin{aligned} \sigma'_{EnK}\{K>1\} = & \frac{1}{H_{nK}} \left\{ \left[\sigma'_{EnK}\{K=1\} \right] + \left[\sum_{i=2}^{K-1} \sigma'_{Ei} (z_{Ti} - z_{Bi}) \right] \right. \\ & \left. + \frac{\pi (H_{nK} - z_{BK})}{360} \left[\left(\frac{\sigma_{ETK} - \sigma_{EBK}}{z_{TK} - z_{BK}} \right) (H_{nK} - z_{BK}) + \sigma_{EBK} \right] \right\} \quad (3-41) \end{aligned}$$

The vertical diffusion coefficient in the layer between H_{nK} and the base of the K^{th} layer is given by the terms

$$\bar{\beta}_K\{K=1\} = \beta_K \quad (3-42)$$

The mean lateral diffusion coefficient in the layer between H_{nK} and the surface is given by the terms

$$\left\{ \begin{array}{l} \bar{\alpha}_K\{K=1\} = \alpha_K \\ \bar{\alpha}_K\{K>1\} = \frac{1}{H_{nK}} \left\{ \left[\sum_{i=1}^{K-1} \alpha_i (z_{Ti} - z_{Bi}) \right] + \left[\alpha_K (H_{nK} - z_{BK}) \right] \right\} \end{array} \right\} \quad (3-47)$$

The number of elementary sources ξ_K required to simulate a uniformly distributed source in the vertical is given by the expression

$$\xi_K = (z_{TK} - z_{BK}) / \Delta h_K \quad (3-48)$$

where

Δh_K = vertical separation of elementary sources in the K^{th} layer

$$= R_c \sigma'_{EH} \left(X_{HK}^2 + Y_{HK}^2 \right)^{1/2} \left(1 + \frac{V_s}{\bar{u}_{HK}} \right)^{1/2}$$

R_c = a constant value depending on the accuracy desired in simulating a vertical line source configuration. A value of $R_c = 0.45$ yields deposition estimates that are within 10 percent of the true value

$$\sigma'_{EH} = \sigma'_{EnK} \text{ with } H_{nK} = \frac{1}{3} (z_{TK} - z_{BK}) + z_{BK}$$

$$X_{HK} = X_{nK} \text{ with } H_{nK} = \frac{1}{3} (z_{TK} - z_{BK}) + z_{BK}$$

$$Y_{NK} = Y_{nK} \text{ with } H_{nK} = \frac{1}{3} (z_{TK} - z_{BK}) + z_{BK}$$

$$\bar{u}_{HK} = \bar{u}_{nK} \text{ with } H_{nK} = \frac{1}{3} (z_{TK} - z_{BK}) + z_{BK}$$

The computer program for calculating gravitational deposition automatically distributes ξ_K sources in the K^{th} layer with uniform vertical spacing. The height H_{nK} in the above equations is the height above the ground of each elementary source.

The angle between the axis of the ground-level deposition pattern and the radial connecting source and receptor for settling velocity V_s is defined by the expressions

$$\begin{aligned}\phi_s &= \left| \theta_1 - 180 + \phi_s - \theta_R \right| \quad (0 < \theta_1 < 180) \\ \phi_s &= \left| \theta_1 + 180 + \phi_s - \theta_R \right| \quad (180 < \theta_1 < 360)\end{aligned}\tag{3-49}$$

where

θ_1 = mean wind direction at the reference height z_R

θ_R = angle between north and a line connecting source and receptor

$$\phi_s = \tan^{-1} \left(\frac{Y_{nK}}{X_{nK}} \right)$$

3.6.2 Gravitational Deposition Model for a Volume Source in the K^{th} Layer

For a volume source at height H_{SK} in the K^{th} layer, the ground-level deposition from gravitational settling is given by the expression

$$DEP_{SK} = \frac{f_i Q_{SK}}{2\pi \sigma_{ySK}} \left\{ M_{SK} + N_{SK} \right\} \left\{ \exp \left[- \frac{1}{2} \left(\frac{y_{SK}}{\sigma_{ySK}} \right)^2 \right] \right\} \tag{3-50}$$

where the subscript SK indicates that the parameters refer to a single source in the K^{th} layer. The subset of equations which define the SK subscripted parameters is the same as the subset defining the terms in Equation (3-36), except the following substitution is made for the term x_s appearing in Equation (3-37):

$$x_s = R_{SK} \cos \phi_{SK} + x_{zSK} \quad (3-51)$$

where

x_{zSK} = the vertical virtual distance for the volume source

$$= \left(\frac{\sigma_{zo}\{SK\}}{\sigma'_{ESK}} \right)^{1/\beta_K}$$

σ'_{ESK} = mean standard deviation of the wind elevation angle
in the layer between H_{SK} and the ground

$\sigma_{zo}\{SK\}$ = vertical source dimension of the volume source

In using Equation (3-50), deposition patterns from all values of V_{SK} representative of the particle or droplet size distribution of the volume source are summed on a reference coordinate system to obtain the total deposition pattern.

SECTION 4

DESCRIPTION OF THE NASA/MSFC MULTILAYER MODEL COMPUTER PROGRAMS

The NASA/MSFC computer programs described in this technical document consist of a Preprocessor Program and the main NASA/MSFC Multilayer Diffusion Models Program - Version 5. Both programs are written in FORTRAN IV and are designed for execution on a UNIVAC 1108 computer. In addition, the automated plotting routines in Version 5 of the program are designed to provide input to the Stromberg-Carlson (SC 4020) machine at MSFC. This section briefly describes the general characteristics of the two programs.

4.1 PREPROCESSOR PROGRAM

The Preprocessor Program is designed to automatically calculate source and meteorological model inputs for use in Models 3 and 4 of the NASA/MSFC Multilayer Diffusion Models Program - Version 5. Requisite inputs to the Preprocessor Program include vertical profiles of meteorological data of the type available from radiosonde soundings, surface turbulence data available at KSC from meteorological towers, and logic information such as the type of launch and vehicle for which the calculation is being performed. The program can currently perform calculations for the following vehicles:

- Space Shuttle
- Titan III
- Delta-Thor
- Minuteman II

Calculations performed by the Preprocessor include:

- Time-height profile of the rise of hot exhaust products contained in the ground cloud

- Position in space of the stabilized ground-cloud
- Source dimensions of the stabilized ground-cloud
- Distribution of pollutant products within the stabilized ground-cloud
- Turbulence input parameters

The cloud-rise models and algorithms used in computing this information are described in Section 2. The data decks produced on option by this program include a complete card deck for each of the four pollutants HCl, CO, CO₂ and Al₂O₃ for use in Models 3 and 4 of the NASA/MSFC Multilayer Diffusion Models Program. The program also provides the option to produce input data for normal and abnormal launches of all four vehicles. A complete description of the program with user's instructions is presented in Appendix A.

4.2 NASA/MSFC MULTILAYER DIFFUSION MODELS PROGRAM - VERSION V

The NASA/MSFC Multilayer Diffusion Models Program is designed to calculate the following quantities downwind from normal and abnormal launches of rocket vehicles:

- Concentration and dosage patterns
- Time - mean concentration patterns
- Average cloud concentration
- Time of cloud passage
- Ground-level deposition patterns due to gravitational settling or precipitation scavenging

Program options include the calculation of concentration, dosage and time-mean concentration patterns with partial reflection of material at the surface, with time-dependent exponential decay, and/or with depletion due to precipitation scavenging. Also, the program is capable of calculating ground-level deposition due to gravitational settling with partial reflection at the surface. Provision is also made (in Model 4) to account for changes in meteorological structure along the cloud trajectory. A

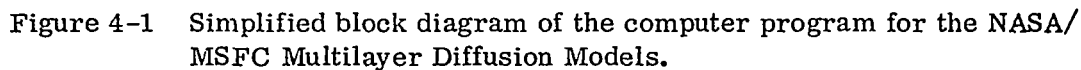
description of the mathematical specifications of the models included in the program is given in Section 3 above.

The multilayer model program is capable of accepting input data from cards from the Preprocessor Program or from input card data sets supplied by the user.

Program output options include:

- Printing of all data inputs
- Printing of the results of all model calculations
- Plotting of maximum centerline concentration, dosage, time-mean concentration and deposition versus distance along the cloud trajectory
- Plotting of concentration, dosage, time-mean concentration and deposition isopleths

A simplified block diagram illustrating major features of the program is shown in Figure 4-1. A description of the NASA/MSFC Multilayer Diffusion Models Program complete with user's instructions is contained in Appendix B.



SECTION 5

EXAMPLE CALCULATIONS

Example calculations have been made for both normal and abnormal launches of the Space Shuttle vehicle to illustrate the use of the Preprocessor Program in conjunction with the NASA/MSFC Multilayer Diffusion Models Program - Version 5. The meteorological data used in the example calculations are described in Section 5.1. The results of the calculations for Model 3 are given in Section 5.2 while those for Model 4 are given in Section 5.3. Computer printout of the example calculations is presented in Appendix D.

5.1 METEOROLOGICAL INPUT DATA

Ground-level concentrations and time-mean concentrations of HCl, CO and Al_2O_3 were calculated for normal and abnormal launches of the Space Shuttle using meteorological measurements made at Kennedy Space Center on 21 October 1972. Surface weather maps show that a cold front approached Florida on 19 October and passed the Cape on the morning of 20 October. By 21 October, the cold front was located just south of Florida. Meteorological data from a radiosonde released at 1115 Z on 21 October are given in Table 5-1. Figure 5-1 shows temperature, wind-speed and wind-direction profiles obtained from these data. The temperature profile shows temperature decreasing with height to about 1600 meters above the surface with the more rapid decrease occurring between 300 and 1400 meters. The wind speed increases from 6 meters per second near the surface to 11 meters per second at 750 meters, remains constant between 750 and 1432 meters, and then decreases with height. As shown in Table 5-1, the relative humidity increases with height to 1432 meters and then decreases. From this information, the depth of the surface mixing layer H_m , which is a required input to the Preprocessor Program, was set equal to 1432 meters. The surface density at the time of the radiosonde

TABLE 5-1
RADIOSONDE MEASUREMENTS FOR 1115 Z ON 21 OCTOBER 1972
USED IN THE EXAMPLE CALCULATIONS

Height (meters)	Wind Direction (degrees)	Wind Speed (m sec ⁻¹)	Temperature (°C)	Pressure (mb)	Relative Humidity (percent)
18	80	6	22.6	1022	57
194	81	8	22.2	1000	57
250	82	9	22.1	994	57
284	82	10	22.0	990	58
500	79	10	19.3	965	65
558	79	10	18.5	959	67
637	78	10	17.8	950	70
750	76	11	16.9	938	74
1000	71	11	14.6	911	86
1098	68	11	13.7	900	93
1135	67	11	13.3	896	94
1250	63	11	12.3	884	97
1432	56	11	10.7	865	97
1500	53	10	10.5	858	90
1577	49	10	10.3	850	79
1716	40	9	9.9	836	55
1750	37	8	10.3	832	55
2000	9	6	12.5	808	49
2259	344	5	11.1	784	44
2500	342	5	9.1	761	54

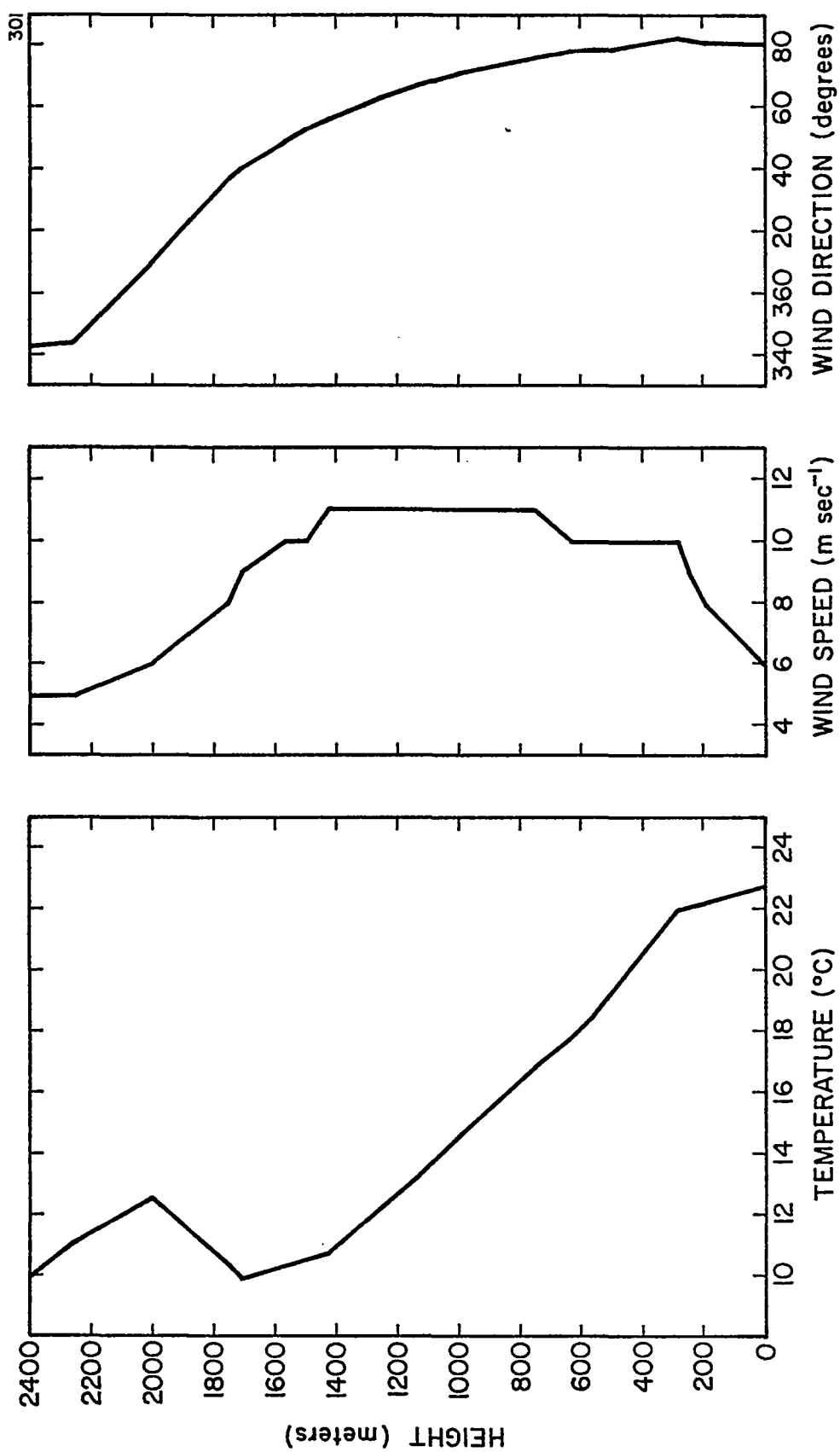


FIGURE 5-1. Vertical profiles of temperature, wind speed and wind direction at Kennedy Space Center for 21 October 1972, 1115 Z.

launch was 1197.07 grams per cubic meter. Bi-directional vane measurements from towers at KSC indicated that the reference standard deviation of the wind azimuth angle σ_{AR} for a ten-minute period was approximately 9 degrees at a height of 18 meters. Other meteorological input parameters required by the Preprocessor Program are supplied by the data given in Table 5-1.

5.2 RESULTS OF CALCULATIONS USING MODEL 3 OF THE NASA/MSFC MULTILAYER DIFFUSION MODELS PROGRAM

Table 5-2 contains the results of the Preprocessor Program calculations of the Model 3 inputs using Space Shuttle source data and the meteorological inputs for 21 October 1972 at KSC. The cloud stabilization height for a normal launch was calculated using Equation (2-2) while Equation (2-8) was used for the two abnormal launches. Effective heights of the stabilized cloud in the surface mixing layer are based on the relationships given by Equation (2-13). The source dimensions were calculated from Equations (2-11) and (2-12). Figure 5-2 shows the configuration of the exhaust product cloud at the time of stabilization for the normal launch. The stippled area in Figure 5-2 represents the dimensions of the stabilized cloud in the surface mixing layer used in the Model 3 calculations. The position of the stabilized cloud in relation to the launch pad and the time required for cloud stabilization are also given in Table 5-2. The source strengths of the three exhaust products HCl, CO and Al_2O_3 were calculated from Equation (2-32).

The results of the Model 3 calculations of the maximum centerline χ_c and ten-minute average $\chi \{10 \text{ min}\}$ HCl concentrations at the surface for a normal launch, single-engine burn and slow-burn on the pad are shown in Figures 5-3 through 5-5. Figure 5-3 shows, for a normal launch, a calculated peak maximum centerline HCl concentration of 0.8 parts per million (ppm) at a distance of about 12.5 kilometers downwind from the launch pad. The profile of centerline ($y=0$) ten-minute average HCl concentration, which is obtained by calculating the mean concentration over the ten-minute period of highest concentrations as the cloud passes a point in space, shows a maximum of 0.21 ppm HCl at a distance of 12.5 kilometers from the

TABLE 5-2

MODEL 3 INPUT PARAMETERS PRODUCED BY THE
PREPROCESSOR PROGRAM

Model 3 Input Parameters	Type of Launch		
	Normal	Single-Engine Burn	Slow Burn
Cloud Stabilization Height, z_m (m) Effective Height, H (m)	a) Cloud Height		
	1790	1910	1772
	1038	1194	1159
Dimension (m) $\sigma_{xo} = \sigma_{yo}$ σ_{zo}	b) Source Dimensions		
	533	444	412
	183	111	127
Time of Cloud Stabilization t^* (sec)	c) Time of Rise		
	447	400	452
Range, R_K (m) Azimuth, A_K (deg)	d) Cloud Position from Pad at Stabilization		
	4103	3646	4218
	235	230	235
Exhaust Product	e) Source Strength in the Surface Mixing Layer		
	8.092×10^9	8.808×10^9	2.555×10^{10}
	1.425×10^{10}	1.551×10^{10}	4.499×10^{10}
	1.802×10^{10}	1.961×10^{10}	5.700×10^{10}

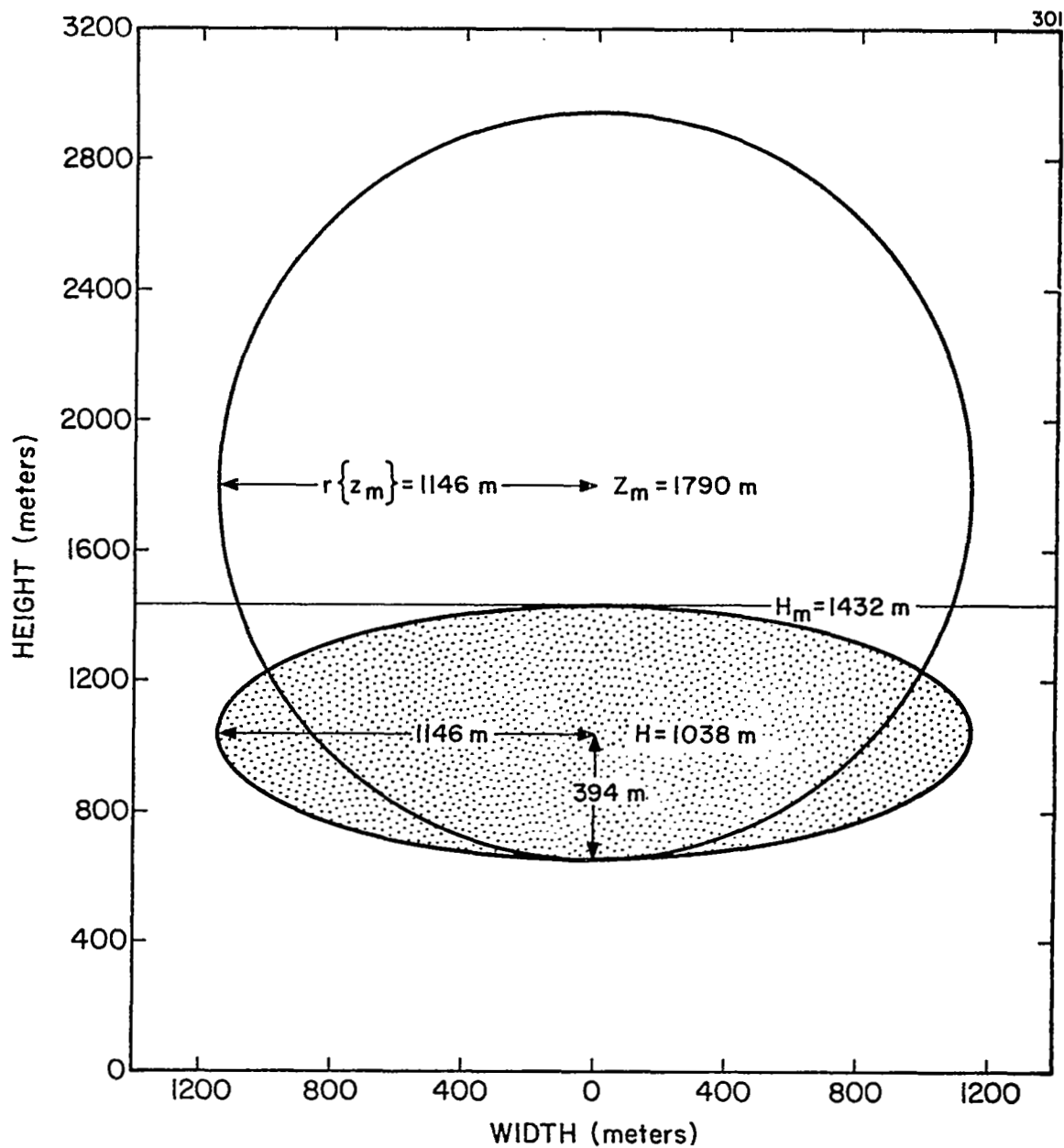


FIGURE 5-2. Configuration of the stabilized cloud of exhaust products used with Model 3 in calculations for a simulated normal launch of a Space Shuttle vehicle on 21 October 1972. Stippled area represents the effective cloud dimensions in the surface mixing layer.

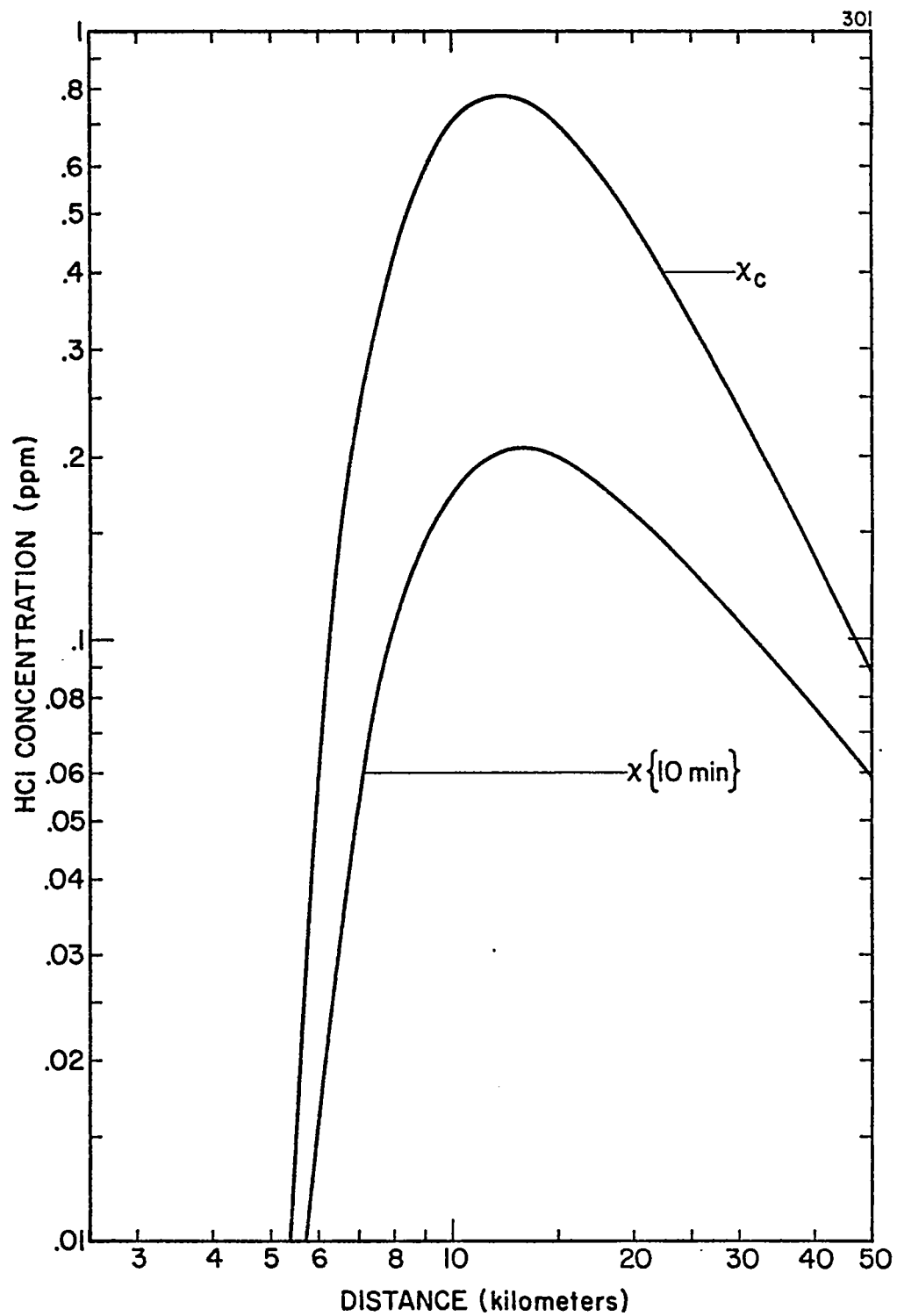


FIGURE 5-3. Maximum centerline χ_c and ten-minute average $\chi\{10 \text{ min}\}$ HCl concentrations at ground-level for the simulated normal launch of the Space Shuttle on 21 October 1972 using Model 3.

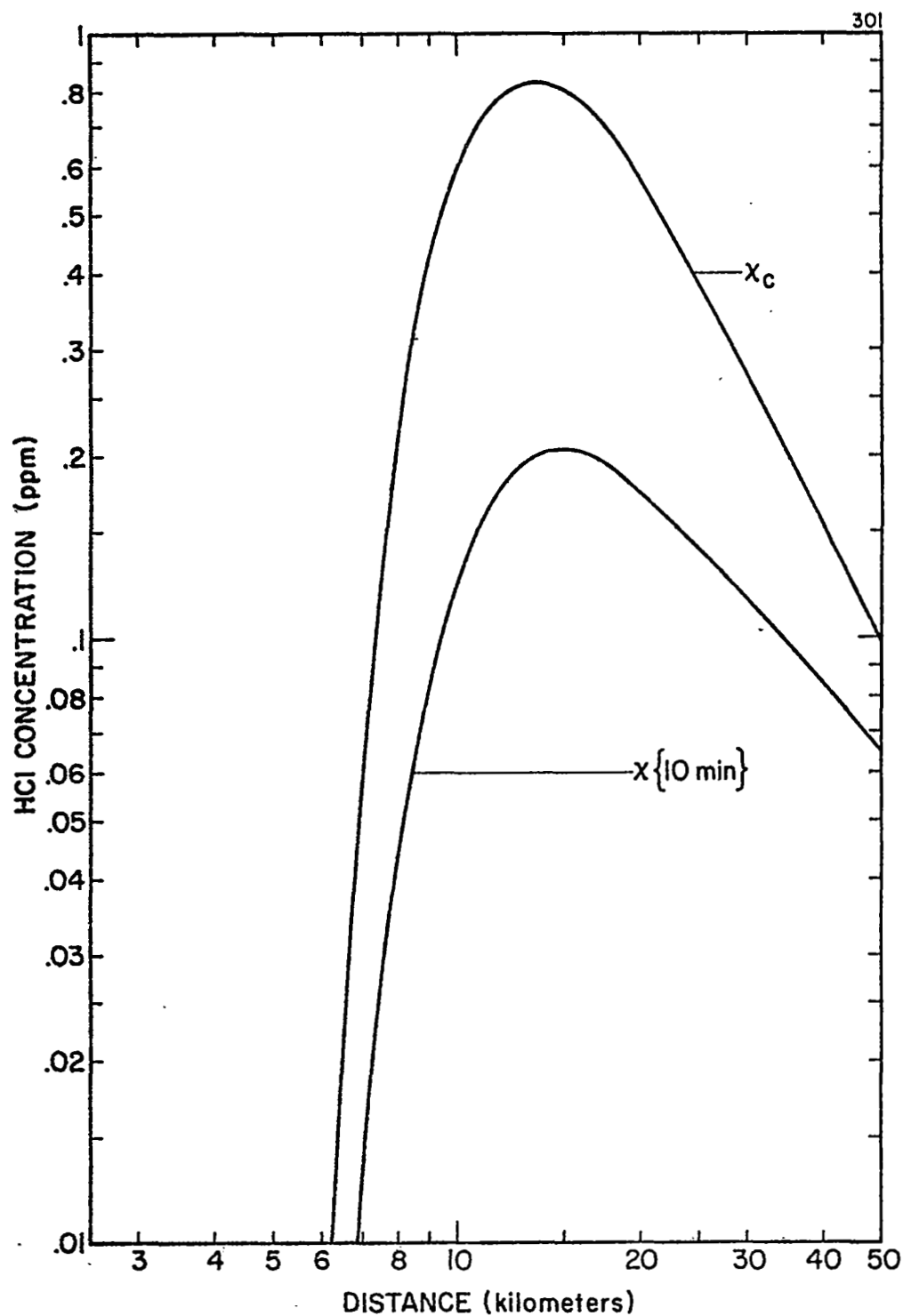


FIGURE 5-4. Maximum centerline x_c and ten-minute average $x\{10 \text{ min}\}$ HCl concentrations at ground-level for the simulated single-engine burn of the Space Shuttle on 21 October 1972 using Model 3.

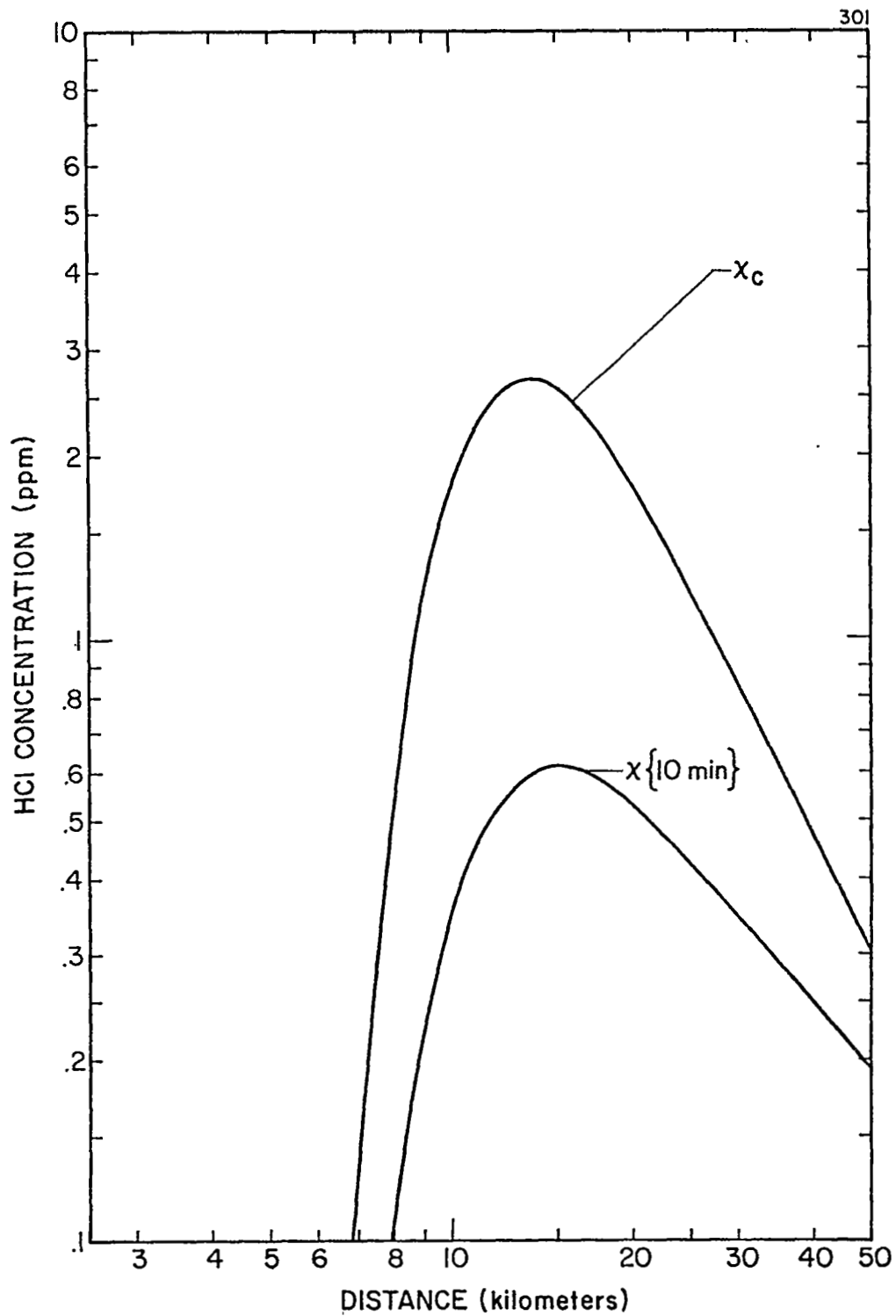


FIGURE 5-5. Maximum centerline x_c and ten-minute average $x \{10 \text{ min}\}$ HCl concentrations at ground-level for the simulated slow-burn of the Space Shuttle on 21 October 1972 using Model 3.

launch pad. The corresponding peak maximum centerline HCl concentration for a single-engine launch abort, as shown in Figure 5-4, is 0.84 ppm at a distance of about 13.5 kilometers from the launch pad. The maximum ten-minute average concentration of .21 ppm occurs at 15 kilometers from the pad. Comparison of the Model 3 calculations for the slow-burn shown in Figure 5-5 with the corresponding results for a normal launch and single-engine burn shows that ground-level concentrations are greater downwind from the launch pad following the slow-burn abort. The maximum centerline HCl concentration for the slow-burn case is 2.7 ppm HCl at 14 kilometers from the pad and the maximum ten-minute average concentration is 0.62 ppm HCl at 15 kilometers from the pad.

Concentration and dosage profiles and isopleths of concentration and dosage for HCl calculated using Model 3 can be found in the computer listing of example solutions in Appendix D.2.

5.3 RESULTS OF CALCULATIONS USING MODEL 4 OF THE NASA/ MSFC MULTILAYER DIFFUSION MODELS PROGRAM

Table 5-3 presents the results of the Preprocessor Program calculations of inputs to Model 4 calculations of HCl, CO and Al_2O_3 concentrations downwind from normal and abnormal simulated launches of the Space Shuttle vehicle on 21 October 1972 at Kennedy Space Center. The source strength distribution shown in Table 5-3 was calculated following the procedures described in Section 2.3.2 which utilizes Equations (2-27) through (2-34). Source dimensions for Model 4 were calculated using Equation (2-14) and the source positions at the time of cloud stabilization were calculated from Equations (2-15) through (2-26). Figure 5-6 shows the configuration of the exhaust cloud at the time of stabilization for the normal launch of the Space Shuttle used in the Model 4 calculations. The abscissa of Figure 5-6 is the range from the launch pad without consideration of the offset due to differences in the azimuth bearings from the launch pad. The calculated cloud stabilization heights and stabilization times for Model 4 are identical to those shown in Table 5-2 for Model 3 calculations.

TABLE 5-3

PREPROCESSOR PROGRAM CALCULATED SOURCE STRENGTHS, SOURCE
DIMENSIONS AND SOURCE POSITION FOR MODEL 4

Layer No.	Height of Layer Top (m)	Source Strength (mg)			Source Dimensions (m) $\sigma_{xo} = \sigma_{yo}$	Source Position	
		HCl	CO	Al ₂ O ₃		Range RK (m)	Azimuth AK (m)
1	194	4.423 x 10 ⁷	7.787 x 10 ⁷	9.849 x 10 ⁷	533	40	261
2	250	1.788 x 10 ⁷	3.147 x 10 ⁷	3.981 x 10 ⁷	533	67	261
3	284	1.383 x 10 ⁷	2.436 x 10 ⁷	3.080 x 10 ⁷	533	88	261
4	500	1.737 x 10 ⁸	3.059 x 10 ⁸	3.869 x 10 ⁸	533	279	261
5	558	8.535 x 10 ⁷	1.503 x 10 ⁸	1.901 x 10 ⁸	533	343	260
6	637	1.565 x 10 ⁸	2.756 x 10 ⁸	3.486 x 10 ⁸	533	439	260
7	750	3.305 x 10 ⁸	5.818 x 10 ⁸	7.359 x 10 ⁸	533	601	259
8	1000	1.407 x 10 ⁹	2.477 x 10 ⁹	3.133 x 10 ⁹	533	1054	257
9	1098	9.012 x 10 ⁸	1.587 x 10 ⁹	2.007 x 10 ⁹	533	1260	256
10	1135	4.021 x 10 ⁸	7.079 x 10 ⁸	8.954 x 10 ⁸	533	1341	255
11	1250	1.482 x 10 ⁹	2.609 x 10 ⁹	3.300 x 10 ⁹	533	1610	253
12	1432	3.078 x 10 ⁹	5.419 x 10 ⁹	6.854 x 10 ⁹	533	2084	250

a) Normal Launch

TABLE 5-3

PREPROCESSOR PROGRAM CALCULATED SOURCE STRENGTHS, SOURCE
DIMENSIONS AND SOURCE POSITION FOR MODEL 4

(Continued)

Layer No.	Height of Layer Top (m)	Source Strength (mg)			Source Dimensions (m) $\sigma_{x0} = \sigma_{y0}$	Source Position	
		HCl	CO	Al ₂ O ₃		Range RK (m)	Azimuth AK (m)

b) Single-Engine Burm

1	194	3.500×10^6	6.162×10^6	7.794×10^6	444	49	261
2	250	2.316×10^6	4.078×10^6	5.158×10^6	444	75	261
3	284	2.047×10^6	3.604×10^6	4.558×10^6	444	93	261
4	500	3.907×10^7	6.878×10^7	8.700×10^7	444	262	261
5	558	2.608×10^7	4.592×10^7	5.808×10^7	444	315	260
6	637	5.692×10^7	1.002×10^8	1.268×10^8	444	394	260
7	750	1.517×10^8	2.672×10^8	3.379×10^8	444	522	259
8	1000	9.836×10^8	1.732×10^9	2.190×10^9	444	867	257
9	1098	8.452×10^8	1.488×10^9	1.882×10^9	444	1019	256
10	1135	4.214×10^8	7.419×10^8	9.384×10^8	444	1078	255
11	1250	1.759×10^9	3.097×10^9	3.917×10^9	444	1272	254
12	1432	4.517×10^9	7.953×10^9	1.006×10^{10}	444	1605	251

TABLE 5-3

PREPROCESSOR PROGRAM CALCULATED SOURCE STRENGTHS, SOURCE
DIMENSIONS AND SOURCE POSITION FOR MODEL 4

(Continued)

Layer No.	Height of Layer Top (m)	Source Strength (mg)			Source Dimensions (m) $\sigma_{x0} = \sigma_{y0}$	Source Position	
		HCl	CO	Al ₂ O ₃		Range Rk (m)	Azimuth Ak (m)

e) Slow Burn

1	194	8.035 x 10 ⁶	1.415 x 10 ⁷	1.789 x 10 ⁷	412	61	261
2	250	5.793 x 10 ⁶	1.020 x 10 ⁷	1.290 x 10 ⁷	412	93	261
3	284	5.234 x 10 ⁶	9.215 x 10 ⁶	1.167 x 10 ⁷	412	117	261
4	500	1.073 x 10 ⁸	1.889 x 10 ⁸	2.390 x 10 ⁸	412	328	261
5	558	7.463 x 10 ⁷	1.314 x 10 ⁸	1.662 x 10 ⁸	412	395	260
6	637	1.662 x 10 ⁸	2.927 x 10 ⁸	3.702 x 10 ⁸	412	494	260
7	750	4.527 x 10 ⁸	7.971 x 10 ⁸	1.008 x 10 ⁹	412	655	259
8	1000	2.988 x 10 ⁹	5.261 x 10 ⁹	6.654 x 10 ⁹	412	1092	257
9	1098	2.553 x 10 ⁹	4.496 x 10 ⁹	5.686 x 10 ⁹	412	1286	256
10	1135	1.262 x 10 ⁹	2.223 x 10 ⁹	2.811 x 10 ⁹	412	1361	255
11	1250	5.187 x 10 ⁹	9.132 x 10 ⁹	1.155 x 10 ¹⁰	412	1611	254
12	1432	1.274 x 10 ¹⁰	2.243 x 10 ¹⁰	2.837 x 10 ¹⁰	412	2044	251

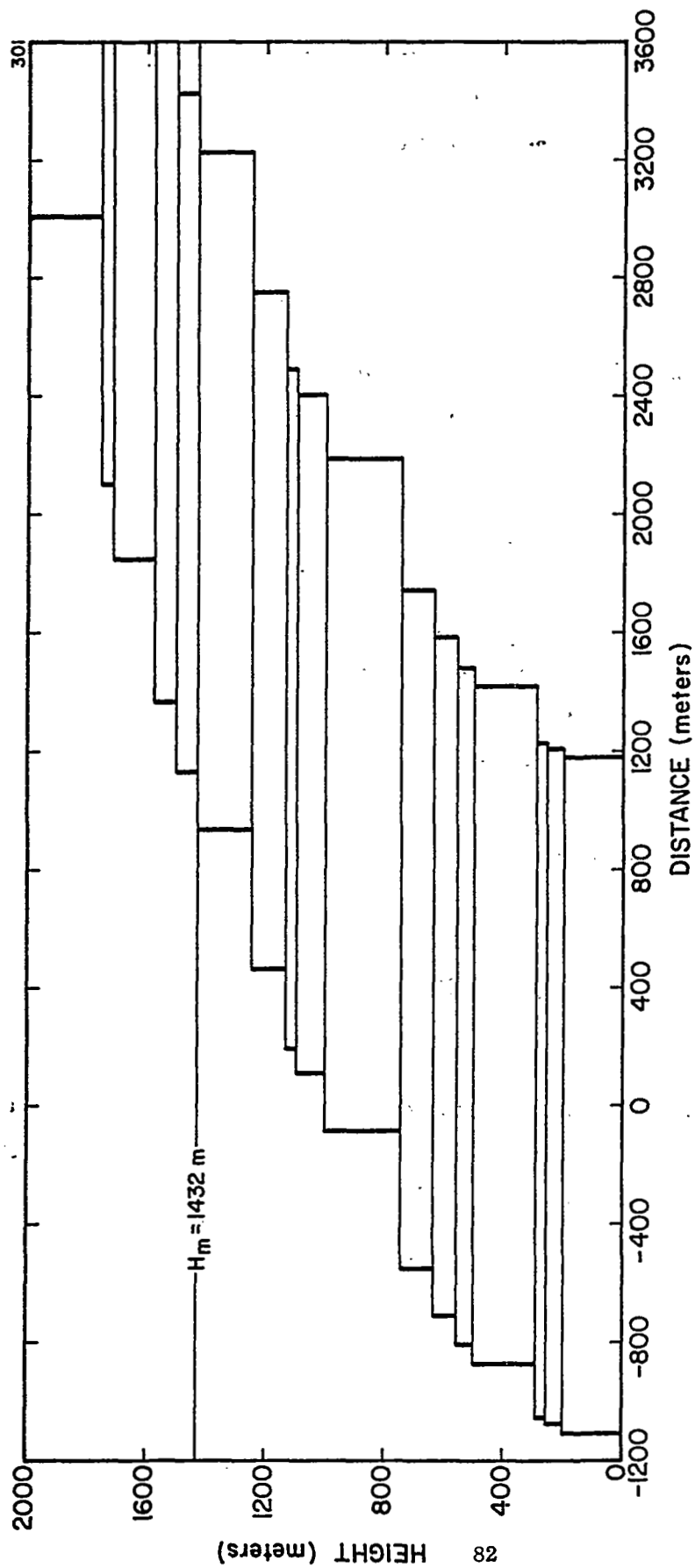


FIGURE 5-6. Configuration of the stabilized cloud of exhaust products used with Model 4 in calculations for a simulated normal launch of a Space Shuttle vehicle at Kennedy Space Center on 21 October 1972.

The results of the Model 4 calculations of the maximum centerline and ten-minute average HCl concentrations at the surface downwind from a simulated normal launch, single-engine burn and slow burn of the Space Shuttle vehicle are respectively shown in Figures 5-7, 5-8 and 5-9. The Model 4 HCl peak centerline concentrations χ_c are 0.6 ppm, .74 ppm and 2.25 ppm for a normal launch, single-engine burn and slow burn on the pad. All of these occur at a distance of approximately 12.5 kilometers downwind from the launch pad. Similarly, the maximum ten-minute average HCl concentrations are .17 ppm, 0.18 and 0.53 ppm respectively for a normal launch, single-engine burn and slow burn on the pad. All of these occur at a distance of approximately 15 kilometers downwind from the launch pad.

Table 5-4 shows a comparison of the results of the Model 4 calculations with those obtained from Model 3. The Model 3 maximum concentrations are 15 to 30 percent greater than the maximum concentrations calculated using Model 4. On the other hand, Model 4 predicts higher concentrations than Model 3 at distances close to the launch pad. These differences are a consequence of the two source configurations as shown in Figure 5-2 (Model 3) and Figure 5-6 (Model 4). Until accurate measurements of the vertical distribution of exhaust products in the stabilized cloud become available, model calculations of concentrations and dosages close to the launch pad are subject to this type of uncertainty. At longer downwind distances from the launch pad, as comparison of the Model 3 and Model 4 calculations shows, there is close agreement in the predicted concentrations because the effects of differing assumptions regarding the source configuration are small.

Calculated concentration and dosage profiles and isopleths of concentration and dosage for HCl obtained by using Model 4 are also presented in the computer listing of example solutions in Appendix D.2.

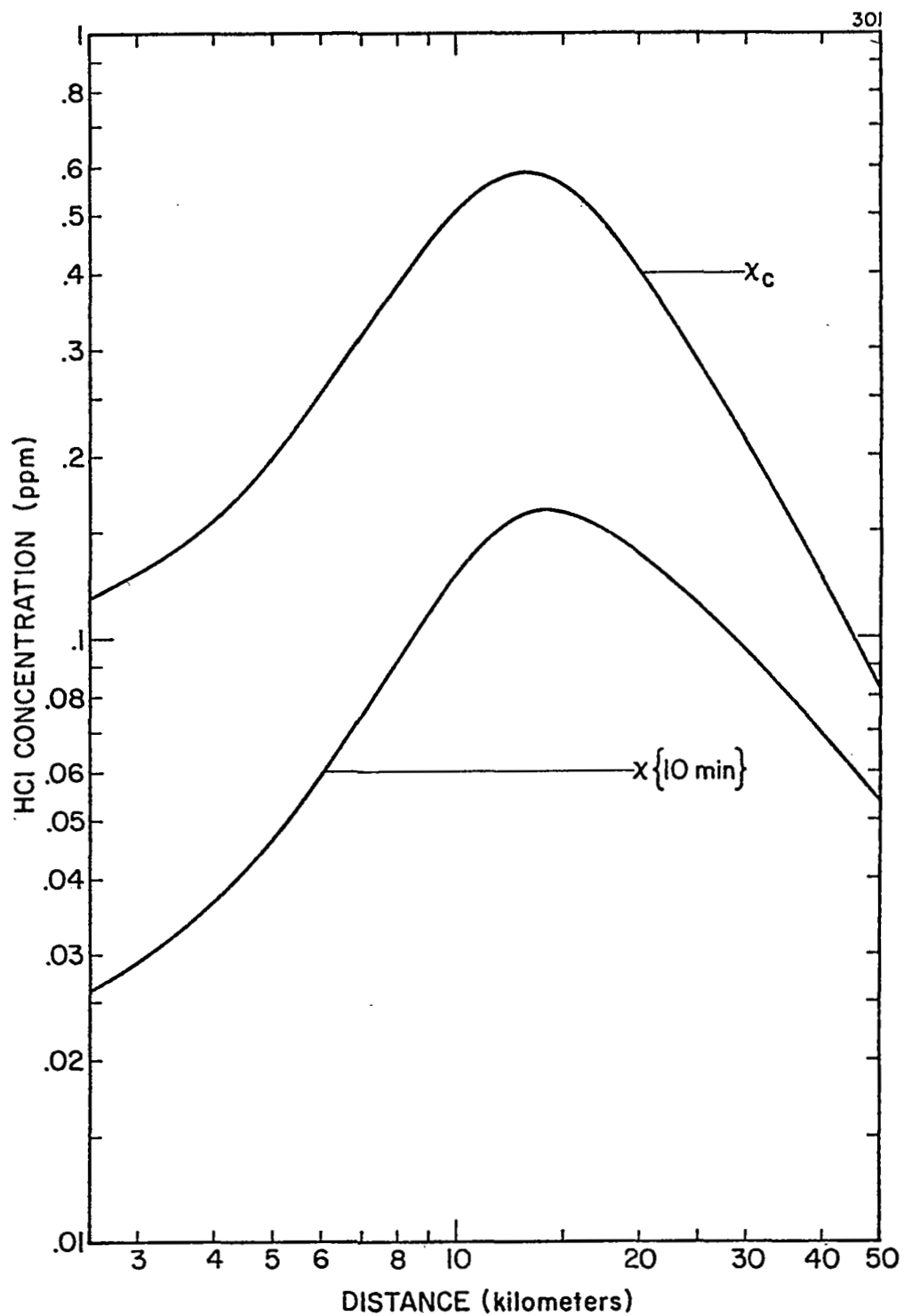


FIGURE 5-7. Maximum centerline χ_c and ten-minute average $\chi\{10 \text{ min}\}$ HCl concentrations at ground-level for the simulated normal launch of the Space Shuttle on 21 October 1972 using Model 4.

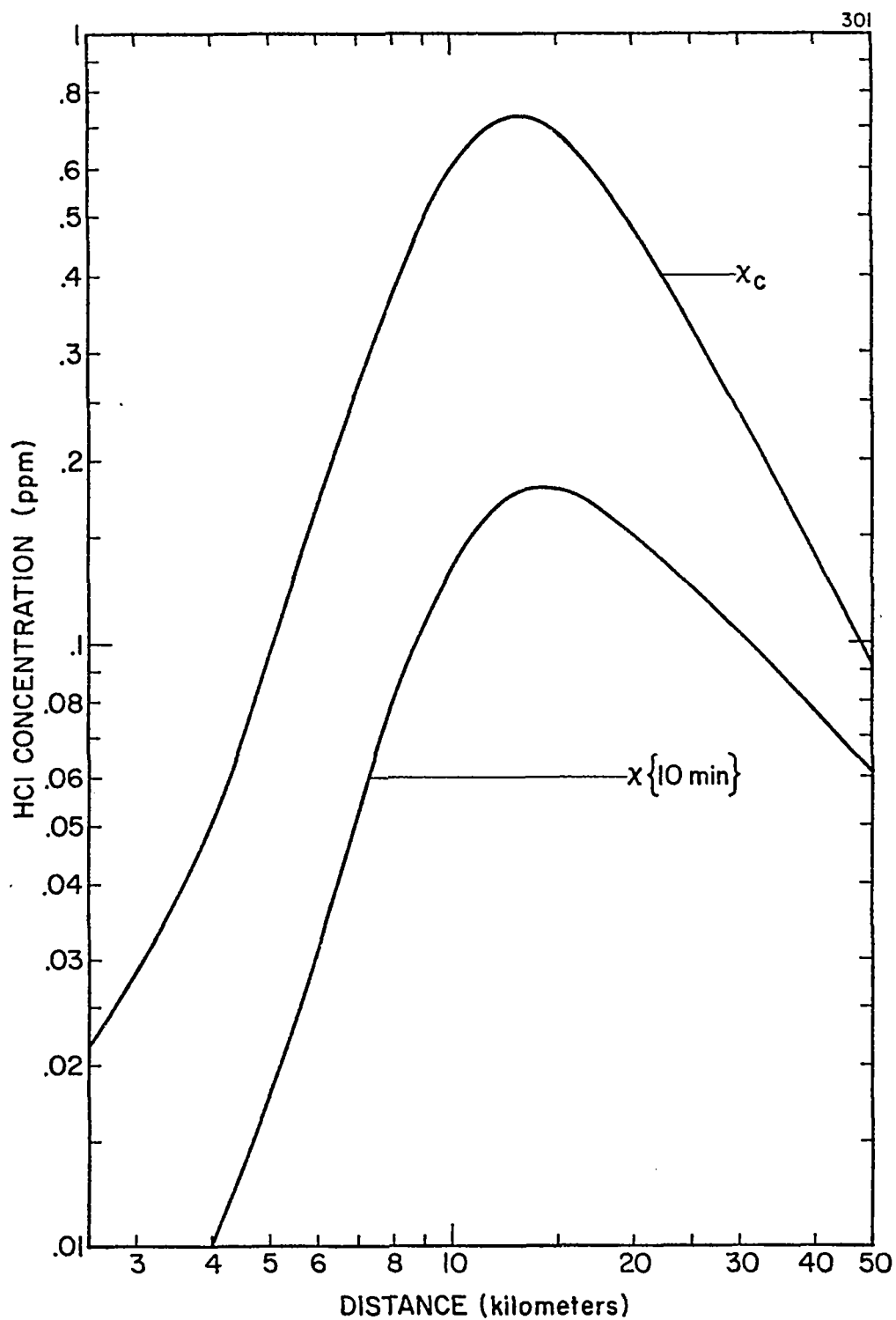


FIGURE 5-8. Maximum centerline x_c and ten-minute average $x\{10 \text{ min}\}$ HCl concentrations at ground-level for the simulated single-engine burn of the Space Shuttle on 21 October 1972 using Model 4.

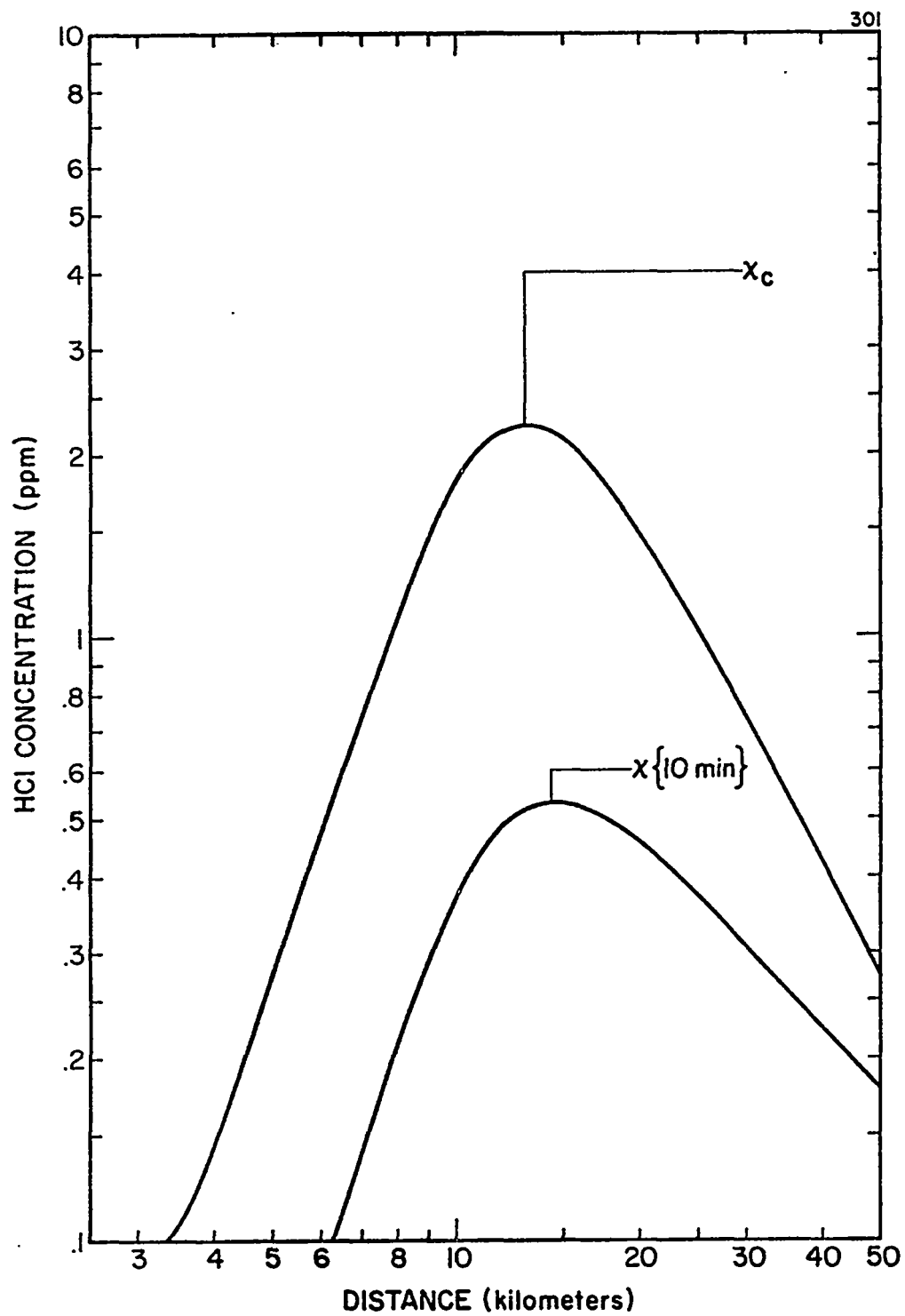


FIGURE 5-9. Maximum centerline x_c and ten-minute average $x\{10 \text{ min}\}$ HCl concentrations at ground-level for the simulated slow-burn of the Space Shuttle on 21 October 1972 using Model 4.

TABLE 5-4

COMPARISON OF PEAK MAXIMUM CENTERLINE χ_c AND
 TEN-MINUTE AVERAGE HCl CONCENTRATIONS (ppm) FROM
 MODEL 3 AND MODEL 4 CALCULATIONS

Model Number	Type of Launch					
	Normal		Single-Engine Burn		Slow Burn	
	χ_c	χ {10 min}	χ_c	χ {10 min}	χ_c	χ {10 min}
3	0.80	0.21	0.84	0.21	2.7	0.62
4	0.60	0.17	0.74	0.18	2.3	0.53

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APPENDIX A

USER INSTRUCTIONS FOR THE PREPROCESSOR PROGRAM FOR USE WITH THE NASA/MSFC MULTILAYER DIFFUSION MODELS COMPUTER PROGRAM -VERSION 5

The Preprocessor Program produces a complete set of data decks for input to the NASA/MSFC Multilayer Diffusion Models Computer Program - Version 5. The Program is specifically designed for use with launches of Space Shuttle, Titan III C, Delta-Thor, and Minuteman II vehicles. The data decks produced on option by this Program include a complete card deck for each of the four pollutants HCl, CO, CO₂ and Al₂O₃ for Models 3 and/or 4 in the NASA/MSFC Multilayer Diffusion Models Computer Program - Version 5.

The Preprocessor Program is written in FORTRAN IV and uses approximately 9700₁₀ locations of computer core storage. The Program requires card input, print and punch output and executes in less than 2 seconds on a UNIVAC 1108 computer. A complete computer listing is given in Appendix C.1 below.

A.1 PROGRAM INPUT PARAMETERS

The Preprocessor Program requires the input of the following meteorological parameters:

- | | | |
|---------------|---|--|
| σ_{AR} | - | Standard deviation of the wind azimuth angle in degrees measured at the first reference height z_1 over a 10 minute time period. |
| ρ | - | Ambient air density in grams per cubic meter measured at z_1 . |
| z | - | Height in feet or meters at which the meteorological measurements are taken. |

θ	-	Wind direction in degrees at z.
u	-	Wind speed in knots or meters per second at z.
T	-	Ambient air temperature in degrees Celsius at z.
P	-	Ambient air pressure in millibars at z.
RH	-	Relative humidity in percent at z.

The Preprocessor Program also requires control information indicating: (1) vehicle type, (2) whether the computer run is for a normal or abnormal launch, (3) whether z is in feet or meters, (4) whether u is in knots or meters per second, (5) height of the surface mixing layer which must coincide with one of the z inputs above, and (6) the model being used and the pollutants for which data decks will be produced.

A.2 PROGRAM INPUT DATA CARD SEQUENCE

The first card in the input data deck contains general case titling information and is used for a page heading in the Preprocessor print output and is also punched in the output data deck for input to the Multilayer Models Program. The second input card contains control information and σ_{AR} and ρ at the surface. An example set of data input cards is shown in Figure A-1. The example problem is described in Section 5 in the main body of this report.

Data Card 1:

Columns 1 - 72 - General data set titling information. If input as blanks the program will use the last information input

Data Card 2:

- Columns 2-4 - Punch these characters indicating the vehicle type.
If left blank Titan III C is assumed.
TTN is the Titan III C vehicle;
STL is the Space Shuttle vehicle;
DTH is the Delta-Thor vehicle;
MIN is the Minuteman II vehicle.
- Columns 5-7 - Punch YES or leave blank if the run is for a normal launch.
Punch NØ1 if the run is for an abnormal launch where a single engine burns on the launch pad. Not produced for the Delta-Thor and Minuteman II vehicles.
Punch NØ2 if the run is for an abnormal launch where a slow burn on the pad occurs.
(Ø is alphabetic)
- Column 8 - Punch M or leave blank if the heights z are in meters.
Punch F if the heights are in feet.
- Column 9 - Punch M or leave blank if the wind speed u is in meters per second.
Punch K if the wind speed is in knots.
- Columns 10-45 - Punch the date of the meteorological case or any case identification information (optional).
- Columns 46-55 - Punch σ_{AR} (see Figure A-1).
- Columns 56-65 - Punch ρ (see Figure A-1).
- Column 67 - Punch a 1 if output for Model 4 is desired; leave blank if not.
- Column 68 - Punch a 1 output for Model 3 is desired; leave blank if not.

Data Card 2 (Continued):

- | | | |
|-----------|---|--|
| Column 69 | - | Punch a 1 if output for HCl is desired; leave blank if not. |
| Column 70 | - | Punch a 1 if output for CO is desired; leave blank if not. |
| Column 71 | - | Punch a 1 if output for Al_2O_3 is desired; leave blank if not. |
| Column 72 | - | Punch a 1 if output for CO_2 is desired; leave blank if not. Produced only for the Titan III C vehicle. |
| Column 73 | - | Punch a 1 if cloud the trajectory range and azimuth bearing are to be calculated; leave blank if not. |

Data Cards 3 to N-1:

- | | | |
|---------------|---|--|
| Columns 1-10 | - | Punch z (see Figure A-1). |
| Columns 11-20 | - | Punch θ (see Figure A-1). |
| Columns 21-30 | - | Punch u (see Figure A-1). |
| Columns 31-40 | - | Punch T (see Figure A-1). |
| Columns 41-50 | - | Punch P (see Figure A-1). |
| Columns 51-60 | - | Punch RH (see Figure A-1). |
| Column 80 | - | Punch and asterisk (*) if this height z coincides with the surface mixing layer height; otherwise, leave blank. If not punched on any card, the program assumes the last z input to be the mixing layer height (See Figure A-1). |

Data Card N:

The last card in the input data deck must be a blank card. Multiple case capability is provided by punching a non-blank character in column 80 of this card. If column 80 is non-blank the Program expects another complete case, otherwise the program stops.

The Preprocessor Program assumes the decimal point in all of the above meteorological parameters to be between the sixth and seventh columns of the 10-column field if it is not punched. For example, if the 10-column field 41 to 50 contained the number $\Delta 121033 \Delta \Delta$ (Δ is a blank), the program would interpret it as 1210.32. To avoid improper alignment of a data value, the decimal should be punched. Also, unless it is certain that the cloud rise height will not exceed the mixing layer height, it is recommended that at least three heights z be input above the height coinciding with the mixing layer height. If the cloud rise height exceeds the last height input, the program will stop and ask for data at greater heights.

A.3 COMPUTER PROGRAM PUNCH OUTPUT

The Preprocessor Program will punch a complete card deck for direct input to the NASA/MSFC Multilayer Models - Version 5. The first card of each of the output data decks contains \$NAM2 and the last card contains \$END. The possible data decks punched are:

Inputs for:	(1)	HCl	-	Model 4
	(2)	CO	-	Model 4
	(3)	CO ₂	-	Model 4
	(4)	Al ₂ O ₃	-	Model 4
	(5)	HCl	-	Model 3
	(6)	CO	-	Model 3
	(7)	CO ₂	-	Model 3
	(8)	Al ₂ O ₃	-	Model 3

EXAMPLE SPACE SHUTTLE NORMAL LAUNCH									
S.TLYESMMSC 21 .OCT 72 .									
18.	80.	6.	22.	6	1022.	9.	119	7.07	1111111
194.	81.	8.	22.	2	1000.			57.	
250.	82.	9.	22.	1	993.66			57.	
284.	82.	10.	22.	0	989.79			58.	
500.	79.	10.	19.	3	965.4			65.	
558.	79.	10.	18.	5	958.94			67.	
637.	78.	10.	17.	8	950.			70.	
750.	76.	11.	16.	9	937.71			74.	
1000.	71.	11.	14.	6	910.61			86.	
1098.	68.	11.	13.	7	900.			93.	
1135.	67.	11.	13.	3	896.25			94.	
1250.	63.	11.	12.	3	884.09			97.	
1432.	56.	11.	10.	7	865.13			97.	*
1500.	53.	10.	10.	5	858.17			90.	
1577.	49.	10.	10.	3	850.			79.	
1716.	40.	9.	9.	9	836.31			55.	
1750.	37.	8.	10.	3	832.87			55.	
2000.	9.	6.	12.	5	808.34			49.	
2259.	344.	5.	11.	1	783.69			44.	
2500.	342.	5.	9.	1	761.39			54.	
BLANK CARD									
FIGURE A-1. Example computer program input for a normal launch of the Space Shuttle vehicle. The heights z are in meters and the wind speeds are in meters per second. The top of the mixing layer is at 1432 meters and the last card									

A.4 COMPUTER OUTPUT FOR THE EXAMPLE DATA

The Preprocessor Program first prints constant model parameters for the vehicle and prints the input data deck for verification. A complete deck for each selected pollutant HCl, CO, CO₂ and Al₂O₃ is printed and punched for Models 3 and/or 4 of the Multilayer Model. Also, the program lists all calculated parameters for Models 3 and/or 4 in a tabularized form for verification of Program calculations. A complete output listing for the example case shown in Figure A-1 is given in Appendix D.1.

A.5 LINKAGE DIAGRAM OF THE PREPROCESSOR PROGRAM

Figure A-2 shows the subroutine linkage diagram for the Preprocessor Program. Each line terminating at a subroutine name represents a subroutine call. The Program also references the FORTRAN library functions ACOS, COS, SIN, ATAN2, EXP not shown in Figure A-2. A description of the subroutines shown in Figure A-2 is given in Section A.6.

A.6 DESCRIPTION OF PROGRAM SUBROUTINES

Subroutine VEHICLE is the main controlling program in the Preprocessor. This routine sets the constants for the particular vehicle desired, inputs all data, converts inputs into proper units and controls all calculations and output.

Subroutine DIM34 calculates the source dimensions for Models 3 and/or 4 and calculates the effective cloud height (see Section 2, Equations (2-11) through (2-14)).

Subroutine PLUME1 calculates the plume rise for instantaneous sources (see Section 2.2.1).

Subroutine PLUME2 calculates the plume rise for continuous sources (see Section 2.2.2).

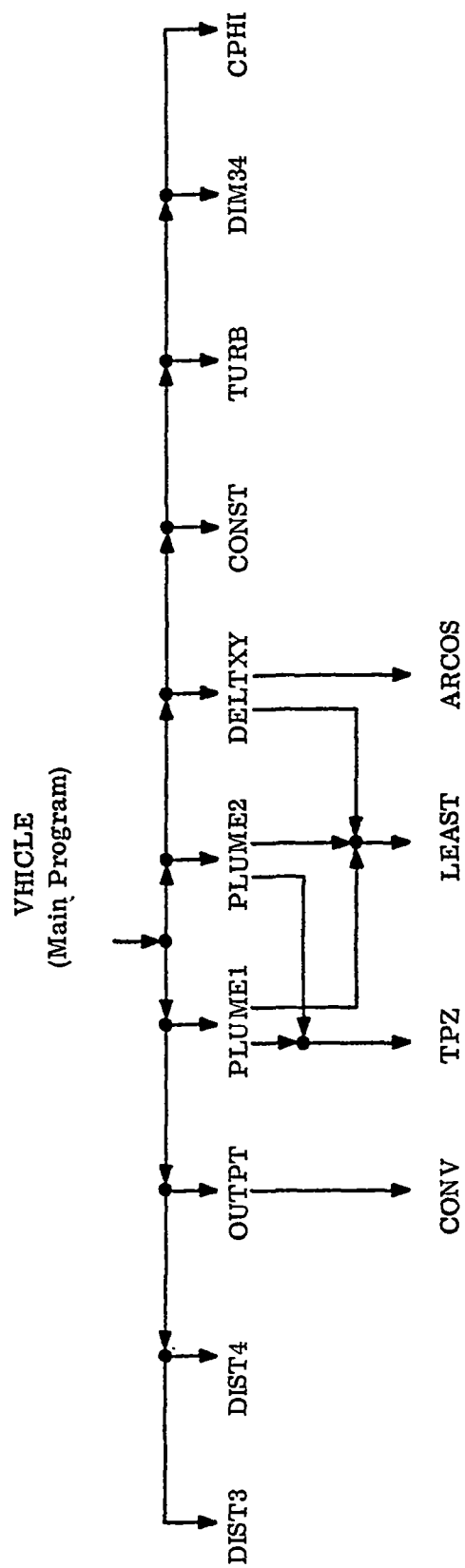


FIGURE A-2. Linkage diagram of the subroutines in the Preprocessor Program.

Subroutine DELTXY calculates the cloud trajectory (see Section 2, Equations (2-15) through (2-26)).

Subroutine LEAST calculates the lapse rate of virtual potential temperature using a least-squares fit (Equation (2-7)).

Subroutine CONV converts the output source strength into a form suitable for punching variable real numbers in a fixed format.

Subroutine TURB inserts the turbulence parameters σ_A and σ_E in the output data and calculates $t_K = t^*$ (Equation (2-4)).

Subroutine TPZ is a linear interpolation function.

Subroutine CPHI is used in calculating the virtual potential temperature (Equation (2-7)).

Subroutine DIST4 calculates the source distribution for Model 4 (see Section 2.3.2).

Subroutine DIST3 calculates the source distribution for Model 3 (see Section 2.3.2).

Subroutine OUTPT prints and punches the output data in a namelist format suitable for input to the NASA/MSFC Multilayer Diffusion Models - Version 5.

Subroutine CONST prints the input parameters and resultant calculations in a format suitable for error check and calculation review.

Subroutine ARCOS is a function reference to ACOS, the UNIVAC 1108 arc-cosine function. It is referenced in this way because the IBM 7044 at MSFC uses ARCOS for the arc-cosine function and program conversion from the UNIVAC 1108 to the IBM 7044 is easily completed by removal of this entire subroutine.

APPENDIX B

USER INSTRUCTIONS FOR THE NASA/MSFC MULTILAYER DIFFUSION MODELS COMPUTER PROGRAM - VERSION 5

B.1 PROGRAM DESCRIPTION

The NASA/MSFC Multilayer Diffusion Models Program - Version 5 is designed to calculate patterns of:

- Concentration
- Dosage
- Time-mean concentration
- Average cloud concentration
- Time of cloud passage
- Ground-level deposition due to precipitation scavenging
- Ground-level deposition due to gravitational settling

Program options include the calculation of concentration, dosage and time-mean concentration with partial reflection, with time dependent decay, and/or with depletion due to precipitation scavenging. Also, the Program is capable of calculating ground-level gravitational deposition with partial reflection of material at the surface. Other Program options include the printing of all data inputs, the printing of all model calculations, the plotting of concentration, dosage and/or time-mean concentration on the printer page, and the plotting of concentration, dosage and/or time-mean concentration on the SC4020 plotter at Marshall Space Flight Center.

The NASA/MSFC Multilayer Diffusion Models Program - Version 5 is written in FORTRAN IV and is designed for use on the UNIVAC 1108 computer at Marshall Space Flight Center, Huntsville, Alabama. The Program requires 33000¹⁰ locations of core storage on the UNIVAC 1108 computer. The FORTRAN source listing is shown in Appendix C.2 and a complete list of Program input parameters and Program options is given in Section B.2 and B.3 below. Also, a linkage diagram of the Program subroutines is shown in Section B.6 and a description of the Program subroutines is given in Section B.7.

B.2 PROGRAM INPUT PARAMETERS

This section gives a complete description of all Program input parameters. A condensed table of the input parameters is given in Section B.3. The data input format is given in Section B.4 and an example input coding sheet is given in Section B.5.

NAMCAS - 72 Hollerith characters of general case identification information. This information is printed in addition to the adjusted cloud stabilization height, range and azimuth bearing and the date and time of the run as a title page to the output listing.

TESTNO - The first 36 Hollerith characters (TESTNO (1) - TESTNO (6)) contain the meteorological case information. This information is printed in the page heading and plot titles following the words "THE METEOROLOGICAL CASE IS".

Characters 37 through 60 (TESTNO (7) - TESTNO (10)) contain the name of the rocket vehicle for use in the page heading. (e.g. TITAN IIC)

Characters 61 through 72 (TESTNO (11) - TESTNO (12)) contain the name of the pollutant only if it is not HCl, CO, CO₂, or Al₂O₃. (e.g. NO_x)

NPS - This parameter is used to indicate multiple cases.

If NPS is set to 0, the Program assumes there is another case to follow.

If NPS is set to 1, the Program assumes this is the last case to process.

ISKIP - Program control option array.

ISKIP (1) - This option, if set non-zero, indicates patterns of concentration, dosage, time-mean concentration, deposition, etc. are to be calculated and printed on the polar reference grid system defined by XX and YY below. The grid system origin is the vehicle launch site and all calculation distances are relative to the origin.

ISKIP (2) - This option, if set non-zero, is used to calculate maximum center-line values of concentration, dosage, time-mean concentration, and/or deposition along the cloud trajectory relative to the launch site.

If ISKIP (2) is set equal to 1, the model calculations are printed.

If ISKIP (2) is set equal to 2, the model calculations are plotted.

If ISKIP (2) is set equal to 3, the model calculations are both printed and plotted.

The maximum centerline concentration, dosage, time-mean concentration and deposition are determined by the use of a spline function. At each radial distance (XX) from the origin, the Program determines a curve via the cubic spline that passes through each angular (azimuth bearing YY) grid coordinate with the

calculated maximum roughly in the midpoint of the curve. The Program will then determine the maximum value and output, the range and azimuth bearing to that maximum (see Section B.7, Subroutine Spline).

- ISKIP (3) - This option, if set non-zero, is used to calculate isopleths of concentration, dosage, time-mean concentration and/or deposition.

If ISKIP (3) is set equal to 1, the isopleths are printed.

If ISKIP (3) is set equal to 2, the isopleths are plotted.

If ISKIP (3) is set equal to 3, the isopleths are both printed and plotted.

- ISKIP (4) - This option is used only with the calculation of ground-level precipitation deposition (Model 5).

If ISKIP (4) is set non-zero, the maximum possible ground-level precipitation deposition is calculated at points downwind from the cloud position at the time of the start of precipitation (TIM1). These calculations are independent of the elapsed time from TIM1 to the calculation point.

If ISKIP (4) is set equal to zero, the calculated precipitation deposition at points downwind from the cloud position at time TIM1 is dependent upon the elapsed time from TIM1 to the points.

- ISKIP (5) - This option controls the pollutant name and units printed in the page heading and plot legend:

If ISKIP (5) is set equal to 1, the units of calculated HCl concentration are in parts per million (ppm) and dosage units are in parts per million seconds.

If ISKIP (5) is set equal to 2, the units of calculated CO concentrations are in parts per million (ppm) and dosage units are in parts per million seconds.

If ISKIP (5) is set equal to 3, the units of calculated CO₂ concentration are in parts per million (ppm) and dosage units are in parts per million seconds.

If ISKIP (5) is set equal to 4, the units of calculated Al₂O₃ concentration are in milligrams per cubic meter (mg/m³) and dosage units are in milligram seconds per cubic meter.

If TESTNO (11) above is non-blank, then ISKIP (5) is used only for units selection and the pollutant name is taken from TESTNO (11). Also, calculated precipitation deposition (Model 5) and gravitational deposition (Model 6) are in units of milligrams per square meter.

- ISKIP (6) - This option is used for printing purposes only and gives the type of vehicle launch for which calculations are being made and inserts the following in the page heading and plot legend.

If ISKIP (6) is set equal to 1, a "STATIC FIRE" is assumed.

If ISKIP (6) is set equal to 0 or 2, a "NORMAL LAUNCH" is assumed.

If ISKIP (6) is set equal to 4, a "SLOW BURN" is assumed.

If ISKIP (6) is set equal to 5, the program omits this option from the page heading and plot legend.

- ISKIP (7) - This option controls the meteorological data used with Model 4.

If ISKIP (7) is set equal to zero, the Program assumes Model 4 is being used to determine concentration, dosage, etc., in a layer where the pollutant distribution at cloud stabilization varies substantially with height. The meteorological data used in Model 4 is automatically determined from the meteorological inputs assigned to the initial layers or sublayers.

If ISKIP (7) is set equal to 1, the Program assumes Model 4 is being used to determine concentration, dosage, etc., resulting from changes in the meteorological layer structure. The meteorological data used in Model 4 after time TAST (time of layer structure change measured from time of cloud stabilization) is taken from the input parameters ALPHL through TEMPL (see page B-22 below).

- | | |
|-----------|--|
| ISKIP (8) | - This option, if set non-zero, prints a detailed listing of all Program inputs. |
| NSX | - Number of radial distances (range) XX in the polar reference grid system. If NXS is set ≤ 0 , the default value of 41 is used for NXS and the array XX is automatically filled from values shown in Table B-3. |
| NYS | - Number of azimuth bearings in the polar reference grid system. If NYS is set ≤ 0 , this parameter is automatically calculated and the array of azimuth bearing coordinates (YY) is automatically filled. The value of NYS includes sufficient points in YY to provide a calculation pattern spanning 100 degrees (see Table B-3, note 9). |

- NZS** - Total number of initial layer boundaries including the ground surface boundary.
- NCI, NDI, NTI** - These parameters each contain two values used in the maximum centerline calculations under ISKIP (2) and in the calculation of isopleths under ISKIP (3).

The total number of isopleth values is given in the hundreds and tens positions of NCI, NDI and/or NTI. If these positions are zero, isopleths for the respective quantity (concentration, dosage, time-mean concentration and/or deposition) is not calculated. The number of critical pollutant levels (air quality standards) to be identified in the plots for maximum centerline calculations is given in the units position of NCI, NDI and/or NTI. If this position is zero, no plot is generated. If set to 9, a plot is generated without indicators for critical pollutant levels (air quality standards).

If the units position of NCI, NDI and/or NTI is greater than zero and not equal to 9, the critical pollutant levels (standards) must be punched as the first values in the arrays CI, DI and/or TI below.

- NPTS** - Number of heights at which calculations are to be made. If NPTS is set equal to zero or omitted, NPTS is defaulted to 1 and ZZL (1) below is set equal to zero.
- NVS** - Number of droplet or particle terminal fall velocities used to calculate ground-level gravitational deposition from all layers except the layer in which a destruct occurs (Model 6 only).

- NVB - Number of droplet or particle terminal fall velocities used to calculate ground-level gravitational deposition from the layer in which a vehicle destruct occurs (Model 6 only).
- XX - Array of radial distances (range) for the coordinates used in calculations on the reference grid system. This array is automatically filled if NXS = 0, (see NXS above). The last 2 points in XX are used only for calculating isopleths; the second to last point should equal 1.2 times the third to the last point and the last point should equal 1.5 times the third to the last point. Space the XX values uniformly and use as many as the program will allow. The user is cautioned to use the default values unless another grid is required.
- YY - Array of azimuth bearings for the coordinates used in calculations on the reference grid system measured clockwise from zero degrees north. This array is automatically filled if NYS = 0 (see NYS above). Space the YY values densely toward the center of the calculation sector and use as many as the program will allow. The user is cautioned to use the default values unless another grid is required.
- Z - Array of layer boundary heights in ascending order beginning with the surface boundary height (the first layer is always the surface layer).
- DELX - Array of the radial distances (range) from the source location (point of cloud stabilization) in each layer to the center of the reference grid system (launch site).
- DELY - Array of azimuth bearings to the source location (point of cloud stabilization) in each layer, measured clockwise from zero degrees north.

TABLE B-1
SOURCE STRENGTH INPUT UNITS

Model	Pollutant	
	HCL, CO, CO ₂	AL ₂ O ₃
1	1	2
2	1	2
3	1	2
4	1	2
5	2	2
6	2	2

Code definition for Table B-1:

- ① $Q = Q' \frac{22.4}{M} \frac{T}{273.16} \frac{1013.2}{P}$

(Concentration output units are parts per million (PPM))

- ② $Q = Q'$

where

Q = Source strength in each initial layer

Q' = Total weight of the material in the layer in milligrams

T = Surface temperature in degrees Kelvin

P = Surface pressure in millibars

M = Molecular weight of the material

TABLE B-1 (Continued)

(Deposition output units for Models 5 and 6 are milligrams per square meter (mg/m^2) and concentration output units for Models 1 through 4 are milligrams per cubic meter (mg/m^3))

Q	- Source strength within each initial layer. The source strength input units depend upon the model used and the pollutant for which calculations are being made. Table B-1 gives the appropriate input units for each model and pollutant combination.
UBARK	- Mean wind speed at ZRK followed by the mean wind speed at the top of each layer.
SIGAK	- Standard deviation of the wind azimuth angle for reference time τ_{oK} at ZRK followed by the standard deviation of the wind azimuth angle at the top of each layer.
SIGEK	- Standard deviation of the wind elevation angle at ZRK followed by the standard deviation of the wind elevation angle at the top of each layer.
SIGXO	- Standard deviation of the alongwind concentration distribution of the source in the layer (alongwind source dimension).
SIGYO	- Standard deviation of the crosswind concentration distribution of the source in the layer at a downwind distance XLRY from the true source (crosswind source dimension). The default value is SIGXO.
SIGZO	- Standard deviation of the vertical concentration distribution of the source in the layer at a downwind distance XLRZ from the true source (vertical source dimension).
ALPHA	- Lateral diffusion coefficient in the layer (default value is 1).
BETA	- Vertical diffusion coefficient in the layer (default value is 1).
ZRK	- Reference height in the surface layer for meteorological measurements (default value is 2 meters).

- TEMPK** - Virtual potential temperature at each layer boundary z.
This parameter is used only in the calculation of the wind speed shear in the layer. If the wind speed shear is negative and the difference between the virtual potential temperature at the top and bottom of the layer is also negative, the Program will use the absolute value of the speed shear. If the temperature difference is positive or zero, the program will use a wind speed shear of zero. If the layer wind speed shear is positive or zero, the virtual potential temperature difference is not used.
- TIMAV** - Time over which time-mean concentration and average cloud concentration are calculated (default value is 600 seconds except for CO, where it is 300 seconds).
- THETAK** - Mean wind direction at ZRK followed by the mean wind direction at the top of each layer.
- TAUK** - Time required for cloud stabilization.
- TAUOK** - Reference time for the standard deviations of the wind azimuth angle SIGAK (default value is 600 seconds).
- H** - Adjusted cloud stabilization height.
- XRY** - Distance downwind from a virtual point source over which rectilinear expansion in the lateral occurs (default value is 100 meters).
- XRZ** - Distance downwind from a virtual point source over which rectilinear expansion in the vertical occurs (default value is 100 meters).
- XLRY** - Reference distance from the true source at which SIGYO is measured (default value is zero).

- XLRZ** - Reference distance from the true source at which SIGZO is measured (default value is zero).
- ZZL** - Vertical calculation heights. This parameter can include any heights within the initial layer structure (default value is zero).
- IZMOD** - This parameter designates the model number or numbers for use in each input layer. A brief description of the six Program models is given below and a complete mathematical description of each model is given in Section 3 of the main body of this report. The possible model number combinations input into IZMOD are given in Table B-2.
- 1 - Model 1, the source extends vertically through the entire initial layer and turbulent mixing is occurring. It is assumed that the vertical distribution of material is uniform with height and the distributions of material along the along-wind and crosswind cloud axes are Gaussian. The digit 1 is included in the array IZMOD for each layer in which Model 1 is to be used. Also, if any digit of IZMOD is 0, the Program assumes Model 1 has been designated.
 - 2 - Model 2 refers to the same source configuration as Model 1 in that the source extends vertically through the entire depth of the layer and the distribution of material is uniform with height. In Model 2, however, it is assumed that no turbulent mixing is occurring. The digit 2 is included in the array IZMOD for each layer in which Model 2 is to be used in the calculations (IZMOD = 2, 2, 2, etc.).
 - 3 - Model 3 differs from Models 1 and 2 in that the vertical extent of the source is less than the depth of the layer. The model equation thus contains vertical expansion terms. The digit 3 is input to IZMOD for Model 3 (IZMOD = 3).

TABLE B-2
POSSIBLE INPUT MODEL NUMBER COMBINATIONS

IZMOD ¹	PROGRAM ASSUMES CALCULATIONS ARE MADE USING:
0	Model 1
1	Model 1
2	Model 2
3	Model 3
14	Model 1 is used prior to layer transition and Model 4 is used after layer transition occurs at time TAST
24	Model 2 is used prior to layer transition and Model 4 is used after layer transition occurs at time TAST
34	Model 3 is used prior to layer transition and Model 4 is used after layer transition occurs at time TAST
4	Model 4 is used to accomodate to a variation of source strength in the layer and layer transition is immediate (TAST=0)
5	Model 5 is used and the layer structure and source distribution is assumed to be that of Model 1 when only the digit 5 is given in IZMOD. The digit 5 can be combined with any of the above digit combinations (145, 45, 35, etc.). When a 5 is combined with any of the above digit combinations the Program assumes the layer structure and source distribution of that combination are used with Model 5.
6	Model 6

¹ The digits under IZMOD can appear in any order. For example, 14 is the same as 41 and 154 is the same as 415.

- 4 - Model 4, the layer-transition model, may be used to calculate concentration and dosage resulting from changes in the meteorological layer structure. Model 4 may also be used to calculate concentration and dosage in a layer where the pollutant distribution at cloud stabilization varies substantially with height.

The application of Model 4 requires the following assumptions:

- The boundaries between adjacent initial layers or sublayers is eliminated (at time TAST) and the layers are replaced by a single layer
- Turbulent mixing is occurring in the resultant single layer
- The material in each of the initial layers or sublayers is (before time TAST) uniformly distributed in the vertical
- Reflection occurs at the upper and lower boundaries of the resultant single layer

If the parameters TAST and ISKIP (7) are both set to zero (or omitted from the inputs) and Model 4 is specified for use, the program assumes the function of the model is to accommodate variations in the pollutant distribution with height in the layer at cloud stabilization. For example, the surface mixing layer can be initially divided into several sublayers where the source strength, although assumed to be vertically uniform in each sublayer, varies from layer to layer. In this case the initial layers are immediately reduced to a single layer and Model 4 calculates the contribution from each of the initial sublayers to the composite concentration and dosage field by permitting turbulent mixing across the initial layer boundaries. IZMOD would contain the digit 4 for each of the respective initial sublayers that comprise the resultant single layer.

If Model 4 is to be used to predict the concentration and dosage fields downwind from a change in meteorological structure, the meteorological parameters of the new resultant layer or layers must be specified. Also, the parameter ISKIP (7) must be set equal to 1 and the parameter TAST set equal to the time (after cloud stabilization) at which the layer transition (meteorological structure change) occurs. Each of the initial sublayers that are to be included in a single layer after layer transition are specified by including the digit 4 in the array IZMOD. For example, assume layers 1 through 4 are to be reduced to a single layer after layer transition and layers 5 and 6 are also reduced to a single layer. The first four values of IZMOD would include a 4, but they would also include the number of the model to be used prior to layer transition (14, 24 or 34). The values of IZMOD (5) and (6) for layers 5 and 6 would include a 9 and 4, respectively. The 9 is a special flag to separate the resultant 2 layers after layer transition. Also, these last two values would include the model number to be used prior to layer transition (14, 24 or 34). If Model 1 was to be used with 4 in the above example the IZMOD inputs would be coded as IZMOD = 4, 4, 4, 4, 9, 4 (or IZMOD = 4*4, 9, 4, or IZMOD = 4*14, 19, 14, etc.).

- 5 - Model 5 is used to calculate the amount of material on the surface by precipitation scavenging. The digit 5 must be included in the array IZMOD for each initial sublayer through which precipitation is occurring. Model 5 uses the layer structure and source distribution defined by any one of Models 1 through 4. Thus, the array IZMOD must include the appropriate model number for each layer that describes the layer structure and source distribution. For example, assume that Model 4 is being used to accommodate to variations in the pollutant distribution with height in the surface mixing layer at cloud stabilization and that the surface mixing layer has been divided into 6 initial sublayers in which the distribution of material can be

considered uniform. Also, assume that precipitation is occurring through all 6 layers. The array IZMOD would then contain six values equal to 45 for each layer 1 through 6 (IZMOD = 6*45).

- 6 - Model 6 is used to calculate the surface deposition due to gravitational settling. The basic source configuration is a volume source of finite lateral extent and unit vertical extent. Other source configurations are treated by summing the deposition at the ground resulting from a number of basic sources arranged to simulate the desired configuration. The model is essentially a tilted plume model in which the effects of wind shear are taken into account. The axis of a particle or droplet cloud of a given settling velocity intersects the ground plane at a distance from the source and at an angle from the mean surface wind direction that are proportional to the total angular wind shear and the residence time of the settling material in the layers between the source and the ground surface. In any layer, the inclination of the cloud axis from the horizontal is given by $\tan^{-1} V_s / \bar{u}$, where V_s is the particle or droplet settling velocity and \bar{u} is the mean transport wind speed in the layer. In all cases, material released in the K^{th} layer and dispersed upwards by turbulence is assumed to be reflected downward at the interface of the K^{th} and $(K + 1)^{\text{th}}$ layers. The basic model is used to calculate the ground-level deposition pattern for a single value of the settling velocity. The total deposition pattern is obtained by summing the results for all settling velocities representative of the particle or droplet-size distribution of the released material on a reference coordinate grid system.

Only IZMOD (1) need be set equal to 6 as no other model can be executed in the same case.

- DECAY - Coefficient of time-dependent decay. If DECAY is set > 0 , then concentration, dosage, time-mean concentration, etc., are calculated with decay (Does not effect Model 5 or Model 6).
- ZLIM - This parameter is the maximum height through which precipitation can occur. If Model 5 is selected, ZLIM is automatically determined from IZMOD. If concentration, dosage, etc., are being calculated with precipitation occurring ($BLAMDA > 0.0$), ZLIM is equal to the upper boundary of the uppermost layer in which precipitation occurs (ZLIM is defaulted to $Z(NZS)$).
- BLAMDA - Precipitation scavenging (washout) coefficient. If Model 5 is selected, this parameter must be greater than 0. Also, if Model 1, 2, 3 or 4 is selected with $BLAMDA > 0$ and without Model 5, the Program assumes concentration, dosage, etc., are to be calculated with precipitation occurring.
- TIM1 - Time of start of precipitation measured from the time of cloud stabilization.
- CI, DI and TI - Arrays of concentration, dosage and time-mean concentration values respectively for which isopleths are calculated. There can be two groups of data in each of these arrays, where both of the groups are arranged in descending order. The values in the first group are critical pollutant levels (air quality standards). The number of values in this group is given in the units position of the parameters NCI, NDI and NTI respectively. The second group of values includes all other isopleth levels desired. The total number of values in CI, DI and TI is given in the hundreds

and tens positions of NCI, NDI and NTI respectively. If precipitation, deposition or gravitational deposition is being calculated, the array DI is used for these quantities.

- TAST - Time of layer structure change (Model 4) measured from the time of cloud stabilization.
- GAMMAP - This parameter is 1 minus the fraction of material reflected at the surface (partial reflection). If this parameter is set to 0, the Program assumes complete reflection; if set equal to .4, 60 percent (.6) reflection is assumed; and, if set equal to 1, no reflection is assumed. If Model 6 is selected and partial reflection is desired, the array GAMMAP must have a value for each particle settling velocity category. For all other models, only GAMMAP (1) need be set.
- VS - Droplet or particle terminal fall velocity distribution used in all layers except a layer in which a vehicle destruct occurs (Model 6 only).
- PERC - Frequency of occurrence of each velocity category VS (Model 6 only).
- ACCUR - Accuracy constant for the line source simulation used in Model 6. A value of 0.45 ensures that the calculated ground deposition is within 10 percent of the deposition expected from a vertical line source. If ACCUR is set to 0.32, the calculated deposition is within 5 percent of that expected from a vertical line source.

- VB - Droplet or particle terminal fall velocity distribution used in the layer in which a vehicle destruct occurs. The layer must be the top layer (Model 6 only).

- PERCB - Frequency of occurrence of each velocity category VB (Model 6 only).

- SCL - Map scale factor in inches for isopleth plots. If the map scale factor is 1 inch = 24000.inches, SCL would be input as 24000. If set to zero, the Program will scale the isopleths within the boundaries defined by XSIZE and YSIZE below.

- XMAXIN - Maximum alongwind distance from the launch site in meters for isopleth plots. If set to zero, the Program will use XX(NXS-2) as the maximum distance.

- YMAXIN - Maximum crosswind distance for isopleth plots in meters. If set to zero, the Program will calculate YMAXIN.

- XSIZE - The number of raster counts on the SC4020 in the X or east-west horizontal plot axis for isopleths. If set to zero, the Program will use 937.

- YSIZE - The number of raster counts on the SC4020 in the Y or north-south vertical plot axis for isopleths. If set to zero, the Program will use 899.

- RASTIN - The number of raster counts per inch on the SC4020 for isopleth plots. If input as zero, the Program uses 163.2.

- XCIZE** - The number of raster counts on the SC4020 on the X or alongwind horizontal axis for maximum centerline plots. If set to zero, the Program uses 937.
- YCIZE** - The number of raster counts on the SC4020 on the vertical axis for maximum centerline plots. If set to zero, the Program uses 899.
- XMAXJN** - Maximum alongwind distance in meters from the launch site for maximum centerline plots. If set to zero, the Program uses XX(NXS-2).
- YMAXJN** - Maximum number of log cycles for the vertical axis of the maximum centerline plots if ISW below equals 0 or 2. Maximum value of the vertical axis if ISW below equals 1. If set to zero, the Program determines YMAXJN.
- ISW** - Maximum centerline plotting flag. If ISW is set to 0 or 2, the Program plots maximum centerline versus distance on a log-log plot. If set to 1, the plot is linear on both axes.
- JSW** - Isopleth plot switch. If JSW is set equal to 0, the Program will fit a cubic spline function to the discrete isopleth points and plot a smooth curve through the points. If JSW is set equal to 1, the Program will not use the spline function but will plot straight lines between adjacent calculated isopleth points. This option has been included because the spline function sometimes fails to fit the data points when the isopleths are sharply curved. These cases are recognized by a high frequency oscillation along the plotted curve and can be corrected by smoothing the curve by hand or replotting with JSW set equal to 1.

The layer step change (transition) parameters below are used only if ISKIP (7) equals 1 and Model 4 has been selected. These parameters are used only when Model 4 is being used to predict the concentration and dosage downwind from a change in meteorological structure (see IZMOD, Model 4 above).

- ALPHL - Lateral diffusion coefficient in each new layer (Default value is 1).
- BETL - Vertical diffusion coefficient in each new layer (Default value is 1).
- TAUL - Time required for cloud stabilization in the new layers.
- TAUOL - Reference time for the standard deviation of the wind azimuth angle SIGAL in the new layers (Default value is 600).
- ZRL - Reference height in the surface layer for meteorological measurements. This must be set only if the new bottom layer includes the initial surface layer (Default value is 2).
- UBARL - Mean wind speed at the bottom and top boundaries of each new layer. These values are input in ascending order of new layers with the value at the top boundary preceded by the bottom. If the new bottom layer contains the initial surface layer, UBARL at ZRL should be input as the bottom value of this layer.
- SIGAL - Standard deviation of the wind azimuth angle for reference time τ_{OL} at the bottom and top boundaries of each new layer. If the new bottom layer contains the initial surface layer, SIGAL at ZRL should be input as the bottom value of this layer.
- THETAL - Mean wind direction at the bottom and top boundaries of each new layer. If the new bottom layer contains the initial surface layer, THETAL at ZRL should be input as the bottom value of this layer.
- TEMPL - Virtual potential temperature at the bottom and top boundaries of each new layer.

B.3 CONDENSED TABLE OF INPUT PARAMETERS

The data input parameters required for the computer Program are given in condensed form in Table B-3. The information categories in the table are defined as follows:

- NAMELIST - Name of the FORTRAN NAMELIST list to which the variables belong.
- FORTRAN - Fortran symbolic notation defining the program input.
- MODEL - Mathematical notation corresponding to the FORTRAN notation.
- UNITS - Dimensional units of the input parameters.
- LIMITS - Numerical limits on input values.
- VALUE - Default value should the parameter have a value of 0.
- ARRAY SIZE - Maximum number of core locations for the input parameter.

B.4 DATA INPUT FORMAT

This Program uses the FORTRAN NAMELIST method to input data. Input data must be in a specific form in order to be read using a NAMELIST list. The first character in each card to be read must be blank. The first card in the NAMELIST list contains the NAMELIST name NAM2 preceded by the character \$ or &. The last card in the NAMELIST list contains \$END (&END) to terminate the list. The form of the remaining data items in the list may be:

a. *Variable Name = Constant* - The *variable name* may be a subscripted array name or a single variable name. Subscripts must be integer constants. The *constant* may be integer, real or Hollerith (nH *alphanumeric characters*) data.

TABLE B-3
TABLE OF INPUT PARAMETERS

NAMLIST	FORTRAN	Model	Units	Limits	Value ^③	Array Size ^⑦
NAM12	TESTNØ	N/A	N/A	N/A	Blanks	12
	NAM1CAS	N/A	N/A	N/A	Blanks	12
	ISKIP	N/A	N/A	①	0	15
	NXS	N/A	N/A	≤41	41	1
	NYS	N/A	N/A	≤41	41	1
	NZS	N/A	N/A	≤16	0	1
	NDI	N/A	N/A	≤103 ⑩	0	1
	NCI	N/A	N/A	≤103 ⑩	0	1
	NTI	N/A	N/A	≤103 ⑩	0	1
	NPTS	N/A	N/A	≤40	1	1
	NVS	N/A	N/A	≤20	0	1
	NVB	N/A	N/A	≤20	0	1
	XX	R	Meters	> 0.0	⑧	41
	YY	A	Degrees	0.0 ≤ Φ ≤ 360.0	⑨	41
	NPS	N/A	N/A	0 or 1	0	1
	Z	z _{Bl} and z _{TK}		≥ 0.0	z(1) = 0.0	16

TABLE B-3
TABLE OF INPUT PARAMETERS
(Continued)

NAMLIST	FORTTRAN	Model	Units	Limits	Value ^③	Array Size ^⑦
NAM2	DELX	R	Meters	≥ 0.0	0.0	15
	DELY	A	Degrees	$0.0 \leq \theta \leq 360.0$	0.0	15
	Q	Q	②	≥ 0.0	0.0	15
	UBARK	\bar{u}_R and \bar{u}_{TK}	Meters Sec ⁻¹	≥ 0.1	0.1	16
	SIGAK	$\{\tau_o\}$ & σ_{AR}	Degrees	≥ 0.5	0.5	16
		σ_{ATK} $\{\tau_o\}$				
	SIGEK	σ_{ER} & σ_{ETK}	Degrees	≥ 0.1	0.1	16
	SIGXØ	σ_{xo} {K}	Meters	> 0.0	N/A	15
	SIGYØ	σ_{yo} {K}	Meters	> 0.0	SIGXØ	15
	SIGZØ	σ_{zo} {K}	Meters	≥ 0.0	0.0	15
	ALPHA	α_K	N/A	≥ 0.0	1.0	15
	BETA	β_K	N/A	≥ 0.0	1.0	15
	ZRK	z_R	Meters	$\geq z(1)$	2.0	1
	TIMAV	T^A	Seconds	≥ 0.0	600 or 360	1
	THETAK	θ_{B1} & θ_{TK}	Degrees	$0.0 \leq \theta_K \leq 360.0$	0.0	16

TABLE B-3
TABLE OF INPUT PARAMETERS
(Continued)

NAMELIST	FORTRAN	Model	Units	Limits	Value ^③	Array Size ^⑦
NAM2	TAUK	τ	Seconds	>0.0	N/A	1
	TAUØK	τ_o	Seconds	≥ 0.0	600.0	1
	H	H	Meters	≥ 0.0	0.0	1
	XRY	x_{ry}	Meters	≥ 0.0	100.0	1
	XRZ	x_{rz}	Meters	≥ 0.0	100.0	1
	XLRY	x_{Ry}	Meters	≥ 0.0	0.0	1
	XLRZ	x_{Rz}	Meters	≥ 0.0	0.0	1
	ZZL	z	Meters	≥ 0.0	0.0	1
	IZNØD	N/A	N/A	⑪	.1	15
	DECAY	k	Seconds ⁻¹	≥ 0.0	0.0	1
	ZLIM	z_{lim}	Meters	$= z_{TK}$	Z (NZS) ^⑬	1
	TIMI	t_1	Seconds	≥ 0.0	⑤	1
	BLAMDA	Λ	Seconds ⁻¹	≥ 0.0	⑤	1
	DI	$D_K \{x_K, y_K, z_K\}$	④	≥ 0.0	⑤	10
	CI	$x_K \{x_K, y_K, z_K\}$	④	≥ 0.0	⑤	10
	TI	$x_K \{x_K, y_K, z_K; z_{T_A}\}$	④	≥ 0.0	⑤	10

TABLE B-3
TABLE OF INPUT PARAMETERS
(Continued)

NAMLIST	FORTRAN	Model	Units	Limits	Value ⁽³⁾	Array Size ⁽⁷⁾
NAM2	TAST	t^*	Seconds	≥ 0.0	0.0	5
	TEMPK	$\phi_{BI} \text{ \& } \phi_{TK}$	Degrees K	≥ 0.0	0.0	16
	VS	V_s	Meters sec^{-1}	≥ 0.0	(5)	20
	PERC	f_i	N/A	> 0.0	(5)	20
	ACCUR	R_c	N/A	(6)	(5)	20
	VB	V_{SK}	Meters sec^{-1}	> 0.0	(5)	20
	PERCB	f_i	N/A	> 0.0	(5)	20
	HB	H_{SK}	Meters	≥ 0.0	0.0	1
	GAMMAP	$1-\gamma_r$	N/A	$\geq 0 \text{ \& } \leq 1$	0.0	20
	ALPHL	α_L	N/A	≥ 0.0	(13)	5
	BETL	β_L	N/A	≥ 0.0	(13)	5
	TAUL	τ	Seconds	> 0.0	TAUK	1
	TAUOL	τ_o	Seconds	≥ 0.0	TAUOK	1
	ZRL	z_{RL}	Meters	≥ 2.0	ZRK	1
	UBARL	$\bar{u}_{BL} \text{ \& } \bar{u}_{TL}$	Meters sec^{-1}	≥ 0.0	(12)	10
	SIGAL	$\sigma_{ABL} \{ \tau_o \} \text{ \& } \sigma_{ATL} \{ \tau_o \}$	Degrees	≥ 0.0	(12)	10

TABLE B-3
TABLE OF INPUT PARAMETERS
(Continued)

NAMLIST	FORTRAN	Model	Units	Limits	Value ^③	Array Size ^⑦
NAM2	SIGEL	σ_{EBL} & σ_{ETL}	Degrees	≥ 0.0	⑫	10
	THETAL	θ_{BL} & θ_{TL}	Degrees	≥ 0.0 & ≤ 360.0	⑫	10
	TEMPL	ϕ_{BL} & ϕ_{TL}	Degrees K	≥ 0.0	0.0	10
	SOL	N/A	Inches	≥ 0	Calculated	1
	XMAXIN	R	Meters	≥ 0	Calculated	1
	YMAXIN	N/A	Meters	≥ 0	Calculated	1
	XSIZE	N/A	Rasters	≥ 0	937	1
	YSIZE	N/A	Rasters	≥ 0	899	1
	RASTIN	N/A	Rasters/ Inch	≥ 0	163.2	1
	XCIZE	N/A	Rasters	≥ 0	937	1
	YCIZE	N/A	Rasters	≥ 0	899	1
	XMAXJN	N/A	Meters	≥ 0	XX(NXS-2)	1
	YMAXJN	N/A	Log Cycles or Meters	≥ 0	Calculated	1
	ISW	N/A	N/A	1 or 2	2	1
	JSW	N/A	N/A	0 or 1	0	1

TABLE B-3

TABLE OF INPUT PARAMETERS

(Continued)

- ① See Section B-2 for the range of values of the ISKIP options.
- ② Units depend on model; see Section B-2 in the definition of Q.
- ③ The column under Value is used to simplify the Program input deck by providing default values should the parameter be intentionally omitted in the first data case or set to zero. All parameters in Table B-3 remain their previous value for all subsequent cases unless changed in the input list.
- ④ Units of dosage and concentration isopleth values must be consistent with Program output units, milligrams/meter³ or parts per million, etc.
- ⑤ These parameters must have values other than zero only if they are used by the model selected and only in the applicable layers.
- ⑥ See Section B-2 for the description of ACCUR.
- ⑦ Several variables are dimensioned to a larger value in the Program, but the extra space is used for other purposes.

TABLE B-3

TABLE OF INPUT PARAMETERS

(Continued)

- ⑧ The default values of XX are: 500, 1250, 2500, 3750, 5000, 6250, 7500, 8750, 10000, 11250, 12500, 13750, 15000, 16250, 17500, 18750, 20000, 21250, 22500, 23750, 25000, 26250, 27500, 28750, 30000, 31250, 32500, 33750, 35000, 36250, 37500, 38750, 40000, 41250, 42500, 43750, 45000, 47500, 50000, 65000, 80000 meters. Default values of XX are used only if NXS is set to 0.
- ⑨ The default values of the YY are the average layer wind direction $\pm 180^\circ$ rounded to the nearest 5° added to each of the following angles: -40, -35, -30, -27, -24, -22, -20, -18, -16, -14, -12, -10, -8, -7, -6, -5, -3, -2, -1, 0, 1, 2, 3, 4, 5, 6, 7, 8, 10, 12, 14, 16, 18, 20, 22, 24, 27, 30, 40 degrees.
- ⑩ The limit values given for NDI, NCI and NTI mean there is a maximum of 10 possible isopleth values with a maximum of 3 critical pollutant levels (air quality standards) within the 10. The total number of values is input in the tens and hundreds positions and the number of critical pollutant levels is input in the units position.
- ⑪ IZMØD is a 3 digit integer where any one of the three digits can be an integer from 0 to 6 or the integer 9. See Section B.2 for a complete explanation of IZMØD.
- ⑫ If these parameters are input, both bottom and top values are input respectively for each new layer in the layer step change.
- ⑬ ZLIM is automatically calculated if IZMØD contains a 5 (Model 5).

b. *Array Name = Set of Constants* (separated by commas) - The *array name* is not subscripted. The *set of constants* consists of constants of the type integer or real. The number of constants must be less than or equal to the array size. Successive occurrences of the same constant can be represented in the form k^* *constant*.

The sequence of the input data parameters within the list is not significant. A more detailed explanation of the FORTRAN NAMELIST can be found in most FORTRAN language manuals. Section B.5 shows an example input data coding sheet. All Program input parameters are set to zero prior to input of the first case. Parameters that are not used or have default values need not appear in the input deck. When multiple cases are stacked, all parameters retain their values from the last case and are changed only by input.

B.5 SAMPLE PROBLEM INPUT DATA

The sample problem is for the hypothetical launch of a Space Shuttle vehicle. The meteorological parameters were measured at Kennedy Space Center for the 21 October 1972 case described in Section 5. The parameter input sheet is shown in Figure B-1 and the Program output is shown in Appendix C.2. The data input parameters shown in Figure B-1 were automatically calculated by the Preprocessor Program and are shown in the Preprocessor output in Appendix D.1. Only the HCl normal launch cases with Model 3 and Model 4 are given here. The first namelist deck shown in Figure B-1 is for Model 4 and the second is for Model 3.

B.6 LINKAGE DIAGRAM OF THE NASA/MSFC MULTILAYER DIFFUSION MODELS PROGRAM - VERSION 5

Figure B-2 shows the subroutine linkage diagram for the NASA/MSFC Multilayer Diffusion Models Program - Version 5. Each line terminating at a subroutine name represents a subroutine call. The asterisk indicates subroutines in the SC4020 plot package used only in the UNIVAC 1108 program copy at Marshall

[illegible]

Space Flight Center, Huntsville, Alabama. The program also references the FORTRAN library functions SIN, COS, ATAN2, EXP, SQRT, ALOG and ALOG10 not shown in Figure B-2. A description of the routines shown in Figure B-2 is given in Section B.8.

B.7 DESCRIPTION OF PROGRAM SUBROUTINES

Subroutine MODEL is the main controlling program in the NASA/MSFC Multilayer Diffusion Models Program - Version 5. This routine controls the calculations for Models 1 through 4 and passes control to subroutines that calculate Models 5 and 6.

Subroutine BREAK completes all calculations for Models 1 through 4 (Section 3.1 to 3.4).

Subroutine READER is the data input routine for the Program. All input default values are set in this routine and most meteorological layer parameters are calculated in this routine.

Subroutine DEPOS controls all calculations for Model 6 (Section 3.6).

Subroutine SGP completes all calculations for Model 6 (Section 3.6).

Subroutine WASHT completes all calculations for Model 5 (Section 3.5).

Subroutine TESTR is used to determine special indices used in calculations with MODEL 4. This routine also calculates the downwind distance at which the layer transition for Model 4 begins.

Subroutine IDENT is used to initialize the SC4020 plotting routines used at Marshall Space Flight Center.

Subroutine ENDJOB is used to close out the SC4020 plotting routines at the end of a job.

Subroutine RB8 is used in the calculation of the wind speed and the standard deviation of the elevation and the azimuth wind angles in the surface layer (q, m, p, ql, ml and pl in Sections 3.1 to 3.4).

Subroutine ERTRAN is a time and date routine. This routine is a UNIVAC system utility routine that returns the data and time. The following statement is used to retrieve the 6 character date (MMDDYY) in the first variable and the 6 character time (HHMMSS) in the second variable:

```
CALL ERTRAN(9,NTFB(1), NTFB(2))
```

Subroutine ISO evaluates the normal error function used in the calculation of Model 4 (Equation 3-18, Section 3.4).

Subroutine COORD calculates the downwind and crosswind distances of a grid system calculation point relative to the alongwind cloud axis and the layer source location.

Subroutine RB11 is used in the calculation of the wind speed and the standard deviation of the elevation and azimuth wind angle in the surface layer. (Equations (3-2), (3-5), (3-17), (3-19), (3-23), and (3-27), Section 3.1 to 3.4).

Subroutine SIGMA calculates the crosswind, alongwind and vertical standard deviations of the dosage distribution for Models 1 through 5.

Subroutine GENPRT controls the printing and/or plotting of all model calculations.

Subrouting LABELS generates the page heading information for all calculations.

Subroutine ISSOPT plots the isopleths of concentration, dosage, deposition, and/or time-mean concentration on the SC4020 plotter.

Subroutine LLPLOT plots maximum concentration, dosage, deposition, and/or time-mean concentration as a function of distance in the form of a printer plot on the UNIVAC 1108.

Subroutine PRTTTL prints all page heading information.

Subroutine NMBRS converts a real number to a sequence of BCD characters for insertion into the page heading and for plotting numeric values.

Subroutine LSSOPT plots maximum centerline concentration, dosage, deposition, and/or time-mean concentration as a function of distance on the SC4020 plotter.

Subroutine PACKS removes extra blank characters from the page heading and packs it into a form suitable for printing.

Subroutine MSFLD extracts a bit string (byte) from one word and stores it into another.

Subroutine HEDING sets up key flags that tell the LABELS subroutine whether the label is concentration, dosage, label or other.

Subroutine INPTS is used to set indices for reading or writing records from mass storage.

Subroutine INTOUT writes or reads records from mass storage.

Subroutine VRTCLE determines the vertical axis label for maximum centerline plots.

Subroutine FSTPLT plots information identifying the plots for a set of input data decks.

Subroutine ILAXES draws and labels the plot grids for maximum centerline and isopleth plots.

Subroutine ILPLOT draws and labels the maximum centerline curve and the isopleth curve.

Subroutine MAXMIN determines the maximum and minimum plot values for maximum centerline plots if they are not input.

Subroutine BOUNDS determines whether the plot curve leaves or enters the plot boundaries and interpolates for the intersection of the curve and boundary.

Subroutine SPLINE calculates the coefficients of a third-degree natural spline function used to calculate maximum centerline values and to plot maximum centerline and isopleth curves. The function derives its name from the fact that such a curve approximates the behavior of a mechanical spline used by draftsman to draw a smooth curve through a set of points. The curve has the properties of having continuous first and second derivatives, and of being the "smoothest" curve through a given set of points in the sense that it minimizes

$$\sigma_K = \int_{x_1}^{x_2} [s^k(x)]^2 dx$$

where

s = dosage, concentration, etc.
x = azimuth bearing or distance.
k = derivative = 1, 2

The function $s(x)$ is the unique curve with these properties and is a piece wise function given by a polynomial of maximum degree of 3 in each of the intervals $[x_i, x_i + 1)$, in general by a different polynomial in each such interval. For a more detailed explanation of the spline function, the reader is referred to the book by T. N. E. Greville, Theory and Applications of Spline Functions.

Subroutine LINE2V draws a line between 2 points (SC4020).

Subroutine PRINTV prints alphanumeric information horizontally (SC4020).

Subroutine FRANIEV advances the camera frame (SC4020).

Subroutine LABLV prints a number (SC4020).

Subroutine APRNTV prints alphanumeric information vertically (SC4020).

Subroutine SETMIV sets the plot margins (SC4020).

APPENDIX C
FORTRAN SOURCE LISTINGS FOR THE PREPROCESSOR
PROGRAM AND THE NASA/MSFC MULTILAYER
DIFFUSION MODELS PROGRAM -
VERSION 5

C.1 FORTRAN SOURCE LISTING FOR THE PREPROCESSOR PROGRAM

This section contains the complete FORTRAN source listing of the Pre-processor Program.

```

1*      C      * PLUME RISE AND SOURCE DISTRIBUTION PREPROCESSOR PROGRAM FOR USE      VHC00100
2*      C      * WITH THE NASA/MSFC MULTILAYER MODEL VERSION V.      VHC00200
3*      C      *      VHC00300
4*      C      *      VHC00400
5*      C      *      VHC00500
6*      C      *      VHC00600
7*      C      *      VHC00700
8*      C      *      VHC00800
9*      C      *      VHC00900
10*     C      *      VHC01000
11*     C      *      VHC01100
12*     C      *      VHC01200
13*     C      *      VHC01300
14*     C      *      VHC01400
15*     C      *      VHC01500
16*     C      *      VHC01600
17*     C      *      VHC01700
18*     C      *      VHC01800
19*     C      *      VHC01900
20*     C      *      VHC02000
21*     C      *      VHC02100
22*     C      *      VHC02200
23*     C      *      VHC02300
24*     C      *      VHC02400
25*     C      *      VHC02500
26*     C      *      VHC02600
27*     C      *      VHC02700
28*     C      *      VHC02800
29*     C      *      VHC02900
30*     C      *      VHC03000
31*     C      *      VHC03100
32*     C      *      VHC03200
33*     C      *      VHC03300
34*     C      *      VHC03400
35*     C      *      VHC03500
36*     C      *      VHC03600
37*     C      *      VHC03700
38*     C      *      VHC03800
39*     C      *      VHC03900
40*     C      *      BURVHC04000

COMMON /PLUME/ QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(
121),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL
2,Tv(21),RH(21),NAMCAS(12),NAMT(12),SIGAR
COMMON /REST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS
COMMON /SIG/ SIGX(20),SIGYO(20),SIGZO(20),H
COMMON /F/ DATE(6),FRQ(4),WTMOL(3)
DIMENSION IPOL(4),I1(160),I2(12),I3(61),I4(13),I5(142),ITP(2),JP(2)
1),ISW(7)
DIMENSION WS(21),I6(44)
INTEGER YES,VEHICL,TYPE,TYPES,UNITS
DIMENSION TYPE(8),QC1(8),QT1(8),QC2(8),QT2(8),QC3(8),QT3(8),AA(8),
1BB(8),HEATN(8),HEATA(8),FRQ1(4,8),FRQ2(4),TYPES(12),UNITS(4),
2LNCH(11),LNTL(8),HEATM(8)
DIMENSION DDI(10,4),CCI(10,4),TTI(10,4)
DATA TYPE/3HTTN,3HSTL,3HOTH,3HMIN,4*0/
DATA QC1/4,174374E6,9.384984E6,9.18E5,3.770856E5,4*0./
DATA QT1/4,6752988E8,9.156269E8,6*0./
DATA QC2/1,742E6,3.75257E6,6*0./
DATA QT2/1,95104E8,4.5781349E8,6*0./
DATA QC3/1,301E6,3.05208994E6,2.958E5,9.567765E4,4*0./
DATA QT3/3,903E8,9.156269E8,8.8741E7,2.8703354E7,4*0./
DATA AA/.63463,.663552,1.32095,.439907,4*0./
DATA BB/.4837,.485477,.39457,.47879,4*0./
DATA HEATN/2500.,2582.,625.88,691.0,4*0.0/
DATA HEATM/1036.,1274.,6*0./
DATA HEATA/4*1000.0,4*0./
FRACTIONAL DIST FOR MINUTEMAN II ABNORMAL LAUNCH
DATA FRQ2/.2042109,.2188377,.0,.2799764/
DATA FRQ1/.21,.279,.029,.304,.207,.28,.0,.304,.137278,.307976,.0,
1,24892,.1973,.22046,.0,.27683,16*0./
DATA TYPES/72HTITAN IIIC      SPACE SHUTTLE      DELTA-THOR
1 MINUTEMAN II      /
DATA UNITS/24H PPM      ML/M**3      /
DATA LNCH/66H      NORMAL LAUNCHSINGLE ENGINE BURN      SLOW BURVHC04000

```

41*	INVEHICLE /	VHC04100
42*	DATA NAMT/12*1H /	VHC04200
43*	DATA CCI/16.0,8.0,4.0,1.0,0.5,0.1,4*0.0,35.0,10.0,4.0,2.0,1.0,0.1,	VHC04300
44*	14*0.0,20.0,10.0,5.0,3.0,1.0,0.5,4*0.0,2.0,1.0,0.4,0.1,0.05,0.01,4*	VHC04400
45*	20.0/	VHC04500
46*	DATA DD1/400.0,200.0,100.0,50.0,25.0,5.0,4*0.0,400.0,200.0,100.0,	VHC04600
47*	150.0,25.0,5.0,4*0.0,400.0,200.0,100.0,50.0,25.0,5.0,4*0.0,40.0,	VHC04700
48*	220.0,10.0,5.0,2.5,0.5,4*0.0/	VHC04800
49*	DATA ITI/30.0,4.0,8.0,2.0,1.0,0.5,4*0.0,150.0,100.0,60.0,30.0,15.0	VHC04900
50*	1,1.0,4*0.0,5.0,2.0,1.0,5.1,0.05,4*0.0,50.0,100.0,25.0,10.0,5.0,1,	VHC05000
51*	20,4*0.0/	VHC05100
52*	DATA JP/12H FEET METERS/,NO/3H NO/,NO1/3HN01/,NO2/3HN02/	VHC05200
53*	DATA YES/3HYES/,IPOL/24H HCL CO CO2 AL203/,ITP/1HF,1HK/	VHC05300
54*	DATA IAST/1H*/	VHC05400
55*	DATA IBLK/3H /,IBNK/1H /,NMS/1HM/	VHC05500
56*	COMMON /OUT/ GF(20,4),WD(21),ISKIP(10),NDI,NCI,NTI,ZRK,JBOT,JTOP,	VHC05600
57*	1IZMOD(21),CI(10),DI(10),TI(10),NPS	VHC05700
58*	COMMON /DISPL/ DXX,DX(21),DYY,DY(21),ILXY	VHC05800
59*	EQUIVALENCE (WS,UBAR),(I1,QC),(I2,ZM),(I3,SIGXO),	VHC05900
60*	1(I4,DATE),(I5,QF),(I6,DXX)	VHC06000
61*		VHC06100
62*	*** PROGRAM INPUTS ***	VHC06200
63*	NAMCAS - GENERAL DATA SET TITLING INFORMATION (CARD 1 COL 1-72)	VHC06300
64*	IF INPUT AS BLANKS THE INFORMATION IN THE LAST CASE INPUT	VHC06400
65*	IS USED	VHC06500
66*	VEHICL - THREE CHARACTERS GIVING THE VEHICLE TYPE (CARD 2 COL 2-4)	VHC06600
67*	TTN IS TITAN IIIC VEHICLE	VHC06700
68*	STL IS SPACE SHUTTLE VEHICLE	VHC06800
69*	DTH IS DELTA-THOR VEHICLE	VHC06900
70*	MIN IS MINUTEMAN II VEHICLE	VHC07000
71*	NORMAL - THREE CHARACTERS GIVING THE TYPE OF LAUNCH(CRD 2 COL 5-7)	VHC07100
72*	YES IS A NORMAL LAUNCH	VHC07200
73*	NO1 IS A SINGLE ENGINE BURN ABNORMAL LAUNCH	VHC07300
74*	NO2 IS A SLOW BURN ABNORMAL LAUNCH	VHC07400
75*	IFLEET - 1 CHARACTER IF Z IS IN FEET PUNCH F, IF Z IS IN METERS	VHC07500
76*	PUNCH M, (CARD 2 COL 6)	VHC07600
77*	KNOTS - 1 CHARACTER IF WS IS IN METERS/SEC PUNCH M, IF WS IS IN	VHC07700
78*	KNOTS PUNCH K, (CARD 2 COL 9)	VHC07800
79*	DATE - 36 CHARACTERS IDENTIFYING THE METEOROLOGICAL DATA CASE	VHC07900
80*	WITHIN THE GENERAL DATA CASE IDENTIFIED IN NAMCAS ABOVE	VHC08000
81*	(CARD 2 COL 10-45)	VHC08100

```

82* C SIGAR - STANDARD DEVIATION OF THE WIND AZIMUTH ANGLE AT THE VHC08200
83* C SURFACE MEASUREMENT HEIGHT (DEGREES) (CARD 2 COL 46-55) VHC08300
84* C F10.0 FORMAT ) VHC08400
85* C RHO - SURFACE AIR DENSITY (G/M**3) (CARD 2 COL 56-65 F10.0 FORMAT) VHC08500
86* C ISW(1) - IF SET TO 1 CALCULATE PARAMETERS FOR MODEL 4 VHC08600
87* C IF SET TO 0 MODEL 4 IS NOT PRODUCED (CARD 2 COL 67) VHC08700
88* C ISW(2) - IF SET TO 1 CALCULATE PARAMETERS FOR MODEL 3 VHC08800
89* C IF SET TO 0 MODEL 3 IS NOT PRODUCED (CARD 2 COL 68) VHC08900
90* C ISW(3) - IF SET TO 1 DATA FOR HCL IS PRODUCED VHC09000
91* C IF SET TO 0 HCL IS NOT PRODUCED (CARD 2 COL 69) VHC09100
92* C ISW(4) - IF SET TO 1 DATA FOR CO IS PRODUCED VHC09200
93* C IF SET TO 0 CO IS NOT PRODUCED (CARD 2 COL 70) VHC09300
94* C ISW(5) - IF SET TO 1 DATA FOR AL203 IS PRODUCED VHC09400
95* C IF SET TO 0 AL203 IS NOT PRODUCED (CARD 2 COL 71) VHC09500
96* C ISW(6) - IF SET TO 1 DATA FOR CO2 IS PRODUCED VHC09600
97* C IF SET TO 0 CO2 IS NOT PRODUCED (CARD 2 COL 72) VHC09700
98* C ISW(7) - IF SET TO 1 THE CLOUD TRAJECTORY COORDINATES DELX,DELY VHC09800
99* C ARE CALCULATED AND PUNCHED FOR EACH LAYER. IF SET TO 0 VHC09900
100* C CLOUD TRAJECTORY COORDINATES ARE NOT CALCULATED VHC10000
101* C (CARD 2 COL 73) VHC10100
102* C THE FOLLOWING PARAMETERS EXCEPT IHM ALL USE AN F10.0 FORMAT VHC10200
103* C Z - HEIGHT OF LAYER BOUNDARIES (FEET OR METERS) COL 1-10 CARDS 3-NVHC10300
104* C WD - WIND DIRECTION AT EACH Z (DEGREES) COL 11-20 CARDS 3-N VHC10400
105* C WS - WIND SPEED AT EACH Z (KNOTS OR METERS/SEC) COL 21-30 CARDS 3-NVHC10500
106* C T - TEMPERATURE AT EACH Z (DEGREES C) COL 31-40 CARDS 3-N VHC10600
107* C P - PRESSURE AT EACH Z (MILLIBARS) COL 41-50 CARDS 3-N VHC10700
108* C RH - RELATIVE HUMIDITY AT EACH Z (PERCENT) COL 51-60 CARDS 3-N VHC10800
109* C IHM - ASTERISK (*) IN COLUMN 80 IF THE HEIGHT Z ON THIS CARD IS VHC10900
110* C THE SURFACE MIXING LAYER HEIGHT HM, IF NOT FOUND THE LAST Z INPUT VHC11000
111* C IS USED FOR HM. VHC11100
112* C VHC11200
113* C VHC11300
114* C VHC11400
115* C VHC11500
116* C VHC11600
117* C VHC11700
118* C VHC11800
119* C VHC11900
120* C VHC12000
121* C VHC12100
122* C VHC12200

5 DO 10 I=1,160
  I1(I) = 0
  IF (I .GT. 142) GO TO 10
  I5(I) = 0
  IF (I .GT. 61) GO TO 10
  I3(I) = 0
  IF (I .GT. 44) GO TO 10
  I6(I) = 0
  IF (I .GT. 12) GO TO 10

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12(I) = 0
IF (I .GT. 13) GO TO 10
14(I) = 0
10 CONTINUE

      *** PROGRAM CONSTANTS ***
      WC1-TOTAL SOURCE OUTPUT RATE IN GRAMS/SEC FOR A NORMAL LAUNCH
      WT1- TOTAL SOURCE STRENGTH IN GRAMS FOR NORMAL LAUNCH
      WC2 - TOTAL SOURCE OUTPUT RATE IN GRAMS/SEC FOR AN ABNORMAL LAUNCH
      WITH ONE ENGINE BURNING ON PAD
      WT2 - TOTAL SOURCE STRENGTH IN GRAMS FOR AN ABNORMAL LAUNCH WITH
      ONE ENGINE BURNING ON PAD
      WC3 - TOTAL SOURCE OUTPUT RATE IN GRAMS/SEC FOR AN ABNORMAL LAUNCH
      WHERE ENGINES EXPLODE AND BURN ON GROUND
      WT3 - TOTAL SOURCE STRENGTH IN GRAMS FOR AN ABNORMAL LAUNCH WHERE
      THE ENGINES EXPLODE AND BURN ON GROUND
      AA AND BB - ROCKET RISE PARAMETERS IN EQUATION TR=AA*Z**BB
      HEATN - HEAT OUTPUT (CAL/G) NORMAL LAUNCH
      HEATM - HEAT OUTPUT (CAL/G) ABNORMAL LAUNCH WITH SINGLE ENGINE
      BURN
      HEATA - HEAT OUTPUT (CAL/G) ABNORMAL LAUNCH WITH SLOW BURN ON PAD
      GAMMAI - ENTRAINMENT PARAMETER FOR NORMAL LAUNCH
      GAMMAI = 0.64
      GAMMAC - ENTRAINMENT PARAMETER FOR ABNORMAL LAUNCH
      GAMMAC = 0.5
      FRG - FRACTIONAL DISTRIBUTION OF MATERIAL FOR HCL, CO, CO2, AL2O3
      WTMOL - MOLECULAR WEIGHTS OF HCL, CO, CO2
      WTMOL(1) = 36.46
      WTMOL(2) = 28.01
      WTMOL(3) = 44.01
      G - ACCELERATION OF GRAVITY (M/SEC SQUARE)
      G = 9.8
      CP - SPECIFIC HEAT OF AIR
      CP = 0.24
      PI - RADIAN IN 180 DEGREES
      PI = 3.1415926
      NDI = 69
      NCI = 69
      ISKIP(1) = 0
      ISKIP(2) = 3
      ISKIP(3) = 3
VHC12300
VHC12400
VHC12500
VHC12600
VHC12700
VHC12800
VHC12900
VHC13000
VHC13100
VHC13200
VHC13300
VHC13400
VHC13500
VHC13600
VHC13700
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VHC15000
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VHC15200
VHC15300
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VHC15800
VHC15900
VHC16000
VHC16100
VHC16200
VHC16300

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NPS = 0
READ 1002, NAMCAS
DO 15 I=1,12
  IF (NAMCAS(I) .NE. IBLK) GO TO 17
15 CONTINUE
DO 16 I=1,12
  16 NAMCAS(I) = NAMT(I)
17 DO 18 I=1,12
  18 NAMT(I) = NAMCAS(I)
  READ 1000, VEHICL,NORMAL,IFEET,KNOTS,DATE,SIGAR,RHO,ISW
  SIGAP(1) = 0.5*SIGAR
  N = 1
  20 READ 1001, Z(K),WD(K),WS(K),T(K),P(K),RH(K),IHM
  IF (IHM .EQ. IAST) HM = Z(K)
  IF (Z(K)+WS(K)) 21,22,21
  21 K = K+1
  GO TO 20
22 NZS = K-1
  IPNPS = 4
  IF (ISW(5) .GT. 0) GO TO 45
  IPNPS = 3
  IF (ISW(6) .GT. 0) GO TO 45
  IPNPS = 2
  IF (ISW(4) .GT. 0) GO TO 45
  IPNPS = 1
45 CONTINUE
  C
  ZKK = HEIGHT AT WHICH SIGAR IS MEASURED (METERS)
  ZKK = Z(1)
  IF (NORMAL .EQ. IBLK) NORMAL = YES
  IF (VEHICL .EQ. TYPE(1)) GO TO 50
  IF (VEHICL .EQ. TYPE(2)) GO TO 51
  IF (VEHICL .EQ. TYPE(3)) GO TO 52
  IF (VEHICL .EQ. TYPE(4)) GO TO 53
  PRINT 2009
50 JV = 1
  GO TO 60
51 JV = 2
  ISW(6) = 0
  GO TO 60
52 JV = 3
  ISW(6) = 0

```

VHC16400
VHC16500
VHC16600
VHC16700
VHC16800
VHC16900
VHC17000
VHC17100
VHC17200
VHC17300
VHC17400
VHC17500
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VHC17700
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VHC17900
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VHC18900
VHC19000
VHC19100
VHC19200
VHC19300
VHC19400
VHC19500
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VHC19700
VHC19800
VHC19900
VHC20000
VHC20100
VHC20200
VHC20300
VHC20400

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IF (NORMAL.EQ. NO1) NORMAL = NO2
GO TO 60
53 JV = 4
    ISW(6) = 0
    IF (NORMAL.EQ. NO1) NORMAL = NO2
    60 IF (JV.NE. 4) GO TO 61
    IF (NORMAL.NE. YES) GO TO 63
    61 DO 62 I=1,4
    62 FRQ(I) = FRQ1(I,JV)
    GO TO 65
    63 DO 64 I=1,4
    64 FRQ(I) = FRQ2(I)
    65 CONTINUE
    IF (IFEET.EQ. IBNK) IFEET = NMS
    IF (KNOTS.EQ. IBNK) KNOTS = NMS
    IF (NORMAL.EQ. YES) GO TO 32
    IF (NORMAL.EQ. NO1) GO TO 31
    IF (NORMAL.EQ. NO2) GO TO 30
    PRINT 2006, NORMAL
    GO TO 500
30 QC = QC3(JV)
    QT = QT3(JV)
    IJM = 7
    GAMMA = GAMMAC
    HEAT = HEATA(JV)
    NORMAL = 0
    ISKIP(6) = 4
    GO TO 33
31 QC = QC2(JV)
    QT = QT2(JV)
    IJM = 4
    GAMMA = GAMMAC
    HEAT = HEATM(JV)
    NORMAL = 0
    ISKIP(6) = 3
    GO TO 33
32 QC = QC1(JV)
    QT = QT1(JV)
    IJM = 1
    GAMMA = GAMMAI
    HEAT = HEATN(JV)

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VHC20500
VHC20600
VHC20700
VHC20800
VHC20900
VHC21000
VHC21100
VHC21200
VHC21300
VHC21400
VHC21500
VHC21600
VHC21700
VHC21800
VHC21900
VHC22000
VHC22100
VHC22200
VHC22300
VHC22400
VHC22500
VHC22600
VHC22700
VHC22800
VHC22900
VHC23000
VHC23100
VHC23200
VHC23300
VHC23400
VHC23500
VHC23600
VHC23700
VHC23800
VHC23900
VHC24000
VHC24100
VHC24200
VHC24300
VHC24400
VHC24500

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286*

NORMAL = 1
1SKIP(0) = 2
33 CONTINUE
A = AA(JV)
B = BB(JV)
N = 3*JV-2
M = N+2
DO 35 I=1,3
J = IJM+I-1
35 LNTL(I) = LNCH(J)
J = 3
DO 36 I=N,M
J = J+1
36 LNTL(J) = TYPES(1)
LNTL(7) = LNCH(10)
LNTL(6) = LNCH(11)
IF (HM .GT. 0.0) GO TO 40
HM = Z(NZS)
40 CONTINUE
PRINT 2005, (LNTL(J),J=1,8)
CALL CONSI(0,NO,YES,IPOL,ITP,IFEET,KNOTS)
CONVERT FEET TO METERS IF IFEET = F
IF (IFEET .NE. ITP(1)) GO TO 76
DO 75 K=1,NZS
75 Z(K) = Z(K)*.3048
HM = HM*.3048
ZRK = ZRK*.3048
76 CONTINUE
CONVERT KNOTS TO METERS/SEC IF KNOTS = K
IF (KNOTS .NE. ITP(2)) GO TO 78
DO 77 K=1,NZS
77 WS(K) = WS(K)*.514791
78 CONTINUE
80 DO 81 K=1,NZS
IF (Z(K)-1.0 .LT. HM.AND.HM .LT. Z(K)+1.0) GO TO 82
81 CONTINUE
GO TO 400
82 KS = K
C
CONVERT TEMPERATURE FROM DEGREES CELSIUS TO ABSOLUTE
DO 90 K=1,NZS
90 T(K) = T(K)+273.16

VHC24600
VHC24700
VHC24800
VHC24900
VHC25000
VHC25100
VHC25200
VHC25300
VHC25400
VHC25500
VHC25600
VHC25700
VHC25800
VHC25900
VHC26000
VHC26100
VHC26200
VHC26300
VHC26400
VHC26500
VHC26600
VHC26700
VHC26800
VHC26900
VHC27000
VHC27100
VHC27200
VHC27300
VHC27400
VHC27500
VHC27600
VHC27700
VHC27800
VHC27900
VHC28000
VHC28100
VHC28200
VHC28300
VHC28400
VHC28500
VHC28600

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287*      C      CALCULATE VIRTUAL POTENTIAL TEMPERATURE
288*      DO 100 K=1,NZS
289*      XT = 1000.0/T(K)
290*      XT = 8.42926604-XT*(1.82717843+.071208271*XT)
291*      XT = RH(K)*.01*10.0**XT
292*      XT = 0.622*XT/(P(K)-XT)
293*      XT = T(K)*(1.0+1.61*XT)/(1.0+XT)
294*      TV(K) = CPHI(XT,P(K))
295*      CALCULATE PLUME RISE
296*      IF (NORMAL.EQ. 0) GO TO 120
297*      CALL PLUME1
298*      IF (IFLG.GT. 0) GO TO 410
299*      IF (JV.NE. 3) GO TO 130
300*      ZMSV = ZM
301*      GAMMA = GAMMAC
302*      120 CONTINUE
303*      CALL PLUME2
304*      IF (IFLG.GT. 0) GO TO 410
305*      IF (JV.NE. 3) GO TO 130
306*      IF (NORMAL.EQ. 0) GO TO 130
307*      GAMMA = .5*(GAMMA1+GAMMAC)
308*      ZM = .5*(ZM+ZMSV)
309*      DO 121 I=2,NZS
310*      IF (ZM.LT. Z(I)) GO TO 122
311*      121 CONTINUE
312*      CALL LEAS1(Z,TV,DPDZ,1,0,0,0,0,U)
313*      IF (DPDZ.LT. 3.322E-4) DPDZ = 3.322E-4
314*      130 CONTINUE
315*      IF (ISW(7).NE. 0) CALL DELTX
316*      CALCULATE TURBULENCE PARAMETERS
317*      CALL TURB
318*      CALCULATE SOURCE DISTRIBUTION FOR MODEL 4
319*      CALL DIST4
320*      CALCULATE SOURCE DIMENSIONS FOR MODEL 4
321*      IFLG = 1
322*      CALL DIM34
323*      JBOT = 1
324*      NMZ = NZS-1
325*      JTOP = KS-1
326*      DO 132 I=JBOT,JTOP
327*      132 IZMOD(I) = 4

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VHC28700
VHC28800
VHC28900
VHC29000
VHC29010
VHC29100
VHC29200
VHC29300
VHC29400
VHC29500
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VHC29700
VHC29800
VHC29900
VHC30000
VHC30100
VHC30200
VHC30300
VHC30400
VHC30500
VHC30600
VHC30700
VHC30800
VHC30900
VHC31000
VHC31100
VHC31200
VHC31300
VHC31400
VHC31500
VHC31600
VHC31700
VHC31800
VHC31900
VHC32000
VHC32100
VHC32200
VHC32300
VHC32400
VHC32500
VHC32600

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328* IF (JTOP .GE. NNZ) GO TO 134
329* J = JTOP+1
330* DO 133 I=J,NNZ
331* 133 I2MOD(I) = 0
332* 134 CONTINUE
333* XX = 1.0E3*22.4*1013.2*T(1)/(273.16*P(1))
334* I1 = 4
335* C OUTPUT NAMELIST NAM2 FOR HCL, CO, CO2, AL2O3 MODEL 4
336* UC 200 I=1,4
337* ISKIP(5) = 1
338* NT1 = 61
339* IF (I .EQ. 1) NT1 = 62
340* IF (I .EQ. 3) NT1 = 69
341* DO 135 J=1,10
342* UI(J) = UI(J,I)
343* TI(J) = TI(J,I)
344* 135 CI(J) = CCI(J,I)
345* C CALCULATE CONVERSION FACTOR TO PPM FOR HCL, CO, CO2 AND TO
346* C MILLIGRAMS PER CUBIC METER FOR AL2O3 AND ADJ FOR PERCENT OF MAT.
347* IF (I .EQ. 4) GO TO 140
348* WK = (XX/WTMOL(I))*FRQ(I)
349* GO TO 150
350* 140 WK = 1.0E3*FRQ(1)
351* C CONVERT % TO PROPER UNITS AND PERCENTAGE OF POLLUTANT
352* 150 DO 160 K=1,NNZ
353* 160 QF(K,I) = QK*Q(K)
354* IF (ISW(1) .EQ. 0) GO TO 200
355* IF (ISW(3) .EQ. 0 .AND. I .EQ. 1) GO TO 200
356* IF (ISW(4) .EQ. 0 .AND. I .EQ. 2) GO TO 200
357* IF (ISW(5) .EQ. 0 .AND. I .EQ. 4) GO TO 200
358* IF (ISW(6) .EQ. 0 .AND. I .EQ. 3) GO TO 200
359* PRINT 2005,NAMCAS,(LNTL(J),J=1,8)
360* PRINT 2007
361* K = 1
362* IF (I .EQ. 4) K = 3
363* IF (ISW(2) .GT. 0) GO TO 170
364* IF (IHM .NE. IBNK) GO TO 170
365* IF (I .EQ. IPNPS) NPS = 1
366* 170 CONTINUE
367* PRINT 2001, (DATE(J),J=1,6),(TYPES(J),J=N,M)
368* PUNCH 2001, (DATE(J),J=1,6),(TYPES(J),J=N,M)

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369*      180 CALL OUTPI(KS,I,4)
370*      200 CONTINUE
371*      IF (ISW(1).EQ.0) GO TO 205
372*      PRINT 2005, NAMCAS, (LNTL(J),J=1,8)
373*      CALL CONST(4,NO,YES,IPOL,ITP,IFEET,KNOTS)
374*      205 CONTINUE
375*      C      OUTPUT NAMELIST NAM2 FOR HCL, CO, CO2, AL2O3 MODEL 3
376*      C      CALCULATE SOURCE DISTRIBUTION FOR MODEL 3
377*      C      CALL DIST3
378*      C      CALCULATE SOURCE DIMENSIONS FOR MODEL 3
379*      IFLG = 0
380*      CALL DIM34
381*      Z(2) = HM
382*      SIGAP(2) = SIGAP(KS)
383*      SIGEP(2) = SIGEP(KS)
384*      WD(2) = WD(KS)
385*      UBAR(2) = UBAR(KS)
386*      IZMOD(1) = 3
387*      DX(1) = DX(ILXY)
388*      DY(1) = DY(ILXY)
389*      IF (SIGZ0(1).LE.0.0) GO TO 420
390*      NZ = 1
391*      NZS = 2
392*      KS = 2
393*      I1 = 3
394*      DO 260 I=1,4
395*      ISNIP(5) = I
396*      NI1 = 61
397*      IF (I.EQ.1) NI1 = 62
398*      IF (I.EQ.3) NI1 = 69
399*      DO 210 J=1,10
400*      UF(J) = UFI(J,I)
401*      TI(J) = TII(J,I)
402*      CI(J) = CCI(J,I)
403*      210 CALCULATE CONVERSION FACTOR TO PPM FOR HCL, CO, CO2 AND TO
404*      C      MILLIGRAMS PER CUBIC METER FOR AL2O3 AND ADJ FOR PERCENT OF MAT.
405*      IF (I.EQ.4) GO TO 220
406*      QF(1,I) = Q(1)*(XX/WTMOL(I))*FRQ(I)
407*      GO TO 230
408*      220 QF(1,I) = Q(1)*1.0E3*FRQ(I)
409*      230 CONTINUE

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VHC36800
VHC36900
VHC37000
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VHC39800
VHC39900
VHC40000
VHC40100
VHC40200
VHC40300
VHC40400
VHC40500
VHC40600
VHC40700
VHC40800

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410*      IF (ISW(2) .EQ. 0) GO TO 260
411*      IF (ISW(3) .EQ. 0.AND.I .EQ. 1) GO TO 260
412*      IF (ISW(4) .EQ. 0.AND.I .EQ. 2) GO TO 260
413*      IF (ISW(5) .EQ. 0.AND.I .EQ. 4) GO TO 260
414*      IF (ISW(6) .EQ. 0.AND.I .EQ. 3) GO TO 260
415*      PRINT 2005,NAMCAS,(LNTL(J),J=1,8)
416*      PRINT 2007
417*      K = 1
418*      IF (I .EQ. 4) K = 3
419*      IF (IHM .NE. IBNK) GO TO 240
420*      IF (I .EQ. IPNPS) NPS = 1
421*      CONTINUE
422*      PRINT 2001, (DATE(J),J=1,6),(TYPES(J),J=N,M)
423*      PUNCH 2001, (DATE(J),J=1,6),(TYPES(J),J=N,M)
424*      CALL OUTP1(KS,I,3)
425*      CONTINUE
426*      IF (ISW(2) .EQ. 0) GO TO 500
427*      PRINT 2003, NAMCAS,(LNTL(J),J=1,8)
428*      CALL CONST(3,NO,YES,IPOL,ITP,IFEET,KNOTS)
429*      GO TO 500
430*      PRINT 2003
431*      GO TO 500
432*      II = 2
433*      IF (IFEET .NE. ITP(1)) GO TO 411
434*      II = 1
435*      ZN = ZM/.3048
436*      PRINT 2004, ZM,JP(II)
437*      GO TO 500
438*      PRINT 2006
439*      IF (IHM .NE. IBNK) GO TO 5
440*      STOP
441*      FORMAT (1X,2A3,2A1,6A6,2F10.0,1X,7I1)
442*      FORMAT (6F10.4,19X,A1)
443*      FORMAT (12A6)
444*      FORMAT (6H $NAM2/11H TESTNO=60H,9A6,7H
445*      FORMAT (84H0 **ERROR** HM MUST BE EQUAL TO ONE OF THE LAYER BOUNDARIES
446*      IRIES 2 AND IN THE SAME UNITS./)
447*      FORMAT (70H0 **ERROR** NOT ENOUGH LAYERS, TOP OF LAST LAYER MUST
448*      BE GREATER THAN ,1PE12.5,1X,A6,19H, INPUT MORE LAYERS/)
449*      FORMAT (1H1,21X,6H*--** ,12A6,8H *--**/29X,8H*--** ,3A6,1X,5VHC44800
450*      1A6,8H *--**/)
VHC40900
VHC41000
VHC41100
VHC41200
VHC41300
VHC41400
VHC41500
VHC41600
VHC41700
VHC41800
VHC41900
VHC42000
VHC42100
VHC42200
VHC42300
VHC42400
VHC42500
VHC42600
VHC42700
VHC42800
VHC42900
VHC43000
VHC43100
VHC43200
VHC43300
VHC43400
VHC43500
VHC43600
VHC43700
VHC43800
VHC43900
VHC44000
VHC44100
VHC44200
VHC44300
VHC44400
VHC44500
VHC44600
VHC44700
VHC44800
VHC44900

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451*	2006	FORMAT (5dH ***EKROR***, COLUMNS 1-3 ARE INCORRECTLY PUNCHED ON CAR)	VHC45000
452*		10 1)	VHC45100
453*	2007	FORMAT (2dX, 75H*--* NAMELIST NAM2 FOR INPUT TO THE NASA/MSFC MULTIL	VHC45200
454*		1AYER MODEL VERSION 5 *--*//)	VHC45300
455*	2008	FORMAT (102H1*--* CLOUD RISE IS WELL ABOVE HM. MODEL 3 PARAMETERS	AVHC45400
456*		ARE NOT CALCULATED. USE MODEL 4 FOR THIS CASE *--*)	VHC45500
457*	2009	FORMAT (5dH *WARNING* VEHICLE TYPE NOT SPECIFIED, TITAN IIIC USED)	VHC45600
458*		END	VHC45700

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SUBROUTINE CONV(W,I,J,K)
DIMENSION ICHAR(12)
DATA ICHAR/1H0,1H1,1H2,1H3,1H4,1H5,1H6,1H7,1H8,1H9,1H+,1H-/
R = 0.0
IP = 0
K = ICHAR(11)
IF (Q) 100,55,10
10 IF (Q-1.0) 20,30,40
20 IP = 0
K = Q
21 IF (R.GT. 1.0) GO TO 50
IP = IP-1
R = R*10.0
GO TO 21
30 K = Q
IP = 0
GO TO 55
40 IP = 0
K = Q
41 IF (R.LT. 10.0) GO TO 50
IP = IP+1
R = R*0.1
GO TO 41
50 CONTINUE
Q = R
K = ICHAR(11)
IF (IP.LI. 0) K = ICHAR(12)
55 I = IABS(IP/10)
J = IABS(IP)-I*10
GO 70 L=1,10
IF (I.NE. L-1) GO TO 70
I = ICHAR(L)
GO TO 71
70 CONTINUE
GO TO 110
71 CONTINUE
DO 80 L=1,10
IF (J.NE. L-1) GO TO 80.
J = ICHAR(L)
GO TO 81

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CNV00100
CNV00200
CNV00300
CNV00400
CNV00500
CNV00600
CNV00700
CNV00800
CNV00900
CNV01000
CNV01100
CNV01200
CNV01300
CNV01400
CNV01500
CNV01600
CNV01700
CNV01800
CNV01900
CNV02000
CNV02100
CNV02200
CNV02300
CNV02400
CNV02500
CNV02600
CNV02700
CNV02800
CNV02900
CNV03000
CNV03100
CNV03200
CNV03300
CNV03400
CNV03500
CNV03600
CNV03700
CNV03800
CNV03900
CNV04000

41*	80	CONTINUE	CNV04100
42*		GO TO 110	CNV04200
43*	61	CONTINUE	CNV04300
44*	90	RETURN	CNV04400
45*	100	PRINT 2000, Q	CNV04500
46*		GO TO 90	CNV04600
47*	110	PRINT 2001, Q	CNV04700
48*		GO TO 90	CNV04800
49*	2000	FORMAT (39H *--ERROR*--* SOURCE STRENGTH NEGATIVE =,E15.8)	CNV04900
50*	2001	FORMAT (47H *--ERROR*--* POWER OF 10 ON Q TOO MANY DIGITS =,E15.8)	CNV05000
51*		END	CNV05100

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21*

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SUBROUTINE TURB
COMMON /PLUME/ QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(
121),NZS,G(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL
2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAR
COMMON /REST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS
PHI1 = G*DPDZ/T(1)
TAUK = PI/SQRT(PHI1)
IF (TAUK .GT. 600.0) TAUK = 600.0
K = 0
10 K = K+1
IF (K .GT. NZS) GO TO 40
IF (Z(K) .GT. HM) GO TO 35
SIGAP(K) = SIGAP(1)
SIGEP(K) = SIGEP(1)
GO TO 10
35 SIGAP(K) = 0.1
SIGEP(K) = 0.1
GO TO 10
40 CONTINUE
RETURN
END

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TRB00100
TRB00200
TRB00300
TRB00400
TRB00500
TRB00600
TRB00700
TRB00800
TRB00900
TRB01000
TRB01100
TRB01200
TRB01300
TRB01400
TRB01500
TRB01600
TRB01700
TRB01800
TRB01900
TRB02000
TRB02100

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1* SUBROUTINE DIM34
2* COMMON /PLUME/ QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(D3400200
3* 121),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,FAUK,NORMAL
4* 2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAR
5* COMMON /SIG/ SIGX(20),SIGY(20),SIGZ(20),H
6* COMMON /REST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS
7* IF(IFLG.EQ.1) GO TO 30
8* SOURCE DIMENSIONS FOR MODEL 3
9* IF (ZM.L1. HM-GAMMA*ZM) GO TO 10
10* SIGZO(1) = (HM-ZM+GAMMA*ZM)*.2325581
11* H = (HM+ZM-GAMMA*ZM)*0.5
12* GO TO 20
13* 10 SIGZO(1) = GAMMA*ZM*.465116279
14* H = ZM
15* 20 SIGX(1) = GAMMA*ZM*.465116279
16* SIGY(1) = SIGX(1)
17* IF (SIGZO(1).GT. 0.0) GO TO 50
18* H = 0.5*(HM-Z(1))
19* SIGZO(1) = GAMMA*H*.465116279
20* GO TO 50
21* SOURCE DIMENSIONS FOR MODEL 4
22* 30 DO 40 K=2,NZS
23* ZK = Z(K-1)
24* IF (K.EQ. 2) ZK = 0.0
25* ZP = 0.5*(Z(K)-ZK)+ZK
26* IF (ZP.GT. ZM) GO TO 35
27* SXO = ZM
28* GO TO 36
29* 35 SXO = (2.0*ZM-ZP)
30* 36 SXO = SXO*GAMMA*.465116279
31* IF (SXO.LT. 0.0) SXO = 0.0
32* IF (NORMAL.EQ. 0) GO TO 38
33* IF (ZP.LE. ZM) GO TO 38
34* IF (SXO.LT. 93.0) SXO = 93.0
35* SIGX(K-1) = SXO
36* SIGY(K-1) = SXO
37* SIGZO(K-1) = 0.0
38* CONTINUE
39* H = ZM
40* RETURN
41* END

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D3400100
D3400200
D3400300
D3400400
D3400500
D3400600
D3400700
D3400800
D3400900
D3401000
D3401100
D3401200
D3401300
D3401400
D3401500
D3401600
D3401700
D3401800
D3401900
D3402000
D3402100
D3402200
D3402300
D3402400
D3402500
D3402600
D3402700
D3402800
D3402900
D3403000
D3403100
D3403200
D3403300
D3403400
D3403500
D3403600
D3403700
D3403800
D3403900
D3404000
D3404100

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PHI00100
PHI00200
PHI00300
PHI00400

FUNCTION CPHI(A,B)
CPHI = A*(1000.0/B)**0.288
RETURN
END

1*
2*
3*
4*

TPZ00100
TPZ00200
TPZ00300
TPZ00400

FUNCTION TPZ(A,B,C,D,E)
TPZ = (A-B)*(C-D)/(A-E)
RETURN
END

1*
2*
3*
4*

```

1* SUBROUTINE DIST4
2* COMMON /PLUME/ WC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(DS400200
3* 121),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL
4* 2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAK
5* COMMON /KEST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS
6* IF (NORMAL.EQ. 0) GO TO 5
7* QQ = WC*A*ZM**B
8* GO TO 6
9* 5 QQ = QT
10* 6 SQ2I = 1.0/(1.41421356*GAMMA*ZM*.465116279)
11* K = 1
12* PHI = 0.0
13* 10 K = K+1
14* IFLG = 0
15* ZP = (Z(K)-ZM)*SQ2I
16* IF (ZP) 20,15,30
17* 15 CONTINUE
18* PZ = 0.5
19* GO TO 60
20* 20 CONTINUE
21* ZP = -ZP
22* IFLG = 1
23* 30 A1 = 0.5/(ZP*ZP)
24* TZ = 1.0
25* IF (ZP.GI. 6.99) GO TO 50
26* B1 = ZP*1.12837917/EXP(ZP*ZP)
27* TZ = B1
28* 40 A1 = A1+1.0/(ZP*ZP)
29* PHI1 = B1/A1
30* TZ = TZ+PHI1
31* IF (PHI1.LE. 1.0E-10) GO TO 50
32* B1 = PHI1
33* GO TO 40
34* 50 PZ = 0.5*(1.0+TZ)
35* IF (IFLG.EQ. 1) PZ = 1.0-PZ
36* 60 Q(K-1) = (PZ-PHI1)*QQ
37* IF (Q(K-1).LT. 0.0) Q(K-1) = 0.0
38* IF (Q(K-1).LT. 1.0E-20).QQ = 0.0
39* PHI = PZ
40* IF (K.LT. NZS) GO TO 10
DS400100
DS400200
DS400300
DS400400
DS400500
DS400600
DS400700
DS400800
DS400900
DS401000
DS401100
DS401200
DS401300
DS401400
DS401500
DS401600
DS401700
DS401800
DS401900
DS402000
DS402100
DS402200
DS402300
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DS402700
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DS403000
DS403100
DS403200
DS403300
DS403400
DS403500
DS403600
DS403700
DS403800
DS403900
DS404000

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IF (NORMAL .EQ. 0) GO TO 90
K = 2
ZP = ZM
70 IF (Z(K) .GE. ZM) GO TO 80
K = K+1
IF (K .LE. NZS) GO TO 70
GO TO 90
80 IF (K .GT. NZS) GO TO 90
Q(K-1) = QC*A*(Z(K)**B-ZP**B)+Q(K-1)
ZP = Z(K)
K = K+1
GO TO 80
90 CONTINUE
RETURN
END

```

DS404100
DS404200
DS404300
DS404400
DS404500
DS404600
DS404700
DS404800
DS404900
DS405000
DS405100
DS405200
DS405300
DS405400
DS405500

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1*      SUBROUTINE DIST3
2*      COMMON /PLUME/ QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(
3*      121),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL
4*      2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAR
5*      COMMON /REST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS
6*      IF (NORMAL.EQ. 0) GO TO 2
7*      QQ = QC*A*ZM**B
8*      GO TO 3
9*      2 QQ = QT
10*      3 CONTINUE
11*      IFLG = 0
12*      ZP = (HM-ZM)/(1.41421356*GAMMA*ZM*.465116279)
13*      PZ = 0.5
14*      IF (ZP) 5,40,10
15*      5 CONTINUE
16*      ZP = -ZP
17*      IFLG = 1
18*      10 A1 = 0.5/(ZP*ZP)
19*      TZ = 1.0
20*      IF (ZP .GT. 6.99) GO TO 30
21*      B1 = ZP*1.12837917/EXP(ZP*ZP)
22*      TZ = B1
23*      20 A1 = A1+1.0/(ZP*ZP)
24*      PHI1 = B1/A1
25*      TZ = TZ+PHI1
26*      IF (PHI1 .LE. 1.0E-10) GO TO 30
27*      B1 = PHI1
28*      GO TO 20
29*      30 PZ = 0.5*(1.0+TZ)
30*      IF (IFLG.EQ. 1) PZ = 1.0-PZ
31*      40 W(1) = PZ*QQ
32*      RETURN
33*      END
DS300100
DS300200
DS300300
DS300400
DS300500
DS300600
DS300700
DS300800
DS300900
DS301000
DS301100
DS301200
DS301300
DS301400
DS301500
DS301600
DS301700
DS301800
DS301900
DS302000
DS302100
DS302200
DS302300
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DS302500
DS302600
DS302700
DS302800
DS302900
DS303000
DS303100
DS303200
DS303300

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1* SUBROUTINE OUTPT(NZ,I,IK)
2* COMMON /PLUME/ QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(OPT00200
3* 121),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL
4* 2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAR
5* COMMON /SIG/ SIGXO(20),SIGYO(20),SIGZO(20),H
6* COMMON /OUT/ QF(20,4),WD(21),ISKIP(10),NDI,NCI,NTI,ZRK,JBOT,JTOP,
7* 11ZMOD(21),CI(10),DI(10),TI(10),NPS
8* COMMON /DISPL/ DX,DX(21),DYY,DY(21),ILXY
9* DIMENSION QL(20),IQ(20),JQ(20),KQ(20)
10* DIMENSION JE(15)
11* DATA JE/9UH 2 Q UBARK SIGAK SIGEK SIGXO SIGYO SIGZOTHETAK
12* 1 DI CI DELX DELY TI TEMPK/
13* 1ZM = NZ
14* IF (NZ.GT. 16) NZ = 16
15* NZ2 = NZ-1
16* NDI = NDI/10
17* NCI = NCI/10
18* NTI = NTI/10
19* ZF = 0.0
20* TIMAV = 600.0
21* IF (I.EQ. 2) TIMAV = 360.0
22* PRINT 1999, NAMCAS
23* 1999 FORMAT (1X,10HNAMCAS=68H,11A6,A2,1H,)
24* PRINT 2000, ISKIP,NPS,NZ,NDI,NCI,NTI,ZRK,TAUK,(IZMOD(K),K=1,NNZ)
25* PRINT 2001, JE(1),ZF,(Z(K),K=2,NZ)
26* PRINT 2002, JE(2),(QF(K,I),K=1,NNZ)
27* PRINT 2001, JE(3),(UBAR(K),K=1,NZ)
28* PRINT 2001, JE(4),(SIGAP(K),K=1,NZ)
29* PRINT 2001, JE(5),(SIGEP(K),K=1,NZ)
30* PRINT 2001, JE(6),(SIGXO(K),K=1,NNZ)
31* PRINT 2001, JE(7),(SIGYO(K),K=1,NNZ)
32* PRINT 2001, JE(8),(SIGZO(K),K=1,NNZ)
33* PRINT 2001, JE(9),(WD(K),K=1,NZ)
34* PRINT 2000, TIMAV
35* PRINT 2001, JE(10),(DI(K),K=1,NDI)
36* PRINT 2001, JE(11),(CI(K),K=1,NCI)
37* PRINT 2001, JE(14),(TI(K),K=1,NTI)
38* PRINT 2001, JE(12),(DX(K),K=1,NNZ)
39* PRINT 2001, JE(13),(DY(K),K=1,NNZ)
40* PRINT 2001, JE(15),(TV(K),K=1,NZ)
OPT00100
OPT00200
OPT00300
OPT00400
OPT00500
OPT00600
OPT00700
OPT00800
OPT00900
OPT01000
OPT01100
OPT01200
OPT01300
OPT01400
OPT01500
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OPT01700
OPT01800
OPT01900
OPT02000
OPT02100
OPT02110
OPT02220
OPT02200
OPT02300
OPT02400
OPT02500
OPT02600
OPT02700
OPT02800
OPT02900
OPT03000
OPT03100
OPT03200
OPT03300
OPT03400
OPT03500
OPT03600
OPT03700
OPT03800

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PRINT 2005, H
PRINT 2003
2000 FORMAT (7H ISKIP=,10(I1,1H),4HNPS=,12,5H,NZS=,12,5H,NDI=,12,5H,NCOPT04100
1I=,12,5H,NTI=,12,5H,2RK=,F5.1,1H,/6H TAU=,F8.3,7H,I2MOD=,15(I1,1H)OPT04200
2,))
2001 FORMAT (1X,A6,1H=,7(F9.3,1H,)/(1X,7(F9.3,1H,)))
2002 FORMAT (1X,A6,1H=,1P4(E14.7,1H,)/1X,5(E14.7,1H,)/(1X,5(E14.7,1H,))OPT04300
1)
2003 FORMAT (5H SEID)
2004 FORMAT (1X,A6,1H=,4(F11.8,1HE,3A1,1H,)/1X,4(F11.8,1HE,3A1,1H,)/
11X,4(F11.8,1HE,3A1,1H,))OPT04400
2005 FORMAT (3H H=,F9.3,1H,))OPT04500
2006 FORMAT (7H TIMAV=,F6.1,1H,))OPT04600
PUNCH 1999, NAMCAS
PUNCH 2000, ISKIP,NPS,NZ,NDI,NCI,NTI,ZRK,TAUK,(I2MOD(K),K=1,NNZ)
PUNCH 2001, JE(1),ZF,(Z(K),K=2,NZ)
DO 20 K=1,NNZ
  QL(K) = GP(K,I)
  20 CALL CONV(QL(K),IG(K),JQ(K),KQ(K))
  PUNCH 2004, JE(2),(QL(K),KQ(K),IG(K),JQ(K),K=1,NNZ)
  PUNCH 2001, JE(3),(UBAR(K),K=1,NZ)
  PUNCH 2001, JE(4),(SIGAP(K),K=1,NZ)
  PUNCH 2001, JE(5),(SIGEP(K),K=1,NZ)
  PUNCH 2001, JE(6),(SIGXO(K),K=1,NNZ)
  PUNCH 2001, JE(7),(SIGYO(K),K=1,NNZ)
  PUNCH 2001, JE(8),(SIGZO(K),K=1,NNZ)
  PUNCH 2001, JE(9),(WD(K),K=1,NZ)
  PUNCH 2006, TIMAV
  PUNCH 2001, JE(10),(DI(K),K=1,NDI)
  PUNCH 2001, JE(11),(CI(K),K=1,NCI)
  PUNCH 2001, JE(14),(TI(K),K=1,NTI)
  PUNCH 2001, JE(12),(DX(K),K=1,NNZ)
  PUNCH 2001, JE(13),(DY(K),K=1,NNZ)
  PUNCH 2001, JE(15),(TV(K),K=1,NZ)
  PUNCH 2005, H
  PUNCH 2007, TAU
2007 FORMAT (1X,6HTAU=,F8.3,1H,))
PUNCH 2003
NZ = NZM
RETURN
END
OPT03900
OPT04000
OPT04100
OPT04200
OPT04300
OPT04400
OPT04500
OPT04600
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OPT05000
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OPT05110
OPT05200
OPT05300
OPT05400
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OPT05600
OPT05700
OPT05800
OPT05900
OPT06000
OPT06100
OPT06200
OPT06300
OPT06400
OPT06500
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OPT06700
OPT06800
OPT06900
OPT07000
OPT07100
OPT07200
OPT07300
OPT07400
OPT07500
OPT07600
OPT07700
OPT07800

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1* SUBROUTINE CONST(JFLG,NO,YES,IPOL,ITP,IFEET,KNOTS) CST00100
2* COMMON /PLUME/QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(2CST00200
3* 11),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL CST00300
4* 2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAR CST00400
5* COMMON /REST/ ZM,DPDZ,K,A1,B1,PH11,ZP,TZ,PZ,PHI,IFLG,KS CST00500
6* COMMON /SIG/ SIGX(20),SIGYO(20),SIGZO(20),H CST00600
7* COMMON /F/ DATE(6),FRQ(4),WTMOL(3) CST00700
8* COMMON /OUT/ QF(20,4),WD(21),ISKIP(10),NDI,NCI,NTI,ZRK,JBOT,JTOP, CST00800
9* 112MOD(21),CI(10),DI(10),TI(10),NPS CST00900
10* COMMON /DISPL/ UXX,DX(21),DYY,DY(21),ILXY CST01000
11* DIMENSION IFQ(4) CST01100
12* DATA IFQ/24H FEET METERSKNOTS MET/S / CST01200
13* DIMENSION IPOL(4) CST01300
14* DIMENSION ITP(2) CST01400
15* INTEGER YES CST01500
16* IF (JFLG.GT. 0) GO TO 20 CST01600
17* PRINT 2000, QC,QT,A,B,HEAT,GAMMA,CP CST01700
18* 2000 FORMAT (1M0,41X, 47H** INITIALIZED DATA USED FOR ABOVE VEHICLE CST01800
19* 1E *-*/20X,69HQC - RATE OF OUTPUT OF EXHAUST MATERIAL FROM VEHICLE CST01900
20* 2 IN GRAMS/SEC IS ,1PE15.8/ CST02000
21* 320X,58HQT - TOTAL AMOUNT OF VEHICLE EXHAUST MATERIAL IN GRAMS IS ,CST02100
22* 4E15.8/ CST02200
23* 520X,64HA AND B - VEHICLE RISE PARAMETERS IN THE EQUATION TR=A*Z**BCST02300
24* 6 ARE ,0PF0.6,5H AND ,F8.0/ CST02400
25* 720X,45HHEAT - TOTAL HEAT OUTPUT IN CALORIES/GRAM IS ,F10.4/ CST02500
26* 820X,33HGAMMA - ENTRAINMENT PARAMETER IS ,F7.4/ CST02600
27* 920X,29HCP - SPECIFIC HEAT OF AIR IS ,F5.3) CST02700
28* PRINT 2001, IPOL,FRQ,WTMOL CST02800
29* 2001 FORMAT (20X,23HPULUTANT MATERIALS ARE ,4(A6,1H,)/ CST02900
30* 120X,50HFRACTIONAL DISTRIBUTION OF THE ABOVE MATERIALS IS ,4(F5.3, CST03000
31* 21H,)/ CST03100
32* 320X,43HMOLECULAR WEIGHT OF THE ABOVE MATERIALS IS ,3(F7.3,1H, ) CST03200
33* J1 = NO CST03300
34* 1F (NORMAL .EQ. 1) J1 = YES CST03400
35* J2 = NO CST03500
36* 1F (IFEET .EQ. ITP(1)) J2 = YES CST03600
37* J3 = NO CST03700
38* 1F (KNOTS .EQ. ITP(2)) J3 = YES CST03800
39* N1 = IFQ(2) CST03900
40* IF (J2 .EQ. YES) N1 = IFQ(1) CST04000

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41* N2 = IFQ(4)
42* IF (J3.EQ. YES) N2 = IFQ(3)
43* PRINT 2002, DATE,J1,J2,J3,SIGAR,RHO,HM,NZS
44* 2002 FORMAT (1H0,52X,26H*-X PROGRAM INPUT DATA *--//25X,11HDATA CARD 1/CST04400
45* 129X,8HTITLE - ,0A6/29X,26HNORMAL - IS LAUNCH NORMAL ?,A3/
46* 229X,54HIFRET - ARE LAYER BOUNDARY HEIGHTS Z, AND HM IN FEET?,A3/
47* 329X,39HKNOTS - IS THE WIND SPEED WS IN KNOTS?,A3/
48* 429X,67HSIGAR - STANDARD DEVIATION OF THE AZIMUTH WIND ANGLE IN DEGCST04800
49* 5REES IS ,F7.3/
50* 029X,42HRHU - AIR DENSITY IN GRAMS/CUBIC METER IS ,F9.3/
51* 729X,39HMM - HEIGHT OF SURFACE MIXING LAYER IS ,F9.3/
52* 825X,20HDATA CARD 2 THROUGH ,12/33X,67H1LAYER BOUNDARY WIND DIRECTCST05200
53* 910H WIND SPEED TEMPERATURE PRESSURE)
54* PRINT 2003, N1,N2
55* 2003 FORMAT (39X,1HZ,1X,1H(,A6,1H),4X,9HWD (DEG),5X,2HWS,1X,1H(,A5,1H)CST05500
56* 1,2X,9HT (DEG C),0X,6HP (MB),3X,12HRH (PERCENT))
57* DO 10 I=1,NZS
58* 10 PRINT 2004, 2(I),WD(I),UBAR(I),T(I),P(I),RH(I)
59* 2004 FORMAT (34X,F9.3,9X,F9.4,4X,F9.4,5X,F9.3,3X,F9.3,5X,F7.3)
60* GO TO 40
61* 20 CONTINUE
62* NZL = NZS-1
63* ZL = 0.0
64* PRINT 2005, JFLG,H,ZM,TAUK,DPDZ,JBOT,JTOP,ZZL
65* 2005 FORMAT (1H0,39X,36H*-X CALCULATED PARAMETERS FOR MODEL ,12,4H *-*/CST06500
66* 1/13X,39HH - ADJUSTED CLOUD HEIGHT IN METERS IS ,F9.3/
67* 213X,36HZM - REAL CLOUD HEIGHT IN METERS IS ,F9.3/
68* 313X,45HTAUK - TIME TO CLOUD STABILIZATION IN SEC IS ,F9.3/
69* 413X,73HDPDZ - VERTICAL GRADIENT OF AMBIENT POTENTIAL TEMP IN DEGREESCST06900
70* 5E5 1/METER IS ,F12.6/
71* 013X,44HJBOT - BOTTOM LAYER FOR USE WITH MODEL 4 IS ,12/
72* 713X,41HJTOP - TOP LAYER FOR USE WITH MODEL 4 IS ,12/
73* 813X,58HZ - BOUNDARY HEIGHT AT THE BOTTOM OF LAYER 1 IN METERS IS ,CST07300
74* 9F6.3)
75* PRINT 2006, ZRK,SIGAP(1),SIGEP(1),IPOL
76* 2006 FORMAT (13X,83HSIGAP - STANDARD DEVIATION OF THE WIND AZIMUTH ANGLCST07600
77* 1E AT THE MEASUREMENT HEIGHT ZRK=F6.2,11H METERS IS ,F8.3/
78* 213X,65HSIGEP - STANDARD DEVIATION OF THE WIND ELEVATION ANGLE AT ZCST07800
79* 3RK IS ,F6.3/19H LAYER PARAMETERS -/11H LAYER Z,21X,21H- SOURCE CST07900
80* 4STRENGTH Q -,19X,59HSIGAP SIGEP SIGZO SIGZO (METER) (METER) (METER)
81* 5 DELX DELY/13H NO. (LAYER,59X,59H(DEG) (DEG) (DEG) (MCST08100

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82* 6ETER) (METER) (DEG)/9X,4HTOP),4X,A6,9X,A6,8X,A6,A6/CST08200
83* 7) CST08300
84* DO 30 K=1,NNZ CST08400
85* 30 PRINT 2007, K,Z(K+1),(QF(K,I),I=1,4),SIGAP(K+1),SIGEP(K+1),SIGXO(KCST08500
86* 1),SIGYO(K),SIGZO(K),DX(K),DY(K) CST08600
87* 2007 FORMAT (1X,I3,1X,F9.3,1P4E14.7,0P2F8.4,3F10.4,F9.2,F7.2) CST08700
88* 40 CONTINUE CST08800
89* RETURN CST08900
90* END CST09000

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1* SUBROUTINE PLUME1
2* COMMON /PLUME/ Q, A, B, HEAT, RHO, CP, PI, GAMMA, T(21), P(21), Z(21), UBAR(PL100200
3* 121), NZS, G(20), QT, HM, SIGEP(21), SIGAP(21), G, TAU, K, NORMAL
4* 2, TV(21), RH(21), NAMCAS(12), NAMT(12), SIGAR
5* COMMON /REST/ ZM, DPDZ, K, A1, B1, PHI1, ZP, TZ, PZ, PHI, IFLG, KS
6* PLUME RISE FOR INSTANTANEOUS SOURCE
7* A1 = 6.0*WC*A*HEAT/(RHO*CP*PI*GAMMA**3)
8* B1 = 1.0/(4.0-B)
9* K = 1
10* K = K+1
11* 20 CALL LEAST(Z, TV, DPDZ, K, 0.0, 0.0, 0.0)
12* IF (DPDZ .LT. 3.322E-4) DPDZ = 3.322E-4
13* ZM = (A1/DPDZ)**B1
14* IF (ZM .LE. Z(K)) GO TO 30
15* K = K+1
16* IF (K .GT. NZS) GO TO 80
17* GO TO 20
18* 30 IF (Z(K)-ZM .LE. 10.0) GO TO 70
19* IF (DPDZ-3.322E-4) 35, 70, 35
20* 35 CONTINUE
21* ZP = Z(K)
22* 40 ZP = ZP-10.0
23* IF (ZP .LT. Z(1)) GO TO 65
24* TVP = TV(K)-TPZ(Z(K), ZP, TV(K), TV(K-1), Z(K-1))
25* CALL LEAST(Z, TV, DPDZ, K-1, 1, ZP, TVP)
26* IF (DPDZ .GT. 3.322E-4) GO TO 60
27* DPDZ = 3.322E-4
28* 50 ZM = ZP
29* GO TO 70
30* 60 ZM = (A1/DPDZ)**B1
31* IF (ZM .GT. ZP) GO TO 50
32* IF (ZM .GT. ZP-10.0) GO TO 70
33* IF (ZP .GE. Z(K-1)) GO TO 40
34* ZM = Z(K-1)
35* RETURN ZM AND DPDZ FOR INSTANTANEOUS SOURCE
36* 70 IFLG = 0
37* GO TO 90
38* C CANNOT CALCULATE ZM AND DPDZ
39* 80 IFLG = 1
40* GO TO 90

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PL104100
PL104200
PL104300

65 IF LG = 2
90 RETURN
END

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1* SUBROUTINE PLUME2
2* COMMON /PLUME/ WC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(PL200100
3* 121),NZS,W(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL PL200200
4* 2,TV(21),RH(21),NAMCAS(12),NAMT(12),SIGAR PL200300
5* COMMON /REST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS PL200400
6* PLUME RISE FOR CONTINUOUS SOURCE PL200500
7* ZSUM = 0.0 PL200600
8* UBARS = 0.0 PL200700
9* A1 = 0.0*WC*HEAT/(RHO*CP*PI*GAMMA**2) PL200800
10* B1 = .333333 PL200900
11* K = 1 PL201000
12* 10 K = K+1 PL201100
13* 20 CALL LEAS1(Z,TV,UPDZ,K,0.0,0.0,0.0) PL201200
14* IF (DPDZ .LT. 3.322E-4) DPDZ = 3.322E-4 PL201300
15* UBARS = UBARS+(Z(K)-Z(K-1))*(UBAR(K)+UBAR(K-1))*0.5 PL201400
16* ZSUM = ZSUM+Z(K)-Z(K-1) PL201500
17* UBARK = UBARS/ZSUM PL201600
18* ZM = (A1/(UBARK*DPDZ))*B1 PL201700
19* IF (ZM .LE. Z(K)) GO TO 30 PL201800
20* K = K+1 PL201900
21* IF (K .GT. NZS) GO TO 80 PL202000
22* GO TO 20 PL202100
23* 30 IF (Z(K)-ZM .LE. 10.0) GO TO 70 PL202200
24* IF (DPDZ-3.322E-4) 35,70,35 PL202300
25* 35 CONTINUE PL202400
26* UBARK = UBARS-(Z(K)-Z(K-1))*(UBAR(K)+UBAR(K-1))*0.5 PL202500
27* ZBARK = ZSUM-(Z(K)-Z(K-1)) PL202600
28* ZP = Z(K) PL202700
29* ZP = ZP-10.0 PL202800
30* IF (ZP .LT. Z(1)) GO TO 85 PL202900
31* TVP = TV(K)-TPZ(Z(K),ZP,TV(K),TV(K-1),Z(K-1)) PL203000
32* CALL LEAS1(Z,TV,UPDZ,K-1,1,ZP,TVP) PL203100
33* IF (DPDZ .GT. 3.322E-4) GO TO 60 PL203200
34* DPDZ = 3.322E-4 PL203300
35* ZM = ZP PL203400
36* GO TO 70 PL203500
37* 60 UBARK = UBAR(K)-TPZ(Z(K),ZP,UBAR(K),UBAR(K-1),Z(K-1)) PL203600
38* UBARK = (UBARK+(ZP-Z(K-1))*(UBARZ+UBAR(K-1))*0.5)/(ZBARK+ZP-Z(K-1)) PL203700
39* 1) PL203800
40* ZM = (A1/(UBARK*DPDZ))*B1 PL203900
    PL204000

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IF (ZM .GT. ZP) GO TO 50
IF (ZM .GT. ZP-10.0) GO TO 70
IF (ZP .GE. Z(K-1)) GO TO 40
ZM = Z(K-1)
RETURN ZM AND DPLZ FOR CONTINUOUS SOURCE
C
70 IFLG = 0
GO TO 90
C CANNOT CALCULATE ZM AND UPDZ
80 IFLG = 1
GO TO 90
85 IFLG = 2
90 RETURN
END

PL204100
PL204200
PL204300
PL204400
PL204500
PL204600
PL204700
PL204800
PL204900
PL205000
PL205100
PL205200
PL205300

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SUBROUTINE DELTX

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COMMON /PLUME/QC,A,B,HEAT,RHO,CP,PI,GAMMA,T(21),P(21),Z(21),UBAR(
121),NZS,Q(20),QT,HM,SIGEP(21),SIGAP(21),G,TAUK,NORMAL
2,TV(21),RH(21),HAMCAS(12),NAMT(12),SIGAR
COMMON /REST/ ZM,DPDZ,K,A1,B1,PHI1,ZP,TZ,PZ,PHI,IFLG,KS
COMMON /OUT/ QF(20,4),WD(21),ISKIP(10),NYS,NDI,NCI,NBK,NPTS,ZRK,
1JBOT,JTOP,IZMOD,CI(10),DI(10),ZZL(2)
COMMON/DISPL/ DX,DX(21),DYY,DY(21),ILXY
IP = 4
L = 3
IF (NORMAL.EQ. 1) GO TO 5
IP = 3
L = 2
5 UF = 0.0
ZF = 0.0
A1 = RHO*CP*PI*GAMMA**L/(3.0*QC*HEAT)
IF (NORMAL.EQ. 1) A1 = A1/A
B1 = G/T(1)
S = 1.0/SQRT(G*DPDZ/T(1))
PPI = PI*5.5555555E-3
TSIR = PI*S
PPII = 1.0/PPI
UXX = 0.0
UY = 0.0
I = 0
TT = 0.0
10 I = I+1
IF (I.GE. NZS) GO TO 30
CALL LEAS1(Z,TV,UPDZS,I+1,0.0,0.0,0.0)
IF (OPDZS.LT. 3.322E-4) OPDZS = 3.322E-4
BK = A1*OPDZS
IF (NORMAL.EQ. 0) GO TO 12
BK = BK/(Z(I+1)**B)
GO TO 15
12 CONTINUE
UFS= UF+(Z(I+1)-Z(I))*5*(UBAR(I+1)+UBAR(I))
ZFS= ZF+(Z(I+1)-Z(I))
BK = BK*UFS/ZFS
15 CONTINUE
ZD = BK*Z(I+1)**IP

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DXY00100
 DXY00200
 DXY00300
 DXY00400
 DXY00500
 DXY00600
 DXY00700
 DXY00800
 DXY00900
 DXY01000
 DXY01100
 DXY01200
 DXY01300
 DXY01400
 DXY01500
 DXY01600
 DXY01700
 DXY01800
 DXY01900
 DXY02000
 DXY02100
 DXY02200
 DXY02300
 DXY02400
 DXY02500
 DXY02600
 DXY02700
 DXY02800
 DXY02900
 DXY03000
 DXY03100
 DXY03200
 DXY03300
 DXY03400
 DXY03500
 DXY03600
 DXY03700
 DXY03800
 DXY03900
 DXY04000

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41* IF (ZD.GT. 2.0) GO TO 20
42* THETAK = (WD(I+1)+WD(I))*0.5
43* IF (ABS(WD(I+1)-WD(I)).GT. 180.0) THETAK = THETAK-180.0
44* BB = 1.0-ZD
45* IF (BB.GT. 1.0) BB = 1.0
46* IF (BB.LT.-1.0) BB = -1.0
47* S = 1.0/SQRT(B1*UPDZS)
48* TK = S*ARCOS(BB)-TT
49* TT = TK+TT
50* IF (TT.LE. TSTR) GO TO 17
51* IF = TT-TK
52* GO TO 20
53* 17 CONTINUE
54* UF = UFS
55* ZF = ZFS
56* IF (NORMAL.EQ. 0) GO TO 18
57* RK = 0.5*(UBAR(I+1)+UBAR(I))*TK
58* GO TO 19
59* 18 RK = UF*TK/ZF
60* 19 CONTINUE
61* BB = THETAK*PPI
62* DY(I) = DY(I-1)-RK*COS(BB)
63* DX(I) = DX(I-1)-RK*SIN(BB)
64* ILAY = I
65* GO TO 10
66* 20 RK = ((ZM-Z(I))/(Z(I+1)-Z(I))*0.5*(UBAR(I+1)-UBAR(I))+UBAR(I))
67* IF (NORMAL.EQ. 1) GO TO 25
68* RK = RK*(ZM-Z(I))+UF
69* ZF = ZF+(ZM-Z(I))
70* RK = RK/ZF
71* 25 RK = RK*(TSTR-TT)
72* BB = WD(I+1)-WD(I)
73* IF (BB.GT. 180.0) BB = BB-360.0
74* IF (BB.LT.-180.0) BB = BB+360.0
75* BB = AMOD(BB,360.0)
76* THETAM = BB/(Z(I+1)-Z(I))*(ZM-Z(I))+WD(I)
77* THETAK = 0.5*(THETAM+WD(I))
78* IF (ABS(THETAM-WD(I)).GT. 180.0) THETAK = THETAK-180.0
79* BB = THETAK*PPI
80* DX(I) = DX(I-1)-RK*SIN(BB)
81* DY(I) = DY(I-1)-RK*COS(BB)

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DXY04100
DXY04200
DXY04300
DXY04400
DXY04500
DXY04600
DXY04700
DXY04800
DXY04900
DXY05000
DXY05100
DXY05200
DXY05300
DXY05400
DXY05500
DXY05600
DXY05700
DXY05800
DXY05900
DXY06000
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DXY06900
DXY07000
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DXY07900
DXY08000
DXY08100

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      ILXY = I
28  I = I+1
      IF (I .GE. NZS) GO TO 30
      RK = TSTR*0.5*(UBAR(I+1)+UBAR(I))
      ZF = 0.5*(WD(I+1)+WD(I))
      IF (ABS(WD(I+1)-WD(I)) .GT. 180.0) ZF = ZF-180.0
      BB = ZF*PPI
      DX(I) = -RK*SIN(BB)
      DY(I) = -RK*COS(BB)
      GO TO 28
30  CONTINUE
      I = NZS-1
      DO 80 J = 1, I
      IF (DX(J)) 50,40,50
40  IF (DY(J)) 50,80,50
50  BB = 270.0-ATAN2(DY(J),DX(J))*PPI
      IF (BB .GT. 360.0) BB = BB-360.0
      IF (BB .GT. 180.0) GO TO 60
      BB = BB+180.0
      GO TO 70
60  BB = BB-180.0
70  DX(J) = SQR(DX(J)*DX(J)+DY(J)*DY(J))
      DY(J) = BB
80  CONTINUE
      RETURN
      END

```

DXY08200
 DXY08300
 DXY08400
 DXY08500
 DXY08600
 DXY08700
 DXY08800
 DXY08900
 DXY09000
 DXY09100
 DXY09200
 DXY09300
 DXY09400
 DXY09500
 DXY09600
 DXY09700
 DXY09800
 DXY09900
 DXY10000
 DXY10100
 DXY10200
 DXY10300
 DXY10400
 DXY10500
 DXY10600
 DXY10700

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SUBROUTINE LEAST(Z,TV,DPDZ,K,ISW,ZP,TVP)
DIMENSION Z(1),TV(1)
IF (K .LE. 1) GO TO 30
L = K
TVB = 0.0
ZB = 0.0
DO 10 I=1,K
TVB = TVB+TV(I)
10 ZB = ZB+Z(I)
IF (ISW .EQ. 0) GO TO 15
TVB = TVB+TVP
ZB = ZB+ZP
L = L+1
15 TVB = TVB/FLOAT(L)
ZB = ZB/FLOAT(L)
S1 = 0.0
S2 = 0.0
DO 20 I=1,K
S1 = S1+(Z(I)-ZB)*(TV(I)-TVB)
20 S2 = S2+(Z(I)-ZB)**2
IF (ISW .EQ. 0) GO TO 25
S1 = S1+(ZP-ZB)*(TVP-TVB)
S2 = S2+(ZP-ZB)**2
25 DPDZ = S1/S2
30 CONTINUE
RETURN
END

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LST00100
LST00200
LST00300
LST00400
LST00500
LST00600
LST00700
LST00800
LST00900
LST01000
LST01100
LST01200
LST01300
LST01400
LST01500
LST01600
LST01700
LST01800
LST01900
LST02000
LST02100
LST02200
LST02300
LST02400
LST02500
LST02600
LST02700

ACS00100
ACS00200
ACS00300
ACS00400

FUNCTION ARCOS(A)
ARCOS = ACOS(A)
RETURN
END

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2*
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**C.2 FORTRAN SOURCE LISTING FOR THE NASA/MSFC MULTILAYER DIFFU-
SION MODELS PROGRAM - VERSION 5**

**This section contains the complete FORTRAN source listing of the NASA/
MSFC Multilayer Models Program - Version 5.**

1*	C	*****	MDL00100
2*	C	* NASA/MSFC MULTILAYER DIFFUSION MODEL, VERSION V	MDL00200
3*	C	*****	MDL00300
4*	C	*****	MDL00400
5*	C	*****	MDL00500
6*	C	*****	MDL00600
7*	C	*****	MDL00700
8*	C	*****	MDL00800
9*	C	*****	MDL00900
10*	C	*****	MDL01000
11*	C	*****	MDL01100
12*	C	*****	MDL01200
13*	C	*****	MDL01300
14*	C	*****	MDL01400
15*	C	*****	MDL01500
16*	C	*****	MDL01600
17*	C	*****	MDL01700
18*	C	*****	MDL01800
19*	C	*****	MDL01900
20*	C	*****	MDL02000
21*	C	*****	MDL02100
22*	C	*****	MDL02200
23*	C	*****	MDL02300
24*	C	*****	MDL02400
25*	C	*****	MDL02500
26*	C	*****	MDL02600
27*	C	*****	MDL02700
28*	C	*****	MDL02800
29*	C	*****	MDL02900
30*	C	*****	MDL03000
31*	C	*****	MDL03100
32*	C	*****	MDL03200
33*	C	*****	MDL03300
34*	C	*****	MDL03400
35*	C	*****	MDL03500
36*	C	*****	MDL03600
37*	C	*****	MDL03700
38*	C	*****	MDL03800
39*	C	*****	MDL03900
40*	C	*****	MDL04000


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SKIP= PROGRAM CONTROL OPTIONS
H = CLOUD STABILIZATION HEIGHT (METERS)
Z = BOUNDARY HEIGHTS OF LAYERS (METERS)
Q = SOURCE STRENGTH IN LAYER
UBAR = CALCULATED TRANSPORT SPEED IN LAYER
ALPHA = LATERAL POWER LAW EXPANSION COEFFICIENT
BETA = VERTICAL POWER LAW EXPANSION COEFFICIENT
SIGYO = STANDARD DEVIATION OF THE LATERAL SOURCE DIMENSION (METER)
SIGAP = CALCULATED LATERAL DIFFUSION COEFFICIENT IN LAYER
SIGXO = STANDARD DEVIATION OF THE ALONG WIND SOURCE DIMENSION
(METERS)
DELTHP = CALCULATED WIND DIRECTION SHEAR IN LAYER
SIGZO = STANDARD DEVIATION OF THE VERTICAL SOURCE DIMENSION
(METERS)
SIGEP = CALCULATED VERTICAL DIFFUSION COEFFICIENT
DELX = RANGE TO SOURCE IN LAYER RELATIVE TO ORIGIN OF REFERENCE
GRID SYSTEM (METERS)
DELY = AZIMUTH BEARING FROM 0 DEGREES NORTH TO SOURCE IN LAYER
(DEGREES)
THETA = CALCULATED MEAN WIND DIRECTION IN LAYER
IZMOD = MODEL OR MODELS TO USE IN LAYER
DELU = CALCULATED WIND SPEED SHEAR
ZZL = CALCULATION HEIGHTS IN LAYER
DOS = CALCULATED VALUE OF DOSAGE
CON = CALCULATED VALUE OF CONCENTRATION
PEAKD = PART OF DOSAGE EQUATION
XX = RANGE TO CALCULATION POINT OF THE POLAR COORDINATE REFERENCE
GRID SYSTEM (METERS)
YY = AZIMUTH BEARING FROM 0 DEGREES NORTH TO CALCULATION POINT OF
THE POLAR COORDINATE REFERENCE GRID
LAT = LATERAL TERM OF DOSAGE EQUATION
VER = VERTICAL TERM OF DOSAGE EQUATION
VREF = REFLECTION TERM OF DOSAGE EQUATION

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41*	C	T	= SOURCE EMISSION TIME IN LAYER FOR GRAVITATIONAL DEP. (SEC)	MDL04100
42*	C	TESTNO	= METEOROLOGICAL CASE INFORMATION	MDL04200
43*	C	DI	= DOSAGE ISOPLETH VALUES OF INTEREST	MDL04300
44*	C	CI	= CONCENTRATION ISOPLETH VALUES OF INTEREST	MDL04400
45*	C	TI	= TIME MEAN CONCENTRATION VALUES OF INTEREST FOR ISOPLETHS	MDL04500
46*	C	SIGZ	= CALCULATED STANDARD DEVIATION OF THE VERTICAL DOSAGE DISTRIBUTION	MDL04600
47*	C	SIGY	= CALCULATED STANDARD DEVIATION OF THE LATERAL DOSAGE DISTRIBUTION	MDL04700
48*	C	SIGX	= CALCULATED STANDARD DEVIATION OF THE ALONG WIND DOSAGE DISTRIBUTION	MDL04800
49*	C	SQR2P	= SQUARE ROOT TWO PI	MDL04900
50*	C	L	= LENGTH OF CLOUD IN ALONG WIND DIRECTION	MDL05000
51*	C	I	= INDEX OF X COORDINATES	MDL05100
52*	C	J	= INDEX OF Y COORDINATES	MDL05200
53*	C	KK	= INDEX OF LAYERS	MDL05300
54*	C	K	= INDEX OVER CALCULATION HEIGHTS ZZL	MDL05400
55*	C	STO1	= TEMP STORAGE	MDL05500
56*	C	STO2	= TEMP STORAGE	MDL05600
57*	C	STO3	= TEMP STORAGE	MDL05700
58*	C	TAST	= TIME OF LAYER STRUCTURE CHANGE (SECONDS)	MDL05800
59*	C	NBK	= NO OF DISTINCT GROUPS OF LAYERS THAT FORM INTO ONE AT TIME TAST.	MDL05900
60*	C	ILK	= INDEX ON NEW LAYERS AFTER TIME TAST	MDL06000
61*	C	NXS	= NO OF X COORDINATES	MDL06100
62*	C	NYS	= NO OF Y COORDINATES	MDL06200
63*	C	NZS	= NO OF LAYER BOUNDARIES	MDL06300
64*	C	NDI	= NO OF DOSAGE ISOPLETHS	MDL06400
65*	C	NCI	= NO OF CONCENTRATION ISOPLETHS	MDL06500
66*	C	NTI	= NO OF TIME MEAN CONCENTRATION ISOPLETHS	MDL06600
67*	C	NPTS	= NO OF CALCULATION HEIGHTS ZZL	MDL06700
68*	C	RAD	= PI/180	MDL06800
69*	C	NNZ	= NZS-1 NO OF LAYERS	MDL06900
70*	C	ITOP	= TOP OF NEW LAYER AFTER TAST IN TERMS OF OLD LAYER STRUCTURE	MDL07000
71*	C	IBOT	= BOTTOM OF NEW LAYER AFTER TAST IN TERMS OF OLD LAYER STRUCTURE (ITOP AND IBOT INDEXES)	MDL07100
72*	C	XAST	= CALCULATE DISTANCE TO TAST	MDL07200
73*	C	SIGXNK	= SIGX OF NEW LAYER STRUCTURE	MDL07300
74*	C	LAMBDA	= WASHOUT COEFFICIENT	MDL07400
75*	C	TIM1	= TIME OF START OF RAIN (SECONDS)	MDL07500
76*	C	ZLIM	= MAXIMUM HEIGHT OF WASHOUT	MDL07600
77*	C			MDL07700
78*	C			MDL07800
79*	C			MDL07900
80*	C			MDL08000
81*	C			MDL08100

82*	C	WASHOU = CALCULATE WASHOUT AT GROUND	MDL08200
83*	C	UBARK = WIND SPEED AT EACH LAYER BOUNDARY, LOWER BOUNDARY OF LAYER	MDL08300
84*	C	1 FOR UBARK IS ASSUMED AT ZRK (METERS/SEC)	MDL08400
85*	C	SIGAK = SIGAP (INITIAL) AT EACH LAYER BOUNDARY, LOWER BOUNDARY OF	MDL08500
86*	C	LAYER 1 FOR SIGAK IS ASSUMED AT ZRK (DEGREES)	MDL08600
87*	C	SIGEK = SIGEP (INITIAL) AT EACH LAYER BOUNDARY, LOWER BOUNDARY OF	MDL08700
88*	C	LAYER 1 FOR SIGEK IS ASSUMED AT ZRK (DEGREES)	MDL08800
89*	C	ZRK = REFERENCE HEIGHT IN SURFACE LAYER (METERS)	MDL08900
90*	C	THETAK = WIND DIRECTION AT LAYER BOUNDARIES (DEGREES)	MDL09000
91*	C	TAUK = TIME IN SECONDS REQUIRED FOR LATERAL CLOUD STABILIZATION	MDL09100
92*	C	TAUOK = SAMPLING PERIOD IN SECONDS AT THE TOP OF THE LAYER	MDL09200
93*	C	DECAY = DECAY COEFFICIENT IN DOSAGE EQUATION	MDL09300
94*	C	UBARL = WIND SPEED AT BOTTOM AND TOP OF EACH NEW LAYER AFTER LAYER	MDL09400
95*	C	CHANGE (METERS/SEC)	MDL09500
96*	C	SIGAL = SIGAP AT BOTTOM AND TOP OF EACH NEW LAYER AFTER LAYER	MDL09600
97*	C	CHANGE (DEGREES)	MDL09700
98*	C	SIGEL = SIGEP AT BOTTOM AND TOP OF EACH NEW LAYER AFTER LAYER	MDL09800
99*	C	CHANGE (DEGREES)	MDL09900
100*	C	ZRL = REFERENCE HEIGHT IN SURFACE LAYER OF NEW STRUCTURE (METERS)	MDL10000
101*	C	THETAL = WIND DIRECTION AT BOTTOM AND TOP OF EACH NEW LAYER AFTER	MDL10100
102*	C	TAUL = TIME IN SECONDS FOR LATERAL CLOUD STABILIZATION IN NEW	MDL10200
103*	C	LAYER STRUCTURE	MDL10300
104*	C	TAUOL = TIME IN SECONDS OF SAMPLING PERIOD AT TOP OF NEW LAYER	MDL10400
105*	C	JBOT = INPUT LAYER NUMBER OF BOTTOM OF NEW LAYER STRUCTURE	MDL10500
106*	C	RELATIVE TO OLD	MDL10600
107*	C	JTOP = INPUT LAYER NUMBER OF TOP OF NEW LAYER STRUCTURE	MDL10700
108*	C	RELATIVE TO OLD	MDL10800
109*	C	VS = SETTLING VELOCITY IN GRAVITATIONAL DEPOSITION MODEL	MDL10900
110*	C	PERC = FREQUENCY OF VS	MDL11000
111*	C	ACCUR = DESIRED ACCURACY COEFFICIENT (.45) INSURES THAT GROUND	MDL11100
112*	C	DEPOSITION FROM NXCI POINT SOURCES IN THE LAYER VARIES	MDL11200
113*	C	LESS THAN TEN PERCENT FROM DEPOSITION EXPECTED FROM A	MDL11300
114*	C	VERTICAL LINE SOURCE IN THE LAYER. FOR (.32) REDUCED TO	MDL11400
115*	C	FIVE PERCENT	MDL11500
116*	C	VB = SETTLING VELOCITIES FROM A BURST OR DESTRUCT IN LAYER NNZ	MDL11600
117*	C	PERCB = FREQUENCY OF VB	MDL11700
118*	C	HB = HEIGHT OF BURST (METERS)	MDL11800
119*	C	PPWR = CALCULATED WIND SPEED POWER LAW EXPONENT	MDL11900
120*	C	QPWR = CALCULATED SIGEP POWER LAW EXPONENT	MDL12000
121*	C	MPWR = CALCULATED SIGAP POWER LAW EXPONENT	MDL12100
122*	C	DTHK = WIND ANGLE SHEAR	MDL12200

123*	C	NVS	=	NUMBER OF SETTLING VELOCITIES VS	MDL12300
124*	C	NVB	=	NUMBER OF SETTLING VELOCITIES VB	MDL12400
125*	C	II	=	INDEX ON VS AND VB	MDL12500
126*	C	DEP	=	TEMP STORAGE	MDL12600
127*	C	YBARY	=	CALCULATED CORRDNATE OF POINT ON CLOUD AXIS OF VS AT INTERSECTION WITH GROUND (DEPOSITION)	MDL12700
128*	C	XBARX	=	CALCULATED CORRDNATE OF POINT ON CLOUD AXIS OF VS AT INTERSECTION WITH GROUND (DEPOSITION)	MDL12800
129*	C	UBARNK	=	CALCULATED WIND SPEED (DEPOSITION)	MDL12900
130*	C	BETANK	=	CALCULATED BETA (DEPOSITION)	MDL13000
131*	C	ALPHNK	=	CALCULATED ALPHA (DEPOSITION)	MDL13100
132*	C	SQBAR	=	TEMP STORAGE	MDL13200
133*	C	ANG	=	ANGLE TO POINT XBARX, YBARY (DEPOSITION)	MDL13300
134*	C	NXCI	=	NUMBER OF POINT SOURCES IN LAYER (DEPOSITION)	MDL13400
135*	C	DEPN	=	CALCULATED VALUE OF GRAVITATIONAL DEPOSITION	MDL13500
136*	C	SIGYNK	=	SIGY OF NEW LAYER STRUCTURE IN CALCULATION OF DOSAGE AND CONCENTRATION	MDL13600
137*	C	SIGENK	=	CALCULATED SIGEP (DEPOSITION)	MDL13700
138*	C	SIGANK	=	CALCULATED SIGAP (DEPOSITION)	MDL13800
139*	C	TIMAV	=	CONCENTRATION AVERAGING TIME (SECONDS)	MDL13900
140*	C	AVCON	=	AVERAGE CONCENTRATION	MDL14000
141*	C	PASSTM	=	TIME OF CLOUD PASSAGE	MDL14100
142*	C	AVMXCN	=	MAXIMUM AVERAGE CONCENTRATION	MDL14200
143*	C	XRY	=	DISTANCE DOWNWIND FROM THE VIRTUAL POINT SOURCE OVER WHICH RECTILINEAR EXPANSION OCCURS Laterally (METERS)	MDL14300
144*	C	XRZ	=	DISTANCE DOWNWIND FROM THE VIRTUAL POINT SOURCE OVER WHICH RECTILINEAR EXPANSION OCCURS Vertically (METERS)	MDL14400
145*	C	XLRY	=	DISTANCE FROM TRUE SOURCE TO POINT OF MEASUREMENT OF SIGYO (METERS)	MDL14500
146*	C	XLZR	=	DISTANCE FROM TRUE SOURCE TO POINT OF MEASUREMENT OF SIGZO (METERS)	MDL14600
147*	C	GAMMA	=	FRACTION OF MATERIAL REFLECTED AT THE SURFACE (=1 FOR COMPLETE REFLECTION, =0 FOR NO REFLECTION)	MDL14700
148*	C	GAMMAP	=	1.0-GAMMA	MDL14800
149*	C	NAMCAS	=	SPECIAL CASE IDENTIFICATION INFORMATION	MDL14900
150*	C	SCL	=	MAP SCALE FACTOR IN INCHES FOR ISOPLETH PLOTS. IF THE MAP SCALE FACTOR IS 1 INCH = 24000 INCHES THEN SCL WOULD BE INPUT AS 24000. (IF 0 THE PROGRAM WILL CALCULATE SCL)	MDL15000
151*	C	ISW	=	SWITCH FOR MAXIMUM CENTERLINE PLOTS. IF SET TO 0 OR 2 LOG-LOG SCALING IS USED. IF SET TO 1 LINEAR IS USED.	MDL15100
152*	C	XMAXJN	=	MAXIMUM ALONGWIND DISTANCE FROM THE LAUNCH SITE FOR	MDL15200
153*	C				MDL15300
154*	C				MDL15400
155*	C				MDL15500
156*	C				MDL15600
157*	C				MDL15700
158*	C				MDL15800
159*	C				MDL15900
160*	C				MDL16000
161*	C				MDL16100
162*	C				MDL16200
163*	C				MDL16300

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164* C      MAXIMUM CENTERLINE PLOTS (METERS) (IF 0 PROG CALCULATES) MDL16400
165* C      - MAXIMUM ALONGWIND DISTANCE FROM THE LAUNCH SITE FOR MDL16500
166* C      ISOPLETHS (METERS) (IF 0 PROGRAM CALCULATES) MDL16600
167* C      - MAXIMUM CROSSWIND DISTANCE FOR ISOPLETHS (METERS) (IF 0 MDL16700
168* C      PROGRAM CALCULATES) MDL16800
169* C      - MAXIMUM NUMBER OF LOG CYCLES FOR THE VERTICAL AXIS OF MDL16900
170* C      THE MAXIMUM CENTERLINE PLOTS IF ISW = 0 OR 2. OR. MAXIMUM MDL17000
171* C      VALUE OF THE VERTICAL AXIS IF ISW = 1. (IF 0 PROGRAM MDL17100
172* C      CALCULATES) MDL17200
173* C      - VIRTUAL POTENTIAL TEMPERATURE AT EACH LAYER BOUNDARY. THIS MDL17300
174* C      ARRAY IS USED TO SEE IF THERE IS A NEGATIVE LAPSE RATE MDL17400
175* C      IN THE LAYER. THE PROG CHECKS TO SEE IF THE WIND SPEED MDL17500
176* C      SHEAR IS NEGATIVE. IF IT IS AND ALSO THE LAPSE RATE IS MDL17600
177* C      NEGATIVE THE PROGRAM USES THE ABSOLUTE VALUE OF THE SPEED MDL17700
178* C      SHEAR. IF THE SPEED SHEAR IS NEGATIVE AND THE LAPSE RATE MDL17800
179* C      IS POSITIVE OR TEMPK IS NOT INPUT THE PROGRAM USES 0 WIND MDL17900
180* C      SPEED SHEAR. MDL18000
181* C      - VIRTUAL POTENTIAL TEMPERATURE AT EACH LAYER BOUNDARY OF MDL18100
182* C      THE NEW LAYER STRUCTURE. MDL18200
183* C      MDL18300
184* C      MDL18400
185* C      - PROGRAM INPUT PARAMETERS - MDL18500
186* C      ISKIP, H, Z, Q, ALPHA, BETA, SIGYO, SIGZO, SIGXO, DELTHP, DELX, MDL18600
187* C      DELY, IZMOD, ZZL, XX, YY, T, TESTNO, DI, CI, TI, TAST, MDL18700
188* C      NX5,NYS,NZS,NDI, NCI, NTI, TIM1, UBARK, SIGAK, ZLIM, MDL18800
189* C      SIGEK, ZRK, THETAK, TAU, TAUOU, DECAY, UBARL, SIGAL, SIGEL, ZRL, MDL18900
190* C      THETAL, TAU, TAUOL, VS, PERC, ACCUR, VB, PERCB, HB,NAMCAS MDL19000
191* C      NVS, NVB, NPTS, TIMAV, LAMBD (BLAMDA), XRY, XRZ, XLRZ, XLRZ MDL19100
192* C      SOME OF THE ABOVE PARAMETERS ARE AUTOMATICALLY DETERMINED BY MDL19200
193* C      THE PROGRAM, CONSULT THE PROGRAM DOCUMENTATION TO DETERMINE MDL19300
194* C      WHICH THEY ARE. ALL INPUTS ARE READ VIA THE FORTRAN NAMELISTS MDL19400
195* C      NAME2 IN SUBROUTINE READER MDL19500
196* C      MDL19600
197* C      MDL19700
198* C      MDL19800
199* C      MDL19900
200* C      COMMON /PARAMT/ TESTNO(12), ISKIP(15),NX5,NYS,NZS,NDI,NCI, MDL20000
201* C      1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15), MDL20100
202* C      2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15), MDL20200
203* C      3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ, MDL20300
204* C      4XLRZ,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBD,DI(10),CI(10), MDL20400

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205* STAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),MDL20500
206* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10),MDL20600
207* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12) MDL20700
208* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),MDL20800
209* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SGR2P,L,TH,I,J,KK,ST01, MDL20900
210* 2ST02,ST03,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR, MDL21000
211* 3MPWR,I,DEP,XBARX,SOBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT, MDL21100
212* 4NCCC,NDDU,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD, MDL21200
213* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42), MDL21300
214* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT, MDL21400
215* 7NYSS,CDAMX(3) MDL21500
216* DIMENSION CON(1),DOS(1),AVCON(1),PASSTM(1) MDL21600
217* EQUIVALENCE (CON,DEPN),(DOS,DEPN(1,2)),(AVCON,DEPN(1,3)),(PASSTM,MDL21700
218* 1EPN(1,4)) MDL21800
219* REAL MPWR,L,LAT,LAMBDA MDL21900
220* INTEGER TESTNO MDL22000
221* *** INPUT SECTION *** MDL22100
222* SQR2P = 2.5066283 MDL22200
223* RAD = .01745329 MDL22300
224* IFF = 1 MDL22400
225* MBR = 0 MDL22500
226* MDL22600
227* 1 CALL READER(IFF) MDL22700
228* IFF = 2 MDL22800
229* IF (KSW(2) .LE. 0) GO TO 5 MDL22900
230* EXECUTE GRAVITATIONAL DEPOSITION MODEL MDL23000
231* CALL DEPOS MDL23100
232* GO TO 700 MDL23200
233* 5 CONTINUE MDL23300
234* IF (ISKIP(2) .LE. 1.AND.ISKIP(3) .LE. 1) GO TO 6 MDL23400
235* IF (MBR .EQ. 5) GO TO 6 MDL23500
236* MBR = 5 MDL23600
237* CALL IDENT(35,'HARD COPY, 1 EACH, PLUS FILM') MDL23700
238* CALL SETMIV(0,0,0,0) MDL23800
239* 6 CONTINUE MDL23900
240* DO 8 I=1,3 MDL24000
241* 8 CDAMX(I) = 0.0 MDL24100
242* ILK = 1 MDL24200
243* DO 10 J=1,41 MDL24300
244* DO 10 I=1,41 MDL24400
245* 10 DEPN(I,J) = 0.0 MDL24500

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287* IF (NBK .EQ. 0.OR.KK.NE. JBOT(ILK)) GO TO 98
288* WRITE (6,923) UBAR(NNZILK),THETA(NNZILK),DELTHP(NNZILK),
289* IDELU(NNZILK),SIGAP(NNZILK),SIGEP(NNZILK)
290* 98 CONTINUE
291* CALL TESTR(KIK)
292* WRITE (6,917)
293* C *** GENERAL GRID PATTERN CALCULATIONS ***
294* 140 CONTINUE
295* JF = NNZ+ILK-1
296* IF (KSW(1) .LE. 0) GO TO 145
297* IF (IFG .EQ. 1) GO TO 500
298* GO TO 148
299* 145 CONTINUE
300* IF (K .GT. NPTS) GO TO 500
301* IF (ZL(K)-Z(KK+1)) 148,500,500
302* 148 MDLS = MODLS(KK)
303* IF (NBK.GT.0.AND.KK.GE.IBOT.AND.KK.LE.ITOP) MDLS = 4
304* IF (NBK .LE. 0) GO TO 149
305* IF (KK .LT. IBOT.OR.KK .GT. ITOP) GO TO 149
306* YT = THETA(JF)+180.0
307* GO TO 150
308* 149 YT = THETA(KK)+180.0
309* 150 CONTINUE
310* C
311* DEFAULT YY (ANGULAR AXES)
312* IF (IMB .EQ. 1) GO TO 153
313* IF (NYS .GT. 0) GO TO 153
314* DEP = YT*0.2+0.5
315* NYSS = DEP
316* DEP = 5*NYSS
317* NYSS = 41
318* DO 152 J=1,NYSS
319* 152 YY(J) = DEP+YSV(J)
320* 153 CONTINUE
321* DO 200 I=1,NXS
322* DO 160 J=1,NYSS
323* IF (KSW(1) .GT. 0) GO TO 155
324* CALL BREAK(K,XX(I),YY(J))
325* CDAMX(1) = AMAX1(CDAMX(1),CON(J))
326* CDAMX(2) = AMAX1(CDAMX(2),DOS(J))
327* CDAMX(3) = AMAX1(CDAMX(3),AVCON(J))
GO TO 160

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328* 155 CALL WASHT
329* 160 CONTINUE
330* IF (KSW(1) .LE. 0) GO TO 170
331* IMB = 1
332* GO TO 200
333* 170 CONTINUE
334* OUTPUT GENERAL GRID PATTERN CALCULATIONS
335* KOUT = 4*I-3
336* CALL INTOUT(CON,KOUT,NYSS,2,1,1)
337* KOUT = 4*I-2
338* CALL INTOUT(DOS,KOUT,NYSS,2,1,1)
339* KOUT = 4*I-1
340* CALL INTOUT(AVCON,KOUT,NYSS,2,1,1)
341* KOUT = 4*I
342* CALL INTOUT(PASSTM,KOUT,NYSS,2,1,1)
343* 200 CONTINUE
344* IF (KSW(1) .LE. 0) GO TO 210
345* IF (Z(KK+1) .LT. ZLIM) GO TO 500
346* IFG = 1
347* OUTPUT WASHOUT DEPOSITION PATTERNS
348* DO 205 J=1,NYSS
349* DO 205 I=1,NXS
350* 205 CDAMX(1) = AMAX1(CDAMX(1),DEPN(I,J))
351* MDLS = 5
352* ZZL(1) = Z(1)
353* CALL GENPRT(1)
354* GO TO 500
355* 210 CONTINUE
356* CALL GENPRT(K)
357* K = K+1
358* IF (K .GT. NPTS) GO TO 500
359* IF (ZZL(K) .LT. Z(KK+1)) GO TO 148
360* 500 CONTINUE
361* *** LOOP FOR NEXT TEST *****
362* 700 CONTINUE
363* IF (NPS .EQ. 0) GO TO 1
364* 777 CONTINUE
365* 800 CONTINUE
366* IF (IMB .EQ. 5) CALL ENDJOB
367* 903 FORMAT (1H0,55X,11H***** LAYER,I2,6H ***** )
368* 904 FORMAT (1H0,57X,16H** INPUT DATA **)

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MDL32800
MDL32900
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MDL33100
MDL33200
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905 FORMAT (4H0 Q=E14.8,6H, ZRK=F7.3,17H, UBAR AT BOTTOM=F8.4,14H, MDL36900
1UBAR AT TOP=F8.4,18H, SIGAK AT BOTTOM=F8.5,14H SIGAK AT TOP=F8.5,14H, MDL37000
25,18H, SIGEK AT BOTTOM=F8.5,15H, SIGEK AT TOP=F8.5,7H, TAUKE=F8.5,7H, MDL37100
33,8H, TAUOK=F8.3,7H SIGXO=F9.4,8H, SIGYO=F9.4,8H, SIGZO=F9.4,1,MDL37200
49H, THETAK AT BOTTOM=F8.3,16H, THETAK AT TOP=F8.3,4H, Z=F9.3,7HMDL37300
5 ALPHA=F4.1,6H BETA=F4.1,4H, H=F9.3,7H, DELX=E14.8,7H, DELY=E14.8,7H, MDL37400
614.8,8H, IZMOD=I3,7H, TIM1=E14.8,6H ZLIM=F9.3, MDL37500
79H, LAMBDA=F7.4,8H, TIMAV=F8.3,6H, XRY=F8.3,6H, XRZ=F8.3,7H, XMDL37600
8LRY=F8.3,7H, XLRZ=F8.3,9H, GAMMAP=F5.3) MDL37700
917 FORMAT (12X,18(6H-----)/) MDL37800
918 FORMAT (4H0 Q=E14.8,17H, UBAR AT BOTTOM=F8.4,14H, UBAR AT TOP=F8.4,14H, MDL37900
18,4,18H, SIGAK AT BOTTOM=F8.5,15H, SIGAK AT TOP=F8.5,17H SIGEK AMDL38000
21 BOTTOM=F8.5,15H, SIGEK AT TOP=F8.5,8H, SIGXO=F9.4,8H, SIGYO=MDL38100
3F9.4,8H, SIGZO=F9.4,19H, THETAK AT BOTTOM=F8.3,15H THETAK AT TOPMDL38200
4=F8.3,4H, Z=F9.3,8H, ALPHA=F4.1,7H, BETA=F4.1, 7H, MDL38300
5DELX=E14.8,7H, DELY=E14.8,7H IZMOD=I3) MDL38400
919 FORMAT (1X,10H Z AT TOP=F10.4) MDL38500
920 FORMAT (6H0 ZRL=F7.3,18H, UBARL AT BOTTOM=F8.4,15H, UBARL AT TOPMDL38600
1=F8.4,18H, SIGAL AT BOTTOM=F8.5,15H, SIGAL AT TOP=F8.5,17H SIGEMDL38700
2L AT BOTTOM=F8.5,15H, SIGEL AT TOP=F8.5,19H, THETAL AT BOTTOM=F8.5,19H, MDL38800
38.3,16H, THETAL AT TOP=F8.3,7H, TAU=L=F8.3,7H TAUOL=F8.3,8H, ALPMDL38900
4HL=F4.1,7H, BETL=F4.1,7H, TAST=E14.8,7H, JBOT=I2,7H, JTOP=I2)MDL39000
921 FORMAT (18H0 UBAKL AT BOTTOM=F8.4,15H, UBARL AT TOP=F8.4,18H, SIMDL39100
1GAL AT BOTTOM=F8.5,15H, SIGAL AT TOP=F8.5,17H SIGEL AT BOTTOM=F8.5,17H, MDL39200
28.5,15H, SIGEL AT TOP=F8.5,19H, THETAL AT BOTTOM=F8.3,16H, THETAMD39300
3L AT TOP=F8.3,7H, TAU=L=F8.3,7H TAUOL=F8.3,8H, ALPHL=F4.1,7H, BMDL39400
4ETL=F4.1,7H, TAST=E14.8,7H, JBOT=I2,7H, JTOP=I2) MDL39500
922 FORMAT (1H0,56HCALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** UMDL39600
1BAR =F10.5,9H, THETA =F10.5,10H, DELTHP =F10.5,8H, DELU =F10.5MDL39700
2/1X,09H, SIGAP =F10.5,9H, SIGEP =F10.5) MDL39800
923 FORMAT (1H0,63HCALCULATED INPUT PARAMETERS FOR LAYER CHANGE MODEL MDL39900
14 *** UBAR =F10.5,9H, THETA =F10.5,10H, DELTHP =F10.5,1X,8H DEMDL40000
2LU =F10.5,9H, SIGAP =F10.5,9H, SIGEP =F10.5) MDL40100
STOP MDL40200
END MDL40300

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1* SUBROUTINE BREAK(K,XO,YO)
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLR,Y,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIN,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10)
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCC,NDD,NTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18* DIMENSION CON(1),DOS(1),AVCON(1),PASSTM(1),ERFX(1)
19* EQUIVALENCE (CON,DEPN),(DOS,DEPN(1,2)),(AVCON,DEPN(1,3)),(PASSTM,DBRK01900
20* 1EPN(1,4)),(ERFX,ANG(10))
21* REAL MPWR,L,LAT,LAMBDA
22* INTEGER TESTNO
23* *** THIS SUBROUTINE CALCULATES DOSAGE,CONCENTRATION AND WASHOUT **BRK02300
24* *** ON A GENERAL GRID WITHIN THE SECTOR DELPHI.
25* DETERMINE LOCATION OF RECEPTOR RELATIVE TO SOURCE AND WIND
26* DIRECTION
27* CALL COORD(N,KK,X,Y,XO,YO,ASP,XS,1)
28* DOS(J) = 0.0
29* CON(J) = 0.0
30* IF (NBK .NE. 0.AND.IBOT .LE. KK.AND.KK .LE. ITOP) GO TO 135
31* IS = 1
32* IF (N .EQ. 9) GO TO 310
33* C 125 CALCULATION OF MODELS 1,2,3
34* CALL SIGMA(X,KK,1)
35* IF (SIGY) 130,130,126
36* 126 IF (SIGZ .LT. 0.0 .AND. MODLS(KK) .EQ. 3) GO TO 130
37* LAT = Y/SIGY
38* LAT = -0.5*LAT*LAT
39* IF (LAT .LT. -60.0) GO TO 130
40* LAT = EXP(LAT)

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41* PEAKD = Q(KK)/(SGR2P*SIGY*UBAR(KK))
42* IF (MODLS(KK).EQ. 3) GO TO 20
43* PEAKD = PEAKD/(Z(KK+1)-Z(KK))
44* GO TO 21
45* 20 PEAKD = PEAKD/(SGR2P*SIGZ)
46* 21 CONTINUE
47* VER = 0.0
48* VREF = 1.0
49* IF (MODLS(KK).NE. 3) GO TO 70
50* VREF = 0.0
51* TMPQ1 = -0.5/(SIGZ*SIGZ)
52* A = H-ZZL(K)
53* B = H-Z(KK)-Z(KK)+ZZL(K)
54* C = B*B
55* C = C*TMPQ1
56* A1 = Z(KK+1)-Z(KK)
57* IF (C.LT. -30.0) GO TO 70
58* D = A*A
59* D = D*TMPQ1
60* IF (D.LT. -30.0) GO TO 50
61* VER = EXP(D)
62* 50 VER = VER+GAMMA(1)*EXP(C)
63* C = 1.0
64* D = GAMMA(1)
65* E = D*D
66* AB = 0.0
67* 60 AB = AB+2.0
68* TR = AB*A1
69* TLIM = TR-B
70* TLIM = TLIM*TLIM*TMPQ1
71* IF (TLIM.LT. -10.0) GO TO 70
72* ST01 = TR+A
73* ST02 = TR-A
74* ST03 = TR+B
75* VREF = VREF+C*EXP(TLIM)+D*(EXP(ST01*ST01*TMPQ1)+EXP(ST02*ST02*TMPQ1)+EXP(ST03*ST03*TMPQ1))
76* 11)*E*EXP(ST03*ST03*TMPQ1)
77* C = D
78* D = E
79* E = E*GAMMA(1)
80* GO TO 60
81* 70 CONTINUE

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BRK04100
BRK04200
BRK04300
BRK04400
BRK04500
BRK04600
BRK04700
BRK04800
BRK04900
BRK05000
BRK05100
BRK05200
BRK05300
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BRK07000
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BRK07400
BRK07500
BRK07600
BRK07700
BRK07800
BRK07900
BRK08000
BRK08100

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82*      TMPQ1 = X/UBAR(KK)
83*      DOS(J) = PEAKD*LAT*(VER+VREF)
84*      IF (DECAY .GT. 0.0) DOS(J) = DOS(J)*EXP(-DECAY*TMPQ1)
85*      IF (LAMBDA .LE. 0.0.OR.TIM1 .GE. TMPQ1) GO TO 127
86*      IF (Z(KK) .GT. ZLIM) GO TO 127
87*      AB = EXP(-LAMBDA*(TMPQ1-TIM1))
88*      DOS(J) = DOS(J)*AB
89*      127 CONTINUE
90*      ANG(1) = UBAR(KK)
91*      ANG(2) = SIGX
92*      IF (SIGX) 129,129,128
93*      128 CONTINUE
94*      CON(J) = DOS(J)*UBAR(KK)/(SQR2P*SIGX)
95*      129 CONTINUE
96*      130 IF (IS .EQ. 1) GO TO 310
97*      GO TO 140
98*      135 IS = 0
99*      IF (N .NE. 9) GO TO 125
100*      C      CALCULATION OF THE FULL TRANSITION MODEL, MODEL4
101*      140 DO 200 M=IBOT,ITOP
102*      CALL COORD(N,M,X,Y,XO,YO,ASP,XS,2)
103*      IF (N .EQ. 9) GO TO 200
104*      CALL SIGMA(X,M,2)
105*      ST01 = 1.414214*SIGZ
106*      TMPQ1 = 1.0/ST01
107*      IF (SIGYNK) 200,200,147
108*      147 IF (SIGZ) 200,200,148
109*      148 LAT = Y/SIGYNK
110*      LAT = -0.5*LAT*LAT
111*      IF (LAT .LT. -60.0) GO TO 200
112*      XBARX = EXP(LAT)
113*      A = Z(M+1)-ZZL(K)
114*      B = ZZL(K)-Z(M)
115*      C = Z(M+1)+ZZL(K)-Z(IBOT)-Z(IBOT)
116*      D = Z(IBOT)+Z(IBOT)-Z(M)-ZZL(K)
117*      ERFX(1) = A*TMPQ1
118*      ERFX(2) = B*TMPQ1
119*      ERFX(3) = C*TMPQ1
120*      ERFX(4) = D*TMPQ1
121*      CALL ISO(1,4)
122*      ST02 = ERFX(1)+ERFX(2)+GAMMA(1)*(ERFX(3)+ERFX(4))

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BRK09500
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BRK09700
BRK09800
BRK09900
BRK10000
BRK10100
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S1 = 0.0
S3 = Z(ITOP+1)-Z(IBOT)
E = 1.0
F = GAMMA(1)
G = F*F
IFL = 0
150 S1 = S1+2.0
S2 = S1*S3
ERFX(3) = (S2+D)*TMPQ1
ERFX(4) = (C-S2)*TMPQ1
IF (IFL.EQ. 0) GO TO 155
IF (ERFX(3) .GT. 3.0 .AND. ERFX(4) .LT. -3.0) GO TO 185
155 IFL = 1
ERFX(1) = (S2+B)*TMPQ1
ERFX(2) = (A-S2)*TMPQ1
CALL ISO(1,4)
STO2 = STO2+F*(ERFX(1)+ERFX(2))+E*(ERFX(3)+ERFX(4))
ERFX(1) = (S2+A)*TMPQ1
ERFX(2) = (B-S2)*TMPQ1
ERFX(3) = (S2+C)*TMPQ1
ERFX(4) = (D-S2)*TMPQ1
CALL ISO(1,4)
STO2 = STO2+F*(ERFX(1)+ERFX(2))+G*(ERFX(3)+ERFX(4))
E = F
F = G
G = G*GAMMA(1)
GO TO 150
165 CONTINUE
STO3 = 1.0/(Z(M+1)-Z(M))
XBARX = EXP(-.5*(Y/SIGYNK)**2)
190 TMPQ2 = X/UBAR(JF)
S1 = (Q(M)*STO3/(2.0*SQR2P*UBAR(JF)*SIGYNK))*XBARX*STO2
IF (DECAY .GT. 0.0) S1 = S1*EXP(-DECAY*TMPQ2)
IF (LAMBDA .LE. 0.0.OR. TIM1.GE. TMPQ2+TAST(ILK-1)) GO TO 195
IF (Z(M) .GT. ZLIM) GO TO 195
S1 = S1*EXP(-LAMBDA*(TMPQ2+TAST(ILK-1)-TIM1))
195 CONTINUE
IF (SIGXNK) 210,210,211
210 S2 = 0.0
GO TO 212
211 CONTINUE

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BRK12400
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BRK18000
BRK18100
BRK18200

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S2 = (S1*UBAR(JF)/(SQR2P*SIGXNK))
212 CONTINUE
  DOS(J) = DOS(J)+S1
  CON(J) = CON(J)+S2
200 CONTINUE
  ANG(1) = UBAR(JF)
  ANG(2) = SIGXNK
310 CONTINUE
  AVCON(J) = 0.0
  PASSTM(J) = 0.0
  IF (ANG(2) .LE. 0.0) GO TO 311
  IF (DOS(J) .LE. 0.0) GO TO 311
  ERFX(1) = ANG(1)*TIMAV/(2.8284271*ANG(2))
  CALL ISO(1,1)
  AVCON(J) = (DOS(J)/TIMAV)*ERFX(1)
  PASSTM(J) = 4.3*ANG(2)/ANG(1)
311 CONTINUE
  RETURN
  END

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1* SUBROUTINE DEPOS
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGX(15),SIGY(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLR,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5IAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10),
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,ST01,
12* 2ST02,ST03,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDDU,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18* ***** THIS SUBROUTINE CALCULATES GRAVITATIONAL DEPOSITION AT GROUNDDEP01800
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41* 10 CONTINUE
42*   WRITE (6,901) (I,VS(I),I,PERC(I),I=1,NVS)
43*   WRITE (6,902) (I,GAMMAP(I),I=1,NVS)
44*   IF (NVB .LE. 0) GO TO 12
45*   WRITE (6,908) HB,(I,VB(I),I,PERCB(I),I=1,NVB)
46* 12 CONTINUE
47*   THETA(1) = THETA(1)
48*   IF (THETA(1) .LT. 180.0) THET = (THETA(1)+180.0)*RAD
49*   IF (THETA(1) .GE. 180.0) THET = (THETA(1)-180.0)*RAD
50*   DTHK(1) = 0.0
51*   DO 20 N=2,NZS
52*   20 UTHK(N) = DTHK(N-1)+DELTHP(N-1)
53*   NYSS = NYS
54*   IF (NYS .GT. 0) GO TO 23
55*   S = THETA(1)+0.5*DTHK(NZS)/FLOAT(NNZ)+180.0
56*   S = AMOD(S,360.0)
57*   S = AMOD(S,360.0)
58*   NYS = 41
59*   NYSS = 41
60*   DO 22 J=1,NYS
61*   22 YY(J) = YSV(J)+S
62* 23 CONTINUE
63*   DO 25 N=2,NZS
64*   25 UTHK(N) = DTHK(N)*RAD
65*   DO 30 J=1,NYS
66*   DO 30 I=1,NXS
67*   30 DEPN(I,J) = 0.0
68*   NIAD = 1
69*   NTAL = 1
70*   IF (NVB .GT. 0) NTAD = 2
71*   IF (NVB .LE. 0.AND.NNZ.EQ. 1) NTAL = 2
72*   DO 73 JF=NTAL,NTAD
73*   NTAP = NVS
74*   IF (JF.EQ. 2) NTAP = NVB
75*   DO 73 II=1,NTAP
76*   IF (JF.EQ. 2.OR.VS(II).LE. 10.0) GO TO 35
77*   WRITE (6,903) VS(II)
78*   RETURN
79* 35 CONTINUE
80*   NTAK = 1
81*   NTAR = NNZ

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IF (NVB .LE. 0) GO TO 45
IF (JF .EQ. 2) GO TO 40
NTAR = NTAR-1
GO TO 45
40 NTAK = NNZ
45 DO 72 KK=NTAK,NTAR
IF (JF .EQ. 2) GO TO 50
IZ = 1
S = ((Z(KK+1)-Z(KK))* .3333333)+Z(KK)
CALL SGP(S,KK,SIGENK(1),1,IDMY,DMY,DY,1)
CALL SGP(S,KK,DY,2,IZ,UBHK,DY,2)
DETERMINE NO. SOURCES IN LINE SOURCE SIMULATION
DHK = ACCUR*SIGENK(1)*SQBAR*SGRT(1.0+VS(11)/UBHK)
IF (DHK .LT. 10.0) DHK = 10.0
S = (Z(KK+1)-Z(KK))/DHK
NXCI = S+1.0
IF (NXCI .LT. 3) NXCI = 3
IF (NXCI .GT. 40) NXCI = 40
IF (JF .EQ. 1) WRITE (6,909) VS(11),KK,NXCI
DHK = (Z(KK+1)-Z(KK))/FLOAT(NXCI)
STO1 = Z(KK)
GO TO 55
50 NXCI = 1
STO1 = 0.0
DHK = HB
55 DO 60 IZ=1,NXCI
STO1 = STO1+DHK
ZL(IZ) = STO1
CALL SGP(ZL(IZ),KK,SIGENK(IZ),1,IDMY,DY,DY,1)
CALL SGP(ZL(IZ),KK,SIGANK(IZ),2,IDMY,DY,DY,1)
CALL SGP(ZL(IZ),KK,DY,2,IZ,UBHK,DY,2)
CALL SGP(ZL(IZ),KK,DY,2,IZ,DY,DY,4)
60 CONTINUE
DO 71 I=1,NXS
DO 71 J=1,NYS
CALL COORD(N,1,X,Y,XX(I),YY(J),ASP,XS,1)
IF (N .EQ. 9) GO TO 71
DO 70 IZ=1,NXCI
PHI = ABS(ASP-(THET+ANG(IZ)))
IF (PHI .GT. 3.1415926536) PHI = 6.2831853072-PHI
Y = XS*SIN(PHI)

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906 FORMAT (8H0 LAYER ,I2,15H, UBARK AT TOP=,F8.4,15H, SIGAK AT TOP=,FDEP16400
18.5,15H, SIGEK AT TOP=,F8.5,4H, Q=,E14.8,7H, DELX=,E14.8/6H DELY=,DEP16500
2E14.8,8H, SIGYO=,F9.4,8H, SIGZO=,F9.4,8H, ALPHA=,F4.1,7H, BETA=,F4DEP16600
3.1,4H, Z=,F9.3/15H THETAK AT TOP=,F8.3)
DEP16700
907 FORMAT (1X,10H Z AT TOP=,F10.4)
DEP16800
908 FORMAT (1X,19HHEIGHT OF BURST HB=,F10.4,3HVB(,I2,2H)=,F10.5,8H, PEDEP16900
1RCB(,I2,2H)=,F10.5,2H, /(1X,3HVB(,I2,2H)=,F10.5,8H, PERCB(,I2,2H)=DEP17000
2,F10.5,5H, VB(,I2,2H)=,F10.5,8H, PERCB(,I2,2H)=,F10.5,5H, VB(,I2, DEP17100
32H)=,F10.5,8H, PERCB(,I2,2H)=,F10.5,2H, ))
DEP17200
909 FORMAT (1H0,10X,4HVS =,F8.4,12H, LAYER NO. ,I2,18H, NO. OF SOURCESDEP17300
1 =,I6)
DEP17400
END
DEP17500

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1* SUBROUTINE WASHT
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLR,Y,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),
8* 6HJ,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARKL(10),SIGAL(10),SIGEL(10)
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STO1,
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDD,NTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18* DIMENSION WASHOU(41,1)
19* EQUIVALENCE (DEPN,WASHOU)
20* EQUIVALENCE (ISW6,SIGENK),(A,SIGENK(2)),(B,SIGENK(3)),(C,SIGENK(4)
21* 1),(D,SIGENK(5)),(E,SIGENK(6)),(G,SIGENK(7))
22* REAL MPWR,L,LAMBDA
23* INTEGER TESTNO
24* C THIS SUBROUTINE CALCULATES PRECIPITATION DEPOSITION - MODEL 5
25* C = 1.0
26* D = 1.0
27* E = 1.0
28* CALL COORD(N,KK,X,Y,XX(I),YY(J),ASP,XS,1)
29* IF (NBK.NE.0.AND.IBOT.LE.KK.AND.KK.LE.ITOP) GO TO 20
30* IF (N.EQ. 9) GO TO 70
31* 10 CALL SIGMA(X,KK,1)
32* A = UBAR(KK)
33* B = SIGY
34* G = TIM1
35* GO TO 30
36* 20 IF (N.NE. 9) GO TO 10
37* CALL COORD(N,KK,X,Y,XX(I),YY(J),ASP,XS,2)
38* IF (N.EQ. 9) GO TO 70
39* CALL SIGMA(X,KK,2)
40* A = UBAR(JF)

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      B = SIGYNK
      G = TIM1-TAST(ILK-1)
      SIGX = SIGXNK
30    IF (X/A .LT. G) GO TO 70
      IF (B .LE. 0.0) GO TO 70
      IF (G .LT. (X-2.15*SIGX)/A) GO TO 40
      IF (ISKIP(4) .EQ. 1) GO TO 40
      E = AMOD(YY(J),360.0)
      IF (E .LT. 0.0) E = 360.0+E
      WRITE (6,80) XX(I),E
40    IF (MODLS(KK) .EQ. 3) GO TO 50
50    E = Y/B
      E = -0.5*E*E
      IF (E .LT. -60.0) GO TO 70
      E = EXP(E)
      IF (ISKIP(4) .EQ. 1) GO TO 60
      C = EXP(-LAMBDA*(X/A-G))
      WASHOU(I,J) = WASHOU(I,J)+(LAMBDA*Q(KK)/(SQR2P*A*B))*C*E
70    RETURN
80    FORMAT (1H0,36H *** PRECIPITATION DEPOSITION AT XX=,F10.3,5H, YY=,
1F10.3,26H MAY BE OVER ESTIMATED ***/)
      END

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WSH04100
WSH04200
WSH04300
WSH04400
WSH04500
WSH04600
WSH04700
WSH04800
WSH04900
WSH05000
WSH05100
WSH05200
WSH05300
WSH05400
WSH05500
WSH05600
WSH05700
WSH05800
WSH05900
WSH06000
WSH06100
WSH06200

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SUBROUTINE GENPRT(K)
COMMON /PARAMT/ TESTNO(12),
1NFK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
4XLKY,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
5STAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCR,VB(20),PERCB(20),
6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10)
7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,
2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
3MPWR,I,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
4NCCC,NDDU,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
7NYSS,CDAMX(3)
THIS PROGRAM CONTROLS PRINTING OF ALL PROGRAM CALCULATIONS
DIMENSION LINE(1),YB(1),DPN(41,1),IX(1)
DIMENSION JLINE(70),KLINE(10)
REAL LAMBDA
COMMON /LBLBL/ J1(9),J2(4),J3(48),J5(6),J7(3),J8(16),J9(13),J10,
1J4(12),J11(2),UNIT(15)
EQUIVALENCE (YBARY,LINE),(YB,SIGENK),(DPN,BETANK),(IX,XI)
COMMON /BNDS/ XRI,XLFT,YBOT,YTOP,XPL,YPL
COMMON /ILPLTS/ XMAX,XMIN,YMAX,YMIN,XLM1,YBM1,HT,CHARF,SCLX,SCLY,
1XSIZE1,YSIZE1
COMMON /PLTLLO/ ISW,XMAXJN,YMAXJN,XCIZE,YCIZE
COMMON /XYXYPT/ YP(41),XP(41),A(41),B(41),C(41),U(41),XI(41),YI(41)
1),NUM(3),NC
DATA ISP/1H0/,JSP/1H/,MS/57/
DATA UNIT/1H,5H(PPM),1H,6H
11H),6H (M,6HG/M**3,1H),5H (MG,6HSEC/M*,3H*3)/
JM = 5
IT1 = 1
IT2 = 7
IT3 = 13
NCV1 = 14
NCV2 = 7
NCV3 = 24

GPT00100
GPT00200
GPT00300
GPT00400
GPT00500
GPT00600
GPT00700
GPT00800
GPT00900
GPT01000
GPT01100
GPT01200
GPT01300
GPT01400
GPT01500
GPT01600
GPT01700
GPT01800
GPT01900
GPT02000
GPT02100
GPT02200
GPT02300
GPT02400
GPT02500
GPT02600
GPT02700
GPT02800
GPT02900
GPT03000
GPT03100
GPT03200
GPT03300
GPT03400
GPT03500
GPT03600
GPT03700
GPT03800
GPT03900
GPT04000

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IF (KSW(1) .LE. 0) GO TO 10
IT1 = 19
GO TO 20
10 IF (KSW(2) .LE. 0) GO TO 30
IT1 = 31
20 NCV1 = 25
   NCV2 = -1
   NCV3 = -1
   JM = 1
30 CONTINUE
   XPRT = TIMAV/60.0
   NXSS = NXS-2
   IF (ISKIP(1) .LE. 0) GO TO 170
   PRINT GENERAL GRID CALCULATIONS
   GET Y IN PROPER INTERVAL
   DO 100 J=1,NYSS
     YB(J) = AMOD(YY(J),360.0)
     IF (YB(J) .LT. 0.0) YB(J) = 360.0+YB(J)
100 CONTINUE
     IB = 0
     DO 160 KS=1,JM
       IF (JM .EQ. 1) GO TO 110
       IF (KS .EQ. 5) IB = 1
       CALL INPTS(KS,IB,NXS,II,NYSS,DEPN,SIGANK)
110  CALL HEDING(KSW,KS,1,0)
       CALL LABELS(K)
       CALL VRTCLE(KS,JM,KSW,SIGANK,ISKIP(5),NCV)
       N1 = -9
120 N1 = N1+10
       N2 = N1+9
       IF (N1 .GT. NYSS) GO TO 160
       IF (N2 .GT. NYSS) N2 = NYSS
       LINES = 80
       DO 150 I=1,NXS
         LINES = LINES+1
         IF (LINES .LT. MS) GO TO 140
         IF (JM .GT. 1) GO TO 125
         CALL PRITTL(NWD,LINES,LINE,0.0,0.0)
         GO TO 126
125 CALL PRITTL(NWD,LINES,LINE,DECAY,LAMBDA)
126 CONTINUE

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GPT04100
GPT04200
GPT04300
GPT04400
GPT04500
GPT04600
GPT04700
GPT04800
GPT04900
GPT05000
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GPT05200
GPT05300
GPT05400
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GPT06400
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GPT06600
GPT06700
GPT06800
GPT06900
GPT07000
GPT07100
GPT07200
GPT07300
GPT07400
GPT07500
GPT07600
GPT07700
GPT07800
GPT07900
GPT08000
GPT08100

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82*      IF (KS .GT. 3) GO TO 130
83*      WRITE (6,900) CDAMX(KS), (SIGANK(J),J=1,NCV)
84*      LINES = LINES+2
85*      130 WRITE (6,901) (YB(J),J=N1,N2)
86*      WRITE (6,902) (SIGANK(J),J=1,NCV)
87*      LINES = LINES+4
88*      140 WRITE (6,903) XX(I), (DEPN(I,J),J=N1,N2)
89*      150 CONTINUE
90*      IF (N2 .LT. NYSS) GO TO 120
91*      CONTINUE
92*      160 CONTINUE
93*      170 CONTINUE
94*      IF (JM .GT. 1) GO TO 190
95*      DO 180 I=1,NXS
96*      KOUT = 4*I-3
97*      CALL INTOUT(DEPN,KOUT,NYSS,2,41,I)
98*      180 CONTINUE
99*      190 CONTINUE
100*      PRINT AND/OR PLOT CENTERLINE CALCULATIONS
101*      IF (ISKIP(2) .LE. 0) GO TO 480
102*      IB = 0
103*      DO 340 KS=1,JM
104*      IF (JM .EQ. 1) GO TO 250
105*      IF (KS .EQ. 5) IB = 1
106*      CALL INPTS(KS,IB,NXS,II,NYSS,DEPN,SIGANK)
107*      250 CONTINUE
108*      DO 340 I=1,NXS
109*      I1 = IX(I)
110*      IF (KS .GT. 1) GO TO 270
111*      IX(I) = NYSS/2+1
112*      YMAX = 0.0
113*      FIND INDEX AT OR CLOSE TO MAXIMUM
114*      DO 260 J=1,NYSS
115*      IF (DEPN(I,J) .LE. YMAX) GO TO 260
116*      IX(I) = J
117*      YMAX = DEPN(I,J)
118*      260 CONTINUE
119*      I1 = IX(I)
120*      YP(I) = YY(I1)
121*      I2 = MAX0(1,I1-3)
122*      I3 = MIN0(NYSS,I1+3)
123*      I3 = I3-I2+1

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123*      DPN(I,KS) = DEPN(I,I1)
124*      IF (I3 .LT. 3) GO TO 340
125*      DO 280 J=1,I3
126*      280 XP(J) = DEPN(I,I2+J-1)
127*      CALL SPLINE(YY(I2),XP,A,B,C,D,I3,IER)
128*      IF (IER .EQ. 1) GO TO 340
129*      IF (KS .GT. 1) GO TO 310
130*      J = 1
131*      YMAX = 0.0
132*      YPL = YY(I2)-0.1
133*      290 YPL = YPL+0.1
134*      IF (YPL .LT. YY(I2+J)) GO TO 300
135*      J = J+1
136*      300 IF (J .GE. I3) GO TO 340
137*      XPL = YPL-YY(I2+J-1)
138*      XPL = XP(J)+XPL*(B(J)+(YPL-YY(I2+J))*(2.0*C(J)+C(J+1))+A(J)*XPL)*
139*      1.166666666)
140*      IF (XPL .LE. YMAX) GO TO 290
141*      YMAX = XPL
142*      YP(I) = YPL
143*      DPN(I,KS) = YMAX
144*      GO TO 290
145*      310 DO 320 J=1,I3
146*      IF (YP(I) .LT. YY(I2+J-1)) GO TO 330
147*      320 CONTINUE
148*      330 XPL = YP(I)-YY(I2+J-2)
149*      DPN(I,KS) = XP(J-1)+XPL*(B(J-1)+(YP(I)-YY(I2+J-1))*(2.0*C(J-1)+C(J-1))+A(J-1)*XPL)*
150*      1)+A(J-1)*XPL)*.166666666)
151*      340 CONTINUE
152*      C
153*      PRINT MAXIMUM CENTERLINE CALCULATIONS
154*      IF (ISKIP(2) .EQ. 2) GO TO 420
155*      CALL HEDING(KSW,1,2,1)
156*      CALL LABELS(K)
157*      LINES = 80
158*      CDAMX(1) = 0.0
159*      CDAMX(2) = 0.0
160*      CDAMX(3) = 0.0
161*      DO 350 I=1,NXSS
162*      DO 350 J=1,3
163*      CDAMX(J) = AMAX1(CDAMX(J),DPN(I,J))
164*      350 CONTINUE
165*      GPT12300
166*      GPT12400
167*      GPT12500
168*      GPT12600
169*      GPT12700
170*      GPT12800
171*      GPT12900
172*      GPT13000
173*      GPT13100
174*      GPT13200
175*      GPT13300
176*      GPT13400
177*      GPT13500
178*      GPT13600
179*      GPT13700
180*      GPT13800
181*      GPT13900
182*      GPT14000
183*      GPT14100
184*      GPT14200
185*      GPT14300
186*      GPT14400
187*      GPT14500
188*      GPT14600
189*      GPT14700
190*      GPT14800
191*      GPT14900
192*      GPT15000
193*      GPT15100
194*      GPT15200
195*      GPT15300
196*      GPT15400
197*      GPT15500
198*      GPT15600
199*      GPT15700
200*      GPT15800
201*      GPT15900
202*      GPT16000
203*      GPT16100
204*      GPT16200
205*      GPT16300

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164*      IF (JM .GT. 1) GO TO 370
165*      IF (KSW(1) .GT. 0) GO TO 360
166*      I1 = 31
167*      I2 = 35
168*      I3 = 33
169*      GO TO 380
170*      I1 = 19
171*      I2 = 23
172*      I3 = 21
173*      GO TO 380
174*      CONTINUE
175*      I1 = 1
176*      I2 = 6
177*      I3 = 3
178*      IF (ISKIP(5) .NE. 4) GO TO 380
179*      I1 = 10
180*      I2 = 15
181*      I3 = 12
182*      CONTINUE
183*      DO 410 I=1,NXSS
184*      IF (DPN(I,1) .LE. 0.0) GO TO 410
185*      LINES = LINES+1
186*      IF (LINES .LT. MS) GO TO 400
187*      IF (JM .GT. 1) GO TO 390
188*      CALL PRTTIL(NWD,LINES,LINE,0.0,0.0)
189*      WRITE (6,904) ISP,CDAMX(1),(J3(J),J=I1,I2)
190*      WRITE (6,905) (J3(J),J=I1,I3)
191*      LINES = LINES+6
192*      GO TO 400
193*      390 CALL PRTTIL(NWD,LINES,LINE,DECAY,LAMBDA)
194*      WRITE (6,904) ISP,CDAMX(1),(J3(J),J=1,3)
195*      WRITE (6,907) JSP,CDAMX(2),(J3(J),J=7,8)
196*      WRITE (6,908) JSP,CDAMX(3),XPRT,(J3(J),J=13,16)
197*      WRITE (6,906) XPRT,(UNIT(J),J=I1,I2),(UNIT(J),J=I1,I3),(UNIT(J),J=17,9),(UNIT(J),J=11,I3)
198*      LINES = LINES+8
199*      400 YPL = AMOD(YP(I),360.0)
200*      IF (YPL .LT. 0.0) YPL = YPL+360.0
201*      WRITE (6,909) XX(I),YPL,(DPN(I,J),J=1,JM)
202*      CONTINUE
203*      410 CONTINUE
204*      420 CONTINUE

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GPT16400
GPT16500
GPT16600
GPT16700
GPT16800
GPT16900
GPT17000
GPT17100
GPT17200
GPT17300
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GPT17500
GPT17600
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GPT17900
GPT18000
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GPT19000
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GPT19200
GPT19300
GPT19400
GPT19500
GPT19600
GPT19700
GPT19800
GPT19900
GPT20000
GPT20100
GPT20200
GPT20300
GPT20400

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205*      C      PLOT MAXIMUM CENTERLINE CALCULATIONS
206*      IF (ISKIP(2) .EQ. 1) GO TO 480
207*      HT = 0.0
208*      XPL = 0.0
209*      YPL = 0.0
210*      DO 430 I=1,NXS
211*      X1 = XX(I)*SIN(YP(I)*RAD)
212*      Y1 = XX(I)*COS(YP(I)*RAD)
213*      YB(I) = Sqrt((X1-XPL)**2+(Y1-YPL)**2)+HT
214*      HT = YB(I)
215*      XPL = X1
216*      YPL = Y1
217*      IF (ISW .NE. 2) GO TO 436
218*      DO 435 I=1,NXS
219*      YB(I) = ALog10(YB(I))
220*      435 CONTINUE
221*      436 CONTINUE
222*      CALL FSTPLT(H,RNG,AZM,NAMCAS,IDATE,ITIME,CDAMX(1),CDAMX(2),CDAMX(3),XPRT)
223*      1),J3(IT1),J3(IT2),J3(IT3),NCV1,NCV2,NCV3,XPRT)
224*      IF (JM .GT. 3) JM = 3
225*      DO 470 KS=1,JM
226*      CALL HEDING(KSW,KS,2,0)
227*      CALL LABELS(K)
228*      CALL VRTCLE(KS,JM,KSW,SIGANK,ISKIP(5),NCV)
229*      NCV = NCV*6
230*      GO TO (440,450,460),KS
231*      440 IF (KSW(1) .GT. 0.OR.KSW(2) .GT. 0) GO TO 450
232*      PLOT MAXIMUM CENTERLINE CONCENTRATION
233*      IF (NCCC .LE. 0) GO TO 470
234*      CALL LLPLOT(DPN(1,KS),YB,NXSS,LINE,CI,NCCC,NWD,SIGANK,NCV)
235*      NWD = NWD*6
236*      CALL LSSOPT(YB,DPN(1,KS),NXSS,NCCC,CI,SIGANK,LINE,NCV,NWD)
237*      GO TO 470
238*      C      CENTERLINE DEPOSITION OR DOSAGE
239*      450 IF (NDUD .LE. 0) GO TO 470
240*      CALL LLPLOT(DPN(1,KS),YB,NXSS,LINE,DI,NDDD,NWD,SIGANK,NCV)
241*      NWD = NWD*6
242*      CALL LSSOPT(YB,DPN(1,KS),NXSS,NDDD,DI,SIGANK,LINE,NCV,NWD)
243*      GO TO 470
244*      C      CENTERLINE TIME-MEAN CONCENTRATION
245*      460 IF (NTTT .LE. 0) GO TO 470

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246* CALL LLPLOT(DPN(1,KS),YB,NXSS,LINE,II,NTTT,NWD,SIGANK,NCV)
247* NWD = NWD*6
248* CALL LSSOPT(YB,DPN(1,KS),NXSS,NTTT,II,SIGANK,LINE,NCV,NWD)
249* 470 CONTINUE
250* 480 CONTINUE
251* C
252* PLOT ISOPLETHS
253* IF (ISKIP(3) .LE. 0) GO TO 540
254* IF (ISKIP(3) .EQ. 1) GO TO 485
255* CALL FSTPLT(H,RNG,AZM,NAMCAS,IDATE,ITIME,CDAMX(1),CUAMX(2),CDAMX(3),
256* 1),J3(IT1),J3(IT2),J3(IT3),NCV1,NCV2,NCV3,XPRT)
257* 485 CONTINUE
258* IF (JM .GT. 3) JM = 3
259* DO 530 KS=1,JM
260* IF (JM .EQ. 1) GO TO 490
261* CALL INPTS(KS,0,NXS,II,NYSS,DEPN,SIGANK)
262* 490 CONTINUE
263* CALL HEDING(KSW,KS,3,0)
264* CALL LABELS(K)
265* CALL VRTGLE(KS,JM,KSW,KLINE,ISKIP(5),NCV)
266* DO 495 J=1,NWD
267* 495 JLINE(J) = LINE(J)
268* NWD = NWD*6
269* GO TO (500,510,520),KS
270* 500 IF (KSW(1) .GT. 0.OR.KSW(2) .GT. 0) GO TO 510
271* IF (NCC .LE. 0) GO TO 530
272* CALL ISSOPT(XX,YY,NXS,NCC,CI,JLINE,NWD,DEPN,NYSS,YBARY,DEPN,II,KS,
273* 1YT,ISKIP(3),KLINE,NCV,JM,DECAY,LAMBDA)
274* GO TO 530
275* 510 IF (NDD .LE. 0) GO TO 530
276* CALL ISSOPT(XX,YY,NXS,NDD,DI,JLINE,NWD,DEPN,NYSS,YBARY,DEPN,II,KS,
277* 1YT,ISKIP(3),KLINE,NCV,JM,DECAY,LAMBDA)
278* GO TO 530
279* 520 IF (NTT .LE. 0) GO TO 530
280* CALL ISSOPT(XX,YY,NXS,NTT,II,JLINE,NWD,DEPN,NYSS,YBARY,DEPN,II,KS,
281* 1YT,ISKIP(3),KLINE,NCV,JM,DECAY,LAMBDA)
282* 530 CONTINUE
283* 540 CONTINUE
284* RETURN
285* 900 FORMAT (1H0,38X,F9.3,21H IS THE MAXIMUM GRID ,5A6)
286* 901 FORMAT (1H0,6H RANGE,44X,29H- AZIMUTH BEARING (DEGREES) -/1X,8H(MEGP
ITERS),10(3X,F7.2,2X))

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287*	902	FORMAT (52X,6A6)	GPT28700
288*	903	FORMAT (1X,F8.1,10(F10.3,2X))	GPT28800
289*	904	FORMAT (A1,38X,F9.3,20H IS THE MAXIMUM PEAK,5A6)	GPT28900
290*	905	FORMAT (1H0,23X,12HMAXIMUM PEAK/18H RANGE AZIMUTH,6X,2A6,A2/11GPT29000	
291*		1X,24HBEARING DEPOSITION/1X,34H(METERS) (DEGREES) (MG/MGPT29100	
292*		2**2)	GPT29200
293*	906	FORMAT (1H0,23X,12HMAXIMUM PEAK,13X,7HMAXIMUM,13X,12HMAXIMUM PEAK/GPT29300	
294*		17H RANGE,4X,7HAZIMUTH,6X,13HCONCENTRATION,13X,6HDOSAGE,8X,F5.1,17GPT29400	
295*		2H MINUTE TIME-MEAN,8X,7HTIME OF,12X,13HAVERAGE CLOUD/11X,7HBEARINGGPT29500	
296*		3,51X,13HCONCENTRATION,9X,13HCLCUD PASSAGE,9X,13HCONCENTRATION/1X,1GPT29600	
297*		49H(METERS) (DEGREES) ,5(2X,3A6,2X))	GPT29700
298*	907	FORMAT (A1,44X,F9.3,15H IS THE MAXIMUM,5A6)	GPT29800
299*	908	FORMAT (A1,32X,F9.3,20H IS THE MAXIMUM PEAK,F5.1,7H MINUTE,5A6)	GPT29900
300*	909	FORMAT (1X,F8.1,2X,F6.1,8X,F10.3,11X,F10.3,13X,F10.3,2(12X,F10.3))GPT30000	
301*		END	GPT30100

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1* SUBROUTINE READER(IFF)
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLRZ,XLRZ,ZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMDDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VR(20),PERCR(20),RDR00700
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,URARL(10),SIGAL(10),SIGEL(10),RDR00800
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),RDR01000
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,OPWR,
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDDD,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18*
19* THIS SUBROUTINE READS ALL INPUT DATA AND CALCULATES NECESSARY
20* LAYER PARAMETERS
21* INTEGER TESTNO
22* REAL MPWR,L,LAMBDA
23* DIMENSION XSV(41),IZR1(1)
24* DIMENSION TEMPK(16),TEMPL(10)
25* DIMENSION NTFB(2)
26* COMMON /PLTISO/ SCL,XMAXIN,YMAXIN,XSIZE,YSIZE,RASTIN,JSW
27* COMMON /PLTLLO/ ISW,XMAXJN,YMAXJN,XCIZE,YCIZE
28* EQUIVALENCE (NTFB,ITOP)
29* EQUIVALENCE (I1,CON),(I2,CON(2)),(I3,CON(3)),(NTAL,CON(4)),(NTAK,
30* 1CON(5)),(INNZ,CON(6)),(S,CON(7)),(S1,CON(8)),(P,CON(9)),(M,CON(10))
31* 2),(NNZI,CON(11)),(N,CON(12)),(DIF1,CON(13)),(DIF2,CON(14))
32* EQUIVALENCE (IZR1,SKIP)
33* DATA YSV/-40.,-35.,-30.,-27.,-24.,-22.,-20.,-18.,-16.,-14.,-12.,
34* 1-10.,-8.,-7.,-6.,-5.,-4.,-3.,-2.,-1.,0.,1.,2.,3.,4.,5.,6.,7.,8.,
35* 210.,12.,14.,16.,18.,20.,22.,24.,27.,30.,35.,40./
36* DATA XSV/500.,1250.,2500.,3750.,5000.,6250.,7500.,8750.,10000.,
37* 111250.,12500.,13750.,15000.,16250.,17500.,18750.,20000.,21250.,
38* 222500.,23750.,25000.,26250.,27500.,28750.,30000.,31250.,32500.,
39* 333750.,35000.,36250.,37500.,38750.,40000.,41250.,42500.,43750.,
40* 445000.,47500.,50000.,65000.,80000./
41* MACHINE DEPENDENT STATEMENT ASSUMES SIX BYTES/WORD

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41* DATA TESTNO/12*6H /,NAMCAS/12*6H /
42* SR121 = 1.0/SQRT(12.0)
43* NAMELIST /NAM2/ TESTNO,ISKIP,NXS,NYS,NZS,NDI,NCI,NPTS,NTI,TI,
44* 1NVS,NVB,XX,YY,Z,DELX,DELY,Q,UBARK,SIGAK,SIGXO,SIGYO,GAMMAP,
45* 2SIGZO,ALPHA,BETA,ZRK,TIMAV,THETAK,TAUK,TAUK,H,XRY,XRZ,XLRZ,
46* 3ZLZ,IZMOD,DECAY,TIM1,BLAMD,DI,CI,TAST,ZLIM,HB,PERCB,VB,
47* 4VS,PERC,ACCUR,ALPHL,BETL,TAUL,TAUL,ISW,XMAXJN,YMAXJN,RASTIN,
48* 5NPS,NAMCAS,SCL,XMAXIN,YMAXIN,ISW,XMAXJN,YMAXJN,RASTIN
49* 6,XSIZE,YSIZE,XCIZE,YCIZE,TEMPK,TEMPL,JSW
50* IF (IFF.GT. 1) GO TO 2
51* ZERO OUT INPUT LISTS FOR PROCESSORS WHERE CORE IS NOT
52* INITIALIZE TO ZERO, 608 IS THE LENGTH OF COMMON /PAKMT/, SUBTRACT
53* 12 FOR TESTNO AND 12 FOR NAMCAS
54* DO 1 I=1,584
55* 1 IZR1(I) = 0
56* 2 READ (5,NAM2)
57* DO 71 I=1,20
58* 71 GAMMA(I) = 1.0-GAMMAP(I)
59* NNZ = NZS-1
60* LAMBDA = BLANDA
61* NCC = NCI/10
62* NDU = NDI/10
63* NTT = NTI/10
64* NCCC = NCI-NCC*10
65* NDDD = NDI-NDU*10
66* NTTT = NTI-NTT*10
67* IF (ISW.LE. 0) ISW = 2
68* IF (RASTIN.LE. 0.0) RASTIN = 163.2
69* IF (XSIZE.LE. 0.0) XSIZE = 937.0
70* IF (YSIZE.LE. 0.0) YSIZE = 899.0
71* IF (XCIZE.LE. 0.0) XCIZE = 937.0
72* IF (YCIZE.LE. 0.0) YCIZE = 899.0
73* IF (NXS.GT. 0) GO TO 5
74* DEFAULT XX
75* NXS = 41
76* DO 4 I=1,NXS
77* 4 XX(I) = XSV(I)
78* 5 CONTINUE
79* IF (TAUOK.GT. 0.0) GO TO 6
80* DEFAULT TAUOK
81* TAUOK = 600.0

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82*	6 CONTINUE		RDR08200
83*	8 DO 16 I=1,NNZ		RDR08300
84*	DEFAULT SIGYO		RDR08400
85*	IF (SIGYO(I) .GT. 0.0) GO TO 9		RDR08500
86*	SIGYO(I) = SIGXO(I)		RDR08600
87*	9 CONTINUE		RDR08700
88*	IF (ALPHA(I) .GT. 0.0) GO TO 10		RDR08800
89*	DEFAULT ALPHA		RDR08900
90*	ALPHA(I) = 1.0		RDR09000
91*	10 IF (BETA(I) .GT. 0.0) GO TO 12		RDR09100
92*	DEFAULT BETA		RDR09200
93*	BETA(I) = 1.0		RDR09300
94*	12 CONTINUE		RDR09400
95*	IF (IZMOD(I) .GT. 0) GO TO 16		RDR09500
96*	DEFAULT IZMOD		RDR09600
97*	IZMOD(I) = 1		RDR09700
98*	16 CONTINUE		RDR09800
99*	IF (XRY .GT. 0.0) GO TO 18		RDR09900
100*	DEFAULT XRY		RDR10000
101*	XRY = 100.0		RDR10100
102*	18 IF (XRZ .GT. 0.0) GO TO 20		RDR10200
103*	DEFAULT XRZ		RDR10300
104*	XRZ = 100.0		RDR10400
105*	20 IF (TIMAV .GT. 0.0) GO TO 24		RDR10500
106*	DEFAULT TIMAV		RDR10600
107*	TIMAV = 600.0		RDR10700
108*	IF (ISKIP(5) .EQ. 2) TIMAV = 360.0		RDR10800
109*	24 IF (ZRK .GT. 0.0) GO TO 26		RDR10900
110*	DEFAULT ZRK		RDR11000
111*	ZRK = 2.0		RDR11100
112*	26 CONTINUE		RDR11200
113*	IF (ISKIP(6) .EQ. 0) ISKIP(6) = 2		RDR11300
114*	CHECK IZMOD		RDR11400
115*	KSW(2) = 0		RDR11500
116*	NBK = 0		RDR11600
117*	KSW(1) = 0		RDR11700
118*	DO 34 I=1,NNZ		RDR11800
119*	I1 = IZMOD(I)/100		RDR11900
120*	I2 = (IZMOD(I)-I1*100)/10		RDR12000
121*	I3 = IZMOD(I)-I1*100-I2*10		RDR12100
122*	IF (I .GT. 1) GO TO 27		RDR12200

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123* RDR12300
124* RDR12400
125* RDR12500
126* RDR12600
127* RDR12700
128* RDR12800
129* RDR12900
130* RDR13000
131* RDR13100
132* RDR13200
133* RDR13300
134* RDR13400
135* RDR13500
136* RDR13600
137* RDR13700
138* RDR13800
139* RDR13900
140* RDR14000
141* RDR14100
142* RDR14200
143* RDR14300
144* RDR14400
145* RDR14500
146* RDR14600
147* RDR14700
148* RDR14800
149* RDR14900
150* RDR15000
151* RDR15100
152* RDR15200
153* RDR15300
154* RDR15400
155* RDR15500
156* RDR15600
157* RDR15700
158* RDR15800
159* RDR15900
160* RDR16000
161* RDR16100
162* RDR16200
163* RDR16300

IF (I1 .NE. 6.AND.I2 .NE. 6.AND.I3 .NE. 6) GO TO 27
KSW(2) = 1
IF (ISKIP(1) .EQ. 0) ISKIP(1) = 1
GO TO 72
27 IF (I1.NE.5.AND.I2.NE.5.AND.I3.NE.5) GO TO 28
ZLIM = Z(I+1)
KSW(1) = 1
II = 1
28 IF (I2.EQ.9.OR.I1.EQ.9.OR.I3.EQ.9) GO TO 29
IF (I2.NE.4.AND.I1.NE.4.AND.I3.NE.4) GO TO 31
IF (NBK .GT. 0) GO TO 30
29 NBK = NBK+1
JBOT(NBK) = I
30 JTOP(NBK) = I
31 NTAL = 0
MODLS(I) = 1
32 NTAL = NTAL+1
IF (NTAL .GT. 3) GO TO 33
IF (I1 .EQ. NTAL.OR.I2 .EQ. NTAL.OR.I3 .EQ. NTAL) GO TO 33
GO TO 32
33 IF (NTAL .LT. 4) MODLS(I) = NTAL
34 CONTINUE
IF (KSW(1) .NE. 1) GO TO 72
IF (ISKIP(1) .EQ. 0) ISKIP(1) = 1
ISKIP(2) = 0
ISKIP(3) = 0
NPTS = II
DO 70 I=1,II
70 ZL(I) = Z(I)
GO TO 73
72 IF (NPTS .GT. 0) GO TO 73
NPTS = 1
ZL(1) = 0.0
73 CONTINUE
IF (LAMBDA .LE. 0.0) GO TO 74
IF (ZLIM .LE. 0.0) ZLIM = Z(NZS)
74 CONTINUE
DO 36 I=1,NZS
CHECK MINIMUM LIMITS
IF (SIGAK(I) .LT. .5) SIGAK(I) = .5
IF (SIGEK(I) .LT. .1) SIGEK(I) = .1

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IF (UBARK(I) .LT. .1) UBARK(I) = .1
36 CONTINUE
IF (NBK .EQ. 0) GO TO 57
IF (ISKIP(7) .GT. 0) GO TO 40
C DETERMINE LAYER CHANGE PARAMETERS
ZRL = ZRK
II = -1
DO 38 I=1,NBK
II = II+2
NTAL = JBOT(I)
NTAK = JTOP(I)
UBARL(II) = UBARK(NTAL)
UBARL(II+1) = UBARK(NTAK+1)
SIGAL(II) = SIGAK(NTAL)
SIGAL(II+1) = SIGAK(NTAK+1)
SIGEL(II) = SIGEK(NTAL)
SIGEL(II+1) = SIGEK(NTAK+1)
THETAL(II) = THETAK(NTAL)
THETAL(II+1) = THETAK(NTAK+1)
ALPHL(I) = ALPHA(NTAL)
BETL(I) = BETA(NTAL)
TEMPK(II) = TEMPK(NTAL)
TEMPK(II+1) = TEMPK(NTAK+1)
38 CONTINUE
TAUOL = TAUOK
TAUL = TAUK
GO TO 52
40 CONTINUE
IF (TAUOL .GT. 0.0) GO TO 42
C DEFAULT TAUOL
TAUOL = 600.0
42 IF (ZRL .GT. 0.0) GO TO 44
C DEFAULT ZRL
ZRL = ZRK
44 DO 48 I=1,NBK
IF (ALPHL(I) .GT. 0.0) GO TO 46
C DEFAULT ALPHL
ALPHL(I) = 1.0
46 IF (BETL(I) .GT. 0.0) GO TO 48
C DEFAULT BETL
BETL(I) = 1.0

RDR16400
RDR16500
RDR16600
RDR16700
RDR16800
RDR16900
RDR17000
RDR17100
RDR17200
RDR17300
RDR17400
RDR17500
RDR17600
RDR17700
RDR17800
RDR17900
RDR18000
RDR18100
RDR18200
RDR18300
RDR18400
RDR18500
RDR18600
RDR18700
RDR18800
RDR18900
RDR19000
RDR19100
RDR19200
RDR19300
RDR19400
RDR19500
RDR19600
RDR19700
RDR19800
RDR19900
RDR20000
RDR20100
RDR20200
RDR20300
RDR20400

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205* 48 CONTINUE
206*   NTAL = 2*NBK
207*   DO 50 I=1,NTAL
208*     CHECK MINIMUM VALUES
209*     IF (SIGAL(I) .LT. .5) SIGAL(I) = .5
210*     IF (UBARL(I) .LT. .1) UBARL(I) = .1
211*     IF (SIGEL(I) .LT. .1) SIGEL(I) = .1
212* 50 CONTINUE
213* 52 NTAK = NNZ+1
214*   NTAL = NNZ+NBK
215* C   COMBINE ALPHA AND BETA WITH ALPHL AND BETL
216*   DO 54 I=NTAK,NTAL
217*     INNZ = I-NNZ
218*     ALPHA(I) = ALPHL(INNZ)
219*     BETA(I) = BETL(INNZ)
220* 54 CONTINUE
221* 57 CONTINUE
222* 58 CONTINUE
223*   ST01 = (TAUK/TAUOK)**(0.2)*RAD
224*   S = (Z(2)/ZRK)
225*   S1 = 1.0/ALOG(S)
226*   P = RB8(UBARK(2),UBARK(1),S1)
227* C   CALCULATE UBAR FOR LAYER 1
228*   UBAR(1) = RB11(UBARK(1),P,Z(2),ZRK)
229*   PPWR = P
230*   IF (NNZ .LT. 2) GO TO 152
231*   DO 150 I=2,NNZ
232* C   CALCULATE UBAR FOR LAYERS 2 TO NNZ
233*   150 UBAR(I) = 0.5*(UBARK(I+1)+UBARK(I))
234*   152 P = RB8(SIGAK(2),SIGAK(1),S1)
235* C   CALCULATE SIGAP FOR LAYER 1
236*   SIGAP(1) = ST01*RB11(SIGAK(1),P,Z(2),ZRK)
237*   MPWR = P
238*   IF (NNZ .LT. 2) GO TO 162
239*   DO 160 I=2,NNZ
240* C   CALCULATE SIGAP FOR LAYERS 2 TO NNZ
241*   160 SIGAP(I) = 0.5*ST01*(SIGAK(I+1)+SIGAK(I))
242*   162 P = RB8(SIGEK(2),SIGEK(1),S1)
243* C   CALCULATE SIGEP FOR LAYER 1
244*   SIGEP(1) = RB11(SIGEK(1),P,Z(2),ZRK)*RAD
245*   IF (NNZ .LT. 2) GO TO 172

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246*      QPWR = P
247*      DO 170 I=2,NNZ
248*      CALCULATE SIGEP FOR LAYERS 2 TO NNZ
249*      170 SIGEP(I) = ((SIGEK(I+1)+SIGEK(I))*RAD)*0.5
250*      172 DO 180 I=1,NNZ
251*      J = I
252*      C CALCULATE THETA FOR ALL LAYERS
253*      THETA(I) = 0.5*(THETAK(J+1)+THETAK(J))
254*      IF (ABS(THETAK(J+1)-THETAK(J)) .GT. 180.0) THETA(I) = THETA(I)-
255*      1180.0
256*      C CALCULATE DELTHP FOR ALL LAYERS
257*      DELTHP(I) = THETAK(J+1)-THETAK(J)
258*      IF (DELTHP(I) .GT. 180.0) DELTHP(I) = 360.0-DELTHP(I)
259*      IF (DELTHP(I) .LT. -180.0) DELTHP(I) = 360.0+DELTHP(I)
260*      180 CONTINUE
261*      DO 185 I=1,NNZ
262*      C CALCULATE DELU FOR ALL LAYERS
263*      DELU(I) = UBARK(I+1)-UBARK(I)
264*      IF (DELU(I) .GE. 0.0) GO TO 185
265*      IF (TEMPK(I+1)-TEMPK(I) .GE. 0.0) GO TO 185
266*      DELU(I) = ABS(DELU(I))
267*      185 CONTINUE
268*      IF (KSW(2) .GT. 0) GO TO 250
269*      IF (NBK .EQ. 0) GO TO 250
270*      ST01 = (TAUL/TAUCL)**(0.2)*RAD
271*      M = JTOP(I)
272*      IF (JBOT(I) .GT. 1) GO TO 186
273*      S = (Z(M+1)/ZRL)
274*      S1 = 1.0/ALOG(S)
275*      186 IF (ISKIP(7) .GT. 0) GO TO 192
276*      DO 188 I=1,NBK
277*      NNZI = NNZ+I
278*      M1 = JBOT(I)
279*      M2 = JTOP(I)
280*      S = 0.0
281*      DO 187 J=M1,M2
282*      S = S+0.5*(UBARK(J)+UBARK(J+1))*(Z(J+1)-Z(J))
283*      UBAR(NNZI) = S/(Z(M2+1)-Z(M1))
284*      188 CONTINUE
285*      GO TO 202
286*      192 CONTINUE

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RDR24600
RDR24700
RDR24800
RDR24900
RDR25000
RDR25100
RDR25200
RDR25300
RDR25400
RDR25500
RDR25600
RDR25700
RDR25800
RDR25900
RDR26000
RDR26100
RDR26200
RDR26300
RDR26400
RDR26500
RDR26600
RDR26700
RDR26800
RDR26900
RDR27000
RDR27100
RDR27200
RDR27300
RDR27400
RDR27500
RDR27600
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RDR27800
RDR27900
RDR28000
RDR28100
RDR28200
RDR28300
RDR28400
RDR28500
RDR28600

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287* IF (JBOT(1) .GT. 1) GO TO 193
288* P = RB8(UBARL(2),UBARL(1),S1)
289* CALCULATE UBAR FOR NEW LAYER 1 (IF CONTAINS SURFACE)
290* UBAR(NNZ+1) = RB11(UBARL(1),P,Z(M+1),ZRL)
291* QPWR = P
292* GO TO 197
293* CALCULATE UBAR FOR NEW LAYER 1 (IF DOES'NT CONTAIN SURFACE)
294* UBAR(NNZ+1) = (UBARL(1)+UBARL(2))*0.5
295* IF (NBK .LT. 2) GO TO 202
296* DO 200 I=2,NBK
297* J = I*2-1
298* CALCULATE UBAR FOR NEW LAYERS 2 TO NBK
299* NNZI = NNZ+I
300* UBAR(NNZI) = (UBARL(J+1)+UBARL(J))*0.5
301* IF (JBOT(1) .GT. 1) GO TO 210
302* P = RB8(SIGEL(2),SIGEL(1),S1)
303* CALCULATE SIGEP FOR NEW LAYER 1 (IF CONTAINS SURFACE)
304* SIGEP(NNZ+1) = RB11(SIGEL(1),P,Z(M+1),ZRL)*RAD
305* GO TO 215
306* CALCULATE SIGEP FOR NEW LAYER 1 (IF DOES'NT CONTAIN SURFACE)
307* SIGEP(NNZ+1) = ((SIGEL(2)+SIGEL(1))*RAD)*0.5
308* IF (NBK .LT. 2) GO TO 222
309* DO 220 I=2,NBK
310* J = I*2-1
311* CALCULATE SIGEP FOR NEW LAYERS 2 TO NBK
312* NNZI = NNZ+I
313* SIGEP(NNZI) = ((SIGEL(J+1)+SIGEL(J))*RAD)*0.5
314* IF (ISKIP(7) .GT. 0) GO TO 226
315* DO 225 I=1,NBK
316* CALCULATE THETA FOR NEW LAYERS 1 TO NBK
317* NNZI = NNZ+I
318* M1 = JBOT(I)
319* M2 = JTOP(I)
320* S = 0.0
321* DO 224 J=M1,M2
322* P = 0.5*(THETAK(J+1)+THETAK(J))
323* IF (ABS(THETAK(J+1)-THETAK(J)) .LT. 180.0) GO TO 226
324* P = P+180.0
325* S = S+P*(Z(J+1)-Z(J))
326* THETA(NNZI) = S/(Z(M2+1)-Z(M1))
327* DELTHP(NNZI) = THETAK(M2+1)-THETAK(M1)

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RDR28700
RDR28800
RDR28900
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RDR29100
RDR29200
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RDR29400
RDR29500
RDR29600
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RDR29800
RDR29900
RDR30000
RDR30100
RDR30200
RDR30300
RDR30400
RDR30500
RDR30600
RDR30700
RDR30800
RDR30900
RDR31000
RDR31100
RDR31200
RDR31300
RDR31400
RDR31500
RDR31600
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RDR32000
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RDR32300
RDR32400
RDR32500
RDR32600
RDR32700

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328* IF (DELTHP(NNZI) .GT. 180.0) DELTHP(NNZI) = 360.0-DELTHP(NNZI)
329* IF (DELTHP(NNZI) .LT. -180.0) DELTHP(NNZI) = 360.0+DELTHP(NNZI)
330* 225 CONTINUE
331* GO TO 230
332* 226 DO 227 I=1,NBK
333* J = 2*I-1
334* NNZI = NNZ+I
335* THETA(NNZI) = 0.5*(THETA(J+1)+THETA(J))
336* IF (ABS(THETA(J+1)-THETA(J)).GT.180.) THETA(NNZI)=THETA(NNZI)+180.
337* DELTHP(NNZI) = THETA(J+1)-THETA(J)
338* IF (DELTHP(NNZI) .GT. 180.0) DELTHP(NNZI) = 360.0-DELTHP(NNZI)
339* IF (DELTHP(NNZI) .LT. -180.0) DELTHP(NNZI) = 360.0+DELTHP(NNZI)
340* 227 CONTINUE
341* 230 CONTINUE
342* DO 235 I=1,NBK
343* J = I*2-1
344* C CALCULATE DELTHP FOR ALL NEW LAYERS
345* NNZI = NNZ+I
346* DELU(NNZI) = UBARK(J+1)-UBARK(J)
347* IF (DELU(NNZI) .GE. 0.0) GO TO 235
348* IF (TEMPL(J+1)-TEMPL(J) .GT. 0.0) GO TO 235
349* DELU(NNZI) = ABS(DELU(NNZI))
350* 235 CONTINUE
351* 237 IF (JBOT(1) .GT. 1) GO TO 242
352* P = RB8(SIGAL(2),SIGAL(1),S1)
353* C CALCULATE SIGAP FOR NEW LAYER 1 (IF CONTAINS SURFACE)
354* SIGAP(NNZ+1) = ST01*RB11(SIGAL(1),P,Z(M+1),ZRL)
355* GO TO 243
356* C CALCULATE SIGAP FOR NEW LAYER 1 (IF DOES'NT CONTAIN SURFACE)
357* SIGAP(NNZ+1) = 0.5*ST01*(SIGAL(1)+SIGAL(2))
358* 242 CONTINUE
359* IF (NBK .LT. 2) GO TO 250
360* DO 245 I=2,NBK
361* J = I*2-1
362* C CALCULATE SIGAP FOR NEW LAYERS 2 TO NBK
363* NNZI = NNZ+I
364* 245 SIGAP(NNZI) = 0.5*ST01*(SIGAL(J+1)+SIGAL(J))
365* 250 CONTINUE
366* DO 395 I=2,NZS
367* IF (H .LE. Z(I)) GO TO 396
368* 395 CONTINUE
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RDR32900
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RDR33200
RDR33300
RDR33400
RDR33500
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RDR33700
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396 RNG = DELX(I-1)
    AZM = DELY(I-1)
CCC  GET DATE AND TIME (UNIVAC 1108 ONLY)
    CALL ERTRAN(9,NTFB(1),NTFB(2))
C    LOAD MM/DD/YY INTO IDATE(1) AND (2) ON FIRST LOOP
C    LOAD HR:MN:SC INTO ITIME(1) AND (2) ON SECOND LOOP
    N = '/'
    DO 400 I=1,2
        J = 2*I-1
        CALL MSFLD(0,12,NTFB(I),0,IDATE(J))
        CALL MSFLD(0,6,N,12,IDATE(J))
        CALL MSFLD(12,12,NTFB(I),18,IDATE(J))
        CALL MSFLD(0,6,N,30,IDATE(J))
        CALL MSFLD(24,12,NTFB(I),0,IDATE(J+1))
    400 CONTINUE
CCC  END DATE AND TIME - PRINT WITH A6,A2 FORMAT
    WRITE (6,1000) NAMCAS,(TESTNO(I),I=1,6),IDATE,ITIME,H,RNG,AZM
    IF (ISKIP(8) .EQ. 1) WRITE (6,NAM2)
    RETURN
1000 FORMAT (1H1,11(/),24X,21(4H***)) /24X,1H*,82X,1H*/24X,1H*,5X,12A6,5RDR38800
    1X,1H*/24X,1H*,82X,1H*/24X,1H*,23X,6A6,23X,1H*/24X,1H*,82X,1H*/
    1 24X,1H*,25X,7HDATE = ,A6,A2,9H, TIME = ,RDR39000
    2A6,A2,25X,1H*,3(/24X,1H*,82X,1H*)/24X,30H* ADJUSTED CLOUD RISE HEIRDR39100
    3GHT = ,F8,2,9H, RANGE = ,F9,2,19H, AZIMUTH BEARING = ,F7,2,2H */24X,1RDR39200
    4H*,82X,1H*/24X,21(4H***)) /1H1
    END
RDR36900
RDR37000
RDR37100
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1* SUBROUTINE SGP(ZH,N,SIG,IN,I2,UBHK,X,IBB)
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLRZ,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),SGP00700
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10)SGP00800
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)SGP00900
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),SGP01000
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,SGP01100
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,SGP01200
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,SGP01300
14* 4NCCC,NDD,NTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,SGP01400
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),SGP01500
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,SGP01600
17* 7NYSS,CDAMX(3)SGP01700
18* DIMENSION DTHK(21)SGP01800
19* EQUIVALENCE (DTHK,XAST)SGP01900
20* INTEGER TESTNOSGP02000
21* REAL MPWR,L,LAMBDASGP02100
22* SUBROUTINE SGP CALCULATES SIGENK AND SIGANK WITH OR WITHOUTSGP02200
23* DESTRUCT IN THE LAYER.SGP02300
24* GO TO (4,44,64,68),IBBSGP02400
25* S = 0.0SGP02500
26* MN = N-1SGP02600
27* HHNK = ZHSGP02700
28* HHKK = 1.0SGP02800
29* IF (N.EQ. 1) GO TO 5SGP02900
30* HPRK = ZHSGP03000
31* HHNK = Z(N+1)SGP03100
32* SG3 = SIGEK(1)SGP03200
33* IF (IN.EQ. 2) SG3 = SIGAP(1)SGP03300
34* IF (N.LE. 2) GO TO 30SGP03400
35* DO 25 M=2,MNSGP03500
36* IF (IN.EQ. 2) GO TO 10SGP03600
37* SG1 = SIGEK(M+1)SGP03700
38* SG2 = SIGEK(M)SGP03800
39* GO TO 20SGP03900
40* SG1 = SIGAP(M+1)SGP04000

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SG2 = SIGAP(M)
20 S = S+(SG1+SG2)*(Z(M+1)-Z(M))*0.5
25 CONTINUE
30 IF (IN.EQ. 2) GO TO 35
SG1 = SIGEK(N+1)
SG2 = SIGEK(N)
PWR = QPWR
GO TO 40
35 SG1 = SIGAP(N+1)
SG2 = SIGAP(N)
PWR = MPWR
40 IF (N.EQ. 1) GO TO 42
S = S+(ZH-Z(N))*((SG1-SG2)/(Z(N+1)-Z(N)))*(ZH-Z(N))+SG2)*0.5
42 SIG = (S+RB11(SG3,PWR,HHNK,ZRK))*RAD/HRK
RETURN
C
ENTRY UBARS(ZH,N,IZ,UBHK)
C
SUBROUTINE UBARS CALCULATES UBARNK, X NK, Y NK, CAP THETA (ANG)
44 XBARX = 0.0
YBARY(IZ) = 0.0
VV = VS(11)
PWR = PPWR
IF (JF.EQ. 2) VV = VB(11)
IF (N.EQ. 1) GO TO 50
MN = N-1
DO 45 M=1,MN
S = DTHK(M+1)-DTHK(M)
IF (S) 46,45,46
46 CONTINUE
S1 = SIN(DTHK(M+1))-SIN(DTHK(M))
S2 = COS(DTHK(M+1))-COS(DTHK(M))
S = UBARK(M)*(Z(M+1)-Z(M))/(VV*S)
XBARX = XBARX+(S1*S)
YBARY(IZ) = YBARY(IZ)+(S2*(-S))
45 CONTINUE
50 TMPQ1 = 1.0/(Z(N+1)-Z(N))
S = ((DTHK(N+1)-DTHK(N))*TMPQ1)*(ZH-Z(N))+DTHK(N)
S1 = SIN(S)-SIN(DTHK(N))
S2 = COS(S)-COS(DTHK(N))
IF (N.EQ. 1) GO TO 52
UBHK = ((UBARK(N+1)-UBARK(N))*TMPQ1)*0.5*(ZH-Z(N))+(0.5*UBARK(N))
GO TO 54

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SGP04300
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SGP06500
SGP06600
SGP06700
SGP06800
SGP06900
SGP07000
SGP07100
SGP07200
SGP07300
SGP07400
SGP07500
SGP07600
SGP07700
SGP07800
SGP07900
SGP08000
SGP08100


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82*      52 UBHK = RB11(UBARK(1),PWR,ZH,ZRK)
83*      54 CONTINUE
84*      S = DTHK(N+1)-DTHK(N)
85*      IF (S) 55,55,53
86*      53 S = UBHK/(VV*S*TMPQ1)
87*      XBARX = XBARX+(S1*S)
88*      YBARY(IZ) = YBARY(IZ)+(S2*(-S))
89*      55 CONTINUE
90*      IF (XBARX) 57,56,57
91*      56 IF (YBARY(IZ)) 57,58,57
92*      57 ANG(IZ) = ATAN2(YBARY(IZ),XBARX)
93*      GO TO 60
94*      58 ANG(IZ) = 0.0
95*      59 UBARNK(IZ) = UBHK
96*      SQBAR = 0.0
97*      GO TO 62
98*      60 IF (XBARX) 61,59,61
99*      61 SQBAR = SQRT(XBARX*XBARX+YBARY(IZ)*YBARY(IZ))
100*      UBARNK(IZ) = SQBAR*VV/ZH
101*      62 CONTINUE
102*      RETURN
103*      ENTRY DEPSO(X,N,IZ)
104*      SUBROUTINE DEPSO CALCULATES ALL OF THE DEPOSITION EQUATION EXCEPT
105*      THE LATERAL TERM
106*      64 ZF = ZZL(IZ)
107*      VV = VS(IZ)
108*      GAMMB = GAMMA(IZ)
109*      XXX = X
110*      PERK = PERC(IZ)
111*      IF (JF.EQ. 1) GO TO 165
112*      ZF = HB
113*      VV = VB(IZ)
114*      XXX = X+(SIGZO(N)/SIGENK(IZ))*(1.0/BETANK(IZ))
115*      PERK = PERCB(IZ)
116*      165 XKNK = 0.0
117*      IF (GAMMB.GE. 1.0) GO TO 69
118*      S1 = 1.0/(SIGENK(IZ)*XXX**BETANK(IZ))
119*      S2 = VV*XXX/UBARNK(IZ)
120*      S3 = -0.5*S1*S1
121*      S4 = BETANK(IZ)*(S2-ZF)-S2
122*      S2 = S2-ZF

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      B = 1.0
      XKNK = -S4*EXP(S2*S2*S3)
      A = 0.0
      65 A = A+2.0
      S5 = A*Z(N+1)
      S6 = S5-S2
      S7 = S5+S2
      S7 = S7*S7*S3
      S6 = S6*S6*S3
      IF (A .LE. 2.0) GO TO 66
      IF (S6 .LT. -10.0.AND.S7 .LT. -10.0) GO TO 67
      66 S5 = S5*BETANK(IZ)
      XKNK = XKNK+B*((S5+S4)*EXP(S7)+GAMMB*(S5-S4)*EXP(S6))
      IF (GAMMB .LE. 0.0) GO TO 67
      B = B*GAMMB
      GO TO 65
      67 CONTINUE
      XY = (SIGYO(N)/SIGANK(IZ))*((1.0/ALPHNK(IZ))
      SIGYNK = SORT((SIGANK(IZ)*(X+XY)**ALPHNK(IZ))**2+(SIGENK(IZ)*XXX**
      1BETANK(IZ)*YBARY(IZ)/ZF)**2)
      IF (SIGYNK .LE. 0.0) GO TO 69
      DEP = Q(N)*PERK*((1.0-GAMMB)*S1*XKNK/(6.2831853*SIGYNK*FLOAT(NXCI))*
      1XXX)
      69 CONTINUE
      RETURN
      ENTRY BETAK(ZH,N,IZ)
      SUBROUTINE BETAK CALCULATES BETA NK AND ALPHA NK
      68 S1 = 0.0
      S2 = 0.0
      IF (N .EQ. 1) GO TO 90
      MN = N-1
      DO 70 M=1,MN
      S1 = S1+BETA(M)*(Z(M+1)-Z(M))
      S2 = S2+ALPHA(M)*(Z(M+1)-Z(M))
      70 CONTINUE
      TMPQ1 = 1.0/ZH
      TMPQ2 = ZH-Z(N)
      BETANK(IZ) = (S1+BETA(N)*TMPQ2)*TMPQ1
      ALPHNK(IZ) = (S2+ALPHA(N)*TMPQ2)*TMPQ1
      GO TO 95
      90 BETANK(IZ) = BETA(N)

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ALPHNK(IZ) = ALPHA(N)
95 CONTINUE
RETURN
END

SGP16400
SGP16500
SGP16600
SGP16700

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1* SUBROUTINE SIGMA(XP,M,MM)
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXQ(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLR,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),SGA00700
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10),SGA00800
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),SGA01000
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,I1,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDDU,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18* INTEGER TESTNO
19* REAL MPWR,L,LAMBDA
20* C ***** THIS SUBROUTINE CALCULATES THE STANDARD DEVIATIONS OF X,Y,ZSGA02000
21* X = XP
22* IF (MM.EQ. 2) X = XAST(M)
23* MM = 1
24* SIGZ = 0.0
25* SIGY = 0.0
26* SIGX = 0.0
27* N = MODLS(M)
28* GO TO (40,20,30),N
29* 20 SIGY = SIGYO(M)
30* SIGX = SIGXO(M)
31* GO TO 220
32* 30 B3 = SIGEP(M)
33* B4 = BETA(M)
34* 40 A1 = 1.0
35* A2 = SIGYO(M)
36* A3 = SIGAP(M)
37* A4 = ALPHA(M)
38* A5 = DELTHP(M)
39* A6 = SIGXO(M)
40* L = 0.0
SGA00100
SGA00200
SGA00300
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SGA00500
SGA00600
SGA00700
SGA00800
SGA00900
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IF (DELU(M) .LE. 0.0) GO TO 45
L = 0.28*X*DELU(M)/UBAR(M)
45 IF (MM .EQ. 1) GO TO 60
N = 1
GO TO 60
50 T1 = (THETA(M)-THETA(JF))*RAD
A1 = 1.0
T2 = SIN(T1)
T1 = COS(T1)
A2 = Sqrt((SIGX*T2)**2+(SIGY*T1)**2)
A3 = SIGAP(JF)
A4 = ALPHA(JF)
A5 = DELTHP(JF)
A6 = Sqrt((SIGX*T1)**2+(SIGY*T2)**2)
B3 = SIGEP(JF)
B4 = BETA(JF)
L = 0.0
IF (DELU(JF) .LE. 0.0) GO TO 60
L = 0.28*X*DELU(JF)/UBAR(JF)
60 IF (A4-1.0) 70,80,70
70 A1 = 1.0/A4
IF (MMM .EQ. 2) GO TO 90
IF (A2-A3*XRY) 80,80,90
80 XY = A2/A3
GO TO 91
90 XY = A4*XRY*(A2/(A3*XRY))*A1+XRY*(1.0-A4)
91 IF (MMM .EQ. 1) XY = XY-XLRY
IF (XY .LT. 0.0) XY = 0.0
IF (A4-1.0) 110,100,110
100 T1 = A3*(X+XY)
GO TO 120
110 T1 = (X+XY-XRY*(1.0-A4))/(XRY*A4)
IF (T1 .LE. 0.0) GO TO 125
T1 = A3*XRY*T1**A4
120 T2 = ABS(A5)*X**4.0589052E-3
SIGY = Sqrt(T1*T1+T2*T2)
125 SIGX = Sqrt(L*L*.05408329+A6*A6)
IF (N .EQ. 1) GO TO 220
GO TO (150,130),MMM
130 IF (B4-1.0) 140,131,140
131 X2 = X

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SGA04100
SGA04200
SGA04300
SGA04400
SGA04500
SGA04600
SGA04700
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GO TO 190
140 T1 = X/XRZ
GO TO 210
150 IF (B4-1.0) 151,160,151
151 B1 = 1.0/B4
IF (SIGZO(M)-B3*XRZ) 160,160,170
160 XZ = SIGZO(M)/B3-XLRZ
GO TO 180
170 XZ = B4*XRZ*(SIGZO(M)/(B3*XRZ))*B1-XLRZ+XRZ*(1.0-B4)
180 IF (XZ .LT. 0.0) XZ = 0.0
XZ = X+XZ
IF (B4-1.0) 200,190,200
190 SIGZ = B3*XZ
GO TO 220
200 T1 = (XZ-XRZ*(1.0-B4))/(B4*XRZ)
IF (T1 .LE. 0.0) GO TO 220
210 SIGZ = B3*XRZ*T1**B4
220 CONTINUE
IF (MM .NE. 2) GO TO 240
IF (MMM .EQ. 2) GO TO 230
N = 2
X = XP
MMM = 2
GO TO 50
230 SIGXNK = SIGX
SIGYNK = SIGY
240 RETURN
END

```

SGA08200
 SGA08300
 SGA08400
 SGA08500
 SGA08600
 SGA08700
 SGA08800
 SGA08900
 SGA09000
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 SGA09400
 SGA09500
 SGA09600
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 SGA10000
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 SGA10200
 SGA10300
 SGA10400
 SGA10500
 SGA10600
 SGA10700
 SGA10800
 SGA10900

```

1* SUBROUTINE TESTR(KTK)
2* COMMON /PARAMT/ TESTNO(12),
3* INBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLR,Y,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10)
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,I1,DEP,XBARX,SOBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDD,NTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7WYSS,CDAMX(3)
18* INTEGER TESTNO
19* REAL MPWR,L,LAMBDA
20* THIS SUBROUTINE DETERMINES THE STRUCTURAL CHANGE IN LAYERS FOR
21* THE PULL TRANSITION MODEL
22* IF (NBK .EQ. 0) GO TO 100
23* IF (KTK .EQ. 0) GO TO 50
24* IF (KK .GE. JBOT(ILK)) GO TO 50
25* IBOT = JBOT(ILK)
26* ITOP = JTOP(ILK)
27* GO TO 61
28* 50 IF (KK .NE. JBOT(ILK)) GO TO 61
29* IBOT = KK
30* ITOP = JTOP(ILK)
31* DO 60 J=IBOT,ITOP
32* 60 XAST(J) = UBAR(J)*TAST(ILK)
33* ILK = ILK+1
34* 61 CONTINUE
35* KTK = 0
36* 100 CONTINUE
37* RETURN
38* END

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```

1* SUBROUTINE ISO(MR,MT)
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLRZ,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10),
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,ST01,
12* 2ST02,ST03,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,I1,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDD,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18* DIMENSION ERFX(6)
19* EQUIVALENCE (ANG(10),ERFX)
20* INTEGER TESTNO
21* REAL MPWR,L,LAMBDA
22* DOUBLE PRECISION A6,A7,A8,A9,A10,A11,DTX
23* THIS SUBROUTINE EVALUATES ERF(X)
24* DATA A6,A7,A8,A9,A10,A11/.0705230784D0,.0422820123D0,.0092705272D0,
25* 1,.0001520143D0,.0002765672D0,.0000430638D0/
26* DO 10 M=MK,MT
27* IN = 0
28* IF (ERFX(M) .LT. 0.0) IN = 1
29* ERFX(M) = ABS(ERFX(M))
30* IF (ERFX(M) .LT. 1.0E-10) GO TO 5
31* IF (ERFX(M) .GT. 5.0) GO TO 6
32* DTX = 1.0D0+ERFX(M)*(A6+ERFX(M)*(A7+ERFX(M)*(A8+ERFX(M)*(A9+ERFX(M)
33* 1)*(A10+ERFX(M)*A11))))
34* ERFX(M) = 1.0D0-(1.0D0/DTX)**16
35* GO TO 7
36* 5 ERFX(M) = 0.0
37* GO TO 10
38* 6 ERFX(M) = 1.0
39* 7 IF (IN .EQ. 1) ERFX(M) = -ERFX(M)
40* 10 CONTINUE

```

C

ISO04100
ISO04200

RETURN
END

41*
42*

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1* SUBROUTINE COORD(N,M,X,Y,XO,YO,ASP,XS,ICK) CRD00100
2* COMMON /PARAMT/ TESTNO(12), ISKIP(15),NXS,NYS,NZS,NDI,NCI, CRD00200
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15), CRD00300
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15), CRD00400
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ, CRD00500
6* 4XLR,Y,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10), CRD00600
7* 5TAST(05),JDOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VR(20),PERCB(20), CRD00700
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,URARL(10),SIGAL(10),SIGEL(10) CRD00800
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12) CRD00900
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20), CRD01000
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI, CRD01100
12* 2STO2,STO3,TRU,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR, CRD01200
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT, CRD01300
14* 4NCCC,NDD,NTTT,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD, CRD01400
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42), CRD01500
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT, CRD01600
17* 7NYSS,CDAMX(3) CRD01700
18* INTEGER TESTNO CRD01800
19* REAL MPWR,L,LAMBDA CRD01900
20* C *****THIS SUBROUTINE TRANSLATES AND ROTATES THE FIXED INPUT *****CRD02000
21* C ***** COORDINATES RELATIVE TO A SYSTEM WITH POSITIVE X AXIS *****CRD02100
22* C ***** ALONG THE WIND DIRECTION THETA. CRD02200
23* N = 0 CRD02300
24* B = AMOD(YO,360.0)*RAD CRD02400
25* IF (ICK.EQ. 2) GO TO 10 CRD02500
26* A = THETA(M) CRD02600
27* GO TO 11 CRD02700
28* 10 A = THETA(JF) CRD02800
29* 11 XP = XO*SIN(B) CRD02900
30* YP = XO*COS(B) CRD03000
31* A = A/RAD CRD03100
32* B = COS(A) CRD03200
33* A = SIN(A) CRD03300
34* DX = 0.0 CRD03400
35* DY = 0.0 CRD03500
36* DY = DELY(M)*RAD CRD03600
37* DX = DELX(M)*SIN(DY) CRD03700
38* DY = DELX(M)*COS(DY) CRD03800
39* 20 IF (ICK.EQ. 2) GO TO 50 CRD03900
40* 21 X1 = XP-DX CRD04000

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Y1 = YP-DY
X = -X1*A-Y1*B
Y = X1*B-Y1*A
IF (X .LE. 0.0) GO TO 80
IF (KSW(2) .EQ. 0) GO TO 40
XS = SQRT(X1*X1+Y1*Y1)
ASP = 0.0
IF (X1) 31,30,31
IF (Y1) 31,90,31
30 ASP = 1.5707963-ATAN2(Y1,X1)
31 IF (ASP .LT. 0.0) ASP = ASP+6.2831853
GO TO 90
40 IF (NRK .EQ. 0) GO TO 90
IF (ICK .EQ. 2) GO TO 90
IF (KK .LT. IBOT.OR.KK .GT. ITOP) GO TO 90
IF (XAST(M) .LE. 0.0) GO TO 80
ASP = THETA(JF)*RAD
50 XS = (THETA(M)+180.0)*RAD
DX = DX+XAST(M)*SIN(XS)
DY = DY+XAST(M)*COS(XS)
IF (ICK .EQ. 2) GO TO 21
X1 = XP-DX
Y1 = YP-DY
XS = -X1*SIN(ASP)-Y1*COS(ASP)
A = ABS(THETA(M)-THETA(JF))
IF (A .GE. 180.0) A = 360.0-A
IF (A .GT. 45.0) GO TO 60
IF (XS .LE. 0.0) GO TO 90
GO TO 80
60 CALL SIGMA(XAST(M),M,3)
ASP = A*RAD
SIGY = 2.15*SQRT((SIGX*SIN(ASP))**2+(SIGY*COS(ASP))**2)
IF (A .GT. 90.0) GO TO 70
IF (X .GT. XAST(M)+SIGY) GO TO 80
IF (XS .LE. 0.0) GO TO 90
IF (X .LT. XAST(M)) GO TO 90
GO TO 80
70 IF (X .LE. XAST(M)+SIGY) GO TO 90
80 N = 9
90 RETURN
END

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CRD04100
CRD04200
CRD04300
CRD04400
CRD04500
CRD04600
CRD04700
CRD04800
CRD04900
CRD05000
CRD05100
CRD05200
CRD05300
CRD05400
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SUBROUTINE PRITTL(NWD,LINES,LINE,A,B)
THIS SUBROUTINE PRINTS THE PAGE HEADING
DIMENSION LINE(1)
DATA IB/1H1/,JB/1H /
N = JB
LINES = 3
N1= 1
N2 = 15
10 IF (N2 .GT. NWD) N2 = NWD
WRITE (6,50) N,(LINE(I),I=N1,N2)
LINES = LINES+1
N = JB
IF (N2 .GE. NWD) GO TO 20
N1 = N2+1
N2 = N2+15
GO TO 10
20 IF (A .GE. 0.0) WRITE (6,80)
LINES = LINES+1
IF (A .LE. 0.0) GO TO 30
WRITE (6,60)
LINES = LINES+1
30 IF (B .LE. 0.0) GO TO 40
WRITE (6,70)
LINES = LINES+1
40 RETURN
50 FORMAT (A1,19X,15A6)
60 FORMAT (42X,45H(DECAY HAS BEEN INCLUDED IN THE CALCULATIONS))
70 FORMAT (35X,64H(PRECIPITATION SCAVENGING HAS BEEN INCLUDED IN THE
1CALCULATIONS))
80 FORMAT ( )
END

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PRT00100
PRT00200
PRT00300
PRT00400
PRT00500
PRT00600
PRT00700
PRT00800
PRT00900
PRT01000
PRT01100
PRT01200
PRT01300
PRT01400
PRT01500
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PRT02300
PRT02400
PRT02500
PRT02600
PRT02700
PRT02800
PRT02900
PRT03000
PRT03100

1*	SUBROUTINE PACKS(LINE,NWD)	PCK00100
2*	THIS SUBROUTINE REMOVES EXCESSIVE BLANKS FROM THE TITLE AND PACKS	PCK00200
3*	IT INTO SUCCESSIVE LINES OF 15 WORDS PER LINE	PCK00300
4*	DIMENSION LINE(1)	PCK00400
5*	DATA IBLK/00000000000005/	PCK00500
6*	DATA IBLK/00000000000000/ - IBM 7044 -	PCK00600
7*	NFLG = 0	PCK00700
8*	NR = 0	PCK00800
9*	JB = 16	PCK00900
10*	JB IS 15+1 OR 15 WORDS PER LINE	PCK01000
11*	IB = 1	PCK01100
12*	LST = IBLK	PCK01200
13*	M = 1	PCK01300
14*	N = 0	PCK01400
15*	J = 0	PCK01500
16*	10 J = J+1	PCK01600
17*	IF (J.LE. NWD) GO TO 15	PCK01700
18*	NFLG = 1	PCK01800
19*	L = IBLK	PCK01900
20*	N1 = N+1	PCK02000
21*	IF (N1.GT. 6) GO TO 80	PCK02100
22*	M1 = M	PCK02200
23*	GO TO 60	PCK02300
24*	15 K = LINE(J)	PCK02400
25*	I = 0	PCK02500
26*	20 I = I+1	PCK02600
27*	IF (I.GT. 6) GO TO 10	PCK02700
28*	II = IABS(6*(I-1))	PCK02800
29*	CALL MSFLD(II,6,K,30,L)	PCK02900
30*	IF (L.NE. IBLK) GO TO 30	PCK03000
31*	IF (LST.EQ. IBLK) GO TO 20	PCK03100
32*	25 IIBLK = I	PCK03200
33*	JIBLK = J	PCK03300
34*	30 N = N+1	PCK03400
35*	NR = NR+1	PCK03500
36*	IF (N.LT. 7) GO TO 40	PCK03600
37*	N = 1	PCK03700
38*	IB = IB+1	PCK03800
39*	M = M+1	PCK03900
40*	40 II = IABS(6*(N-1))	PCK04000

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IF (L .NE. IBLK) GO TO 50
NNBLK = N
MMBLK = M
50 IF (IB .LT. JB) GO TO 70
IF (LST .EQ. IBLK.OR.L .EQ. IBLK) GO TO 69
L = IBLK
NR = NR-1
N1 = NNBLK+1
M1 = MMBLK
IF (N1 .LT. 7) GO TO 60
M1 = M1+1
N1 = 1
60 II = IABS(6*(N1-1))
CALL MSFLD(30,6,L,II,LINE(M1))
IF (NFLG .EQ. 1) GO TO 63
GO TO 64
63 NR = NR+1
64 CONTINUE
N1 = N1+1
IF (N1 .LT. 7) GO TO 60
IF (NFLG .EQ. 1) GO TO 80
N1 = 1
M1 = M1+1
IF (M1 .LT. M) GO TO 60
J = JBLK
I = IBLK
N = 7
M = M-1
LST = IBLK
IB = 0
K = LINE(J)
GO TO 20
69 IB = 1
LST = IBLK
IF (L .NE. IBLK) GO TO 70
NR = NR-1
IB = 0
N = 7
M = M-1
GO TO 20
70 LST = L

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PCK04100
PCK04200
PCK04300
PCK04400
PCK04500
PCK04600
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PCK04800
PCK04900
PCK05000
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PCK08000
PCK08100

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75 CALL MSFLD(30,6,L,II,LINE(M))
GO TO 20
80 NWD = NR/6
IB = NWD*6
IF (IB .LT. NR) NWD = NWD+1
RETURN
END

PCK08200
PCK08300
PCK08400
PCK08500
PCK08600
PCK08700
PCK08800

HED00100
HED00200
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HED00500
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HED00700
HED00800
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HED01900
HED02000
HED02100
HED02200
HED02300

C
SUBROUTINE HEDING(KSW,KS,JSW,LSW)
SET ALL PARAMETERS NEC TO BUILDING PAGE HEADING
DIMENSION KSW(1)
KSW(3) = 3*JSW-3
GO TO (10,40,50,60,70),KS
10 IF (KSW(1).EQ. 0) GO TO 20
KSW(4) = 18
GO TO 80
20 IF (KSW(2).EQ. 0) GO TO 30
KSW(4) = 30
GO TO 80
30 KSW(4) = 0
IF (LSW.EQ. 1) KSW(4) = 24
GO TO 80
40 KSW(4) = 6
GO TO 80
50 KSW(4) = 12
GO TO 80
60 KSW(4) = 36
GO TO 80
70 KSW(4) = 42
80 RETURN
END

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BND04100
BND04200
BND04300
BND04400
BND04500

C UPPER RIGHT HAND CORNER ASSUME CROSSES XRIT
100 XPL = XRIT
110 GO TO 80
110 RETURN
END

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1*      SUBROUTINE VRTICLE(KS,JM,КСW,ISTOR,ISKIP5,NCV)
2*      DIMENSION KSW(1),ISTOR(1)
3*      COMMON /LBLBL/ J1(9),J2(4),J3(48),J5(6),J7(3),J8(16),J9(13),J10,
4*      1J4(12),J11(2),UNIT(15)
5*      INTEGER UNIT
6*      IF (JM .GT. 1) GO TO 20
7*      I1 = 31
8*      IF (КСW(2) .GT. 0) GO TO 10
9*      I1 = 19
10*     I2 = I1+4
11*     I3 = 8
12*     GO TO 82
13*     20 GO TO (30,40,50,60,70),KS
14*     30 I1 = 1
15*     I2 = 3
16*     GO TO 80
17*     40 I1 = 7
18*     I2 = 8
19*     I3 = 3
20*     GO TO 81
21*     50 I1 = 13
22*     I2 = 17
23*     GO TO 80
24*     60 I1 = 37
25*     I2 = 40
26*     I3 = 6
27*     GO TO 82
28*     70 I1 = 43
29*     I2 = 47
30*     I3 = 0
31*     81 IF (ISKIP5 .EQ. 4) I3 = I3+9
32*     82 ISTOR(1) = J3(4)
33*     DO 90 I=I1,I2
34*     90 ISTOR(I-I1+2) = J3(I)
35*     NCV = I2-I1+2
36*     IF (JM .GT. 1) GO TO 110
37*     DO 100 I=1,2
38*     100 ISTOR(NCV+I) = J4(I3+I)
39*     NCV = NCV+2
40*     GO TO 130

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VRT00100
VRT00200
VRT00300
VRT00400
VRT00500
VRT00600
VRT00700
VRT00800
VRT00900
VRT01000
VRT01100
VRT01200
VRT01300
VRT01400
VRT01500
VRT01600
VRT01700
VRT01800
VRT01900
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VRT03900
VRT04000

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110 DO 120 I=1,3
120 ISTOR(NCV+I) = UNIT(I3+I)
NCV = NCV+3
130 CALL PACKS(ISTOR,NCV)
RETURN
END

VRT04100
VRT04200
VRT04300
VRT04400
VRT04500
VRT04600

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1* SUBROUTINE LLPLOT(YAR,XAR,N,TITLE,CRIT,NCRIT,NWD,VERTCL,NCV) LLP00100
2* COMMON /XYXYPT/ YP(41),XP(41),A(41),B(41),C(41),D(41),XI(41),YI(41) LLP00200
3* 1),NUM(3),NC LLP00300
4* COMMON /ILPLTS/ XMAX,XMIN,YMAX,YMIN,XLM1,YBM1,HT,CHARF,SCLX,SCLY, LLP00400
5* 1XSIZE1,YSIZE1 LLP00500
6* DIMENSION XAR(1),YAR(1),LINE(120),TITLE(1),CRIT(1),XF(1),YF(1), LLP00600
7* 1FLOX(5),LEXP(3),VERTCL(1) LLP00700
8* COMMON /ILALPH/ LCRT(10),IBLANK,ISTAR,IP1,IP2,IP3,HLABEL(5),NCH LLP00800
9* COMMON /PLTLO/ ISW,XMAXJN,YMAXJN,XCIZE,YCIZE LLP00900
10* DATA HLABEL/27HALONGWIND DISTANCE (METERS)/,NCH/27/ LLP01000
11* DATA IBLANK/1H /,LCRT/2HA=,2HB=,2HC=,2HD=,2HE=,2HF=,2HG=,2HH=,2H LLP01100
12* 1=,2HJ=,ISTAR/1H*/ LLP01200
13* EQUIVALENCE (FLOX,HLABEL),(LINE,B),(XF,X1),(YF,YI) LLP01300
14* EQUIVALENCE (XLM1,NND),(YBM1,NST),(HT,IPWRX),(CHARF,IPWRY),(SCLX,FLP LLP01400
15* 11),(SCLY,I1),(XSIZE1,IEXP),(YSIZE1,PWRY) LLP01500
16* DATA LEXP/2H5=,2H2=,2H / LLP01600
17* 2010 FORMAT (7X,1H(,20(6H-----),1H) LLP01700
18* 2020 FORMAT (4X,3H10=,1H(,3(11(1H-),1H1,15(1H-),1H1,11(1H-),1H1)/8X LLP01800
19* 1,3(11,39X),11/6X,3(2H10,11X,1H2,15X,1H5,10X),2H10/54X,5A6) LLP01900
20* 2030 FORMAT (1X,A1,3X,A2,1H(,120A1,1H)) LLP02000
21* 2040 FORMAT (1X,A1,3X,A2,1H(,120A1,1H)) LLP02100
22* 2050 FORMAT (1X,A1,2X,4H10-(,120A1,1H)) LLP02200
23* 2080 FORMAT (2(5(10X,A1,1H=E10.3,1H)/)) LLP02300
24* 2090 FORMAT (80H0 ** NO PLOTS THIS CASE -- DOSAGE OR CONCENTRATION VAL LLP02400
25* 1ES ARE PROBABLY OUT OF RANGE **//) LLP02500
26* FIND XMIN, XMAX, YMIN, YMAX LLP02600
27* YMAX = -1.E20 LLP02700
28* NND = N+1 LLP02800
29* NST = 0 LLP02900
30* 5 NST = NST+1 LLP03000
31* IF (NST .GE. N) GO TO 500 LLP03100
32* IF (XAR(NST) .LE. 0.0 .OR. YAR(NST) .LE. 0.0) GO TO 5 LLP03200
33* 10 NND = NND-1 LLP03300
34* IF (NND .LE. 1) GO TO 500 LLP03400
35* IF (XAR(NND) .LE. 0.0 .OR. YAR(NND) .LE. 0.0) GO TO 10 LLP03500
36* IF (NND .LE. NST) GO TO 500 LLP03600
37* DO 15 I=NST,NND LLP03700
38* IF (XAR(I) .LE. 0.0) GO TO 500 LLP03800
39* XF(I) = XAR(I) LLP03900
40* IF (ISW .NE. 2) XF(I) = ALOG10(XAR(I)) LLP04000

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41* IF (YAR(I) .LE. 0.0) GO TO 14
42* YF(I) = ALOG10(YAR(I))
43* IF (YMAX .LT. YF(I)) YMAX = YF(I)
44* GO TO 15
45* YF(I) = -1.E20
46* 14 CONTINUE
47* IPWRX = INT(XF(NST)+100.0)-100
48* IPWRY = INT(YMAX+100.0)-102
49* IF (IPWRX .LT. 2) IPWRX = 2
50* PWRY = FLOAT(IPWRY)
51* PRINT TITLE INFORMATION
52* CALL PRITTL(NWD,LINES,TITLE,-1.0,0.0)
53* WRITE (6,2010)
54* 15 LOOP FOR 48 PRINTER LINES
55* JK1 = 0
56* JK2 = 1
57* JT = 0
58* IST = (48-NCV)/2
59* DO 220 I=1,48
60* IAI = LEXP(3)
61* IF (I .LT. IST) GO TO 17
62* JT = JT+1
63* IF (JT .GT. NCV) GO TO 17
64* JK1 = JK1+1
65* IF (JK1 .LT. 7) GO TO 16
66* JK1 = 1
67* JK2 = JK2+1
68* 16 CALL MSFLU(IABS(6*(JK1-1)),6,VERICL(JK2),0,IAI)
69* 17 CONTINUE
70* II=48-1
71* FII = FLOAT(II)*0.0625
72* IF (NCRII) 60,60,20
73* 20 IF (NCRII .EQ. 9) GO TO 60
74* DO 40 KK=1,NCRII
75* IF (CRIT(KK) .LE. 0.0) GO TO 40
76* IF (ABS(ALOG10(CRIT(KK))-FII-PWRY) .GT. 0.031255) GO TO 40
77* DO 30 LL=1,120
78* 30 LINE(LL) = LCRIT(KK)
79* GO TO 67
80* 40 CONTINUE
81* 60 DO 65 J=1,120

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82*	65	LINE(J) = IBLANK	LLP08200
83*	67	DO 70 J=NST,NND	LLP08300
84*		IF (ABS(YF(J)-F11-PWRY) .GT. 0.031255) GO TO 70	LLP08400
85*		L = INT(XF(J)*40.0+0.5)-40*IPWRX	LLP08500
86*		IF (L .LT. 1 .OR. L .GT. 120) GO TO 70	LLP08600
87*		LINE(L)=ISTAR	LLP08700
88*	70	CONTINUE	LLP08800
89*		IF (15-I) 90,80,90	LLP08900
90*	80	IEXP=IPWRY+2	LLP09000
91*		GO TO 130	LLP09100
92*	90	IF (31-I) 110,100,110	LLP09200
93*	100	IEXP=IPWRY+1	LLP09300
94*		GO TO 130	LLP09400
95*	110	IF (47-I) 140,120,140	LLP09500
96*	120	IEXP=IPWRY	LLP09600
97*	130	IF (I .GT. 1) WRITE (6,2040) IAI,IEXP,LINE	LLP09700
98*		GO TO 220	LLP09800
99*	140	IF (16-I) 150,170,150	LLP09900
100*	150	IF (32-I) 160,170,160	LLP10000
101*	160	IF (48-I) 175,170,175	LLP10100
102*	170	IF (I .EQ. 1) GO TO 220	LLP10200
103*		IF (I .EQ. 48) GO TO 171	LLP10300
104*		WRITE (6,2050) IAI,LINE	LLP10400
105*		GO TO 220	LLP10500
106*	171	IP1 = IPWRX+1	LLP10600
107*		IP2 = IPWRX+2	LLP10700
108*		IP3 = IPWRX+3	LLP10800
109*		WRITE (6,2020) IPWRX,IP1,IP2,IP3,FLDX	LLP10900
110*		GO TO 220	LLP11000
111*	175	IEXP = 3	LLP11100
112*		IF (MOD(I,16) .NE. 5) GO TO 180	LLP11200
113*		IEXP = 1	LLP11300
114*		GO TO 190	LLP11400
115*	180	IF (MOD(I,16) .EQ. 11) IEXP = 2	LLP11500
116*	190	IF (I .GT. 1) WRITE (6,2030) IAI,LEXP(IEXP),LINE	LLP11600
117*	220	CONTINUE	LLP11700
118*		IF (NCRIT .GT. 0 .AND. NCRIT .LT. 9) WRITE (6,2080) (LCRIT(I),CRIT(I),I=1,NCRIT)	LLP11800
119*	370	RETURN	LLP11900
120*	500	WRITE(6,2090)	LLP12000
121*		RETURN	LLP12100
122*		END	LLP12200
123*			LLP12300

```

1* SUBROUTINE LABELS(K)
2* COMMON /PARAMT/ TESTNO(12),
3* 1NBK,NPTS,NVS,NVB,XX(41),YY(41),Z(16),DELX(15),DELY(15),Q(15),
4* 2UBARK(16),SIGAK(16),SIGEK(16),SIGXO(15),SIGYO(15),SIGZO(15),
5* 3ALPHA(20),BETA(20),ZRK,TIMAV,THETAK(16),TAUK,TAUOK,H,XRY,XRZ,
6* 4XLRX,XLRZ,ZZL(40),IZMOD(15),DECAY,ZLIM,TIM1,LAMBDA,DI(10),CI(10),
7* 5TAST(05),JBOT(05),JTOP(05),VS(20),PERC(20),ACCUR,VB(20),PERCB(20),
8* 6HB,ALPHL(05),BETL(05),TAUL,TAUOL,ZRL,UBARL(10),SIGAL(10),SIGEL(10),
9* 7,THETAL(10),GAMMAP(20),NTI,TI(10),NPS,NAMCAS(12)
10* COMMON /PARAMS/ UBAR(20),SIGAP(20),DELTHP(20),SIGEP(20),THETA(20),
11* 1DELU(20),VER,VREF,PEAKD,SIGZ,SIGY,SIGX,SQR2P,L,TH,I,J,KK,STOI,
12* 2STO2,STO3,TRD,ILK,RAD,NNZ,ITOP,IBOT,XAST(21),SIGXNK,JF,PPWR,QPWR,
13* 3MPWR,II,DEP,XBARX,SQBAR,NXCI,LAT,SIGYNK,GAMMA(20),NCC,NDD,NTT,
14* 4NCCC,NDDD,NTTI,NSW2,MODLS(15),KSW(5),LINES,IM1,MDLS,NWD,
15* 5YSV(41),YBARY(41),UBARNK(41),BETANK(41),ALPHNK(41),ANG(42),
16* 6SIGENK(41),SIGANK(41),DEPN(41,41),RNG,AZM,IDATE(2),ITIME(2),YT,
17* 7NYSS,CDAMX(3)
18* DIMENSION LINE(1)
19* COMMON /LBLBL/ J1(9),J2(4),J3(48),J5(6),J7(3),J8(16),J9(13),J10,
20* 1J4(12),J11(2),UNIT(15)
21* EQUIVALENCE (YBARY,LINE)
22* INTEGER TESTNO
23* DATA IBLNK/6H /
24* C CHANGE FOLLOWING TWO STATEMENTS FOR 7044
25* C DATA IBLK/0000000000060/,IBLP/00000000000033/
26* DATA IBLK/0000000000005/,IBLP/00000000000075/
27* DATA J1/54H/CALCULATIONS OF MAXIMUM CENTERLINE ISOPLETHS
28* DATA J11/7H MINUTE/
29* DATA J2/24H HCL CO CO2 AL2O3/
30* DATA J3/14H CONCENTRATION,3*1H,7H DOSAGE,4*1H,24H TIME-MEAN CONCLBL03000
31* 1ENTRATION,2*2H,25H GRAVITATIONAL DEPOSITION,1H,13H CALCULATIONSBL03100
32* 2,3*1H,25H PRECIPITATION DEPOSITION,1H,22H TIME OF CLOUD PASSAGE,
33* 32*1H,28H AVERAGE CLOUD CONCENTRATION,1H /
34* DATA J10/3H IN/
35* DATA J4/3HPPM,1H,6HPPM SE,1HC,6HMG/M**,1H3,6HMG SEC,5H/M**,3,6HMG/LBL03500
36* 1M**,1H2,6HSECOND,1HS/
37* DATA J5/36H AT A HEIGHT OF METERS/
38* DATA J7/18H DOWNWIND FROM A /
39* DATA J8/96H STATIC FIRE. NORMAL LAUNCH.
40* 1 ENGINE BURN, SLOW BURN. /

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DATA J9/78XMODEL
IOLOGICAL CASE IS
DO 10 J=1,70
10 LINE(J) = IBLNK
DO 20 J=1,3
N = KSW(3)+J
20 LINE(J+3) = J1(N)
30 IF (TESTNO(11) .NE. IBLNK) GO TO 40
N = ISKIP(5)
LINE(8) = J2(N)
GO TO 50
40 LINE(8) = TESTNO(11)
LINE(9) = TESTNO(12)
50 IF (KSW(4) .NE. 12) GO TO 60
B = TIMAV/60.0
CALL NMBRS(B,LINE(10),IDUM)
LINE(12) = J11(1)
LINE(13) = J11(2)
60 DO 70 J=1,6
N = KSW(4)+J
70 LINE(J+13) = J3(N)
IF (KSW(4) .NE. 24) LINE(20) = J10
IF (KSW(4) .EQ. 24) GO TO 100
M = 8
IF (KSW(4) .EQ. 30.OR.KSW(4) .EQ. 18) GO TO 80
M = 0
IF (ISKIP(5) .EQ. 4) M = 4
IF (KSW(4) .EQ. 6) M = M+2
IF (KSW(4) .EQ. 36) M = 10
80 DO 90 J=1,2
N = M+J
90 LINE(J+20) = J4(N)
100 DO 110 J=1,6
110 LINE(J+22) = J5(J)
CALL NMBRS(ZZL(K),LINE(26),IDUM)
130 DO 140 J=1,3
140 LINE(J+28) = J7(J)
DO 150 J=1,4
150 LINE(J+31) = TESTNO(J+6)
IF (ISKIP(6) .EQ. 5) GO TO 165
DO 160 J=1,4

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WAS USED IN THE CALCULATIONS AND THE METEOR

LBL04100
LBL04200
LBL04300
LBL04400
LBL04500
LBL04600
LBL04700
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LBL07700
LBL07800
LBL07900
LBL08000
LBL08100

82*	N = ISKIP(6) *4-4+J	LBL08200
83*	160 LINE(J+35) = J8(N)	LBL08300
84*	165 J = 39	LBL08400
85*	NWD = 1	LBL08500
86*	GO TO 190	LBL08600
87*	166 IF (N.EQ. 6) JS = JS+1	LBL08700
88*	DO 170 J=1,13	LBL08800
89*	170 LINE(J+JS) = J9(J)	LBL08900
90*	B = MDLS	LBL09000
91*	CALL NMBS(B,LINE(JS+2),IDUM)	LBL09100
92*	JS = JS+13	LBL09200
93*	DO 180 J=1,6	LBL09300
94*	180 LINE(J+JS) = TESTNO(J)	LBL09400
95*	J = JS+6	LBL09500
96*	NWD = 2	LBL09600
97*	190 N = 7	LBL09700
98*	191 N = N-1	LBL09800
99*	IF (N.GT. 0) GO TO 200	LBL09900
100*	N = 6	LBL10000
101*	J = J-1	LBL10100
102*	200 JS = IABS(6*(N-1))	LBL10200
103*	M = 0	LBL10300
104*	CALL MSFLD(JS,6,LINE(J),30,M)	LBL10400
105*	IF (M.EQ. IBLK) GO TO 191	LBL10500
106*	IF (M.EQ. IBLP) GO TO 220	LBL10600
107*	N = N+1	LBL10700
108*	IF (N.LT. 7) GO TO 210	LBL10800
109*	N = 1	LBL10900
110*	J = J+1	LBL11000
111*	210 JS = IABS(6*(N-1))	LBL11100
112*	CALL MSFLD(30,6,IBLP,JS,LINE(J))	LBL11200
113*	220 JS = J	LBL11300
114*	GO TO (160,230),NWD	LBL11400
115*	230 NWD = JS	LBL11500
116*	CALL PACKS(LINE,NWD)	LBL11600
117*	RETURN	LBL11700
118*	END	LBL11800

```

1* SUBROUTINE MAXMIN(A,B,C,D,E,F,ISW)
2* THIS SUBROUTINE DETERMINES MAX & MIN FOR PLOTTING FUNCTION VS.
3* DISTANCE A,B ARE INPUT MAX AND MIN, C AND D ARE CALC MAX AND MIN
4* DETERMINE MAX AND MIN
5* IF (ISW .EQ. 2) GO TO 10
6* IF (A .GT. 0.0) GO TO 80
7* LINEAR SCALING
8* C = E
9* D = F
10* GO TO 90
11* 10 CONTINUE
12* XX = 4.0
13* IF (A .GT. 0.0) XX = A
14* LOG-LOG SCALING
15* C = ALOG10(E)
16* K = C
17* X = K
18* IF (X-C) 20,30,20
19* 20 K = K+1
20* 30 C = 10.0**K
21* D = 1.0
22* IF (F .LE. 0.0) GO TO 40
23* D = F
24* 40 D = ALOG10(D)
25* J = D
26* X = J
27* IF (X-D) 50,60,50
28* 50 IF (D .LT. 0.0) J = J-1
29* 60 IF (FLOAT(K-J) .LE. XX) GO TO 70
30* J = J+1
31* GO TO 60
32* 70 D = 10.0**J
33* GO TO 90
34* 80 C = A
35* D = B
36* 90 RETURN
37* END

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1* SUBROUTINE LSSOPT(X,Y,NX,NY,FI,VLABEL,LEGEND,NCV,NCHAR)
2* IS ISW = 2 LOG-LOG, IF ISW = 1 LINEAR
3* COMMON /PLTLLO/ ISW,XMAXJN,YMAXJN,XCIZE,YCIZE
4* COMMON /BND5/ XRIT,XLFT,YBOT,YTOP,XPL,YPL
5* DIMENSION X(1),Y(1),FI(1)
6* COMMON /ILALPH/LCRIT(10),IBLANK,ISTAR,IP1,IP2,IP3,HLABEL(5),NCH
7* COMMON /XYXYPT/ YP(41),XP(41),A(41),B(41),C(41),D(41),XI(41),YI(41)
8* 1),NUM(3),NC
9* COMMON /ILPLTS/ XMAX,XMIN,YMAX,YMIN,XLM1,YBM1,HT,CHARF,SCLX,SCLY,
10* 1XSIZE1,YSIZE1
11* DATA DISP,XLN,YBN,XRN,YTN/3.,62.,102.,24.,22./
12* XLM1 = XLN
13* YBM1 = YBN
14* XRM1 = XRN
15* YTM1 = YTN
16* XSIZE1 = XCIZE
17* YSIZE1 = YCIZE
18* DETERMINE MAX AND MIN FOR BOTH AXES
19* Y1 = 0.0
20* J1 = 0
21* J2 = 0
22* Y2 = 1.0E20
23* J3 = 0
24* DO 50 I=1,NX
25* IF (Y(I) .LE. 0.0) GO TO 40
26* IF (J1 .GT. 0) GO TO 30
27* J1 = I
28* GO TO 31
29* J2 = I
30* IF (Y(I) .GT. Y1) Y1 = Y(I)
31* IF (Y(I) .LT. Y2) Y2 = Y(I)
32* GO TO 50
33* J3 = I
34* CONTINUE
35* IF (ISW .EQ. 2) GO TO 60
36* IF (J3 .EQ. 1) Y2 = 0.0
37* J1 = 1
38* J2 = NX
39* CALL MAXMIN(XMAXJN,X(J1),XMAX,XMIN,X(J2),X(J1),1)
40* CALL MAXMIN(YMAXJN,Y2,YMAX,YMIN,Y1,Y2,1)
LSS00100
LSS00200
LSS00300
LSS00400
LSS00500
LSS00600
LSS00700
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GO TO 80
60 IF (Y1 .LE. 0.0) GO TO 230
61 IF (X(J1) .GT. 0.0) GO TO 70
J1 = J1+1
GO TO 61
70 XPL = 10.0**X(J1)
YPL = 10.0**X(J2)
CALL MAXMIN(XMAXJN,XPL,XMAX,XMIN,YPL,XPL,2)
CALL MAXMIN(YMAXJN,Y2,YMAX,YMIN,Y1,Y2,2)
80 CONTINUE
C DETERMINE PLOT SCALE
90 IF (ISW .EQ. 2) GO TO 100
SCLX = XSIZE1/(XMAX-XMIN)
SCLY = YSIZE1/(YMAX-YMIN)
GO TO 110
100 SCLX = XSIZE1/(ALOG10(XMAX)-ALOG10(XMIN))
SCLY = YSIZE1/(ALOG10(YMAX)-ALOG10(YMIN))
110 CONTINUE
IF (ISW .EQ. 2) GO TO 115
XLFT = XMIN
XRIT = XMAX
YTOP = YMAX
YBOT = YMIN
GO TO 116
115 CONTINUE
XLFT = ALOG10(XMIN)
XRIT = ALOG10(XMAX)
YTOP = ALOG10(YMAX)
YBOT = ALOG10(YMIN)
116 CONTINUE
CALL SETMIV(0,0,0,0)
CALL FRAMEV(0)
HT = 12
CHARF = 8
C
C DRAW AXES
C CALL ILAXES(ISW,VLABEL,HLABEL,NCV,NCH,LEGEND,NCHAR)
C
C PLOT CURVE
C NC = J2-J1+1
IF (NC .LE. 0) GO TO 220

```

LSS04100
LSS04200
LSS04300
LSS04400
LSS04500
LSS04600
LSS04700
LSS04800
LSS04900
LSS05000
LSS05100
LSS05200
LSS05300
LSS05400
LSS05500
LSS05600
LSS05700
LSS05800
LSS05900
LSS06000
LSS06100
LSS06200
LSS06300
LSS06400
LSS06500
LSS06600
LSS06700
LSS06800
LSS06900
LSS07000
LSS07100
LSS07200
LSS07300
LSS07400
LSS07500
LSS07600
LSS07700
LSS07800
LSS07900
LSS08000
LSS08100

```

82* IF (ISW .NE. 2) GO TO 121
83* DO 120 I=J1,J2
84* Y(I) = ALOG10(Y(I))
85* 120 CONTINUE
86* 121 CONTINUE
87* IF (NC .LT. 3) GO TO 125
88* CALL SPLINE(X(J1),Y(J1),A,B,C,D,NC,IER)
89* IF (IER .EQ. 1) GO TO 125
90* DX = (X(J2)-X(J1))/82.0
91* XPL = X(J1)-DX
92* N = 0
93* I = 1
94* 123 XPL = XPL+DX
95* IF (XPL .LT. X(J1+I)) GO TO 124
96* I = I+1
97* 124 IF (I+J1 .GT. J2.OR.N .GE. 82) GO TO 127
98* N = N+1
99* YPL = XPL-X(J1+I-1)
100* YP(N) = Y(J1+I-1)+YPL*(B(I)+(XPL-X(J1+I))*(2.0*C(I)+C(I+1))+A(I))*
101* 1YPL)*.16666667)
102* XI(N) = XPL
103* GO TO 123
104* 125 DO 126 I=1,NC
105* XI(I) = X(J1+I-1)
106* YP(I) = Y(J1+I-1)
107* N = NC
108* 127 CONTINUE
109* CALL ILPLOT(XI,YP,N,2,A)
110* IF (NY .EQ. 0.OR.NY .EQ. 9) GO TO 220
111* XINC = 8
112* IF (ISW .EQ. 2) YMIN = ALOG10(YMIN)
113* DO 210 I=1,NY
114* IF (ISW .EQ. 2) GO TO 130
115* YS = (FI(I)-YMIN)*SCLY+YBM1
116* GO TO 131
117* 130 YS = (ALOG10(FI(I))-YMIN)*SCLY+YBM1
118* 131 IF (YS .LT. YBM1) GO TO 210
119* IF (YS .GT. YBM1+YSIZE1) GO TO 210
120* CALL NMBS(FI(I),NUM,NC)
121* X1 = XLM1-XINC
122* X2 = XLM1+XSIZE1-FLOAT(NC)*CHARF-DISP

```

```

LSS08200
LSS08300
LSS08400
LSS08500
LSS08600
LSS08700
LSS08800
LSS08900
LSS09000
LSS09100
LSS09200
LSS09300
LSS09400
LSS09500
LSS09600
LSS09700
LSS09800
LSS09900
LSS10000
LSS10100
LSS10200
LSS10300
LSS10400
LSS10500
LSS10600
LSS10700
LSS10800
LSS10900
LSS11000
LSS11100
LSS11200
LSS11300
LSS11400
LSS11500
LSS11600
LSS11700
LSS11800
LSS11900
LSS12000
LSS12100
LSS12200

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123*
124*
125*
126*
127*
128*
129*
130*
131*
132*
133*
134*
135*
136*
137*
138*
139*
140*

```

      IB = 2
      XPL = XLM1
140  X1 = X1+XINC
      IF (X1 .GE. X2) X1 = X2
      IF (IB .EQ. 3) GO TO 150
      CALL LINE2V(IFIX(XPL),IFIX(YS),IFIX(X1-XPL),0)
150  IF (X1 .GE. X2) GO TO 170
      XPL = X1
      IF (IB .EQ. 2) GO TO 160
      IB = 2
      GO TO 140
160  IB = 3
      GO TO 140
170  CALL PRINTV(NC,NUM,IFIX(X2+XINC),IFIX(YS-4,0))
210  CONTINUE
220  CONTINUE
230  RETURN
      END

```

LSS12300
LSS12400
LSS12500
LSS12600
LSS12700
LSS12800
LSS12900
LSS13000
LSS13100
LSS13200
LSS13300
LSS13400
LSS13500
LSS13600
LSS13700
LSS13800
LSS13900
LSS14000

```

1* SUBROUTINE FSTPLT(A,B,C,I,J,K,D,E,F,V1,V2,V3,N1,N2,N3,XP)
2* DIMENSION I(1),J(1),K(1),LSTK(2),V1(1),V2(1),V3(1)
3* DATA MTRS/6HMETERS/
4* IF (K(1).EQ. LSTK(1).AND.K(2).EQ. LSTK(2)) GO TO 40
5* CALL SETMIV(0,0,0,0)
6* CALL FRAMEV(0)
7* CALL PRINTV(72,I,200,800)
8* CALL PRINTV(37,37HADJUSTED CLOUD STABILIZATION HEIGHT =,200,750)
9* CALL LABLV(A,504,750,7,1,4)
10* CALL PRINTV(6,MTRS,568,750)
11* CALL PRINTV(7,7HRANGE =,200,725)
12* CALL LABLV(B,272,725,7,1,5)
13* CALL PRINTV(6,MTRS,336,725)
14* CALL PRINTV(17,17HAZIMUTH BEARING =,200,700)
15* CALL LABLV(C,352,700,6,1,3)
16* CALL PRINTV(7,7HDEGREES,416,700)
17* CALL PRINTV(8,8HRUN DATE,200,650)
18* CALL PRINTV(8,J,288,650)
19* CALL PRINTV(8,8HRUN TIME,400,650)
20* CALL PRINTV(8,K,488,650)
21* IDY = 625
22* IF (N1.LT. 0) GO TO 10
23* IDY = IDY-25
24* CALL PRINTV(7,7HMAXIMUM,200,IDY)
25* CALL PRINTV(N1,V1,264,IDY)
26* CALL LABLV(D,264+8*(N1+1),IDY,-6,1,0)
27* 10 IF (N2.LT. 0) GO TO 20
28* IDY = IDY-25
29* CALL PRINTV(7,7HMAXIMUM,200,IDY)
30* CALL PRINTV(N2,V2,264,IDY)
31* CALL LABLV(E,264+8*(N2+1),IDY,-6,1,0)
32* 20 IF (N3.LT. 0) GO TO 30
33* IDY = IDY-25
34* CALL PRINTV(7,7HMAXIMUM,200,IDY)
35* CALL LABLV(XP,264,IDY,4,1,2)
36* CALL PRINTV(6,6HMINUTE,304,IDY)
37* CALL PRINTV(N3,V3,352,IDY)
38* CALL LABLV(F,352+8*(N3+1),IDY,-6,1,0)
39* 30 CONTINUE
40* LSTK(1) = K(1)
FST00100
FST00200
FST00300
FST00400
FST00500
FST00600
FST00700
FST00800
FST00900
FST01000
FST01100
FST01200
FST01300
FST01400
FST01500
FST01600
FST01700
FST01800
FST01900
FST02000
FST02100
FST02200
FST02300
FST02400
FST02500
FST02600
FST02700
FST02800
FST02900
FST03000
FST03100
FST03200
FST03300
FST03400
FST03500
FST03600
FST03700
FST03800
FST03900
FST04000

```


41*
42*
43*

LSTK(2) = K(2)
40 RETURN
END

FST04100
FST04200
FST04300

1*
2*
3*
4*
5*
6*
7*
8*
9*
10*
11*
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13*
14*
15*
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36*
37*
38*
39*
40*

```

SUBROUTINE NMBRS(A,NUM,NC)
DIMENSION IM(15),NUM(3)
NC = 0
IF (A) 20,10,30
10 NC = 1
NUM(1) = 0
GO TO 110
20 NC = NC+1
IM(1) = 1
30 B = ABS(A)
K = 0
M = B
IF (M.EQ. 0) GO TO 41
M = ALOG10(B)
M = M+1
MM = B
DO 40 I=1,M
NC = NC+1
K = MM/10**(M-I)
MM = MM-K*10**(M-I)
40 IM(NC) = K+48
K = 3
41 M = B
C = M
IF (B-C) 50,80,50
50 NC = NC+1
IM(NC) =
B = B-M
I = 0
B = B+1.0E-7
60 I = I+1
NC = NC+1
B = B*10.0
M = B
B = B-M
IM(NC) = M+48
IF (I.LT. 6) GO TO 60
70 IF (IM(NC).GT. 48.AND.IM(NC).LT. 58) GO TO 80
NC = NC-1
IF (NC.LE. 2) GO TO 80

```

NMB00100
NMB00200
NMB00300
NMB00400
NMB00500
NMB00600
NMB00700
NMB00800
NMB00900
NMB01000
NMB01100
NMB01200
NMB01300
NMB01400
NMB01500
NMB01600
NMB01700
NMB01800
NMB01900
NMB02000
NMB02100
NMB02200
NMB02300
NMB02400
NMB02500
NMB02600
NMB02700
NMB02800
NMB02900
NMB03000
NMB03100
NMB03200
NMB03300
NMB03400
NMB03500
NMB03600
NMB03700
NMB03800
NMB03900
NMB04000

NMB04100
 NMB04200
 NMB04300
 NMB04400
 NMB04500
 NMB04600
 NMB04700
 NMB04800
 NMB04900
 NMB05000
 NMB05100
 NMB05200
 NMB05300

```

      GO TO 70
80  K = 1
    M = 0
    DO 100 I=1,NC
      M = M+1
      IF (M .LT. 7) GO TO 90
      M = 1
      K = K+1
      90 CALL MSFLD(30,6,IM(I),IABS(6*(M-1)),NUM(K))
    100 CONTINUE
    110 CONTINUE
      RETURN
      END
  
```

41*
 42*
 43*
 44*
 45*
 46*
 47*
 48*
 49*
 50*
 51*
 52*
 53*

```

1* SUBROUTINE ILAXES(ISW,VLABEL,HLABEL,NCV,NCH,LEGEND,NCHAR)
2* COMMON /ILPLTS/ XMAX,XMIN,YMAX,YMIN,XLM1,YBM1,HT,CHARF,SCLX,SCLY,
3* 1XSIZE1,YSIZE1
4* DIMENSION NUM(3),LEGEND(1),VLABEL(1),HLABEL(1)
5* ISW = 1 LINEAR AXES
6* ISW = 2 LOG-LOG AXES
7* DATA TIC1/8.0/,TIC2/4.0/,DISP/3.0/
8* IF (ISW .NE. 2) GO TO 40
9* XST = ALOG10(XMIN)
10* XINC = 1.0
11* K = XST
12* XP = K
13* IF (XST-XP) 20,60,20
14* 20 IF (XST) 21,21,30
15* 21 K = K-1
16* GO TO 60
17* 30 K = K+1
18* GO TO 60
19* 40 CONTINUE
20* C DETERMINE INCREMENT BETWEEN MINOR TIC MARKS
21* XINC = (XMAX-XMIN)/(XSIZE1*10.0)
22* IF (XINC .LT. 1.0) XINC = 1.0
23* J = ALOG10(XINC)
24* K = XINC*10.0*(-J)
25* XINC = K*10**J
26* XST = 0.0
27* 50 IF (XST .LE. XMIN) GO TO 60
28* XST = XST-10.0*XINC
29* GO TO 50
30* 60 CONTINUE
31* CALL LINE2V(IFIX(XLM1),IFIX(YBM1),0,IFIX(YSIZE1))
32* CALL LINE2V(IFIX(XLM1),IFIX(YBM1+YSIZE1),IFIX(XSIZE1),0)
33* CALL LINE2V(IFIX(XLM1+XSIZE1),IFIX(YBM1+YSIZE1),0,-IFIX(YSIZE1))
34* CALL LINE2V(IFIX(XLM1+XSIZE1),IFIX(YBM1),-IFIX(XSIZE1),0)
35* PLOT AND LABEL X AXES
36* YP = YEM1
37* J = 1
38* DO 150 I=1,2
39* IF (ISW .EQ. 2) GO TO 70
40* X = XST-XINC

```

ILA00100
 ILA00200
 ILA00300
 ILA00400
 ILA00500
 ILA00600
 ILA00700
 ILA00800
 ILA00900
 ILA01000
 ILA01100
 ILA01200
 ILA01300
 ILA01400
 ILA01500
 ILA01600
 ILA01700
 ILA01800
 ILA01900
 ILA02000
 ILA02100
 ILA02200
 ILA02300
 ILA02400
 ILA02500
 ILA02600
 ILA02700
 ILA02800
 ILA02900
 ILA03000
 ILA03100
 ILA03200
 ILA03300
 ILA03400
 ILA03500
 ILA03600
 ILA03700
 ILA03800
 ILA03900
 ILA04000

41*
42*
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70*
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72*
73*
74*
75*
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77*
78*
79*
80*
81*

```

      GO TO 80
70  L = K -1
      X = 9.0
80  IB = 9
90  X = X+XINC
      IB = IB+1
      IF (IB .LE. 10) GO TO 95
      IB = 1
      IF (ISW .NE. 2) GO TO 95
      IB = 2
      L = L+1
      X = 2.0
95  CONTINUE
      IF (ISW .EQ. 2) GO TO 100
      XP = (X-XMIN)*SCLX+XLM1
      GO TO 105
100  XP = (ALOG10(X*10.0**L)-XST)*SCLX+XLM1
105  IF (XP .LT. XLM1) GO TO 90
      IF (XP .GT. XLM1+XSIZE1) GO TO 140
110  CONTINUE
      IF (IB .LT. 10) GO TO 130
      CALL LINE2V(IFIX(XP),IFIX(YP),0,IFIX(TIC1)*J)
      IF (ISW .EQ. 2) GO TO 120
      IB = 0
      IF (J .LT. 0) GO TO 140
      CALL NMBS(X,NUM,NCHT)
      CALL PRINTV(NCHT,NUM,IFIX(XP-.5*FLOAT(NCHT)*CHARF),IFIX(YP-HT-NISRIL
1))
120  X = L+1
      IB = 1
      IF (J .LT. 0) GO TO 125
      CALL PRINTV(2,'10',IFIX(XP-2.0*CHARF),IFIX(YP-HT-3.0*DISP))
      CALL LABLV(X,IFIX(XP),IFIX(YP-HT-DISP),2,1,2)
125  CONTINUE
      X = 1.0
      L = L+1
      GO TO 140
130  CALL LINE2V(IFIX(XP),IFIX(YP),0,IFIX(TIC2)*J)
140  CONTINUE
      IF (XP .LT. XLM1+XSIZE1) GO TO 90

```

ILA04100
ILA04200
ILA04300
ILA04400
ILA04500
ILA04600
ILA04700
ILA04800
ILA04900
ILA05000
ILA05100
ILA05200
ILA05300
ILA05400
ILA05500
ILA05600
ILA05700
ILA05800
ILA05900
ILA06000
ILA06100
ILA06200
ILA06300
ILA06400
ILA06500
ILA06600
ILASRILA06700
ILA06800
ILA06900
ILA07000
ILA07100
ILA07200
ILA07300
ILA07400
ILA07500
ILA07600
ILA07700
ILA07800
ILA07900
ILA08000
ILA08100

```

82*      YP = YBM1+YSIZE1
83*      J = -1
84*      150 CONTINUE
85*      C      PLOT AND LABEL Y AXES
86*      IF (ISW .NE. 2) GO TO 154
87*      XST = ALOG10(YMIN)
88*      XINC = 1.0
89*      K = XST
90*      XP = K
91*      IF (XST-XP) 151,156,151
92*      151 IF (XST) 152,152,153
93*      152 K = K-1
94*      GO TO 156
95*      153 K = K+1
96*      GO TO 156
97*      154 XINC = (YMAX-YMIN)/(YSIZE1*10.0)
98*      IF (XINC .LT. 1.0) XINC = 1.0
99*      J = ALOG10(XINC)
100*      K = XINC*10.0*(-J)
101*      XINC = K*10**J
102*      XST = 0.0
103*      155 IF (XST .LE. YMIN) GO TO 156
104*      XST = XST-XINC*10.0
105*      GO TO 155
106*      156 CONTINUE
107*      XP = XLM1
108*      XD = 5.0
109*      IF (ISW .EQ. 2) XD = 4.0
110*      XD = XP-XD*CHARF-DISP
111*      J = 1
112*      DO 250 I=1,2
113*      IF (ISW .EQ. 2) GO TO 160
114*      X = XST-XINC
115*      GO TO 170
116*      160 L = K-1
117*      X = 9.0
118*      170 IB = 9
119*      160 X = X+XINC
120*      IB = IB+1
121*      IF (IB .LE. 10) GO TO 165
122*      IB = 1

```

```

ILA08200
ILA08300
ILA08400
ILA08500
ILA08600
ILA08700
ILA08800
ILA08900
ILA09000
ILA09100
ILA09200
ILA09300
ILA09400
ILA09500
ILA09600
ILA09700
ILA09800
ILA09900
ILA10000
ILA10100
ILA10200
ILA10300
ILA10400
ILA10500
ILA10600
ILA10700
ILA10800
ILA10900
ILA11000
ILA11100
ILA11200
ILA11300
ILA11400
ILA11500
ILA11600
ILA11700
ILA11800
ILA11900
ILA12000
ILA12100
ILA12200

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ILA12300
 ILA12400
 ILA12500
 ILA12600
 ILA12700
 ILA12800
 ILA12900
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 ILA13700
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 ILA13900
 ILA14000
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 ILA14700
 ILA14800
 ILA14900
 ILA15000
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 ILA15500
 ILA15600
 ILA15700
 ILA15800
 ILA15900
 ILA16000
 ILA16100
 ILA16200
 ILA16300

```

123* IF (ISW .NE. 2) GO TO 185
124* IB = 2
125* L = L+1
126* X = 2.0
127* 185 CONTINUE
128* IF (ISW .EQ. 2) GO TO 190
129* YP = (X-YMIN)*SCLY+YBM1
130* GO TO 200
131* 190 YP = (ALOG10(X*10.0**L)-XST)*SCLY+YBM1
132* 200 IF (YP .LT. YBM1) GO TO 180
133* IF (YP .GT. YBM1+YSIZE1) GO TO 240
134* 210 CONTINUE
135* IF (IB .LT. 10) GO TO 230
136* CALL LINE2V(IFIX(XP),IFIX(YP),IFIX(TIC1)*J,0)
137* IF (ISW .EQ. 2) GO TO 220
138* IB = 0
139* IF (J .LT. 0) GO TO 240
140* CALL NMBS(X,NUM,NCHT)
141* XF = XP-FLOAT(NCHT)*CHARF-DISP
142* CALL PRINTV(NCHT,NUM,IFIX(XF),IFIX(YP-.5*HT))
143* IF (XF .LT. XD) XD = XF
144* GO TO 240
145* 220 X = L+1
146* IB = 1
147* IF (J .LT. 0) GO TO 225
148* CALL PRINTV(2,'10',IFIX(XP-4.0*CHARF-DISP),IFIX(YP-.5*HT))
149* CALL LABLV(X,IFIX(XP-2.0*CHARF-DISP),IFIX(YP+DISP),2,1,2)
150* 225 CONTINUE
151* X = 1.0
152* L = L+1
153* GO TO 240
154* 230 CALL LINE2V(IFIX(XP),IFIX(YP),IFIX(TIC2)*J,0)
155* 240 CONTINUE
156* IF (YP .LT. YBM1+YSIZE1) GO TO 180
157* XP = XLM1+XSIZE1
158* J = -1
159* 250 CONTINUE
160* C DRAW VERTICAL AXIS LABEL
161* XP = XD-DISP-CHARF
162* YP = (YSIZE1+FLOAT(NCV)*(HT+DISP))*0.5+YBM1
163* CALL APRNTV(0,-IFIX(HT+DISP),NCV,VLABEL,IFIX(XP),IFIX(YP))

```

164*
165*
166*
167*
168*
169*
170*
171*
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174*
175*
176*
177*
178*
179*
180*

```

C      DRAW HORIZONTAL AXIS LABEL
      XP = (XSIZE1-FLOAT(NCH)*CHARF)*0.5+XLM1
      YP = YBM1-2.0*(2.5*DISP+HT)
      CALL PRINTV(NCH,HLABEL,IFIX(XP),IFIX(YP))
      DRAW LEGEND FOR PLOT
      XP = (XSIZE1-90.0*CHARF)*0.5+XLM1
      YP = YP-2.0*HT
      J = -89
      260 YP = YP-(HT+DISP)
      J = J+90
      K = J+89
      IF (K.GT. NCHAR) K = NCHAR
      I = (J/6)+1
      CALL PRINTV(K-J+1,LEGEND(I),IFIX(XP),IFIX(YP))
      IF (K.LT. NCHAR) GO TO 260
      RETURN
      END

```

ILA16400
ILA16500
ILA16600
ILA16700
ILA16800
ILA16900
ILA17000
ILA17100
ILA17200
ILA17300
ILA17400
ILA17500
ILA17600
ILA17700
ILA17800
ILA17900
ILA18000


```

1* SUBROUTINE SPLINE(X,Y,A,B,C,D,N,IER)
2* DIMENSION X(1),Y(1),A(1),B(1),C(1),D(1)
3* IER = 0
4* C(1) = 0.0
5* C(N) = 0.0
6* Q = 1.07179677
7* Q = 4.0*(2.0-SQRT(3.0))
8* NP = N-1
9* DO 10 I=1,NP
10* A(I) = X(I+1)-X(I)
11* B(I) = (Y(I+1)-Y(I))/A(I)
12* IF (I .LT. 2) GO TO 10
13* C(I) = 2.0*(B(I)-B(I-1))/(A(I-1)+A(I))
14* D(I) = C(I)*1.5
15* D(I) = C(I)*3.0/2.0
16*
17* 10 CONTINUE
18* NTM = 0
19* XM = 0.0
20* DO 30 I=2,NP
21* YP = C(I+1)
22* YP = Q*((YP-C(I-1))/(1.0+A(I)/A(I-1))-YP)*0.5-C(I)+D(I)
23* IF (ABS(YP) .GT. XM) XM = ABS(YP)
24* C(I) = C(I)+YP
25* 30 CONTINUE
26* NTM = NTM+1
27* IF (NTM .LT. 80) GO TO 35
28* IER = 1
29* GO TO 36
30* 35 CONTINUE
31* IF (1.0E-3 .LE. XM) GO TO 20
32* 36 CONTINUE
33* DO 40 I=1,NP
34* A(I) = (C(I+1)-C(I))/A(I)
35* 40 CONTINUE
36* RETURN
37* END

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1* SUBROUTINE ISSOPT(X,Y,NX,NY,FI,LEGEND,NCHAR,DP,NYS,XX,DR,II,IREC,YISS00100
2* 1T,ISW3,KLINE,NCV,JM,DECAY,LAMBDA) ISS00200
3* COMMON /PLTISO/ SCL,XMAXIN,YMAXIN,XSIZE,YSIZE,RASTIN,JSW ISS00300
4* COMMON /BNDIS/ XRIT,XLFT,YBOT,YTOP,XPL,YPL ISS00400
5* DIMENSION X(1),Y(1),FI(1),DP(41,1),XX(1),DR(245,1),NPP(4),YY(1) ISS00500
6* DIMENSION TLABEL(3) ISS00600
7* DIMENSION LEGEND(1),KLINE(1) ISS00700
8* REAL LAMBDA ISS00800
9* DIMENSION KB(4) ISS00900
10* COMMON /XYXYPT/ YP(41),XP(41),A(41),B(41),C(41),D(41),XI(41),YI(41) ISS01000
11* 1),NUM(3),NC ISS01100
12* EQUIVALENCE (YP,YY) ISS01200
13* DATA XLM,YBM,XRM,YTM,DISP/62.,102.,24.,22.,3./ ISS01300
14* DATA RADII/57.295779/ ISS01400
15* DATA TLABEL/17HDISTANCE (METERS)/,NCTH/17/ ISS01500
16* COMMON /ILPLTS/ XMAX,XMIN,YMAX,YMIN,XLM1,YRM1,HT,CHARF,SCLX,SCLY, ISS01600
17* 1XSIZE1,YSIZE1 ISS01700
18* COMMON /ILALPH/ LCRIT(10),IBLANK,ISTAR,IP1,IP2,IP3,HLABEL(5),NCH ISS01800
19* DATA RAD/.017453293/ ISS01900
20* XYTERP(A,B,C,D,E) = B-(D-E)*(B-A)/(D-C) ISS02000
21* NWD = NCHAP/6 ISS02100
22* IF (ISW3.EQ. 1) GO TO 50 ISS02200
23* XLM1 = XLM ISS02300
24* YBM1 = YBM ISS02400
25* XRM1 = XRM ISS02500
26* YTM1 = YTM ISS02600
27* XSIZE1 = XSIZE ISS02700
28* YSIZE1 = YSIZE ISS02800
29* ISW = 1 ISS02900
30* DETERMINE MAX AND MIN FOR BOTH AXES ISS03000
31* DO 10 I=1,NX ISS03100
32* DO 10 J=1,NYS ISS03200
33* DO 10 K=1,NY ISS03300
34* IF (DP(I,J).GT. FI(K)) GO TO 11 ISS03400
35* 10 CONTINUE ISS03500
36* GO TO 800 ISS03600
37* 11 CONTINUE ISS03700
38* XMX = X(NX-2) ISS03800
39* IF (XMAXIN.GT. 0.0) XMX = XMAXIN ISS03900
40* XMAX = 0.0 ISS04000

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41* XMIN = 0.0
42* YMAX = 0.0
43* YMIN = 0.0
44* XI(1) = XM* SIN(YT* RAD)
45* YI(1) = XM* COS(YT* RAD)
46* YPL = 1.0E8
47* DO 12 N=1,NY
48* 12 YPL = AMINI(YPL,FI(N))
49* I1 = 0
50* I2 = 0
51* DO 15 J=1,NYS
52* DO 14 I=1,NX
53* IF (DP(I,J) .LT. YPL) GO TO 14
54* IF (I1 .GT. 0) GO TO 13
55* I1 = I
56* I2 = I
57* 13 CONTINUE
58* IF (I1 .GT. 0) GO TO 16
59* 14 CONTINUE
60* J1 = J
61* I3 = 0
62* I4 = 0
63* DO 19 J=1,NYS
64* DO 18 I=1,NX
65* IF (DP(I,NYS-J+1) .LT. YPL) GO TO 18
66* IF (I3 .GT. 0) GO TO 17
67* I3 = I
68* I4 = I
69* 17 CONTINUE
70* IF (I3 .GT. 0) GO TO 20
71* 18 CONTINUE
72* J2 = NYS-J+1
73* IF (I1 .EQ. 0) I1 = 1
74* IF (I2 .EQ. 0) I2 = NX-2
75* IF (I3 .EQ. 0) I3 = 1
76* IF (I4 .EQ. 0) I4 = NX-2
77* IF (I2 .GT. NX-2) I2 = NX-2
78* IF (I4 .GT. NX-2) I4 = NX-2
79* Y1 = Y(J1)
80* Y2 = Y(J2)
81* IF (YMAXIN .LE. 0.0) GO TO 21

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82*      XPL = 1.0-YMAXIN/(2.0*XMX*XX)
83*      IF (XPL .GT. 1.0) XPL = 1.0
84*      IF (XPL .LT. -1.0) XPL = -1.0
85*      XPL = ACOS(XPL)*RADI
86*      Y1 = YT+0.5*XPL
87*      Y2 = YT-0.5*XPL
88*      21 XI(2) = X(I1)*SIN(Y1*RAD)
89*      YI(2) = X(I1)*COS(Y1*RAD)
90*      XI(3) = X(I3)*SIN(Y2*RAD)
91*      YI(3) = X(I3)*COS(Y2*RAD)
92*      XI(4) = X(I2)*SIN(Y1*RAD)
93*      YI(4) = X(I2)*COS(Y1*RAD)
94*      XI(5) = X(I4)*SIN(Y2*RAD)
95*      YI(5) = X(I4)*COS(Y2*RAD)
96*      DO 22 I=1,5
97*      XMAX = AMAX1(XMAX,XI(I))
98*      YMAX = AMAX1(YMAX,YI(I))
99*      XMIN = AMIN1(XMIN,XI(I))
100*      YMIN = AMIN1(YMIN,YI(I))
101*      C
102*      22 DETERMINE PLOT SCALE
103*      IF (SCL .LE. 0.0) GO TO 30
104*      SCLX = 12.0/(SCL*.3048)
105*      SCLY = SCLX
106*      XPL = (XMAX-XMIN)*SCLX
107*      IF (XPL .LE. XSIZE1) GO TO 24
108*      YPL = XPL-XSIZE1
109*      XMAX = XMAX-0.5*YPL
110*      XMIN = XMIN+0.5*YPL
111*      IF (XMAX .GE. 0.0) GO TO 23
112*      XMIN = XMIN-XMAX
113*      XMAX = 0.0
114*      GO TO 24
115*      23 IF (XMIN .LE. 0.0) GO TO 24
116*      XMAX = XMAX-XMIN
117*      XMIN = 0.0
118*      24 XPL = (YMAX-YMIN)*SCLY
119*      IF (XPL .LE. YSIZE1) GO TO 26
120*      YPL = XPL-YSIZE1
121*      YMAX = YMAX-0.5*YPL
122*      YMIN = YMIN+0.5*YPL

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IF (YMAX .GE. 0.0) GO TO 25
YMIN = YMIN-YMAX
YMAX = 0.0
GO TO 26
25 IF (YMIN .LE. 0.0) GO TO 26
YMAX = YMAX-YMIN
YMIN = 0.0
26 CONTINUE
GO TO 40
30 CONTINUE
SCLX = XSIZE1/(XMAX-XMIN)
SCLY = YSIZE1/(YMAX-YMIN)
40 CONTINUE
HT = 12
CHARF = 8
CALL SETMIV(0,0,0,0)
CALL FRAMEV(0)
DRAW AXES
CALL ILAXES(1,TLABEL,TLABEL,NCTH,NCTH,LEGEND,NCHAR)
XRIT = XMAX
XLFT = XMIN
YBOT = YMIN
YTOP = YMAX
50 CONTINUE
LINES = 57
DO 710 N=1,NY
IF (N.EQ. 1) GO TO 240
DO 230 I=1,NX
KOUT = 4*I-4+IREC
CALL INTOUT(DP,KOUT,NYS,1,41,I)
230 CONTINUE
240 CONTINUE
CALL NMBS(FI(N),NUM,NC)
DO 400 I=1,4
400 NPP(I) = 0
NP = 3
L = 0
K = 1
DO 470 I=1,NX
JB = 0
DO 430 J=2,NYS

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164* IF (DP(I,J-1) .LE. FI(N).AND.FI(N) .LE. DP(I,J)) GO TO 410
165* IF (DP(I,J-1) .GE. FI(N).AND.FI(N) .GE. DP(I,J)) GO TO 410
166* GO TO 430
167* JB = 1
168* L = L+1
169* NP = NP+1
170* Y2 = Y(J)
171* IF (ABS(Y(J-1)-Y2) .LT. 180.0) GO TO 420
172* Y2 = 360.0-ABS(Y(J-1)-Y2)+Y(J-1)
173* YY(NP) = XYTERP(Y(J-1),Y2,DP(I,J-1),DP(I,J),FI(N))
174* XX(NP) = X(I)
175* IF (NP .GE. 245) GO TO 475
176* 430 CONTINUE
177* IF (JB .EQ. 1) GO TO 440
178* IF (L .EQ. 0) GO TO 440
179* NPP(K) = L
180* L = 0
181* K = K+1
182* 440 IF (I .EQ. NX) GO TO 470
183* DO 460 J=1,NYS
184* IF (DP(I,J) .LE. FI(N).AND.FI(N) .LE. DP(I+1,J)) GO TO 450
185* IF (DP(I,J) .GE. FI(N).AND.FI(N) .GE. DP(I+1,J)) GO TO 450
186* GO TO 460
187* L = L+1
188* NP = NP+1
189* XX(NP) = XYTERP(X(I),X(I+1),DP(I,J),DP(I+1,J),FI(N))
190* YY(NP) = Y(J)
191* IF (NP .GE. 245) GO TO 475
192* 460 CONTINUE
193* 470 CONTINUE
194* 475 CONTINUE
195* NP = NP-3
196* NPP(K) = L
197* 480 IF (NPP(K) .GT. 1) GO TO 490
198* K = K-1
199* IF (K .LE. 0) GO TO 710
200* GO TO 480
201* 490 CONTINUE
202* C DETERMINE IF CLOSED CURVE OR NO, KR(L)=0 IS YES, KR(L) NOT 0 IS NO
203* IP1 = 3
204* DO 497 L=1,K

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205*	KB(L) = 0	ISS20500
206*	MP = NPP(L)	ISS20600
207*	J2 = 1	ISS20700
208*	DO 495 I2=1,2	ISS20800
209*	DO 491 I=1,MP	ISS20900
210*	IF (YY(I+IP1)-Y(J2)) 491,492,491	ISS21000
211*	491 CONTINUE	ISS21100
212*	GO TO 495	ISS21200
213*	492 X1 = 1.0E8	ISS21300
214*	DO 493 J=1,MP	ISS21400
215*	IF (J+IP1).EQ. I+IP1) GO TO 493	ISS21500
216*	Y1 = ABS(Y(J2)-YY(J+IP1))	ISS21600
217*	IF (Y1 .GT. 180.0) Y1 = 360.0-Y1	ISS21700
218*	IF (Y1 .GE. X1) GO TO 493	ISS21800
219*	X1 = Y1	ISS21900
220*	J1 = J	ISS22000
221*	493 CONTINUE	ISS22100
222*	IF (XX(I+IP1) .GT. XX(J1+IP1)) GO TO 494	ISS22200
223*	KB(L) = KB(L)+3	ISS22300
224*	GO TO 495	ISS22400
225*	494 KB(L) = KB(L)+1	ISS22500
226*	495 J2 = NYS	ISS22600
227*	496 IP1 = IP1+NPP(L)	ISS22700
228*	IF (KB(L) .EQ. 1.OR.KB(L) .EQ. 3) KB(L) = 0	ISS22800
229*	497 CONTINUE	ISS22900
230*	IF (K .LE. 1) GO TO 501	ISS23000
231*	L = 1	ISS23100
232*	498 L = L+1	ISS23200
233*	IF (L .GT. K) GO TO 501	ISS23300
234*	IF (KB(L-1) .NE. 2) GO TO 498	ISS23400
235*	IF (KB(L) .NE. 6) GO TO 498	ISS23500
236*	NPP(L-1) = NPP(L-1)+NPP(L)	ISS23600
237*	KB(L-1) = 0	ISS23700
238*	MP = 0	ISS23800
239*	499 IF (L+MP .GE. K) GO TO 500	ISS23900
240*	NPP(L+MP) = NPP(L+MP+1)	ISS24000
241*	KB(L+MP) = KB(L+MP+1)	ISS24100
242*	MP = MP+1	ISS24200
243*	GO TO 499	ISS24300
244*	500 K = K-1	ISS24400
245*	GO TO 498	ISS24500

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246* 501 CONTINUE
247*   IP1 = 3
248*   DO 700 L=1,K
249*     NP = NPP(L)
250*     SHIFT POINTS TO START OF ARRAY
251*     I = 0
252*     DO 502 J=1,NP
253*       I = I+1
254*       XX(I) = XX(J+IP1)
255*       YY(I) = YY(J+IP1)
256*       IF (XX(I) .LT. X(NX)) GO TO 502
257*     CONTINUE
258*     CALL CALCS(XX,YY,NP,RAD,RADI,XSHFT,YSHFT,0.0,0.0,0.0)
259*     I = 1
260*     I = I+1
261*     IF (I .GT. NP) GO TO 546
262*     IF (YY(I)-YY(I-1) .GE. 1.0) GO TO 543
263*     I = I+1
264*     IF (I .GT. NP) GO TO 545
265*     DO 544 J=I,NP
266*       YY(J-1) = YY(J)
267*       XX(J-1) = XX(J)
268*     I = I-2
269*     NP = NP-1
270*     CONTINUE
271*   C FIND START POINT OF CURVE
272*   IF (KB(L) .EQ. 0) GO TO 577
273*   C CALC MAX DIFF AND START INDEX FOR OPEN CURVE
274*   YRM = YT
275*   IF (KB(L) .EQ. 6) YRM = YT-180.0
276*   550 IF (YRM .GE. YY(1)) GO TO 551
277*   YRM = YRM+360.0
278*   GO TO 550
279*   551 IF (YRM .LE. YY(NP)) GO TO 552
280*   YRM = YRM-360.0
281*   GO TO 551
282*   552 CONTINUE
283*   DO 553 I=1,NP
284*     IF (YY(I) .LT. YRM) GO TO 553
285*     MP = I-1
286*   553 CONTINUE

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GO TO 582
577 YRM = YT+180.0
578 IF (YRM .GT. YY(1)) GO TO 579
YRM = YRM+360.0
GO TO 578
579 IF (YRM .LE. YY(NP)) GO TO 580
YRM = YRM-360.0
GO TO 579
580 MP = 0
DO 581 I=1,NP
IF (YY(I) .LT. YRM) GO TO 581
MP = I-1
GO TO 582
581 CONTINUE
582 IF (MP .LE. 0) GO TO 594
DO 592 I=1,MP
X1 = XX(1)
Y1 = YY(1)
DO 591 J=1,NP
XX(J-1) = XX(J)
YY(J-1) = YY(J)
591 XX(NP) = X1
592 YY(NP) = Y1
C
MAKE SURE IN ASCENDING ORDER
Y1 = YY(1)
DO 593 I=2,NP
DIF = ABS(YY(I)-Y1)
IF (DIF .GT. 180.0) DIF = 360.0-DIF
Y1 = YY(I)
593 YY(I) = YY(I-1)+DIF
594 CONTINUE
IF (NP .GT. 245) NP = 245
DIF = 0.0
DO 595 I=1,NP
IF (XX(I) .LE. DIF) GO TO 595
DIF = XX(I)
N1 = I
595 CONTINUE
X1 = YY(N1)
IF (N1-2 .LT. 1) GO TO 596
IF (N1+2 .GT. NP) GO TO 596

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328*      X1 = (YY(N1-2)+YY(N1-1)+YY(N1)+YY(N1+1)+YY(N1+2))/5
329*      596 CONTINUE
330*      NPI = NP
331*      IER = 1
332*      IF (NP.LT. 6) GO TO 650
333*      IF (JSW.NE. 0) GO TO 650
334*      CALL SPLINE(YY,XX,DR,DR(1,2),DR(1,3),DR(1,4),NP,IER)
335*      IF (IER.EQ. 1) GO TO 650
336*      XPL = (YY(NP)-YY(1))/200.0
337*      XPL2 = XPL*0.1
338*      XPI = XPL
339*      J = 0
340*      M = 1
341*      YPL = YY(1)-XPI
342*      631 YPL = YPL+XPI
343*      IF (YPL.LE. X1-2.0*XPL) GO TO 632
344*      XPI = XPL2
345*      IF (YPL.GE. X1+2.0*XPL) XPI = XPL
346*      632 IF (YPL.LT. YY(M+1)) GO TO 634
347*      633 M = M+1
348*      IF (M.GE. NP) GO TO 670
349*      IF (YPL.GE. YY(M+1)) GO TO 633
350*      Y1 = YPL-YY(M)
351*      Y1 = XX(M)+Y1*(DR(M,2)+(YPL-YY(M+1))*(2.*DR(M,3)+DR(M,1)
352*      1*Y1)*.16666667)
353*      J = J+1
354*      DR(J,5) = Y1
355*      DR(J,6) = YPL
356*      IF (J.GE. 244) GO TO 670
357*      GO TO 631
358*      650 DO 660 M=1,NP
359*      DR(M,5) = XX(M)
360*      660 DR(M,6) = YY(M)
361*      NPI = NP
362*      670 MP = 1
363*      IF (IER.EQ. 0) NP = J
364*      IF (KB(L).NE. 0) GO TO 680
365*      NP = NP+1
366*      DR(NP,5) = DR(1,5)
367*      DR(NP,6) = DR(1,6)
368*      680 CONTINUE
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369* DO 681 J=1,NP
370* I = J+MP-1
371* YPL = DR(I,6)*RAD
372* DR(I,6) = DR(I,5)*COS(YPL)+YSHFT
373* DR(I,5) = DR(I,5)*SIN(YPL)+XSHFT
374* PLOT CURVE
375* IF (ISW3.EQ. 2) GO TO 694
376* M = NPI
377* CALL CALCS(XX,YY,M,RAD,RADI,X1,Y1,XSHFT,YSHFT,1)
378* IF (LINES.GT. 52) GO TO 686
379* WRITE (6,2000) FI(N),(KLINE(J),J=1,NCV)
380* WRITE (6,2003)
381* LINES = LINES+3
382* GO TO 687
383* LINES = 57
384* CONTINUE
385* M1 = -5
386* M1 = M1+6
387* IF (M1.GT. M) GO TO 692
388* M2 = M1+5
389* IF (M2.GT. M) M2 = M
390* LINES = LINES+1
391* IF (LINES.LT. 57) GO TO 691
392* IF (JM.GT. 1) GO TO 689
393* CALL PRTTTL(NWD,LINES,LEGEND,0.0,0.0)
394* GO TO 690
395* CALL PRTTTL(NWD,LINES,LEGEND,DECAY,LAMBDA)
396* WRITE (6,2000) FI(N),(KLINE(J),J=1,NCV)
397* WRITE (6,2001)
398* LINES = LINES+7
399* WRITE (6,2002) (XX(J),YY(J),J=M1,M2)
400* GO TO 688
401* CONTINUE
402* CONTINUE
403* IF (ISW3.EQ. 1) GO TO 695
404* CALL ILPLOT(DR(MP,5),DR(MP,6),NP,1,DR)
405* CONTINUE
406* IPI = IPI+MPP(L)
407* CONTINUE
408* CONTINUE
409* IF (ISW3.EQ. 1) GO TO 800

```

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ISS36900
ISS37000
ISS37100
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ISS40000
ISS40100
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ISS40900

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410*      XPL = -XMIN*SCLX+XLM1-0.5*HT
411*      YPL = -YMIN*SCLY+YBM1-0.5*HT
412*      CALL PRINTV(1,1H*,IFIX(XPL),IFIX(YPL))
413*      800 CONTINUE
414*      RETURN
415*      2000 FORMAT (1H0,40X,22H*--* ISOPLETH LEVEL =,F9.3,2H, ,9A6)
416*      2001 FORMAT(1H0,6(19H RANGE AZIMUTH )/1X,6(19H (METERS) BEARING )/
417*      11X,6(10X,9H(DEGREES))/1X,19(6H-----))
418*      2002 FORMAT (1X,6(F10.3,F8.3,1X))
419*      2003 FORMAT ( )
420*      END
ISS41000
ISS41100
ISS41200
ISS41300
ISS41400
ISS41500
ISS41600
ISS41700
ISS41800
ISS41900
ISS42000

```

```

1* SUBROUTINE ILPLOT(X,Y,N,ISW,IJ)
2* THIS SUBROUTINE PLOTS AND LABELS CURVES
3* COMMON /BNDX/ XRIT,XLFT,YBOT,YTOP,XPL,YPL
4* COMMON /ILPLTS/ XMAX,XMIN,YMAX,YMIN,XLM1,YBM1,HT,CHARF,SCLX,SCLY,
5* IXSIZE1,YSIZE1
6* DIMENSION X(1),Y(1),IJ(1)
7* COMMON /XYXYPT/ YP(41),XP(41),A(41),B(41),C(41),D(41),XI(41),YI(41)
8* 1),NUM(3),NC
9* IFLG = 0
10* JFLG = 0
11* DO 100 I=1,N
12* X1 = X(I)
13* Y1 = Y(I)
14* IF (XLFT .LE. X1.AND.X1 .LE. XRIT) GO TO 20
15* GO TO 30
16* IF (YBOT .LE. Y1.AND.Y1 .LE. YTOP) GO TO 60
17* IF (JFLG .EQ. 0) GO TO 50
18* LAST POINT WAS OUT, THIS POINT IS OUT
19* IJ(I) = 3
20* GO TO 90
21* JFLG = 1
22* IF (I.EQ. 1) GO TO 40
23* THIS POINT OUT LAST POINT IN
24* INTERP FOR PLOT POINT
25* CALL BOUNDS(X1,Y1,XLST,YLST)
26* ITAG = 1
27* GO TO 80
28* IF (JFLG .EQ. 0) GO TO 70
29* THIS POINT IN LAST POINT OUT
30* JFLG = 0
31* CALL BOUNDS(XLST,YLST,X1,Y1)
32* IJ(I-1) = 0
33* X(I-1) = (XPL-XLFT)*SCLX+XLM1
34* Y(I-1) = (YPL-YBOT)*SCLY+YBM1
35* XPL = X1
36* YPL = Y1
37* ITAG = 2
38* X(I) = (XPL-XLFT)*SCLX+XLM1
39* Y(I) = (YPL-YBOT)*SCLY+YBM1
40* IJ(I) = ITAG

```

ILP00100
ILP00200
ILP00300
ILP00400
ILP00500
ILP00600
ILP00700
ILP00800
ILP00900
ILP01000
ILP01100
ILP01200
ILP01300
ILP01400
ILP01500
ILP01600
ILP01700
ILP01800
ILP01900
ILP02000
ILP02100
ILP02200
ILP02300
ILP02400
ILP02500
ILP02600
ILP02700
ILP02800
ILP02900
ILP03000
ILP03100
ILP03200
ILP03300
ILP03400
ILP03500
ILP03600
ILP03700
ILP03800
ILP03900
ILP04000

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81*

```

90 XLST = X1
   YLST = Y1
100 CONTINUE
   IF (ISW.EQ. 2) GO TO 160
   IJ = 0 CURVE ENTERS GRAPH - FIRST POINT
   IJ = 1 CURVE LEAVES GRAPH - LAST POINT
   IJ = 2 CURVE CONTINUES WITHIN GRAPH
   IJ = 3 CURVE OUTSIDE OF GRAPH DO NOT PLOT

   FIND POINTS FOR ISOPLETH LABELS
   FIND ALL POINTS WHERE CURVE LEAVES GRAPH
   M = 0
   DO 110 I=1,N
     IF (IJ(I) .NE. 1) GO TO 110
     M = M+1
     B(M) = X(I)
     C(M) = Y(I)+.02
110 CONTINUE
   FIND ALL POINTS WHERE CURVE ENTERS GRAPH
   DO 120 I=1,N
     IF (IJ(I) .NE. 0) GO TO 120
     M = M+1
     B(M) = X(I)
     C(M) = Y(I)+.02
120 CONTINUE
   L = N/2
   IF (IJ(L) .EQ. 2) GO TO 130
   L = 1
   IF (IJ(L) .EQ. 2) GO TO 130
   L = N/4
   IF (IJ(L) .EQ. 2) GO TO 130
   L = 3*N/4
   IF (IJ(L) .NE. 2) GO TO 140
   M = M+1
   B(M) = X(L)
   C(M) = Y(L)+.02
140 CONTINUE
   PLOT LABELS
   XPL = -XLFT*SCLX+XLM1
   YPL = -YBOT*SCLY+YBM1
   DO 150 I=1,M

```

IILP04100
 IILP04200
 IILP04300
 IILP04400
 IILP04500
 IILP04600
 IILP04700
 IILP04800
 IILP04900
 IILP05000
 IILP05100
 IILP05200
 IILP05300
 IILP05400
 IILP05500
 IILP05600
 IILP05700
 IILP05800
 IILP05900
 IILP06000
 IILP06100
 IILP06200
 IILP06300
 IILP06400
 IILP06500
 IILP06600
 IILP06700
 IILP06800
 IILP06900
 IILP07000
 IILP07100
 IILP07200
 IILP07300
 IILP07400
 IILP07500
 IILP07600
 IILP07700
 IILP07800
 IILP07900
 IILP08000
 IILP08100

82*	IF (ABS(XPL-B(I)) .GT. 0.2) GO TO 145	ILP08200
83*	IF (ABS(YPL-C(I)) .LE. 0.2) GO TO 150	ILP08300
84*	145 CONTINUE	ILP08400
85*	CALL PRINTV(NC,NUM,IFIX(B(I)),IFIX(C(I)))	ILP08500
86*	150 CONTINUE	ILP08600
87*	160 CONTINUE	ILP08700
88*	PLOT THE CURVE	ILP08800
89*	IF (IJ(1) .NE. 3) IJ(1) = 0	ILP08900
90*	N = N-1	ILP09000
91*	DO 170 I=1,N	ILP09100
92*	IF (IJ(I) .EQ. 3) GO TO 170	ILP09200
93*	IF (IJ(I+1) .EQ. 3) GO TO 170	ILP09300
94*	CALL LINE2V(IFIX(X(I)),IFIX(Y(I)),IFIX(X(I+1))-X(I),IFIX(Y(I+1))-Y(I)))	ILP09400
95*	170 CONTINUE	ILP09500
96*	180 CONTINUE	ILP09600
97*	RETURN	ILP09700
98*	END	ILP09800
99*		ILP09900

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```

SUBROUTINE CALCS(XX,YY,NP,RAD,RADI,XSHIFT,YSHIFT,X1,Y1,LLSW)
DIMENSION XX(1),YY(1)
IF (LLSW .EQ. 1) GO TO 5
XSHIFT = 0.0
YSHIFT = 0.0
5 DO 10 I=1,NP
  YPL = YY(I)*RAD
  YY(I) = XX(I)*COS(YPL)+Y1
  XX(I) = XX(I)*SIN(YPL)+X1
  IF (LLSW .EQ. 1) GO TO 10
  XSHIFT = XSHIFT+XX(I)
  YSHIFT = YSHIFT+YY(I)
10 CONTINUE
  IF (LLSW .EQ. 1) GO TO 11
  XSHIFT = XSHIFT/FLOAT(NP)
  YSHIFT = YSHIFT/FLOAT(NP)
11 CONTINUE
  DO 20 I=1,NP
    IF (LLSW .EQ. 1) GO TO 17
    XX(I) = XX(I)-XSHIFT
    YY(I) = YY(I)-YSHIFT
17 CONTINUE
    IF (XX(I)) 19,18,19
    18 IF (YY(I)) 19,20,19
    19 YPL = 90.0-ATAN2(YY(I),XX(I))*RADI
    IF (YPL .LT. 0.0) YPL = YPL+360.0
    XX(I) = SQR(XX(I)*XX(I)+YY(I)*YY(I))
    YY(I) = YPL
    IF (LLSW .EQ. 1) YY(I) = AMOD(YPL,360.0)
20 CONTINUE
    IF (LLSW .EQ. 1) GO TO 31
    DO 30 M=2,NP
      DO 30 I=2,NP
        IF (YY(I) .GE. YY(I-1)) GO TO 30
        YPL = YY(I)
        YY(I) = YY(I-1)
        YY(I-1) = YPL
        YPL = XX(I)
        XX(I) = XX(I-1)
        XX(I-1) = YPL
30 CONTINUE
31*
32*
33*
34*
35*
36*
37*
38*
39*
40*

```

CAL00100
CAL00200
CAL00300
CAL00400
CAL00500
CAL00600
CAL00700
CAL00800
CAL00900
CAL01000
CAL01100
CAL01200
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CAL01400
CAL01500
CAL01600
CAL01700
CAL01800
CAL01900
CAL02000
CAL02100
CAL02200
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CAL02400
CAL02500
CAL02600
CAL02700
CAL02800
CAL02900
CAL03000
CAL03100
CAL03200
CAL03300
CAL03400
CAL03500
CAL03600
CAL03700
CAL03800
CAL03900
CAL04000

CAL04100
CAL04200
CAL04300

30 CONTINUE
31 RETURN
END

41*
42*
43*

```

1*
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6*
7*
8*
C
C
C
C
C
SUBROUTINE MSFLD(I1,I2,IWRD,J1,JWRD)
THIS PROG EXTRACS AN I2 BIT BYTE FROM IWRD STARTING AT BIT I1 ANDMSF00100
STORES IT IN JWRD STARTING AT BIT J1. THE REMAINING BITS OF JWRD MSF00200
ARE UNCHANGED. I1 AND J1 ARE COUNTED RIGHT FROM THE SIGN BIT AND MSF00300
THE SIGN BIT IS BIT ZERO. MSF00400
FLD(J1,I2,JWRD) = FLD(I1,I2,IWRD) MSF00500
RETURN MSF00600
END MSF00700
MSF00800

```

```

1*
2*
3*
4*
FUNCTION RB11(PARM,P,Z,ZRK)
RB11 = PARM*(Z**P-ZRK**P)/(P*(Z-ZRK)*ZRK**(P-1.0))
RETURN
END
R1100100
R1100200
R1100300
R1100400

```

```

1*
2*
3*
4*
5*
6*
7*
FUNCTION RBB(A,B,C)
RBB = ALOG(A/B)*C
IF (RBB+1.0) 20,10,20
10 RBB = -.99999999
20 RBB = RBB+1.0
RETURN
END
RB800100
RB800200
RB800300
RB800400
RB800500
RB800600
RB800700

```

1*
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14*
15*
16*
17*

```

SUBROUTINE INPTS(KS,IB,NX,II,NY,D,T)
DIMENSION D(41,1),T(1)
DO 10 I=1,NX
  KOUT = 4*I-4+KS-IB
  CALL INTOUT(U,KOUT,NY,1,41,I)
10 CONTINUE
  IF (KS .LT. 5) GO TO 30
  DO 20 I=1,NX
    KOUT = 4*I-2
    CALL INTOUT(T,KOUT,NY,1,1,1)
  DO 20 J=1,NY
    TMP = 0.0
    IF (T(J) .LE. 0.0.OR.D(I,J) .LE. 0.0) GO TO 20
    TMP = T(J)/D(I,J)
  20 U(I,J) = TMP
  30 RETURN
  END

```

INP00100
INP00200
INP00300
INP00400
INP00500
INP00600
INP00700
INP00800
INP00900
INP01000
INP01100
INP01200
INP01300
INP01400
INP01500
INP01600
INP01700

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15*
16*

```

SUBROUTINE INTOUT(D,I,N,L,IDM,K)
DIMENSION D(IDM,1)
INTEGER RECLTH
IF (ISP .NE. 0) GO TO 10
ISP = 1
RECLTH = 41
DEFINE FILE 12(164,RECLTH,U,KOUK)
10 CONTINUE
  KOUK = I
  60 FORMAT (1X,8I6)
  GO TO (20,30),L
  20 READ (12,KOUK) (U(K,J),J=1,N)
  GO TO 40
  30 WRITE (12,KOUK) (D(K,J),J=1,N)
  40 RETURN
  END

```

INT00100
INT00200
INT00300
INT00400
INT00500
INT00600
INT00700
INT00800
INT00900
INT01000
INT01100
INT01200
INT01300
INT01400
INT01500
INT01600

APPENDIX D
PROGRAM OUTPUT FOR THE PREPROCESSOR PROGRAM
AND THE NASA/MSFC MULTILAYER DIFFUSION
MODELS PROGRAM -- VERSION 5

D.1 PREPROCESSOR PROGRAM EXAMPLE OUTPUT

The Preprocessor Program produces both printed output and punched data-card output. The Program first prints a detailed list of all inputs including the constants pertaining to the rocket vehicle for which calculations are being made. The first page of the example output listing below shows the constants used for the example problem described in Section 5 of the main text for a simulated normal launch of a Space Shuttle vehicle at Kennedy Space Center on 21 October 1972. The Program next prints the FORTRAN NAMELIST data which are subsequently punched on cards for Model 4 calculations using the Multilayer Diffusion Models Program. In this example, the first printed list and punched card deck are for calculation of HCl concentrations and dosages; the second list and punched card deck are for CO concentrations and dosages; the third list and punched card deck are for Al_2O_3 concentrations and dosages. A printed list and punched card deck for the calculation of CO_2 concentrations and dosages are not produced for the Space Shuttle vehicle. The Program output for the Model 4 calculations ends with a summary table of the Program calculations. The Program then produces similar printed listings and sets of punched card decks in the same format for use in the Model 3 calculations.

***** NORMAL LAUNCHSPACE SHUTTLE VEHICLE

*** INITIALIZED DATA USED FOR ABOVE VEHICLE ***

GC - RATE OF OUTPUT OF EXHAUST MATERIAL FROM VEHICLE IN GRAMS/SEC IS 9.38498400+06
 GT - TOTAL AMOUNT OF VEHICLE EXHAUST MATERIAL IN GRAMS IS 9.15626984+08
 A AND B - VEHICLE RISE PARAMETERS IN THE EQUATION $TR=A*Z**B$ ARE .663552 AND .485477
 HEAT - TOTAL HEAT OUTPUT IN CALORIES/GRAM IS 2582.0000
 GAMMA - ENTRAINMENT PARAMETER IS .6400
 CP - SPECIFIC HEAT OF AIR IS .240
 POLLUTANT MATERIALS ARE HCL, CO, CO2, AL2O3,
 FRACTIONAL DISTRIBUTION OF THE ABOVE MATERIALS IS .207, .280, .000, .304,
 MOLECULAR WEIGHT OF THE ABOVE MATERIALS IS 36.460, 28.010, 44.010,

**-X PROGRAM INPUT DATA **

DATA CARD 1

TITLE - KSC 21 OCT 72.

NORMAL - IS LAUNCH NORMAL ? YES

IFEET - ARE LAYER BOUNDARY HEIGHTS Z, AND HM IN FEET? NO

KNOTS - IS THE WIND SPEED WS IN KNOTS? NO

SIGAR - STANDARD DEVIATION OF THE AZIMUTH WIND ANGLE IN DEGREES IS 9.000

RHO - AIR DENSITY IN GRAMS/CUBIC METER IS 1197.070

HM - HEIGHT OF SURFACE MIXING LAYER IS 1432.000

DATA CARD 2 THROUGH 20

LAYER	BOUNDARY Z (METERS)	WIND DIRECTION WD (DEG)	WIND SPEED WS (MET/S)	TEMPERATURE T (DEG C)	PRESSURE P (MB)	RH (PERCENT)
1	18.000	80.0000	6.0000	22.600	1022.000	57.000
2	194.000	81.0000	8.0000	22.200	1000.000	57.000
3	250.000	82.0000	9.0000	22.100	993.660	57.000
4	284.000	82.0000	10.0000	22.000	989.790	58.000
5	500.000	79.0000	10.0000	19.300	965.400	65.000
6	558.000	79.0000	10.0000	18.500	958.940	67.000
7	637.000	78.0000	10.0000	17.800	950.000	70.000
8	750.000	76.0000	11.0000	16.900	937.710	74.000
9	1000.000	71.0000	11.0000	14.600	910.610	86.000
10	1098.000	68.0000	11.0000	13.700	900.000	93.000
11	1135.000	67.0000	11.0000	13.300	896.250	94.000
12	1250.000	63.0000	11.0000	12.300	884.090	97.000
13	1432.000	56.0000	11.0000	10.700	865.130	97.000
14	1500.000	53.0000	10.0000	10.500	858.170	90.000
15	1577.000	49.0000	10.0000	10.300	850.000	79.000
16	1716.000	40.0000	9.0000	9.900	836.310	55.000
17	1750.000	37.0000	8.0000	10.300	832.310	55.000
18	2000.000	9.0000	6.0000	12.500	808.340	49.000
19	2259.000	344.0000	5.0000	11.100	783.690	44.000
20	2500.000	342.0000	5.0000	9.100	761.390	54.000

***** EXAMPLE SPACE SHUTTLE NORMAL LAUNCH

 NORMAL LAUNCH SPACE SHUTTLE

VEHICLE

*** CALCULATED PARAMETERS FOR MODEL 4 ***

H - ADJUSTED CLOUD HEIGHT IN METERS IS 1790.000

ZM - REAL CLOUD HEIGHT IN METERS IS 1790.000

TAUK - TIME TO CLOUD STABILIZATION IN SEC IS 447.273

DPDZ - VERTICAL GRADIENT OF AMBIENT POTENTIAL TEMP IN DEGREES K/METER IS .001489

JBOT - BOTTOM LAYER FOR USE WITH MODEL 4 IS 1

JTOP - TOP LAYER FOR USE WITH MODEL 4 IS 12

Z - BOUNDARY HEIGHT AT THE BOTTOM OF LAYER 1 IN METERS IS .000

SIGAP - STANDARD DEVIATION OF THE WIND AZIMUTH ANGLE AT THE MEASUREMENT HEIGHT ZRKE= 18.00 METERS IS 4.500

SIGEP - STANDARD DEVIATION OF THE WIND ELEVATION ANGLE AT ZRK IS 4.500

LAYER PARAMETERS -

LAYER NO.	LAYER Z (LAYER TOP)	HCL	CO	CO2	SIGAP (DEG)	SIGEP (DEG)	SIGXO (METER)	SIGYO (METER)	SIGZO (METER)	DELX (METER)	DELY (DEG)
1	194.000	4.4227630+07	7.7872643+07	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	39.92	260.50
2	250.000	1.7876125+07	3.1474920+07	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	67.18	260.91
3	284.000	1.3832882+07	2.4355885+07	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	88.49	261.17
4	500.000	1.7371957+08	3.0587220+08	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	279.08	260.71
5	558.000	8.5352765+07	1.5028265+08	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	343.02	260.39
6	637.000	1.5651768+08	2.7558477+08	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	438.96	259.98
7	750.000	3.3045618+08	5.8184208+08	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	601.04	259.18
8	1000.000	1.4067362+09	2.4768740+09	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	1053.55	256.73
9	1098.000	9.0119715+08	1.5867593+09	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	1259.95	255.54
10	1135.000	4.0207076+08	7.0793558+08	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	1340.74	255.06
11	1250.000	1.4819777+09	2.6095335+09	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	1610.33	253.36
12	1432.000	3.0778224+09	5.4191954+09	0.0000000	4.5000	4.5000	532.8372	532.8372	.0000	2084.00	250.17
13	1500.000	1.3646208+09	2.4027204+09	0.0000000	.1000	.1000	532.8372	532.8372	.0000	2278.11	248.80
14	1577.000	1.6626373+09	2.9274452+09	0.0000000	.1000	.1000	532.8372	532.8372	.0000	2516.03	247.07
15	1716.000	3.2292515+09	5.6858202+09	0.0000000	.1000	.1000	532.8372	532.8372	.0000	3001.70	243.27
16	1750.000	8.1639239+08	1.4374416+09	0.0000000	.1000	.1000	532.8372	532.8372	.0000	3246.75	241.29
17	2000.000	7.6941059+09	1.3547196+10	0.0000000	.1000	.1000	507.5349	507.5349	.0000	4103.23	235.46
18	2259.000	7.1501190+09	1.2569385+10	0.0000000	.1000	.1000	431.7767	431.7767	.0000	2460.00	176.50
19	2500.000	4.9842630+09	8.7759108+09	0.0000000	.1000	.1000	357.3581	357.3581	.0000	2236.36	163.00

***** EXAMPLE SPACE SHUTTLE NORMAL LAUNCH
 ***** NORMAL LAUNCH SPACE SHUTTLE VEHICLE *****

*** NAMELIST NAME2 FOR INPUT TO THE NASA/MSFC MULTILAYER MODEL VERSION 5 ***

```

>NAME2
TESTNO=60HKSC 21 OCT 72.
NAMCAS=6BH EXAMPLE SPACE SHUTTLE NORMAL LAUNCH
ISLIP=0,3,3,0,1,2,0,0,0,0,NPSE=0,NLSE=2,NUI=69,NCI=69,NTI=62,ZRK=18.0,
TAUK=447.273,IZNUJ=3,
Z= .000, 1432.000,
U= 8.0917873+09,
UBARKE 6.000, 11.000,
SIGAKE 4.500, 4.500,
SIGEKE 4.500, 4.500,
SIGXOE 532.837,
SIGYOE 532.837,
SIGZOE 183.163,
THETAKE 80.000, 56.000,
TIMEAVE 600.0,
DI= 400.000, 200.000, 100.000, 50.000, 25.000, 5.000,
CI= 16.000, 6.000, 4.000, 1.000, .500, .100,
TI= 30.000, 4.000, 8.000, 2.000, 1.000, .500,
DELXE 4103.229,
DELYE 235.455,
TEMPKE 295.624, 297.076,
H= 1038.200,
$END
    
```

***** EXAMPLE SPACE SHUTTLE NORMAL LAUNCH *****
***** NORMAL LAUNCH SPACE SHUTTLE VEHICLE *****

*** NAMELIST NAME2 FOR INPUT TO THE NASA/MSFC MULTILAYER MODEL VERSION 5 ***

\$NAME2
TESTNO=60HKSC 21 OCT 72.
NAMEAS=6BH EXAMPLE SPACE SHUTTLE NORMAL LAUNCH SPACE SHUTTLE
ISKIP=0,3,3,0,2,2,0,0,0,0,NPS= 0,NCS= 2,NDI=69,NCI=69,NTI=61,ZRK= 18.0,
TAUK= 447.273,12MOD=3,
Z= .000, 1432.000,
Q= 1.4247403+10,
UBARK= 6.000, 11.000,
SIGAK= 4.500, 4.500,
SIGEK= 4.500, 4.500,
SIGEX= 532.837,
SIGXO= 532.837,
SIGYO= 183.163,
SIGZO= 80.000, 56.000,
THETAK= 360.0,
TIMAV= 400.000, 200.000, 100.000, 50.000, 25.000, 5.000,
DI= 35.000, 10.000, 4.000, 2.000, 1.000, .100,
CI= .150,000, 100.000, 60.000, 30.000, 15.000, 1.000,
DELX= 4103.229,
DELY= 235.455,
TEMPK= 295.624, 297.076,
HE= 1038.200,
\$END

***** EXAMPLE SPACE SHUTTLE NORMAL LAUNCH *****

*** NAMELIST NAME2 FOR INPUT TO THE NASA/MSFC MULTILAYER MODEL VERSION 5 ***

NAME2
 TESTNU=60HKSC 21 OCT 72, SPACE SHUTTLE
 NAMECAS=68H EXAMPLE SPACE SHUTTLE NORMAL LAUNCH
 ISKIP=0,3,3,0,4,2,0,0,0,0,0,NPSE=0,NZS=2,HDI=69,NCI=69,NTI=61,ZRK=18.0,
 TAUKE=447.273,IZMOD=3,
 Z= .000, 1432.000,
 Q= 1.8019791+10,
 UBAKE= 6.000, 11.000,
 SIGAK= 4.500, 4.500,
 SIGEK= 4.500, 4.500,
 SIGXO= 532.837,
 SIGYO= 532.837,
 SIGZO= 183.163,
 THETAKE= 80.000, 56.000,
 TIMAV= 600.0,
 DI= 40.000, 20.000, 10.000, 5.000, 2.500, .500,
 CI= 2.000, 1.000, .400, .100, .050, .010,
 TI= 50.000, 100.000, 25.000, 10.000, 5.000, 1.000,
 DELX= 4103.229,
 DELY= 235.455,
 TEMPK= 295.624, 297.076,
 HE= 1038.200,
 SEND

CLOUD RISE AND SOURCE DISTRIBUTION NASA/MSFC MULTILAYER MODELS V-5

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DATE 07/16/75

*** CALCULATED PARAMETERS FOR MODEL 3 ***

H - ADJUSTED CLOUD HEIGHT IN METERS IS 1038.200

ZM - REAL CLOUD HEIGHT IN METERS IS 1790.000

TAUK - TIME TO CLOUD STABILIZATION IN SEC IS 447.273

DPVZ - VERTICAL GRADIENT OF AMBIENT POTENTIAL TEMP IN DEGREES K/METER IS .001489

JBOT - BOTTOM LAYER FOR USE WITH MODEL 4 IS 1

JTOP - TOP LAYER FOR USE WITH MODEL 4 IS 12

Z - BOUNDARY HEIGHT AT THE BOTTOM OF LAYER 1 IN METERS IS .000

SIGAP - STANDARD DEVIATION OF THE WIND AZIMUTH ANGLE AT THE MEASUREMENT HEIGHT ZRK= 18.00 METERS IS 4.500

SIGEP - STANDARD DEVIATION OF THE WIND ELEVATION ANGLE AT ZRK IS 4.500

LAYER PARAMETERS -

LAYER NO. (LAYER TOP)

HCL

CU

- SOURCE STRENGTH Q -

CO2

AL203

SIGAP (DEG)

SIGEP (DEG)

SIGXO (METER)

SIGYO (METER)

SIGZO (METER)

DELX (METER)

DELY (DEG)

1	1432.000	8.0917873+09	1.4247403+10	0.0000000	1.8019791+10	4.5000	4.5000	532.8372	532.8372	183.1628	4103.23	235.46
---	----------	--------------	--------------	-----------	--------------	--------	--------	----------	----------	----------	---------	--------

D.2 NASA/MSFC MULTILAYER DIFFUSION MODELS PROGRAM EXAMPLE OUTPUT

This section contains a listing of the output produced by the NASA/MSFC Multilayer Diffusion Models Program using the example input data shown in Figure B-1, Appendix B and that portion of the data deck produced by the Preprocessor Program for HCl concentrations for a normal launch of a Space Shuttle vehicle. The input data shown in Figure B-1 instructed the Multilayer Diffusion Models Program to use Models 3 and 4 to calculate maximum centerline HCl concentrations, dosages and time-mean concentrations and to calculate the corresponding isopleths.

As shown below, the Program first lists a title page containing the data set title, the meteorological case, date and time, and the height, range and azimuth bearing of the stabilized cloud. Next, the Program prints the data parameters for each layer and the results of the parameter calculations (for example, mean wind speed, wind speed shears, etc.) in each layer. The inputs for Layer 1 are shown as the second page of the example listing. This page is followed by a list of maximum centerline concentrations, dosages, time-mean concentrations, the time of cloud passage and average concentration at various distances (ranges) along the cloud trajectory (azimuth bearing). The Program then produces a series of pages containing printer plots on log-log scales of the maximum centerline concentrations, dosages and time-mean concentrations at selected distances along the cloud trajectory.

After the printer plots are generated, the Program lists the range and azimuth bearing from the launch pad to selected isopleths of concentration, dosage and time-mean concentration. The printed ranges and azimuth bearings are ordered such that the output values begin at a range close to the launch pad and move in an azimuth direction clockwise around the isopleth. The listing for the Model 4 calculations ends with a printout of model input parameters for the layers that con-

tributed to the ground-level concentrations and dosages previously listed. The computer Program then lists similar output for the Model 3 calculations except that model input parameters are given only for the sample layer used in the calculations.

The NASA/MSFC Multilayer Diffusion Models Program also contains provision for generating plots of maximum centerline concentrations, dosages, time-mean concentrations and isopleths of these parameters. Figures D-1 and D-2 respectively show computer plots of the Model 3 and Model 4 calculations of maximum centerline concentrations for the example case. Figures D-3 and D-4 respectively show plots of Model 3 and Model 4 HCl dosage isopleths for the example case.

```
*****
*
*   EXAMPLE SPACE SHUTTLE NORMAL LAUNCH
*
*   KSC 21 OCT 72.
*
*   DATE = 07/17/75, TIME = 16/07/48
*
*   ADJUSTED CLOUD RISE HEIGHT = 1790.00, RANGE = 2084.00, AZIMUTH BEARING = 250.17
*
*****
```


***** LAYER 1 *****

** INPUT DATA **

W= .4427627+08, ZR= 14.000, UBAR AT BOTTOM= 6.0000, SIGAK AT TOP= 8.0000, SIGAK AT BOTTOM= 4.50000
 SIGAK AT TOP= 4.50000, SIGEK AT BOTTOM= 4.50000, TAU= 447.273, TAUOK= 447.273
 SIGX= 532.8370, SIGY= 532.8370, SIGZ= .0000, THETA AT BOTTOM= 80.000, THETA AT TOP= 81.000, Z= .000
 ALPHA= 1.0 BETA= 1.0, H= 1790.000, DELX= .7992200+02, DELY= .2605000+03, IZMODE= 4, TIM1= .00000000
 ZLIM= .000, LAMBDA= .0000, TIME= 600.000, XRY= 100.000, XLRZ= .000, GAMMA= .000

ZRL= 18.000, UBARL AT BOTTOM= 6.0000, UBARL AT TOP= 11.0000, SIGAL AT BOTTOM= 4.50000, SIGAL AT TOP= 4.50000
 SIGEL AT BOTTOM= 4.50000, SIGEL AT TOP= 4.50000, THETA AT BOTTOM= 80.000, THETA AT TOP= 56.000, TAU= 447.273
 TAUOL= 447.273, ALPHA= 1.0, BLT= 1.0, TAST= .00000000, JHOT= 1, JTOP= 12

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** UBAR = 7.31893, THETA = 80.50000, DELTHP = 1.00000, DELU = 2.00000
 , SIGAP = .07854, SIGEP = .07854

CALCULATED INPUT PARAMETERS FOR LAYER CHANGE MODEL 4 *** UBAR = 10.03876, THETA = 68.00000, DELTHP = -24.00000
 DELU = 5.00000, SIGAP = .07854, SIGEP = .07854

MAXIMUM CENTERLINE HCL CALCULATIONS AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE
NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21
OCT 72.

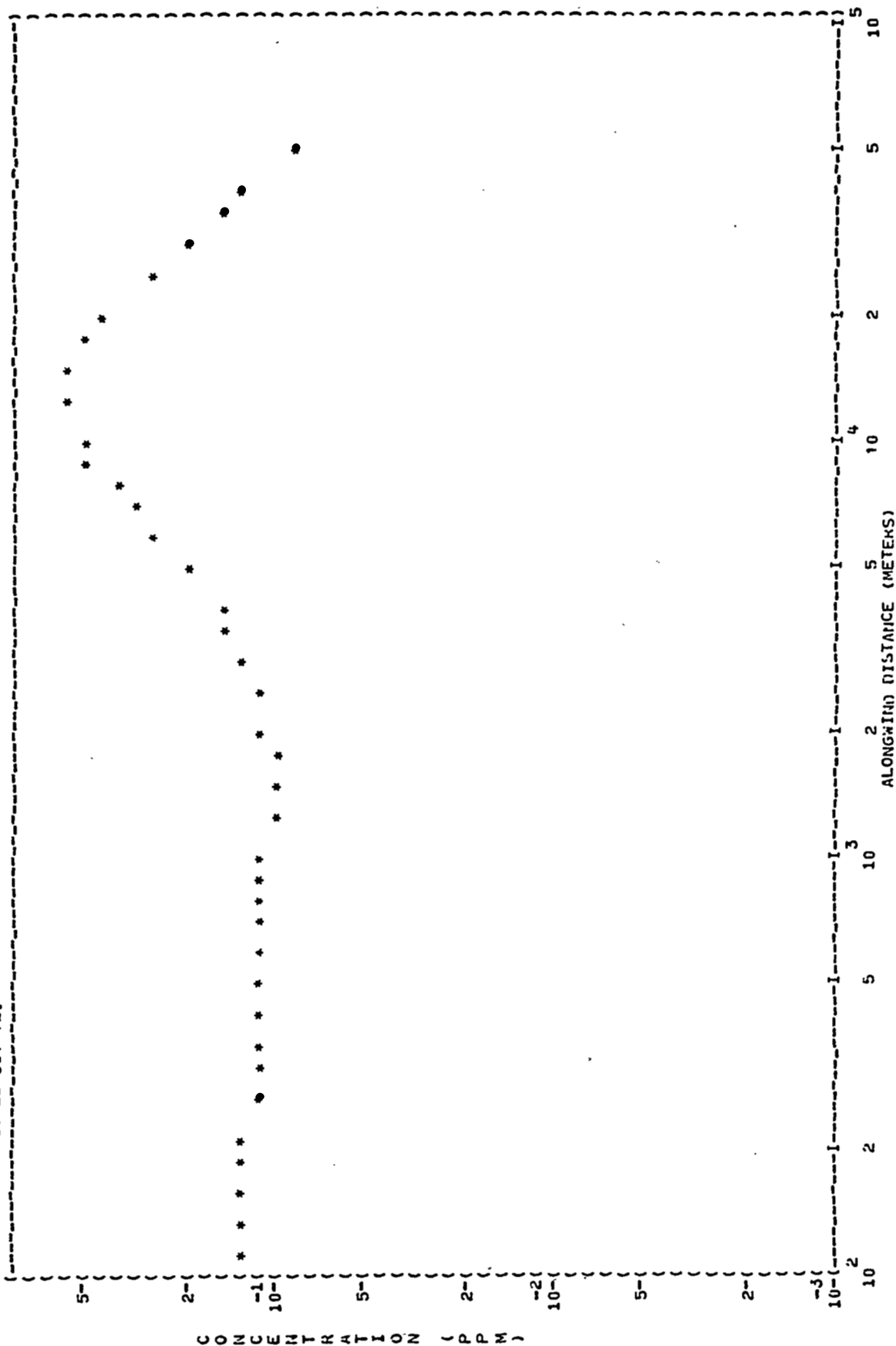
.597 IS THE MAXIMUM PEAK CONCENTRATION

97.569 IS THE MAXIMUM DOSAGE

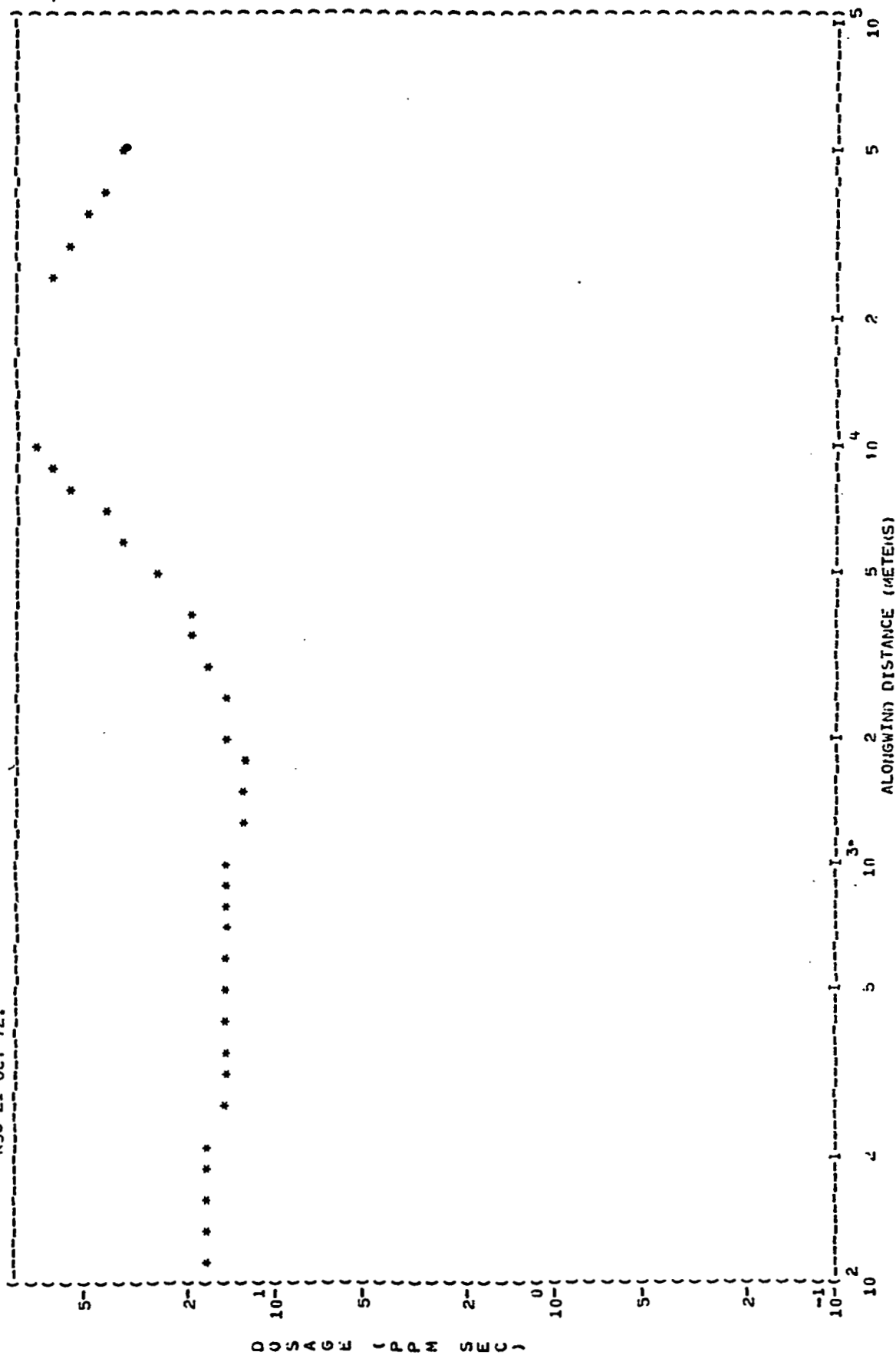
.163 IS THE MAXIMUM PEAK 10.0 MINUTE TIME-MEAN CONCENTRATION

RANGE (METERS)	AZIMUTH BEARING (DEGREES)	MAXIMUM PEAK CONCENTRATION (PPM)	MAXIMUM DOSAGE (PPM SEC)	10.0 MINUTE TIME-MEAN CONCENTRATION (PPM)	TIME OF CLOUD PASSAGE (SECONDS)	AVERAGE CLOUD CONCENTRATION (PPM)
50.0	284.2	.128	16.322	.027	228.235	.072
100.0	256.6	.127	16.852	.028	228.235	.074
125.0	254.1	.126	16.789	.028	228.236	.074
150.0	252.7	.126	16.727	.028	228.237	.073
175.0	251.8	.125	16.665	.028	228.239	.073
200.0	251.2	.125	16.603	.028	228.241	.073
250.0	250.4	.124	16.479	.027	228.247	.072
300.0	250.0	.123	16.356	.027	228.236	.072
350.0	249.7	.122	16.234	.027	228.235	.071
400.0	249.4	.121	16.113	.027	228.237	.071
500.0	249.1	.119	15.872	.026	228.237	.070
600.0	248.9	.117	15.634	.026	228.235	.069
700.0	248.8	.116	15.400	.026	228.240	.067
800.0	248.7	.114	15.168	.025	228.254	.066
900.0	248.6	.112	14.945	.025	228.276	.065
1000.0	248.5	.111	14.738	.025	228.307	.065
1250.0	248.4	.107	14.309	.024	228.232	.063
1500.0	248.4	.105	14.047	.023	228.247	.062
1750.0	248.4	.105	14.069	.023	228.244	.062
2000.0	248.5	.106	14.420	.024	228.302	.063
2500.0	248.6	.117	15.785	.026	228.309	.069
3000.0	248.7	.130	17.553	.029	228.591	.077
3500.0	248.7	.144	19.559	.033	229.083	.085
4000.0	248.8	.160	21.816	.038	229.784	.095
5000.0	248.9	.201	27.765	.048	231.605	.120
6000.0	249.0	.256	35.829	.060	234.631	.153
7000.0	249.0	.318	45.247	.075	238.235	.190
8000.0	248.9	.387	55.976	.093	242.584	.231
9000.0	248.8	.453	67.608	.113	247.637	.273
10000.0	248.7	.522	78.636	.131	253.351	.310
12500.0	248.6	.597	95.777	.160	270.245	.354
15000.0	248.5	.567	97.569	.163	290.329	.336
17500.0	248.4	.492	91.376	.152	312.989	.292
20000.0	248.3	.415	82.976	.138	337.707	.246
25000.0	248.3	.293	68.043	.113	391.725	.174
30000.0	248.2	.215	57.279	.095	450.003	.127
35000.0	248.2	.164	49.428	.081	511.082	.097
40000.0	248.2	.128	43.461	.071	574.070	.076
50000.0	248.1	.084	35.005	.054	703.713	.050

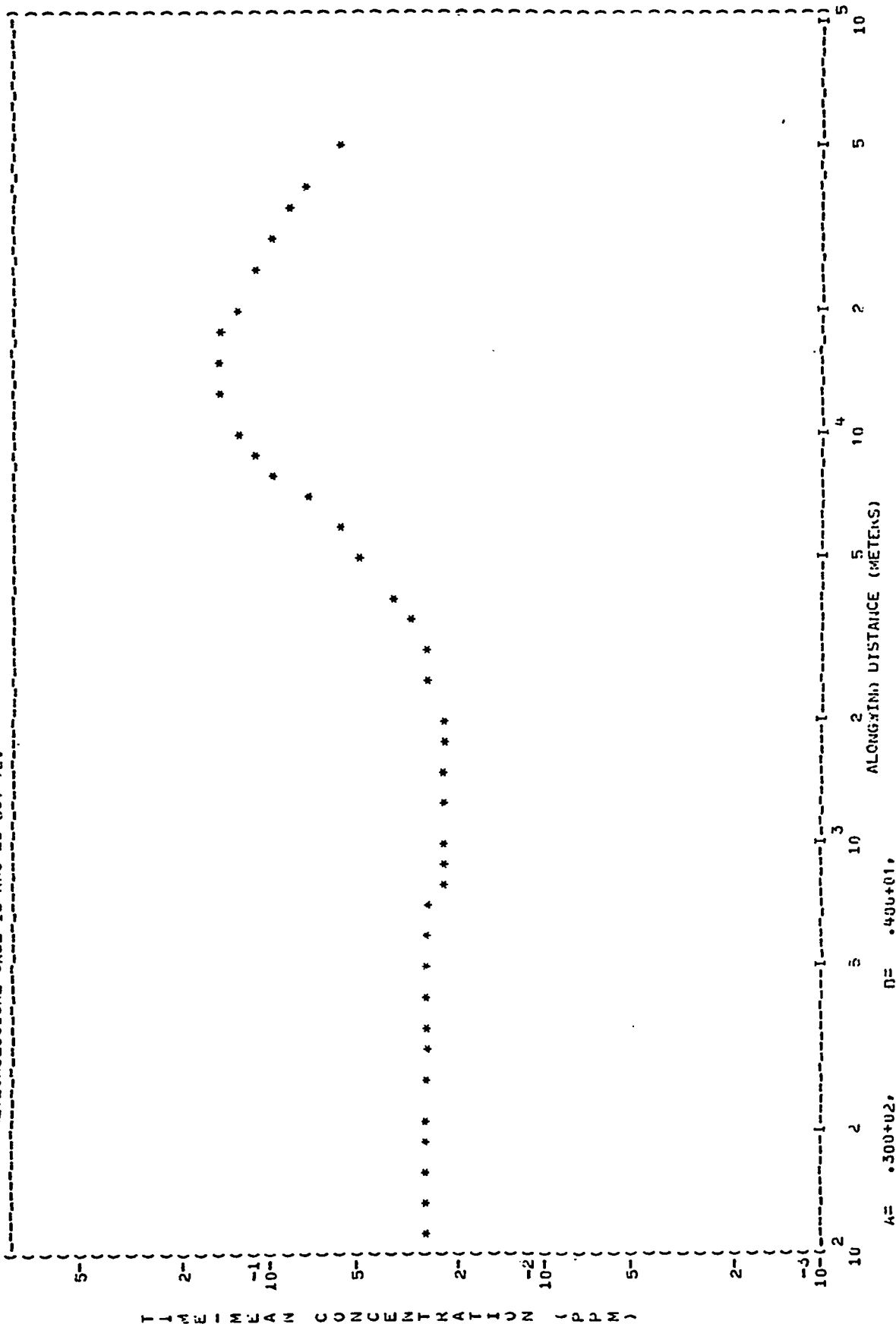
MAXIMUM CENTERLINE HCL CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21 OCT 72.



MAXIMUM CENTERLINE HCL DOSAGE IN PPM SEC AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21 OCT 72.



MAXIMUM CENTERLINE HCL 10 MINUTE TIME-MEAN CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS
 DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE
 METEOROLOGICAL CASE IS KSC 21 OCT 72.



ISOPLETHS incl CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS DOWNING FROM A SPACE SHUTTLE
 HORIZONTAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21
 OCT 72.

*** ISOPLETH LEVEL = .500, CONCENTRATION (PPM)									
RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)
9740.952	247.500	10000.000	246.548	10972.021	245.000	12500.000	243.857	15560.100	245.000
17131.179	247.500	10898.565	250.000	15000.000	252.396	14810.763	252.500	11233.255	252.500
10000.000	250.869	9749.320	250.000						
*** ISOPLETH LEVEL = .100, CONCENTRATION (PPM)									
50.001	209.461	923.045	230.000	3000.001	237.762	3500.001	237.007	5000.001	235.655
5977.467	235.000	7000.000	234.450	8000.001	234.037	9000.000	233.702	12500.000	233.384
15000.000	233.803	17500.000	234.559	18707.787	235.000	25000.000	237.121	30000.000	238.864
39299.433	242.500	44325.712	245.000	46367.489	247.500	45756.911	250.000	42311.098	252.500
30196.775	257.500	23030.000	259.443	20000.000	261.358	17500.000	262.210	15000.000	263.076
10000.000	263.921	9000.000	263.836	8000.000	263.654	7000.000	263.355	6000.000	262.829
5000.000	262.012	4000.000	260.992	3500.000	260.386	3000.000	259.514	2500.001	258.142

ISOPLETHS INCL USAGE IN PPM SEC AT A HEIGHT OF 0 METERS DOWNING FROM A SPACE SHUTTLE
NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21
OCT 72.

*** ISOPLETH LEVEL = 50.000, DOSAGE (PPM SEC)

RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)
7891.711	245.000	6776.813	242.500	10000.000	240.757	11251.232	240.000
17500.000	239.873	17936.565	240.000	20000.000	240.621	25000.000	242.325
33719.494	250.000	30000.000	252.064	25000.000	254.182	23184.092	255.000
15148.421	257.500	13000.000	257.530	12500.000	257.700	11950.255	257.500
8300.000	253.050	7475.542	250.000				

*** ISOPLETH LEVEL = 25.000, DOSAGE (PPM SEC)

5000.001	243.883	7000.001	238.583	9000.001	236.600	12500.001	235.436
20000.000	230.434	24501.625	237.500	25000.000	237.599	30000.000	238.552
40000.000	240.380	50000.000	242.096	64543.615	245.000	71020.357	247.500
50000.000	254.147	40035.337	255.000	40000.000	255.983	34999.999	256.900
24999.999	259.009	20763.984	260.000	20000.000	260.192	17500.000	260.881
10000.000	261.375	6000.001	260.249	6000.000	257.553	4753.132	252.500

D-20

*** ISOPLETH LEVEL = 5.000, DOSAGE (PPM SEC)

1191.870	200.000	3000.001	224.765	4000.001	227.024	5000.001	228.148
8000.001	229.104	9000.001	229.156	10000.001	229.164	12500.000	229.268
17500.000	230.077	20000.000	230.408	25000.000	231.123	30000.001	231.836
50000.000	233.657	60000.000	234.758	79999.998	260.630	50000.000	262.468
30000.000	264.590	26636.448	265.000	24999.999	265.300	20000.000	266.355
13401.642	267.500	12500.000	267.726	10000.000	268.180	9000.000	268.335
6000.000	268.960	5000.000	269.362	4250.778	270.000	3500.000	271.184

***** LAYER 2 *****

** INPUT DATA **

U = .1706124+08, URAR AT BOTTOM = 8.0000, URAR AT TOP = 9.0000, SIGAK AT BOTTOM = 4.50000, SIGAK AT TOP = 4.50000
SIGAK AT BOTTOM = 4.50000, SIGAK AT TOP = 4.50000, SIGXO = 532.8170, SIGYO = 532.8370, SIGZO = .0000, THETAK AT BOTTOM = 81.000
THETAK AT TOP = 82.000, Z = 194.000, ALPHA = 1.0, BETA = 1.0, DELXE = .67179000+02, DELYE = .26090600+03
LMODE = 4

CALCULATE INPUT PARAMETERS FOR MODEL 1.2.1 *** URAR = 8.50000, THETA = 81.50000, DELTHP = 1.00000, DELU = 1.00000
, SIGAP = .07834, SIGEP = .07854

***** LAYER 3 *****

** INPUT DATA **

U = .1302881+08, URAR AT BOTTOM = 9.0000, URAR AT TOP = 10.0000, SIGAK AT BOTTOM = 4.50000, SIGAK AT TOP = 4.50000
SIGAK AT BOTTOM = 4.50000, SIGAK AT TOP = 4.50000, SIGXO = 532.8170, SIGYO = 532.8370, SIGZO = .0000, THETAK AT BOTTOM = 82.000
THETAK AT TOP = 82.000, Z = 200.000, ALPHA = 1.0, BETA = 1.0, DELXE = .88489000+02, DELYE = .26116900+03

LZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 **** URAR = 9.50000, THETA = 82.00000, DELTHP = .00000, DELU = 1.00000
 , SIGAP = .07854, SIGEP = .07854

***** LAYER 4 *****

** INPUT DATA **

Q= .17371956+09, UBAR AT BOTTOM= 10.00000, URAR AT TOP= 10.00000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
 SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 82.000
 THETAK AT TOP= 79.0000, Z= 234.000, ALPHA= 1.0, RETA= 1.0, DELX= .27907700+03, DELY= .26071200+03
 LZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 **** URAR = 10.00000, THETA = 80.50000, DELTHP = -3.00000, DELU = .00000
 , SIGAP = .07854, SIGEP = .07854

***** LAYER 5 *****

** INPUT DATA **

Q= .85352756+08, UBAR AT BOTTOM= 10.00000, URAR AT TOP= 10.00000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
 SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 79.000
 THETAK AT TOP= 79.0000, Z= 500.000, ALPHA= 1.0, RETA= 1.0, DELX= .34302300+03, DELY= .26039300+03
 LZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 **** URAR = 10.00000, THETA = 79.00000, DELTHP = .00000, DELU = .00000
 , SIGAP = .07854, SIGEP = .07854

***** LAYER 6 *****

** INPUT DATA **

Q= .150051787+09, UBAR AT BOTTOM= 10.00000, URAR AT TOP= 10.00000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
 SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 79.000
 THETAK AT TOP= 73.0000, Z= 558.000, ALPHA= 1.0, RETA= 1.0, DELX= .43896500+03, DELY= .25997900+03
 LZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 **** URAR = 10.00000, THETA = 78.50000, DELTHP = -1.00000, DELU = .00000
 , SIGAP = .07854, SIGEP = .07854

***** LAYER 7 *****

** INPUT DATA **

Q= .33045615+09, UBAR AT BOTTOM= 10.00000, URAR AT TOP= 11.00000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
 SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 78.000
 THETAK AT TOP= 70.0000, Z= 637.000, ALPHA= 1.0, RETA= 1.0, DELX= .60103300+03, DELY= .25917500+03
 LZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** URAR = 10.50000, THETA = 77.00000, DELTHP = -2.00000, DELU = 1.00000
SIGAP = .07854, SIGEP = .07854

***** LAYER 8 *****
** INPUT DATA **

U= .14007361+10, UBAR AT BOTTOM= 11.0000, UBAR AT TOP= 4.50000, SIGAK AT TOP= 4.50000
SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 76.0000
THETAK AT TOP= 71.0000, Z= 750.000, ALPHA= 1.0, BETA= 1.0, DELX= .10535510+04, DELY= .25673400+03
IZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** URAR = 11.00000, THETA = 73.50000, DELTHP = -5.00000, DELU = .00000
SIGAP = .07854, SIGEP = .07854

***** LAYER 9 *****
** INPUT DATA **

U= .90119700+09, UBAR AT BOTTOM= 11.0000, UBAR AT TOP= 11.0000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 71.0000
THETAK AT TOP= 68.0000, Z= 1000.000, ALPHA= 1.0, BETA= 1.0, DELX= .12599460+04, DELY= .25554400+03
IZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** URAR = 11.00000, THETA = 69.50000, DELTHP = -3.00000, DELU = .00000
SIGAP = .07854, SIGEP = .07854

***** LAYER 10 *****
** INPUT DATA **

U= .40007074+09, UBAR AT BOTTOM= 11.0000, UBAR AT TOP= 11.0000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 68.0000
THETAK AT TOP= 67.0000, Z= 1076.000, ALPHA= 1.0, BETA= 1.0, DELX= .13407370+04, DELY= .25505700+03
IZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** URAR = 11.00000, THETA = 67.50000, DELTHP = -1.00000, DELU = .00000
SIGAP = .07854, SIGEP = .07854

***** LAYER 11 *****
** INPUT DATA **

U= .14019776+10, UBAR AT BOTTOM= 11.0000, UBAR AT TOP= 11.0000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 67.0000
THETAK AT TOP= 63.0000, Z= 1125.000, ALPHA= 1.0, BETA= 1.0, DELX= .10103340+04, DELY= .25336000+03
IZMODE= 4

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** URAR = 11.00000, THETA = 65.00000, DELTHP = -4.00000, DELU = .00000

, SIGAP = .07854, SIGEP = .07854

***** LAYER12 *****

** INPUT DATA **

Q= .30778222+10, UBAR AT BOTTOM= 11.0000, UBAR AT TOP= 11.0000, SIGAK AT BOTTOM= 4.50000, SIGAK AT TOP= 4.50000
 SIGEK AT BOTTOM= 4.50000, SIGLK AT TOP= 4.50000, SIGXO= 532.8370, SIGYO= 532.8370, SIGZO= .0000, THETAK AT BOTTOM= 63.000
 THETAK AT TOP= 56.000, Z= 1250.000, ALPHA= 1.0, BETA= 1.0, DELX= .20839970+04, DELY= .25016700+03
 IZMODE= 4
 Z AT TOP= 1432.0000

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** UBAR = 11.00000, THETA = 59.50000, DELTHP = -7.00000, DELU = .00000
 , SIGAP = .07854, SIGEP = .07854

```

*****
*
*   EXAMPLE SPACE SHUTTLE NORMAL LAUNCH
*
*
*   KSC 21 OCT 72.
*
*   DATE = 07/17/75, TIME = 16/14/18
*
*
*   ADJUSTED CLOUD H1SF HEIGHT = 1038.20, RANGE = 4103.23, AZIMUTH BEARING = 235.45
*
*****

```

***** LAYER 1 *****

** INPUT DATA **

W= .80917866+10, ZRK= 1.4.000, USAR AT BOTTOM= 6.0000, SIGAK AT TOP= 11.0000, SIGAK AT BOTTOM= 4.50000
 SIGAK AT TOP= 4.50000, SIGEK AT BOTTOM= 4.50000, SIGEK AT TOP= 4.50000, TAUKE= 447.273, TAUOK= 447.273
 SIGXO= 524.8370, SIGYU= 532.8370, SIGZO= 183.1630, THETAK AT BOTTOM= 80.000, THETAK AT TOP= 56.000, Z= .000
 ALPHA= 1.0, BETA= 1.0, I= 1038.200, DELX= .41032290+04, DELY= .23545500+03, IZMOD= 3, TIMI= .00000000
 ELIME= .000, LAMBDA= .000, TMAV= 600.000, XRY= 100.000, XRPY= .000, XLRZ= .000, GAMMAPE= .000
 Z AT TOP= 1432.0000

CALCULATED INPUT PARAMETERS FOR MODELS 1,2,3 *** USAR = 9.71775, THETA = 68.00000, DELTHP = -24.00000, DELU = 5.00000
 , SIGAP = .07854, SIGEP = .07854

MAXIMUM CENTERLINE HCL CALCULATIONS AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE
NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21
001 72.

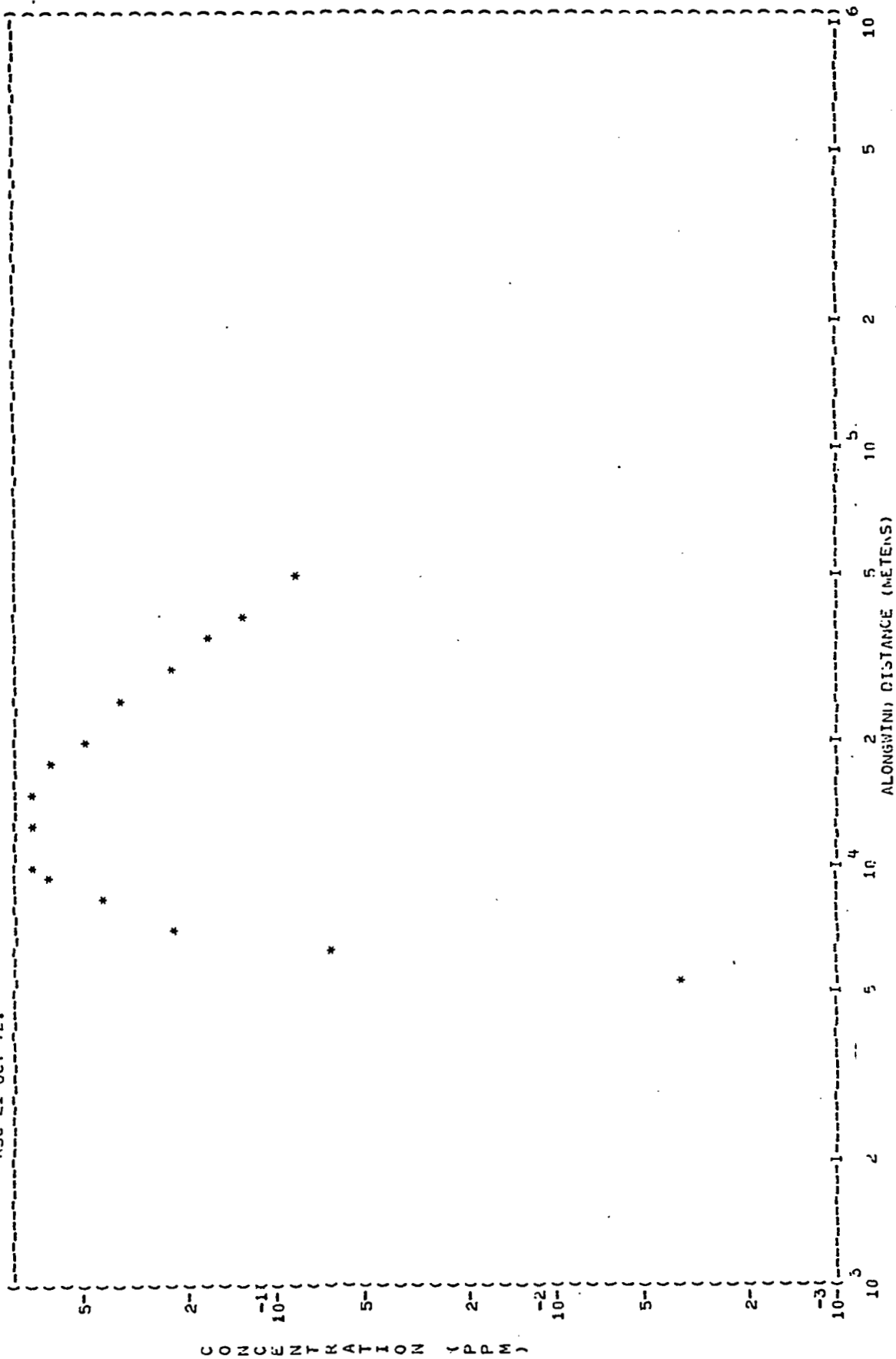
.788 IS THE MAXIMUM PEAK CONCENTRATION

122.735 IS THE MAXIMUM DOSAGE

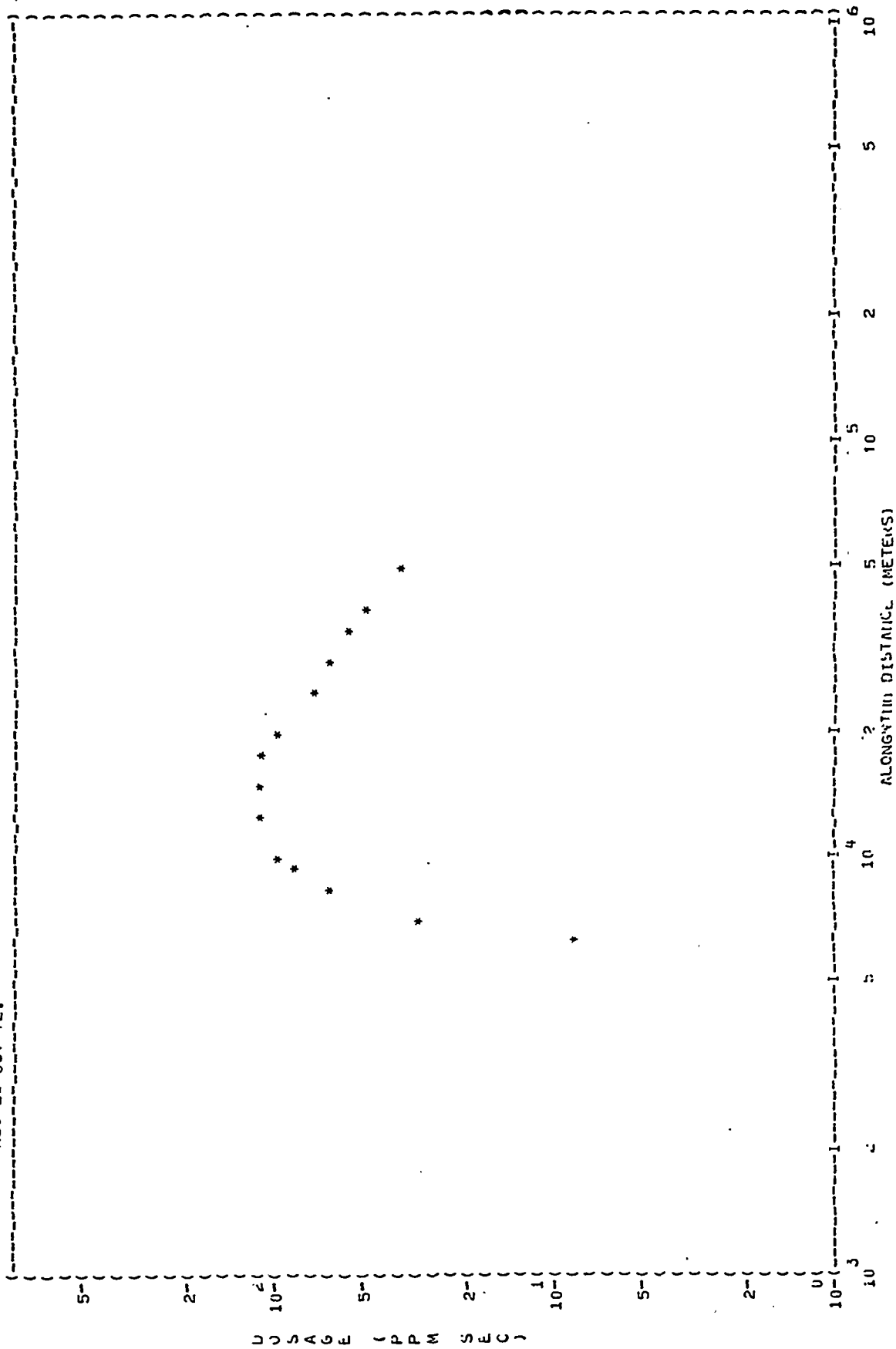
.205 IS THE MAXIMUM PEAK 10.0 MINUTE TIME-MEAN CONCENTRATION

RANGE (METERS)	AZIMUTH BEARING (DEGREES)	MAXIMUM PEAK CONCENTRATION (PPM)	MAXIMUM DOSAGE (PPM SEC)	10.0 MINUTE TIME-MEAN CONCENTRATION (PPM)	TIME OF CLOUD PASSAGE (SECONDS)	AVERAGE CLOUD CONCENTRATION (PPM)
5000.0	240.4	.003	.470	.001	236.196	.002
6000.0	240.6	.067	9.218	.015	237.531	.039
7000.0	241.2	.243	33.956	.057	239.784	.142
8000.0	241.9	.451	63.878	.106	242.936	.263
9000.0	242.5	.615	88.541	.143	246.946	.359
10000.0	243.0	.719	105.552	.170	251.770	.419
12500.0	243.9	.788	122.735	.205	267.077	.460
15000.0	244.6	.717	119.633	.199	286.410	.418
17500.0	245.0	.605	108.895	.181	308.985	.352
20000.0	245.4	.493	96.999	.162	334.174	.290
25000.0	245.9	.341	77.579	.129	390.269	.199
30000.0	246.3	.244	64.253	.107	451.609	.142
35000.0	246.5	.182	54.807	.090	516.298	.106
40000.0	246.7	.141	47.774	.077	583.250	.082
50000.0	246.9	.090	38.006	.059	721.357	.053

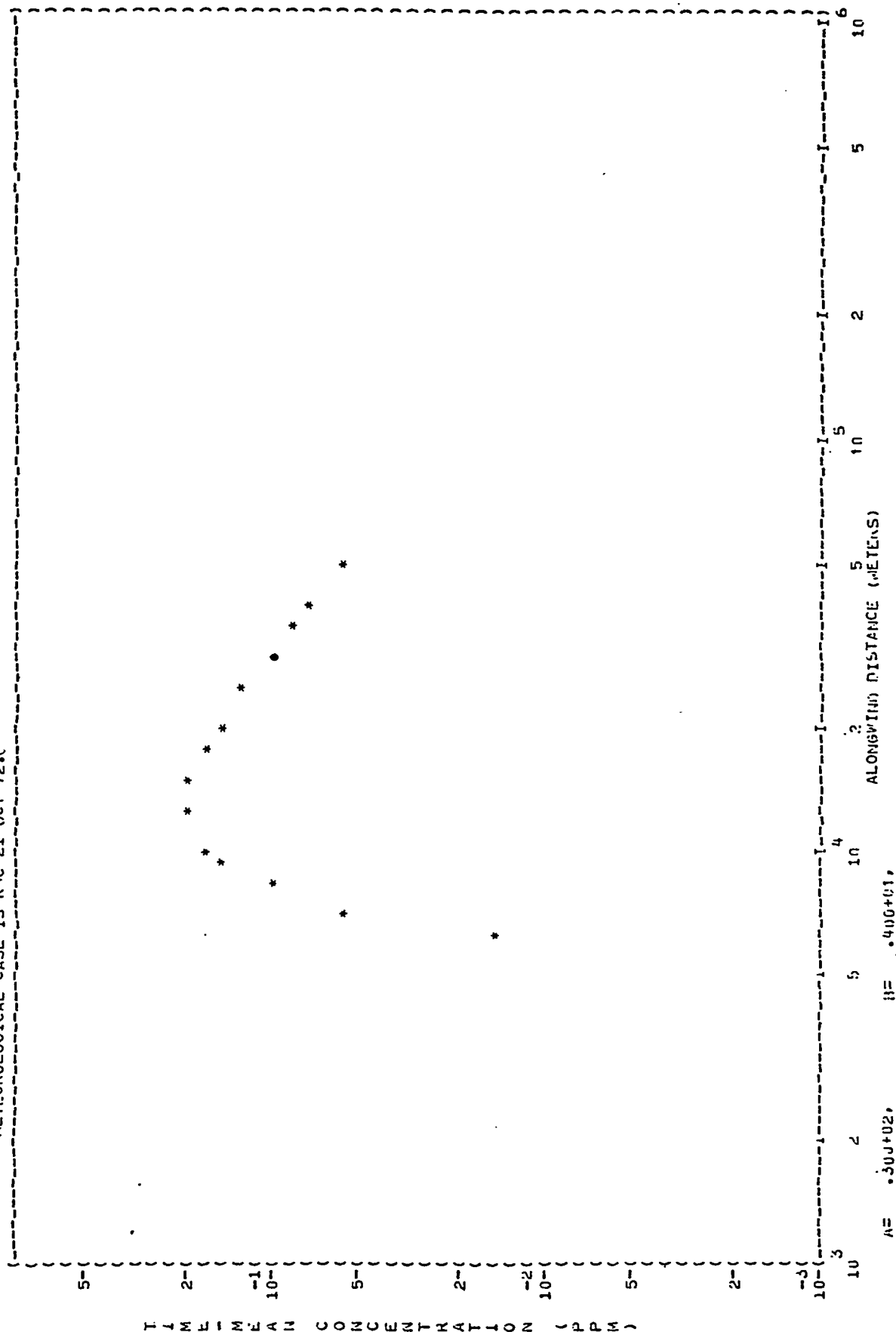
MAXIMUM CENTERLINE HCL CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21 OCT 72.



MAXIMUM CENTERLINE HCL LOSAGE IN PPM SEC AT A FLIGHT OF 0 METERS DOWNWIND FROM A SPACE
 SHUTTLE NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS
 KSC 21 OCT 72.



MAXIMUM CENTERLINE HCL 10 MINUTE TIME-MEAN CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS
 DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE
 METEOROLOGICAL CASE IS KSC 21 OCT 72.



ISOPLETHS ARE CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE
NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21
UCT 72.

*** ISOPLETH LEVEL = .500, CONCENTRATION (PPM)

RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)
8479.742	240.000	9000.000	238.377	9824.534	237.500	11502.030	237.500	12500.000	237.582
16503.409	240.000	17500.000	240.965	18900.599	242.500	19914.082	245.000	19314.856	247.500
15825.102	250.000	15000.000	250.281	12500.000	250.289	12046.868	250.000	10000.000	248.563
9000.000	246.581	6576.366	245.000	8306.618	242.500				

*** ISOPLETH LEVEL = .100, CONCENTRATION (PPM)

6195.893	246.000	6522.493	235.000	8000.001	230.711	9000.001	230.219	10000.001	230.159
15000.000	231.319	17500.000	232.367	17812.657	232.500	20000.000	233.224	24580.074	235.000
30000.000	236.974	31566.847	237.500	35000.000	238.847	38137.632	240.000	40000.000	240.944
47385.351	245.000	40002.933	247.500	45950.121	250.000	40000.000	252.494	35000.000	254.170
30000.000	255.581	25000.000	256.765	20471.090	257.500	20000.000	257.587	17500.000	257.904
13278.011	257.500	12500.000	257.387	10000.000	256.160	9065.872	255.000	8000.001	253.306
6310.190	245.000								

DATE 07/17/75

ISOPLETHS HCL DOSAGE IN PPM SEC AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE
NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21
OCT 72.

*** ISOPLETH LEVEL = 100.000, DOSAGE (PPM SEC)

RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)	RANGE (METERS)	AZIMUTH BEARING (DEGREES)
10000.000	241.173	11499.040	240.000	12500.000	239.704	13443.331	240.000
15342.909	245.003	14042.315	247.500	17500.000	247.777	15000.000	248.494
10000.000	245.102	9684.162	242.500			15000.000	248.128
						12500.000	248.128
						17500.000	248.128
						11806.448	247.500

*** ISOPLETH LEVEL = 50.000, DOSAGE (PPM SEC)

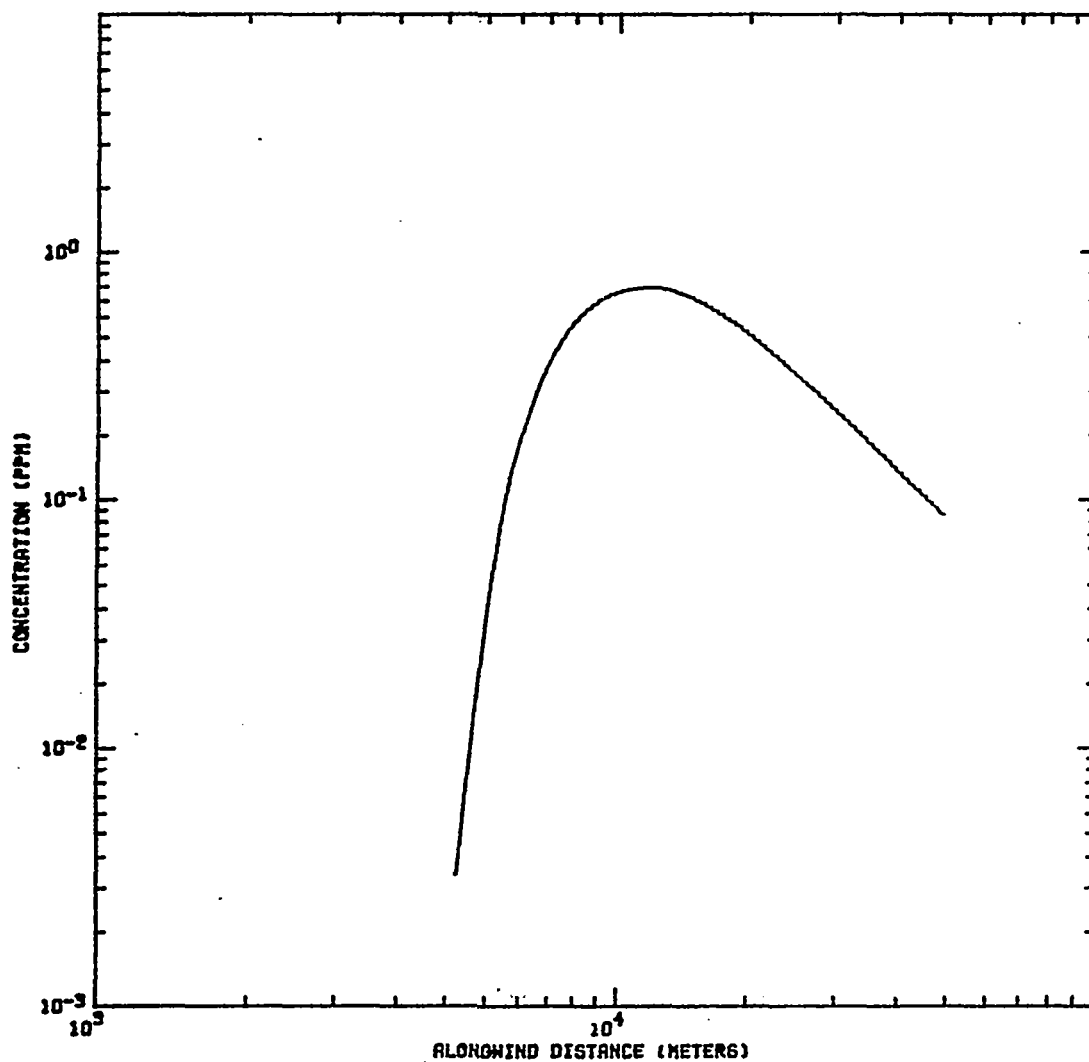
7965.976	237.500	5000.001	235.542	10000.000	235.015	12500.001	235.100
19725.600	237.500	20000.000	237.514	25000.000	239.499	26582.323	240.000
35000.000	243.760	37473.169	245.000	36143.867	247.500	35000.000	249.321
24462.403	252.500	20000.000	253.295	17500.000	253.551	15000.000	253.512
10000.000	251.082	9000.001	249.492	8264.691	247.500	7550.387	242.500
						15000.001	235.746
						30000.000	241.447
						30000.000	251.110
						12500.000	252.883
						17500.000	236.620
						33007.956	242.500
						25000.000	252.376
						11964.034	252.500

*** ISOPLETH LEVEL = 25.000, DOSAGE (PPM SEC)

7264.799	235.000	6663.306	232.500	12500.001	232.233	15000.001	232.841
22668.500	235.000	25000.001	235.626	30000.000	236.778	33346.294	237.500
47366.076	240.000	61725.603	242.500	72490.601	245.000	76031.759	247.500
49999.999	253.380	40000.000	254.645	37018.661	255.000	34999.999	255.237
19999.999	250.682	17499.999	256.718	15000.000	256.534	12500.000	255.871
7160.786	247.500	6655.104	240.000			11236.577	255.000
						17500.001	233.498
						35000.000	237.843
						70775.987	250.000
						24999.999	256.328
						9000.000	252.956
						20000.001	234.251
						40000.000	238.767
						57299.939	252.500

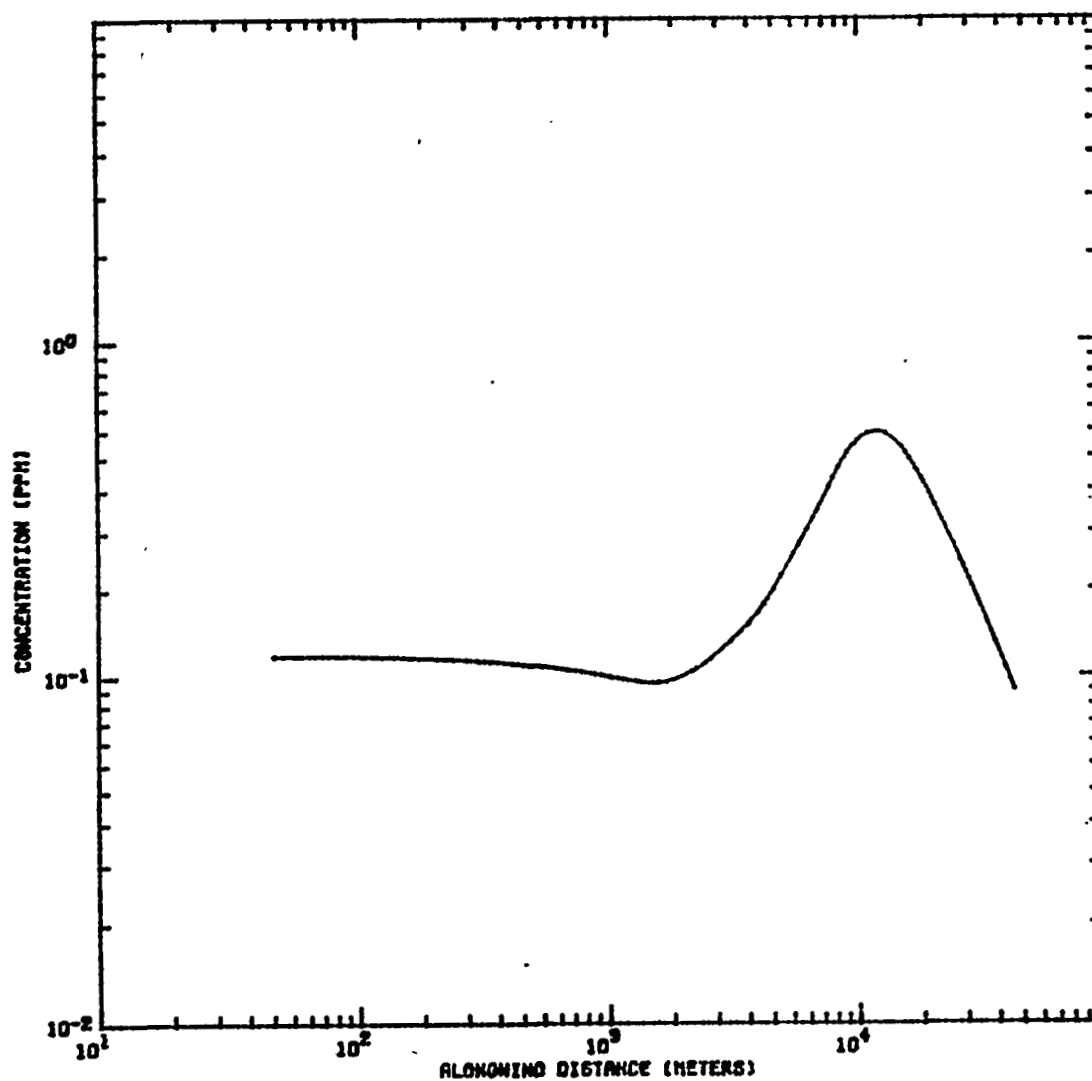
*** ISOPLETH LEVEL = 5.000, DOSAGE (PPM SEC)

5799.861	235.000	6603.690	230.000	8000.001	227.631	9000.001	227.250
12500.001	227.642	15000.001	228.049	17500.001	228.553	20000.000	229.134
30000.001	230.830	35000.000	231.460	40000.000	232.100	43747.968	232.500
79283.034	235.000	79999.999	259.735	49999.999	260.352	49999.999	261.005
29999.999	261.821	24999.998	261.916	19999.999	261.099	17500.000	261.778
11498.750	260.000	10000.000	259.260	9000.000	258.207	8000.000	256.699
5543.303	242.500					7000.001	254.031
						10000.001	227.226
						23712.746	230.000
						25000.001	230.202
						65000.000	234.024
						34999.999	261.670
						12500.000	260.681
						6179.228	250.000



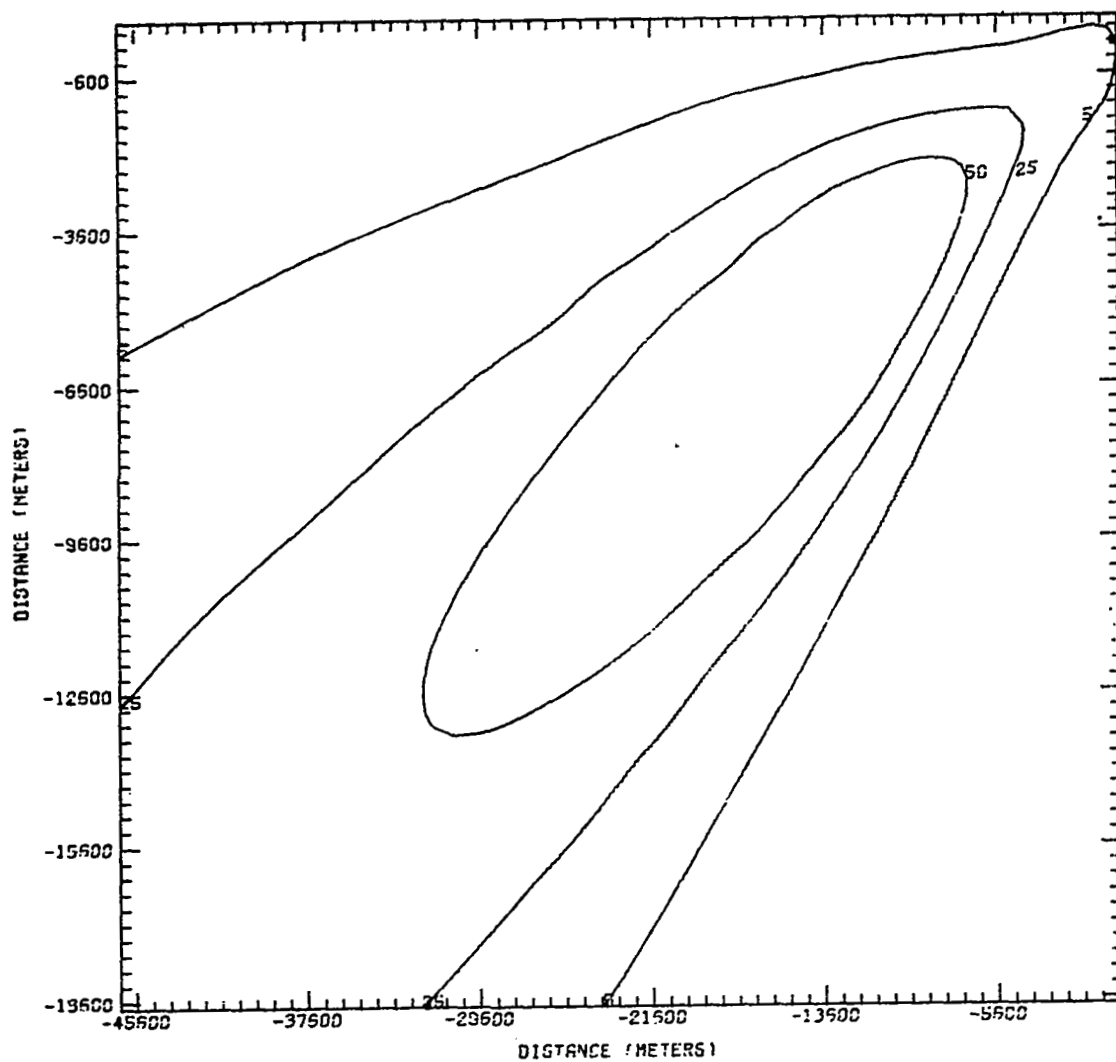
MAXIMUM CENTERLINE HCL CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21 OCT 72.

FIGURE D-1. Maximum centerline HCL concentration for Model 3 generated by the plot routines for the NASA/MSFC Multilayer Diffusion Models Program.



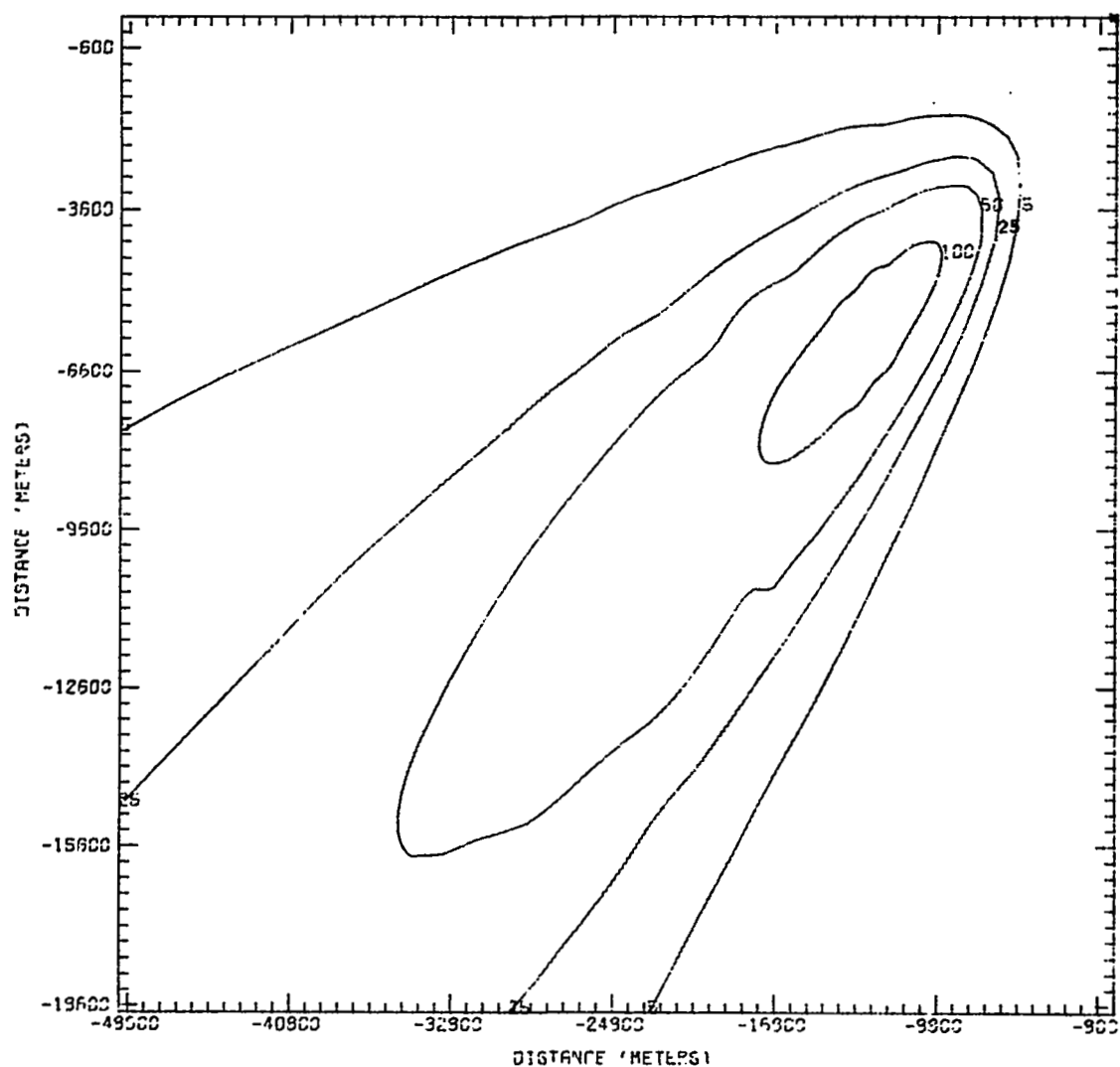
MAXIMUM CENTERLINE HCL CONCENTRATION IN PPM AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21 OCT 72.

FIGURE D-2. Maximum centerline HCL concentration from Model 4 generated by the plot routines for the NASA/MSFC Multilayer Diffusion Models Program.



ISOPLETHS HCL DOSAGE IN PPM SEC AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE NORMAL LAUNCH. MODEL 4 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 21 OCT 72.

FIGURE D-3. Isopleths of HCL dosage for Model 3 generated by the plot routines for the NASA/MSFC Multilayer Diffusion Models Program.



ISOPLETHS HCL DOSAGE IN PPM SL AT A HEIGHT OF 0 METERS DOWNWIND FROM A SPACE SHUTTLE
NORMAL LAUNCH. MODEL 3 WAS USED IN THE CALCULATIONS AND THE METEOROLOGICAL CASE IS KSC 7:
OCT 72.

**FIGURE D-4. Isopleths of HCL dosage for Model 4 generated by the plot routines
for the NASA/MSFC Multilayer Diffusion Models Program.**

APPENDIX E

DERIVATION OF THE VERTICAL TERM APPEARING IN EQUATION (3-18), SECTION 3

The Vertical Term defined by the summation in Equation (3-18), Section 3 of the main body of this report was derived using the method of image sources (Slade, 1968, p. 346) to account for the reflection of gases and aerosols at the earth's surface and at the bases of elevated inversions. Consider a point source located at a height H above the surface ($z = 0$) within a surface mixing layer of depth H_m , as shown in Figure E-1. The numbers appearing at the height z in Figure E-1 denote the intersections of rays from the real and image sources with the height z at various distances downwind from the source. Thus, the number 1 indicates the ray intersection from the real source at height H with the level z and the number 4 indicates the intersection of the ray from the image source at height $(2H_m + H - z)$. The heights of other image sources above the z plane are indicated at the left of Figure E-1. The parameter γ_r is the fraction of material reflected at the earth's surface. For complete reflection, γ_r is set equal to unity while, for no reflection, γ_r is zero. Note that the height term for each image source is multiplied by γ_r for each ray intersection with the earth's surface that occurs prior to the intersection of the ray with height z . Therefore, the image source at the height $(2H_m + H - z)$ is multiplied by γ_r for the reflection at the earth's surface from the image source at $(H + z)$; the image source at $(2H_m + H + z)$ is multiplied by γ_r^2 for the reflection at the earth's surface from the image sources at $(H + z)$ and $(2H_m + H + z)$. The contributions to the vertical term in Equation (3-18) from the real and image sources, assuming a Gaussian distribution of material in the vertical, are seen from an inspection of Figure E-1 to be given by the following expressions:

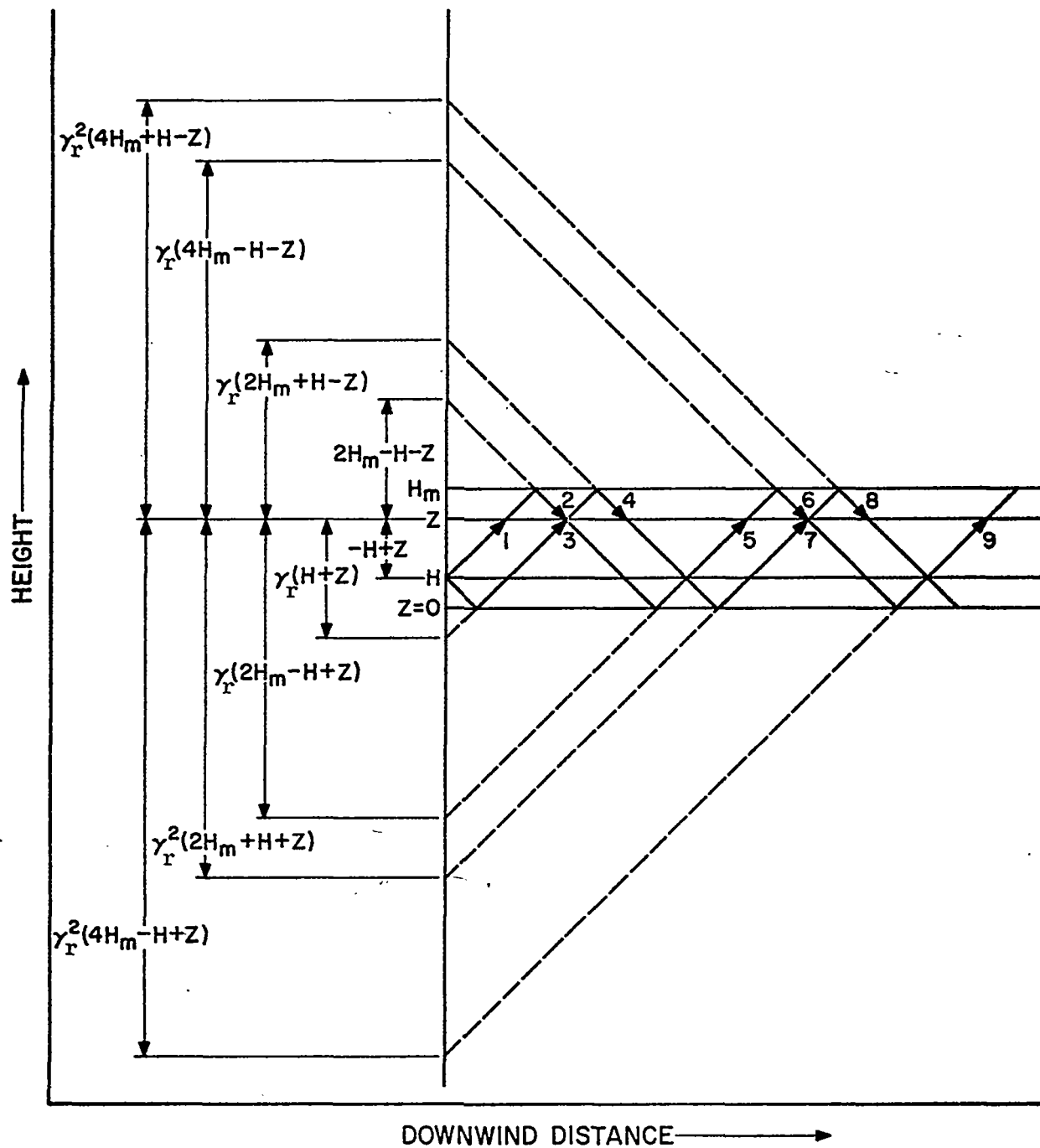


FIGURE E-1. Schematic diagram of image source configurations.

**Contribution at
Point Number**

Term

1

$$\exp \left\{ \frac{-(H-z)^2}{2\sigma_z^2} \right\}$$

2

$$\exp \left\{ \frac{-(2H_m - H - z)^2}{2\sigma_z^2} \right\}$$

3

$$\gamma_r \exp \left\{ \frac{-(H+z)^2}{2\sigma_z^2} \right\}$$

4

$$\gamma_r \exp \left\{ \frac{-(2H_m + H - z)^2}{2\sigma_z^2} \right\}$$

5

$$\gamma_r \exp \left\{ \frac{-(2H_m - H + z)^2}{2\sigma_z^2} \right\}$$

6

$$\gamma_r \exp \left\{ \frac{-(4H_m - H - z)^2}{2\sigma_z^2} \right\}$$

7

$$\gamma_r^2 \exp \left\{ \frac{-(2H_m + H + z)^2}{2\sigma_z^2} \right\}$$

8

$$\gamma_r^2 \exp \left\{ \frac{-(4H_m + H - z)^2}{2\sigma_z^2} \right\}$$

Contribution at
Point Number

Term

$$\begin{array}{c}
 9 \\
 \vdots \\
 i
 \end{array}
 \gamma_r^2 \exp \left\{ \frac{-(4H_m - H + z)^2}{2\sigma_z^2} \right\}$$

If we sum over all sources and combine terms, the total Vertical Term becomes

$$\begin{aligned}
 & \sum_{i=0}^{\infty} \left\{ \gamma_r^i \exp \left[-1/2 \left(\frac{2_i H_m + H - z}{\sigma_z} \right)^2 \right] + \gamma_r^{i+1} \exp \left[-1/2 \left(\frac{2_i H_m + H + z}{\sigma_z} \right)^2 \right] \right\} \\
 & + \sum_{i=1}^{\infty} \left\{ \gamma_r^i \exp \left[-1/2 \left(\frac{2_i H_m - H + z}{\sigma_z} \right)^2 \right] + \gamma_r^{i-1} \exp \left[-1/2 \left(\frac{2_i H_m - H - z}{\sigma_z} \right)^2 \right] \right\} \quad (E-1)
 \end{aligned}$$

where, for convenience in writing Equation (E-1), the quantity 0^0 is defined to be unity.

Next, consider a vertical line source of length L centered about height H in Figure E-1. Assuming that the line source is comprised of an infinite number of point sources, the Vertical Term for the real line source V_H centered at height H is given by the expression

$$V_H = \frac{1}{\sqrt{2\pi} \sigma_z} \int_{H-L/2}^{H+L/2} \exp \left\{ \frac{-(z'-z)^2}{2\sigma_z^2} \right\} dz' \quad (E-2)$$

If the substitution

$$\xi = \frac{z' - z}{\sqrt{2} \sigma_z}$$

is made in Equation (E-2), the resulting expression is

$$V_H = \frac{1}{\sqrt{\pi}} \int_{\frac{-L/2+(H-z)}{\sqrt{2} \sigma_z}}^{\frac{L/2+(H-z)}{\sqrt{2} \sigma_z}} \exp(-\xi^2) d\xi \quad (E-3)$$

Using the definition of error functions (Abramowitz and Stegun, 1964, p. 297), Equation (E-3) can be written in the form

$$\begin{aligned} V_H &= \frac{1}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (H-z)}{\sqrt{2} \sigma_z} \right) - \operatorname{erf} \left(\frac{-L/2 + (H-z)}{\sqrt{2} \sigma_z} \right) \right\} \\ &= \frac{1}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (H-z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (H-z)}{\sqrt{2} \sigma_z} \right) \right\} \end{aligned} \quad (E-4)$$

Inspection of Equation (E-4) shows that the numerators of the error function arguments express, respectively, the distance of the top and base of the line source from the height z . The corresponding vertical terms for point and line sources in the surface mixing layer, including provision for partial reflection at the earth's surface, are given in Table E-1.

The Vertical Term of Equation (3-18) can be obtained from the line source terms in Table E-1 through use of the following relationships:

$$2H_m = 2(z_{TL} - z_{BL}) \quad (E-5)$$

TABLE E-1
CORRESPONDING VERTICAL TERMS OF A POINT SOURCE AND LINE
SOURCE IN THE SURFACE MIXING LAYER

Point Source	Line Source
$\exp \left[-\frac{1}{2} \left(\frac{H-z}{\sigma_z} \right)^2 \right]$	$\frac{1}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (H-z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (H-z)}{\sqrt{2} \sigma_z} \right) \right\}$
$\gamma_r \exp \left[-\frac{1}{2} \left(\frac{H+z}{\sigma_z} \right)^2 \right]$	$\frac{\gamma_r}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (H+z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (H+z)}{\sqrt{2} \sigma_z} \right) \right\}$
$\exp \left[-\frac{1}{2} \left(\frac{2iH_m - H-z}{\sigma_z} \right)^2 \right]$	$\frac{1}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (2iH_m - H-z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (2iH_m - H-z)}{\sqrt{2} \sigma_z} \right) \right\}$
$\gamma_r \exp \left[-\frac{1}{2} \left(\frac{2iH_m - H+z}{\sigma_z} \right)^2 \right]$	$\frac{\gamma_r}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (2iH_m - H+z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (2iH_m - H+z)}{\sqrt{2} \sigma_z} \right) \right\}$
$\gamma_r \exp \left[-\frac{1}{2} \left(\frac{2iH_m + H-z}{\sigma_z} \right)^2 \right]$	$\frac{\gamma_r}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (2iH_m + H-z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (2iH_m + H-z)}{\sqrt{2} \sigma_z} \right) \right\}$
$\gamma_r^2 \exp \left[-\frac{1}{2} \left(\frac{2iH_m + H+z}{\sigma_z} \right)^2 \right]$	$\frac{\gamma_r^2}{2} \left\{ \operatorname{erf} \left(\frac{L/2 + (2iH_m + H+z)}{\sqrt{2} \sigma_z} \right) + \operatorname{erf} \left(\frac{L/2 - (2iH_m + H+z)}{\sqrt{2} \sigma_z} \right) \right\}$

$$H = \left(\frac{z_{TK} - z_{BK}}{2} + z_{BK} - z_{BL} \right) \quad (E-6)$$

$$z = (z_L - z_{BL}) \quad (E-7)$$

$$\frac{L}{2} = \frac{z_{TK} - z_{BK}}{2} \quad (E-8)$$

where Equations (E-5) through (E-8) express the relationship between the height coordinate system of Figure E-1 and the generalized height coordinate system used in developing Equation (3-18).

Substituting Equations (E-5) through (E-8) into the line source terms in the right-hand column of Table E-1 and collecting terms results in the expression

$$\begin{aligned} & \frac{1}{2} \left\{ \sum_{i=0}^{\infty} \left[\gamma_r^i \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) + z_{TK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) - z_{BK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \right. \right. \\ & + \gamma_r^{i+1} \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) - 2z_{BL} + z_{TK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) + 2z_{BL} - z_{BK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \left. \right] \\ & + \gamma_r^{i-1} \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) + 2z_{BL} - z_{BK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) - 2z_{BL} + z_{TK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \left. \right\} \\ & + \sum_{i=1}^{\infty} \left[\gamma_r^i \left[\operatorname{erf} \left(\frac{2i(z_{TL} - z_{BL}) - z_{BK} + z_L}{\sqrt{2} \sigma_{zLK}} \right) + \operatorname{erf} \left(\frac{-2i(z_{TL} - z_{BL}) + z_{TK} - z_L}{\sqrt{2} \sigma_{zLK}} \right) \right] \right] \end{aligned} \quad (E-9)$$

where, again 0^0 is defined equal to unity. In Equation (3-18), the factor $1/2$ appearing before the bracket in Equation (E-9) is contained in the form factor $1/2 \sqrt{2\pi}$ appearing in the denominator of the first term on the right following the equal sign.